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A complete 0D thermodynamic predictive model for Direct Injection Diesel engines

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Abstract

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Ideal models provide the simplest way to reproduce internal combustion engine (ICE) cycles, but usually they do not represent with sufficient accuracy the actual behaviour of an ICE. A suitable alternative for research and development applications is provided by zero-dimensional (0D) thermodynamic models. Such models are very useful to predict the instantaneous pressure and temperature in the combustion chamber, which in turn allow the prediction of engine operation characteristics. However, their simplifying hypotheses led in some cases to a leak of accuracy or a limited predictive capability.

This paper describes a 0D single-zone thermodynamic model that takes into account the heat transfer to the chamber walls, the blow-by leakage, the fuel injection and engine deformations, along with the instantaneous change of the gas properties. Special attention have been paid to the description of the specific sub-models that have been used for the calculation of the energy and mass equations terms. Also the procedures followed for the estimation of some mechanical and heat transfer parameters and the combustion model fitting is detailed. After the fitting, the model was validated in a large amount of operation points in a 4-cylinder 2-litre DI diesel engine, showing a good capability for accurate predictions of the gas state during the closed cycle and engine performance.

Keywords: real cycles, thermodynamic model, heat transfer, blow-by leakage, engine deformations

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25 Nomenclature

α	Crank angle [cad]
c_v	Specific heat at constant volume $[J/kg \ K]$
D	Cylinder bore [m]
d_{comb}	Combustion duration [cad]
dHR	Heat Released in a calculation step $\dots [J]$
EVO	Exhaust Valve Opening
F_s	Stoichiometric fuel-air equivalence ratio
HRF	Cummulated Heat Release Fraction [-]
γ	Adiabatic index
h	Specific enthalphy $[J/kg]$
IVC	Intake Valve Closing
IMEP	Indicated Mean Effective Pressure
m	Mass
p	In-cylinder pressure [Pa]
Q	Heat transfer to the walls
R	Specific gas constant $[J/kg \ K]$
SOC	Start of Combustion
SOI	Start of Injection
T	Temperature $[K]$
U, u	Internal energy $[J]$, $[J/kg]$
Y	Mass fraction [—]

26 Subscripts

a Relative to the air

b Relative to the stoichiometric combustion products

bb Relative to the blow-by

c Relative to the gas mean properties in the chamber

cool Relative to coolant

exh Relative to exhaust conditions

f Relative to the fuel

f, ev Relative to the evaporated fuel f, g Relative to the gaseous fuel

f, inj Relative to the liquid fuel at injection conditions

in Relative to inlet conditions evap Relative to the evaporation

1. Introduction

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Nowadays the main challenges in the internal combustion engines (ICE) consist on the reduction of fuel consumption and emissions. With this purpose different techniques have appeared to better optimize the combustion process: high pressure fuel injection systems [1], multiple injections [2], high boost pressure [3], EGR [4], variable valve timing [5], high swirl ratios [6], new clean fuels [7], Nevertheless, the great quantity of possible configurations for each engine condition as well as the necessity of shorter engine set up times, impose the use of predictive calculation models that allow to speed-up the search for the optimum configuration.

In this framework, ideal models represent the first approach to analyze the main trends of efficiency and engine power. Despite the claims by several researchers [8–10], it is generally accepted [11] that, due to their intrinsic limitations, do not represent with sufficient accuracy the behaviour of a current ICE.

To improve the ideal models performances, Bhattacharyya [12] included thermal and friction losses into an equivalent friction term in order to find the maximum power output varying compression ratio and cut-off ratio. On other hand, Akash [13] used a linear expression to estimate the heat added to the working fluid during combustion. Chen [14] conjugate previous works, i.e. he applied a linear expression for the heat transfer during combustion and estimate friction losses assuming that the friction losses force was a linear function of the velocity, as Angulo-Brown [15, 16] who used it for air-standard Otto cycle. Al-Sharki [17] included the temperature-dependent specific heats in order to improve previous works. Zhao [18] first included a new approach for determining the heat leakage level as a percentage of the fuel's chemical energy; and later on he included the specific heats dependence with temperature [19]. Other improvement in ideal model is carried out by [20] who used the Taylor model [21] to estimate the rate of heat loss. The most recent study [22] presents a diesel engine simulation where a dual Wiebe function [23] was used to model the heat release while the convective heat transfer coefficient is given by the Woschni model [24]. There are similar approaches to other cycles: Otto [25], Atkinson [26], Miller [27] and Dual [28].

In spite of the improvements performed in the ideal models by the cited authors and others, they have important assumptions to solve. A suitable alternative for research and development applications is provided by 0D thermodynamic models [11]. In these models, the mass and energy conservation equations are solved so that the instantaneous state of the gas inside the combustion chamber (pressure and temperature) is obtained. However, their simplifying hypotheses led in some cases to a leak of accuracy or a limited predictive capability. Their assumptions are related to the close system statement, i.e. the no consideration of blow-by through the piston rings and fuel addition, the use of correlations for air specific heat without considering the changes in the gas composition due to the fuel injection and chemical reactions that take place during the combustion process, i.e. the combustion products; and finally the consideration of a realistic heat release law. All this issues are dealt with in this work.

The objective of the present work is to develop a 0D single-zone thermodynamic model covering all the previous statements (the simulation of the rate of heat release law, heat transfer and gas properties sub-models as well as blow-by leakage and the fuel injection contributions to the diesel engine cycles) that have been partially covered by previous researchers. 0D models do not have spatial resolution, however they can consider one or more zones in the combustion chamber (mass and energy conservation equations are solved for each zone). In some applications, such as emissions prediction, or when the combustion details must be accounted for [29–31], it is imperative to consider more that one zone in order to obtain acceptable results. However, correctly calibrated and validated the proposed 0D single-zone model is suitable for the accurate calculation of the pressure evolution with the purpose of predicting engine performances and fuel economy or obtaining boundary conditions for specific combustion models with a high computational efficiency. Starting from Intake Valve Closing (IVC) conditions (p, m) and T) the proposed model calculates the gas state up to the Exhaust Valve Opening (EVO) thus being a predictive tool that allows rapid parametric explorations of different engine operating conditions and geometric configurations.

4 2. Thermodynamic model description

In this section, the main hypothesis in which the thermodynamic model is based, the energy and mass balances as well as a description of sub-models required to calculate such balances are described.

2.1. Basic hypothesis

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- 1. Uniform pressure in the chamber is assumed. This is normally assumed in combustion calculations because both fluid and flame velocities are much smaller than the sound speed [32].
- 2. Three species are considered: air, fuel vapour and stoichiometric combustion products. In a DI diesel engine operating with conventional combustion, during the mixing-controlled burning phase the flame is located at the stoichiometric fuel-air ratio region [33] and thus this hypothesis is quite realistic.
- 3. **Perfect gas behaviour is assumed**. This assumption is reasonable since negligible errors are committed as shown by Lapuerta et al. [35].

- 4. Specific heats depend on temperature and gas composition. This hypothesis is consistent with the two previous hypotheses.
- 5. Internal energy is calculated assuming mean uniform temperature in the chamber. This is perhaps the hardest hypothesis, as it affects the calculation of the internal energy of the gas. This can be specially important for the burned products at the beginning of the combustion. However, the error diminishes as the combustion progresses because dilution and heat transfer tend to make the temperature uniform.
- 6. Heat transfer to the chamber walls is considered. The fraction of the fuel energy lost by the heat transfer to the chamber walls depends on engine size (the larger the engine is the more adiabatic it is) and the operating conditions (the higher the load and the engine speed are, the more adiabatic it is). For a high speed DI diesel engine typical values range from 10% at full load and 4000 rpm to 30% at low load and 1000 rpm. Therefore, consideration of heat transfer, depending on operating conditions, is necessary in order to obtain accurate predictions of indicated parameters or the thermodynamic state of the charge [36].
- 7. Blow-by leakage is considered. The blow-by mass flow is a good indicator of the integrity of the piston rings and lubricant. In normal operating conditions the blow-by mass flow is not important, and thus it is not usually measured. However, in small DI diesel engines (0.35 l of displacement) at low engine speeds, blow-by mass rates about 4% of the trapped mass have been measured; moreover, during the cold start it is usual to have more than 20% [37, 38]. It is thus advisable to consider this term in the mass and the energy balances, as it will be discussed below.
- 8. Fuel injection is considered. Althought the empirical combustion model that will be described later is not based on the injection rate, the consideration of the injected fuel is an important issue for both the mass and the energy balances. It has been checked that the error in the energy balance (that includes the temperature and gas properties errors due to the mass and composition errors) can reach a 7% in the case of rich fuel-air equivalence ratio. Thus the consideration of the injected mass is highly recommendable.
- 9. Engine deformation is considered. Usually, simple 0D models do not include engine deformations for the instantaneous volume calculation. However, the deformations of diesel engine, where pressure can easily reach 150 bar at the top dead center, can be higher than 2% at this point. Thus a simple deformation model will be used to estimate the real in-cylinder volume.

2.2. Energy balance

The energy and mass balance are solved between IVC and EVO, considering the combustion chamber as an open system because of the blow-by and the fuel injection. The main results are the instantaneous pressure and temperature. Accordingly to the hypotheses 1 and 5, no spatial resolution of the thermodynamic properties is considered.

The most general expression of the First Law for an open system is:

$$dU_{c} = -dQ + dW + h_{f,inj} dm_{f,inj} - h_{c} dm_{bb}$$

= $-dQ - p dV + h_{f,inj} dm_{f,inj} - h_{c} dm_{bb}$ (1)

where U_c is the internal energy of the charge, Q is the heat transfer to the walls, W is the work, $h_{f,inj}$ is the specific enthalpy of the fuel at the injection conditions, $m_{f,inj}$ is the injected fuel mass, h_c is the specific enthalpy of the charge and m_{bb} is the blow-by mass.

As the model considers only one zone in the chamber, a single gas phase is considered, and thus the injected fuel is taken into account in the mass and energy balances assuming that the evaporation takes place instantaneously. This assumption is realistic for main injections in which the time for atomization, heating and evaporation is negligible, whereas this is less realistic for early pilot injections. In order to describe the process rigorously it would be necessary to follow the fuel parcel during all its evolution from the injection conditions up to the mean chamber temperature. This would require the consideration of at least two zones (liquid and gaseous phases), and thus the model complication; however, it has been checked that minor improvements in the results would be obtained.

Accordingly to the previous comments, Eq. (1) may be written as

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$$dU_c = -dQ - p \ dV + h_{f,inj} \cdot dm_{f,ev} - h_c \ dm_{bb}$$
 (2)

where the evaporated fuel mass $m_{f,ev}$ has been considered instead of injected fuel mass $m_{f,inj}$. Taking into account the hypothesis 2, the left-hand side of Eq. (2) can be expressed as:

$$dU_c = d(m_c \ u_c) = d(m_a \ u_a + m_{f,g} \ u_{f,g} + m_b \ u_b)$$

$$= m_a \ du_a + m_{f,g} \ du_{f,g} + m_b \ du_b +$$

$$+ u_a \ dm_a + u_{f,a} \ dm_{f,g} + u_b \ dm_b$$
(3)

where the subscripts c, a, f, g and b refer to chamber, air, gaseous fuel and stoichiometric burned products, respectively. The terms with du_a , $du_{f,g}$ and du_b in Eq. (3) correspond to the variation of internal sensible energy due to the change in the chamber temperature:

$$m_a du_a + m_{f,g} du_{f,g} + m_b du_b =$$

$$= m_a c_{v,a} dT + m_{f,g} c_{v,f} dT + m_b c_{v,b} dT =$$

$$= m_c (Y_a c_{v,a} + Y_f c_{v,f} + Y_b c_{v,b}) dT = m_c c_{v,c} dT$$
(4)

where Y represents the fuel mass fraction. In Eq. (3), the terms with dm_a , $dm_{f,g}$ and dm_b represent, respectively, the variation of the internal energy due to the composition change associated with combustion, the fuel injection and blow-by losses:

$$dm_a = -dm_{a,b} - Y_a dm_{bb} (5)$$

$$dm_{f,g} = dm_{f,ev} - dm_{f,b} - Y_{f,g} dm_{bb}$$
 (6)

$$dm_b = dm_{a,b} + dm_{f,b} - Y_b dm_{bb} \tag{7}$$

where $dm_{a,b}$ y $dm_{f,b}$ are the mass of air and fuel burned in stoichiometric conditions, so that

$$dm_{f,b} = F_s \cdot dm_{a,b} \tag{8}$$

Taking into account Eqs. (7) and (8), the variations of the air and fuel masses (Eqs. (5) and (6)) can be expressed in terms of the variation of the stoichiometric burned

mass. Substituting into Eq. (3), taking into account Eq. (4), after some algebra Eq. (1) becomes:

$$\left[u_b - \frac{u_a + u_{f,g} F_e}{F_e + 1} \right] \cdot (dm_b + Y_b dm_{bb}) =$$

$$= -m_c c_{v,c} dT - dQ - p dV + (h_{f,inj} - u_{f,g}) \cdot dm_{f,ev} - R_c T dm_{bb}$$
(9)

where the first term in Eq. (9) is the heat released by combustion -dHR.

The negative sign before dHR is used for consistency with the sign criterion used for the heat transfer to the wall: positive means lost by the charge and negative supplied to the charge. In this way, dHR is positive and should be interpreted as the heat supplied to the gas as a result of combustion. If the negative sign had not been included, dHR would be the reaction energy (which would be negative since the internal energy of the combustion products is lower than that of the reactants). With this criterion, one may write:

$$dHR = m_c c_{v,c} dT + dQ + p dV - (h_{f,inj} - u_{f,g}) \cdot dm_{f,ev} + R_c T dm_{bb}$$
 (10)

In Eq. (10) all the involved phenomena can be easily identified: at the left-hand side dHR is the heat released by combustion in a calculation step (the RoHR can be directly obtained by dividing dHR by the angle step, RoHR = dHR/d α), whereas the terms in the right-hand side are, respectively, the sensible internal energy of the gas, the heat transfer to the walls, the work done by the gas, the energy related to fuel injection, evaporation and heating, and the flow work associated with the blow-by leakage.

The fuel injection term can be also decomposed as follows:

$$h_{f,inj} - u_{f,g} = h_{f,l}(T_{inj}) - u_{f,g}(T) =$$

$$= [h_{f,l}(T_{inj}) - u_{f,l}(T_{inj})] - [u_{f,l}(T_{evap}) - u_{f,l}(T_{inj})] -$$

$$- [u_{f,g}(T_{evap}) - u_{f,l}(T_{evap})] - [u_{f,g}(T) - u_{f,g}(T_{evap})]$$
(11)

where the temperature at which enthalpies and internal energies are evaluated has been indicated in brackets. The four terms in the right-hand side of Eq. (11) corresponds to:

- Specific flow work of the injected fuel
- Liquid fuel heating from the injection temperature (T_{inj}) up to the evaporation temperature (T_{evap})
- Internal energy of vaporization

• Gaseous fuel heating from the vaporization temperature up to the mean gas temperature (T_c) .

To finish the description of the energy balance, it is necessary to remark that all the terms in Eq. (9) depend on different variables but all of them depends only on pressure and temperature. Thus Eq. (9) has two unknown variables and an additional equation is required to solve it: the equation of state. Accordingly with hypothesis 3 the ideal gas equation is used to close the equations system:

$$p V = m R_c T \tag{12}$$

Eqs. (9) and (12) are solved for each calculation step from IVC to EVO, providing instantaneous pressure and temperature as the main results. Along with them, other results are obtained of which those most important are: instantaneous mass and composition, heat flux to the walls, and indicated parameters (pressure, work, consumption, etc.).

The sub-models used for the calculation of each term of Eq. (9) are described in the section 2.4.

2.3. Mass balance

The trapped mass at IVC is the addition of the induced mass of fresh air (m_a) , the exhaust gas recirculation mass (m_{EGR}) and the residual gas of the previous cycle (m_{res}) , to which the short-circuiting mass (m_{sc}) during valve overlap (which normally can be neglected in a 4 stroke engine) is subtracted. All of them are model inputs. The incoming masses m_a and m_{EGR} are obtained from experimental measurements, whereas m_{res} and m_{sc} are calculated with a simple filling and emptying model such as that described in [39].

Starting from IVC, the model calculates the instantaneous change of the mass of each species (air, fuel and stoichiometric burned products) associated with blow-by, fuel injection and combustion. Therefore, both instantaneous gas mass and composition can be obtained. The instaneous blow-by and fuel injected models are described in 2.4.1 and 2.4.2

In order to calculate the thermodynamic properties of the gas (see 2.4.3), it is necessary to know the instantaneous gas composition [40]. Taking into account the masses involved, the mass fraction of the burned products at EVO can be expressed as:

$$Y_{b,EVO} = \frac{m_f + \frac{m_f}{F_s} + (m_{EGR} + m_{res}) \cdot Y_{b,EVO} - m_{sc} \cdot Y_{b,IVC} - m_{bb} \left(\frac{Y_{b,IVC} + Y_{b,EVO}}{2}\right)}{m_a + m_f + m_{EGR} + m_{res} - m_{sc} - m_{bb}}$$

$$= \frac{1 + \frac{1}{F_s} - Y_{b,IVC} \left(\frac{m_{sc} + m_{bb}/2}{m_f}\right)}{1 + \frac{1}{F} - \frac{m_{sc} + m_{bb}/2}{m_f}}$$
(13)

where the mean of IVC and EVO compositions has been assumed for the *blow-by* mass leakage. At IVC the composition can be expressed as:

$$Y_{b,IVC} = \frac{m_{EGR} \cdot Y_{b,EVO} + m_{res} \cdot Y_{b,EVO} - m_{sc} \cdot Y_{b,IVC}}{m_a + m_{EGR} + m_{res} - m_{sc}}$$

$$= \frac{Y_{b,EVO}}{1 + \frac{m_a}{m_{EGR} + m_{res}}}$$
(14)

Substitution of Eq. (14) into Eq. (13) gives $Y_{b,EVO}$ expressed in terms of known parameters:

$$Y_{b,EVO} = \frac{1 + \frac{1}{F_s}}{1 + \frac{1}{F} - \left(\frac{m_{sc} + m_{bb}/2}{m_f}\right) \frac{1}{1 + \frac{m_{EGR} + m_{res}}{m_a}}}$$
(15)

At any intermediate instant i between IVC and EVO, the instantaneous mass fraction of burnt products can be written as a function of the heat release fraction (see section 2.4.4):

$$Y_{b} = \frac{\left(m_{f} + \frac{m_{f}}{F_{s}}\right) \cdot \text{HRF} + \left(m_{EGR} + m_{res}\right) \cdot Y_{b,EVO} - m_{sc} \cdot Y_{b,IVC} - \int_{IVC}^{i} Y_{b} \cdot dm_{bb}}{m_{a} + \int_{IVC}^{i} dm_{f} + m_{EGR} + m_{res} - m_{sc} - \int_{IVC}^{i} dm_{bb}}$$
(16)

The unburnt gaseous fuel mass fraction at any intermediate instant can also be obtained as a function of the instantaneous heat release fraction:

$$Y_{f} = \frac{\int_{IVC}^{i} m_{f} - m_{f} \cdot HRF}{m_{a} + \int_{IVC}^{i} dm_{f} + m_{EGR} + m_{res} - m_{sc} - \int_{IVC}^{i} dm_{bb}}$$
(17)

An hence the instantaneous air mass fraction is calculated as $Y_a = 1 - Y_b - Y_f$.

The fuel mass injected per cycle, m_f , is obtained from experimental measurements (along with m_a and m_{EGR}). The instantaneous mass of evaporated fuel (dm_f) is obtained as described in paragraph 2.4.2.

2.4. Sub-models description

A detailed description of the sub-models used to calculate the different terms in Eq. (9) are included in the following section.

238 2.4.1. Blow-by model

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The instantaneous mass flow of blow-by to the crankcase is estimated by means of the equation of an adiabatic nozzle as in [41]:

$$\dot{m}_{bb} = c_{bb} A_{ref} p \sqrt{\frac{x}{R_c T}}$$
(18)

where $x = \frac{2\gamma}{(\gamma - 1)} \left[\left(\frac{p_{crk}}{p} \right)^{\frac{2}{\gamma}} - \left(\frac{p_{crk}}{p} \right)^{\frac{\gamma + 1}{\gamma}} \right]$. The discharge coefficient of the nozzle, c_{bb} , was adjusted with experimental measurements so that the cumulated blow-by coincided with the actual flow, A_{ref} is the reference section proposed by Hohenberg [42] $(3.5 \cdot 10^{-6}D)$ and p_{crk} is the crankcase pressure.

When the pressure ratio accomplishes:

$$\frac{p_{crk}}{p} < \left(\frac{2}{\gamma + 1}\right)^{\frac{\gamma}{\gamma - 1}} = \frac{p_{crit}}{p} \tag{19}$$

sonic conditions is reached and then the nozzle flow must be calculated considering the critical pressure instead of the crankcase pressure.

2.4.2. Fuel injection model

As it is justified later, the consideration of the fuel injection has an important effect on the mass balance, gas composition and energy balance and thus it must be considered to get accurate simulation results. However, as the rate of heat release (see section 2.4.4) contains the main combustion information the accuracy of the injection rate shape is not a critical issue, thus a square-shaped law is assumed for the fuel injection rate. It has been checked that consideration of a more realistic injection rate produces only minor improvements in the description of the thermodynamic evolution in the combustion chamber.

It must be born in mind that the model is not a predictive combustion model and thus it does not account for the real effect of the injection rate on the heat release. Such an effect should be taken into account with a physical combustion model such as that proposed by Arrègle et al. [43, 44]. However, this is out of the scope of this work.

2.4.3. Gas properties model

The thermodynamic properties of the gas are calculated in the following way:

$$R_c = R_a \cdot Y_a + R_f \cdot Y_f + R_b \cdot Y_b \tag{20}$$

$$c_{v,c} = c_{v,a} \cdot Y_a + c_{v,f} \cdot Y_f + c_{v,b} \cdot Y_b \tag{21}$$

where subscripts c, a, f and b refer to the mean properties of the chamber, air, fuel and stoichiometric burned products, respectively. R_a , R_f and R_b are the specific constants of each species, and the specific heats at constant volume, $c_{v,a}$, $c_{v,f}$ y $c_{v,b}$ are obtained of polynomial expressions as a function of the temperature [40].

2.4.4. Heat release law model

The most complex process to be considered when performing the energy balance is combustion. Different proposals can be found in literature for the modelling of this phenomenon, from the simplest heat release laws proposed by Wiebe [23] or Watson [45] to the considerably more complex phenomenological models proposed by Barba [46], Hiroyasu [29] or Arrègle et al. [43, 44].

As stated, the objective of the proposed model is not the prediction of the physical phenomena involved in the combustion process, but the provision of accurate thermodynamic conditions in the chamber. Therefore, following the proposal of Serrano et al. [47], a simple phenomenological model with several Wiebe functions have been used in order to describe the RoHR: in operating point with pilot injection, one Wiebe function (W_1) is used to reproduce it, another three Wiebe functions $(W_2, W_3 \text{ and } W_4)$ are always used to simulate the premixed, diffusion and late combustion phases respectively. Hence, the mathematical expression of the rate of heat release is:

$$RoHR = \sum_{i=1}^{4} \left[\frac{C_i \left(m_i + 1 \right)}{d_{comb,i}} \left(\frac{\alpha - SOC_i}{d_{comb,i}} \right)^{m_i} e^{\left[-C_i \left(\frac{\alpha - SOC_i}{d_{comb,i}} \right)^{(m_i + 1)} \right]} \cdot \beta_i \right]$$
(22)

where α is the crank angle, $d_{comb,i}$ is the duration of the combustion phase i (pilot, premixed, diffusion or late combustion), SOC_i is the start of the combustion phase i, β_i is the proportion of the combustion phase i with respect to the whole combustion (hence its value varies between 0 and 1), C_i is the completion parameter that is traditionally

used to represent the combustion efficiency and m_i is the shape parameter that controls the gradient of the combustion phase i.

The methods follow to fit the values of $d_{comb,i}$, SOC_i , β_i and m_i will be discussed later.

2.4.5. Volume calculation model

The instantaneous volume in the cylinder is calculated as the addition of combustion chamber volume, the instantaneous displaced volume and the deformation due to pressure and inertial forces:

$$V = V_{cc} + V_{d,inst} + \Delta V_p + \Delta V_i \tag{23}$$

where

$$V_{cc} = \frac{V_d}{r_c - 1} \tag{24}$$

being V_d the total displaced volume and r_c the compression ratio. The instantaneous displaced volume $V_{d,inst}$ is calculated taking into account the engine geometry, including the piston eccentricity. The two deformation terms in Eq. (23) are based on a previous work [34]:

$$\Delta V_p = k_{def} \cdot \frac{\pi D^2}{4} \cdot \frac{p}{E_{steel}} \cdot \left(\frac{D}{D_{pp}}\right)^2 \cdot L_0 \tag{25}$$

$$\Delta V_i = k_{def} \cdot \frac{m_{alt} \cdot a}{E_{steel}} \cdot \left(\frac{D}{D_{pp}}\right)^2 \cdot L_0 \tag{26}$$

where k_{def} is a deformation coefficient that is experimentally fitted using motoring tests, as described in section 3.1 (typical values range from 1 to 3), $E_{steel} = 2.1 \cdot 10^{11} \ N/m^2$ is the elasticity coefficient of the steel, D_{pp} is the piston pin diameter, m_{alt} is the mass with reciprocating motion, a is the instantaneous piston acceleration and L_0 is a characteristic length defined as:

$$L_0 = h_{pis} + L_c + \frac{S}{2} \tag{27}$$

where h_{pis} is the distance from the piston pin axis to the top surface of the piston, L_c is the connecting rod length and S is the stroke.

2.4.6. Heat transfer model

Concerning the heat transfer to the combustion chamber walls, in motoring tests (or during compression stroke and after the end of combustion), this process is essentially governed by convection (even though gas radiation to the walls also occurs, in those conditions this mode of heat transfer is negligible in comparison with convection [48]). While combustion is taking place, additionally to convection there is also radiation from the gas and the soot particles formed. There is no agreement with respect to the fraction of the heat transfer that is transferred by radiation: Morel and Keribar [49] obtained values ranging from 4% to 20%, whereas Heywood [11] states that this fraction can be higher than 20%. Anyway, any accurate radiative model would require the calculation of the soot formed in the spray [11, 50], which is out of the scope of this work.

In the present model, a variation of the expression proposed by Woschni [24, 51] is used to calculate the heat transfer coefficient. However, several efforts have been done in order to improve the original model proposed by Woschni. As described in detail in

two previous works [52, 53], the values of the constants in the model have been modified, and the way in which the swirl effect is considered is also different, so that more realistic predictions in direct injection Diesel engines can be achieved. Those improvements were performed by means of CFD calculations, as it is described in the cited references. The expression used for the heat transfer coefficient is:

$$h = CD^{-0.2}p^{0.8}T^{-0.53} \left[C_{W1} c_m + C_{W2} c_u + C_2 \frac{V_d T_{IVC}}{V_{IVC} p_{IVC}} (p - p_0) \right]^{0.8}$$
 (28)

where C = 0.12 and $C_2 = 0.001$, c_m is the mean piston speed, c_u is the instantaneous tangential velocity of the gas in the chamber (see [52] for the details of the calculation), C_{W1} and C_{W2} are constants whose values are fitted (along with k_{def}) from motoring tests on a specific engine, and p_0 is the motoring pressure assuming polytropic evolution. Finally, the calculation of the heat flux to the wall requires the estimation of the wall temperatures. A nodal model that calculates the mean temperature of the liner, piston

3. Experimental study: adjustment and validation

and cylinder head temperatures is used [54, 55].

In the following section the adjustment process for the different models is described as well as the validation performed. The experimental measurements were carried out in a high speed direct injection diesel engine with 2.0-liter displacement. It is a currently in production engine produced by an European manufacturer. It is a four-cylinder turbocharged engine equipped with a common rail injection system. Its main characteristics are given in Table 1. The engine was directly coupled to an electric dyno. An AVL tests system collects the mean variables (acquired at a constant sample frequency of 100 Hz) necessary for controlling the engine operating point and also for the combustion diagnosis. In-cylinder pressure was measured in one of the cylinders by means of a Kistler 6055B glow-plug piezoelectric transducer with a measurement range of 0-250 bar.

3.1. Adjustment process

Some of the models described in the previous section have a number of constants that have to be adjusted. In particular is is necessary to set the values of the heat transfer model constant C_{W1} , C_{W2} (section 2.4.6) and the deformation model constant k_{def} (section 2.4.5); also it is necessary to fit the values of the Wiebe constants in the combustion model described in section 2.4.4. The adjustment was done using experimental measurements obtained through a broad parametric study of the engine, both in motoring and combustion operation conditions as described below.

3.1.1. Mechanical and heat transfer characterization

The values of the constants C_{W1} , C_{W2} and k_{def} are fitted using experimental motoring tests performed on the engine to be modeled. The adjustment method is based on the sensitivity of the thermodynamic cycle to uncertainties in motoring conditions, as Lapuerta et al. studied [56, 57]. In motoring conditions, dHR is zero and the heat transfer to the walls, dQ, can be solved in the energy balance Eq. (10), which provides an "experimental" measurement of the heat transfer based on in-cylinder pressure [57]. An equivalent analysis can be done by imposing the heat transfer of the Woschni-like

model in Eq. (10), and then computing the error in dHR [56] (theoretically it should be zero in motoring conditions). The constant k_{def} has a characteristic effect on the thermodynamic cycle (both in terms of heat transfer to the wall and the error in dHR) and the heat transfer calculated with the Woschni-like model depends on C_{W1} , C_{W2} . It is thus possible to adjust the real value of the uncertainties so that the "experimental" heat transfer to the walls coincides with the results of the Woschni-like model and thus the error in dHR is minimised.

The characterization procedure is performed on a set of motoring tests with the typical range of the main Diesel engine parameters as detailed in the Table 2. The results obtained from the fitting process are $C_{W1}=1.95$, $C_{W2}=1.15$ and $k_{def}=2.64$.

3.1.2. Combustion model adjustment

The values of parameters $d_{comb,i}$, SOC_i , β_i and m_i in equation 22 are obtained by fitting the rate of heat release to experimental ones obtained with the combustion analysis tool CALMEC [34]. With the aim of providing more generality to the mathematical Wiebe combustion model, phenomenological criteria proposed by Serrano et al. [47] was followed in order to diminish the number of parameters to fit:

```
\beta_4 = 1 - \beta_1 - \beta_2 - \beta_3
376
            \beta_1 > 0, \, \beta_2 > 0, \, \beta_3 > 0
377
            C_i = 6.9 \ \forall i
378
            SOC_1 < SOC_2 < SOC_3
379
            SOC_3 = SOC_4
380
            d_{comb,2} < 25 cad
381
           d_{comb,4} = 130 \text{ cad}
382
           m_i > 0 \ \forall i
383
            m_2 = 0.8
384
```

The value of the completion parameter C_i is 6.9 so that the percentage of energy released within the combustion duration is 99.9%. This simplifications has enabled to reduce the number of adjusting parameters from 20 to 12. The rest of parameters were adjusted in 25 steady operating points (at different speed and loads) distributed in the complete engine map described in Table 3. The operating points used for the model fitting are one half of the complete experimental matrix while the other half is aimed to validate the experimental results in the following section. The Levenberg-Marquardt algorithm [58, 59] was used to perform the parameters fitting.

3.2. Validation of results

For the model validation the complementary operating points to that used for the model fitting were used. Both fitting and validation operating point are equally distributed in the engine map, thus the values of the combustion model parameters in the validation operating points were interpolated from values obtained in the fitted operating points, as described in the previous section.

Figs. 1, 2 and 3 show the evolution of in-cylinder pressure, heat release for three different test points with a single injection (Fig. 3), a multiple injection (Fig. 2), both without EGR, and a case with EGR and two injections (Fig. 1), respectively. As can be seen, the agreement between measured and predicted heat release laws and cylinder traces is good. Ignition delay as well as premixed and diffusion combustion phases are well predicted.

To illustrate the general behaviour of the model, Figs. 4, 5 and 6 show the maximum in-cylinder pressure, the pressure at exhaust valve opening and indicated mean effective pressure for all operating points used for both the model adjustment and model validation. The maximum pressure is a good parameter to account the performance of the model during the compression and the first stage of the combustion, similarly, the pressure at the EVO is a good indicator of the model behaviour during the last part of the combustion and expansion. Finally, the IMEP is the key parameter in all engine analysis and summarises well the general accuracy of the model. As can be seen in the stated figures, once the different models are adjusted using a first series of experimental tests ("Model fitting" in the figure legends), it is confirmed that the predictive character of the proposed tool is good ("Model validation" in the figure legends). In fact, the global error is just slightly higher in the validation tests. The difference can be attributed to the rate of heat release simulation; if a more accurate combustion model was used, the results would improve.

From the practical point of view, to get accurate results in agreement with experimental values is the key issue of a predictive model. However, it is also interesting to estimate which is the contribution of each submodel to the global accuracy to identify the main phenomena. With this purpose a short parametric study was performed. Its objective is to analyse the sensitivity of the results to different parameters included in the proposed submodels and its comparison with the effect of usual engine and operating parameters (compression ratio and injection timing). Six parameters were separately modified: blow-by rate, the constants of the heat transfer model $(C_{W1} \text{ and } C_{W2}), k_{def}$, compression ratio and start of injection (SOI), also the effect of including the injection rate was considered. A mean load (imep=12.5 bar) and mean engine speed (2000 rpm) operating point was used for the study. For each parameter (except fuel injection) three values were assumed as shown in Table 4. The effect of the injected fuel was considered by means of the cycle simulation with and without including this term (in both the cases the rate of heat release was considered). The parameter variation ranges were set according to realistic experimental and modelled results in real engines. Although it is difficult to ensure the coherence of the variation of such different parameters, it was checked that the result sensitivity is almost linear and thus it is easy to extrapolate if different parameter range were considered. The values of C_{W1} and C_{W2} were changed simultaneously; the blow-by percentage is referred to trapped mass at IVC. The effect of each parameter variation on indicated pressure and maximum pressure are shown in

The global trends obtained can be easily justified as follows. If no injection fuel is included the cooling effect of the evaporation is not taken into account and the specific heat of the gas in the chamber will be underestimated as the specific heat of the fuel is higher than those of the air and burnt products, thus higher imep and maximum pressures are obtained. The effect of advancing the SOI will produce a higher pressure peak and also a slightly higher imep value due to the change in the indicated diagram. The increase of the heat transfer parameter leads to higher heat losses and thus both imep and pressure peak diminish. If the engine deformation increases, the real compression ratio will diminish and lower in-cylinder pressures will be reached, thus diminishing the indicated performance of the engine. Similar conclusion can be stated if a higher blow-by leakage is considered. Finally, to increase the compression ratio has the opposite effect than the deformations: the higher r_c is the higher the peak pressure and imep are due

to the change in the instantaneous volume and indicated cycle.

At this point it is interesting to highly that the most influencing parameters for the imep prediction are the heat transfer, the blow-by leakage and the fuel injection and all of them are more important that the variation of the compression ratio or SOI. The heat transfer coefficients C_{W1} and C_{W2} have an important effect on the engine performance as they affects directly the heat transfer. It can be stated that if no heat transfer were considered, the error in the imep prediction is similar to varying the compression ratio about two points or modifying the SOI 3 cad. Concerning the blow-by, even if a lower leakage of about 2% (which is completely normal in real engines) is considered, its effects is higher than changing one point the compression ratio and similar to change 2 cad the SOI. Similar importance has the consideration of the injected fuel. Regarding the peak pressure sensitivity, the relative effect of these three parameters (fuel injection, heat transfer coefficients and blow-by) is not so big as in the case of imep, anyway they have a similar weight than compression ratio or SOI. In this case it is also clear that the deformation has an important effect similar to blow-by or fuel injection.

From this comments it is easy to conclude that all the described specific submodels are important to account for the corresponding phenomena, and if one or several of them were not included, the accuracy of the predictions shown in Figs. 4, 5 and 6 would be clearly worse.

4. Conclusions

In the proposed 0D thermodynamic model, the mass and energy conservation equations are solved in order to obtain the instantaneous gas state in the combustion chamber. A detailed description of the sub-models used for the calculation of each term appearing in the energy balance has been provided. In particular, specific sub-models to reproduce the blow-by leakage, the chamber deformation due to pressure and inertial efforts, the heat transfer to the chamber walls and the fuel injection have been presented.

Some constants in the stated sub-models needed to be adjusted and a thermodynamic-based procedure has been described in short. Also the combustion heat release laws needed to be estimated so that the model could have accurate predictive capability. The methodology followed is based on the fitting of several Wiebe function and allowed to estimate the rate of heat release in all the engine operating points.

Once the model adjusted, it has shown to be able to accurately reproduce the incylinder pressure evolution in a complete matrix of combustion tests in a DI Diesel engine.

Finally, it has been justified the necessity of including the proposed submodels for accurate prediction of the engine performance and it has been quantified the error in the simulation results if such submodel were not included.

5. Acknowledgements

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493 References

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621 List of figures

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- $_{623}$ $\,$ Fig. 1. Experimental and simulated RoHR and pressure at 2000 rpm, 80 Nm (25% load),
- $_{624}$ $\,$ 15% EGR and 2 injections.
- $_{625}$ Fig. 2. Experimental and simulated RoHR and pressure at 3000 rpm, 280 Nm (100% $_{\rm }$
- 626 load) and 2 injections.
- $_{627}$ Fig. 3. Experimental and simulated RoHR and pressure at 4000 rpm, 120 Nm (50% $_{628}$ load) and single injection.
- ⁶²⁹ Fig. 4. Experimental and simulated maximum pressure.
- ⁶³⁰ Fig. 5. Experimental and simulated pressure at the exhaust valve opening.
- Fig. 6. Experimental and simulated indicated mean effective pressure.
- Fig. 7. Effect of the parameters on imep and maximum pressure.

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- $_{\it 636}$ Table 1. Engine main characteristics.
- Table 2. Motoring tests.
- 638 Table 3. Combustion tests.
- Table 4. Parameters variation.

Table 1: Engine main characteristics

Displaced volume	2.0	[l]
Cylinders	4	
Bore	85	[mm]
Stroke	88	[mm]
Connecting rod length	145	[mm]
Compression ratio	17.8	
Maximum power	93	[kW]

Table 2: Motoring tests.

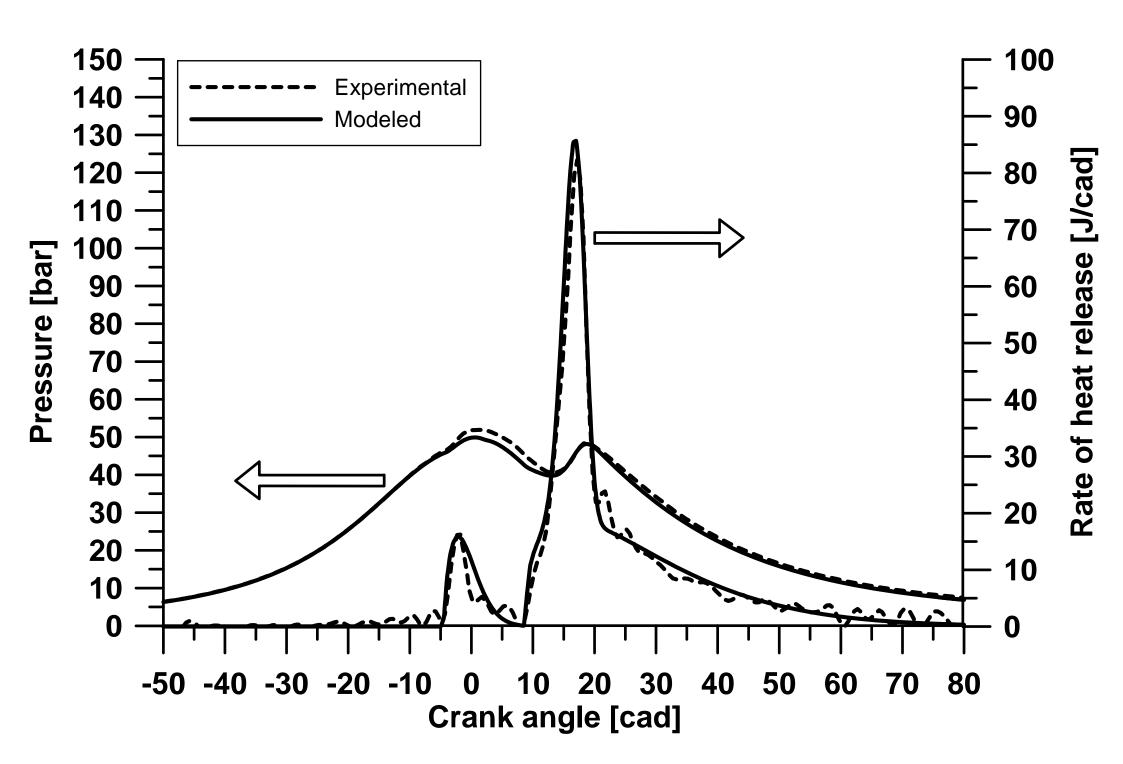
	1	2	3	4	Units
Speed	1000	2000	3000	4000	[rpm]
Inlet pressure	1026	1166	1129	1190	[mbar]
Inlet temperature	36	49	49	56	$[^{\circ}C]$
Exhaust pressure	1115	1720	1620	1841	[mbar]
Exhaust temperature	32	31	32	32	$[^{\circ}C]$

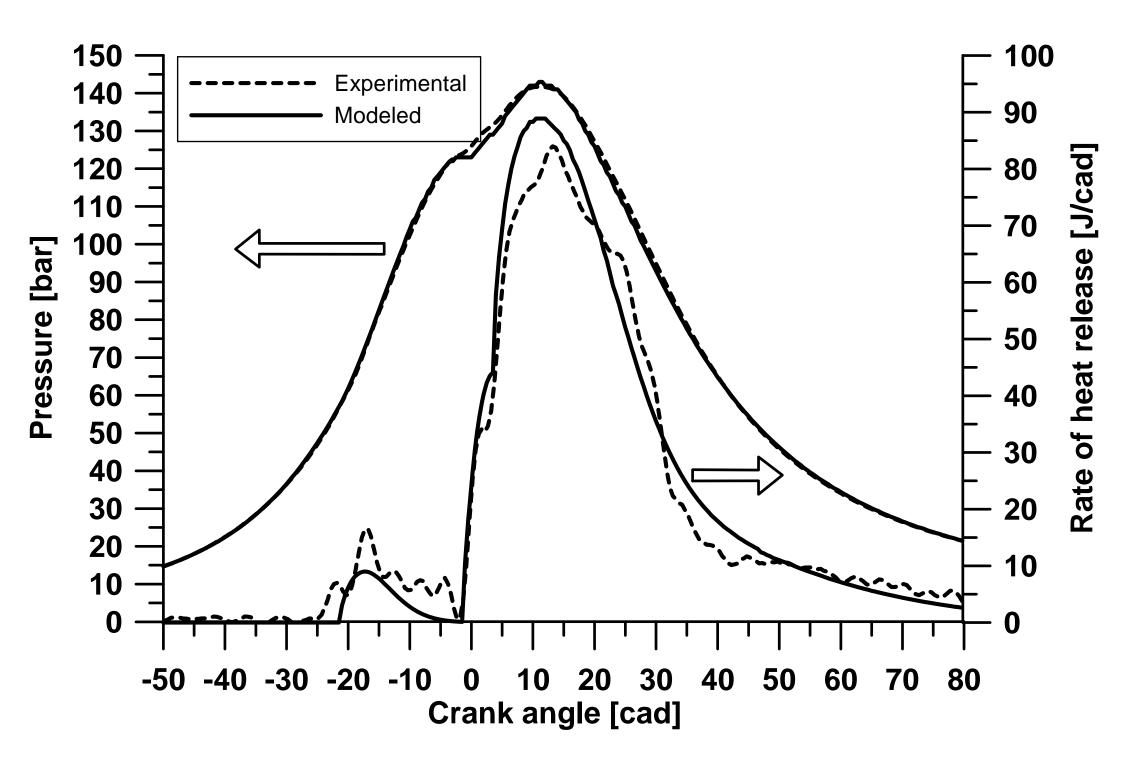
Table 3: Combustion tests.

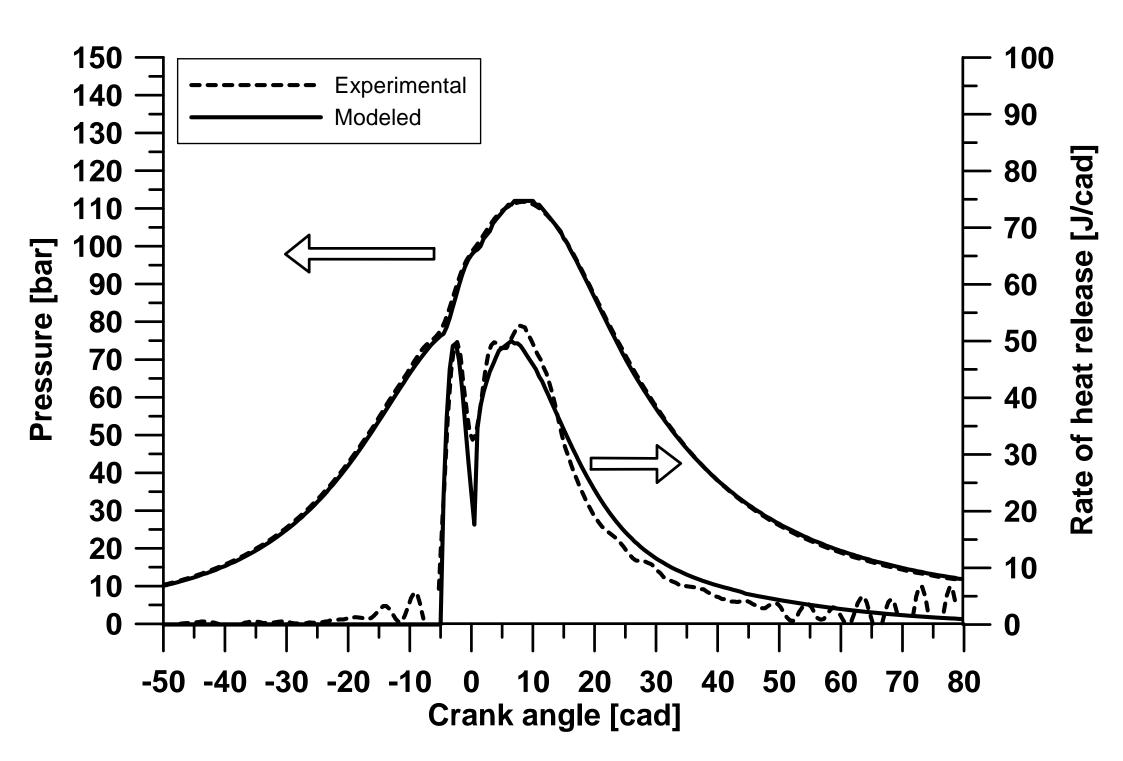
Speed	1000-4500 (steps of 500)	[rpm]
Load	13-100	[%]
Torque	40-320 (steps of 40)	[Nm]
EGR	0-35	[%]
Rail pressure	325-1600	[bar]
Injections	1-2	[-]

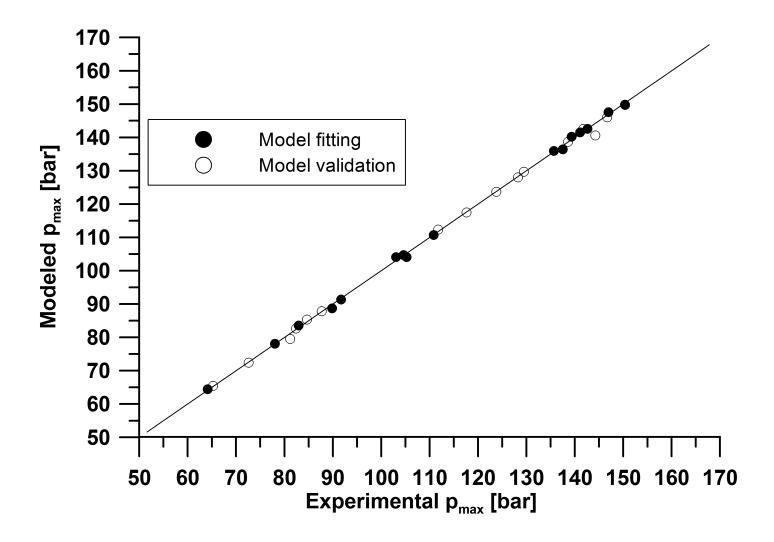
Table 4: Parameters variation. Simulation

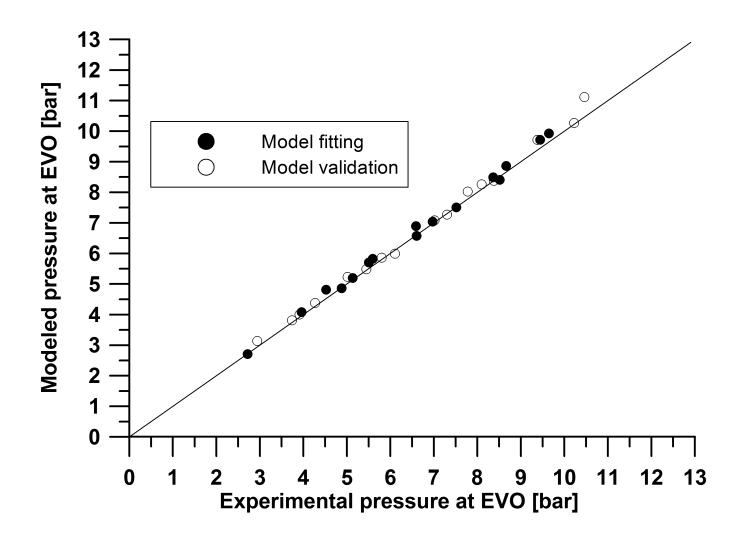
	Simulation			
	1	2	3	
m_{bb}	0%	2%	4%	
C_{W1}	0.85	1.7	2.55	
C_{W2}	0.5	1	1.5	
k_{def}	0	2	4	
r_c	15.3	16.3	17.3	
SOI	-2.4	-3.4	-4.4	

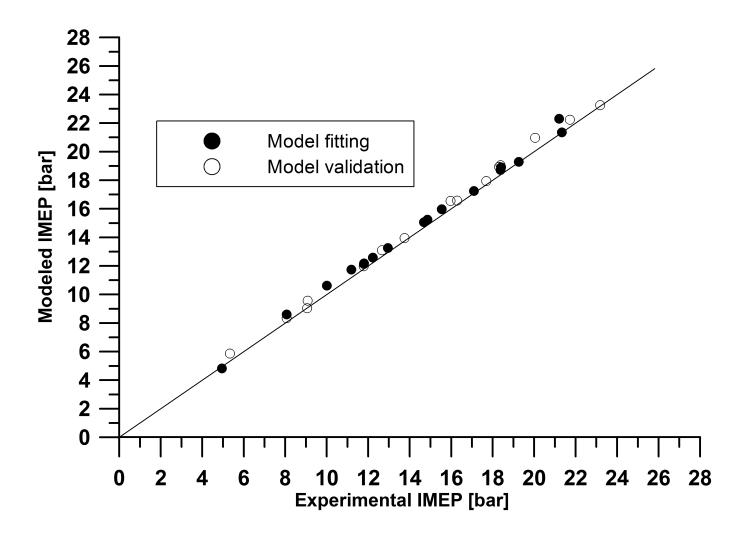


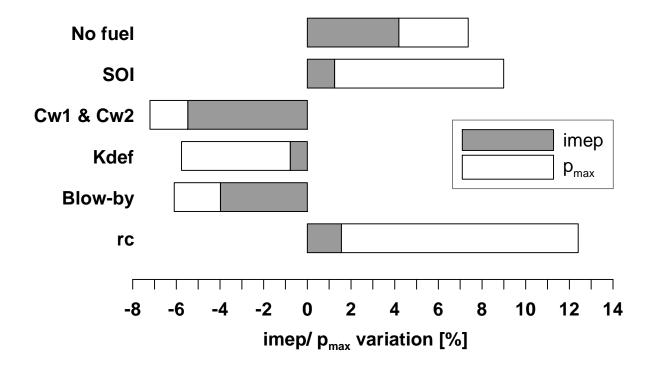
















Valencia, February 2011

Dear Editor,

Please find enclosed the modified manuscript and figures of our article entitled "Complete 0D thermodynamic predictive model for Direct Injection Diesel engines", by Francisco Payri, Pablo Olmeda, Jaime Martin and Antonio García to be considered for publication in the Journal Applied Energy

In the following pages we have included all the comments made by the reviewers together with our detailed answers to their points.

We hope that the modified version of the paper will be considered worth publishing in your Journal.

Yours sincerely,

Dr. Jaime Martín

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Ref: APEN-D-10-01212

Title: complete 0D thermodynamic predictive model for Direct Injection Diesel engines

Authors: Francisco Payri, Pablo Olmeda, Jaime Martin and Antonio García

Reviewer #1:

First of all, we would like to thank you for your comments. They helped us to address some weak points in the initial manuscript.

In the following paragraphs we try to answer more specifically your comments:

Comments:

This paper was aimed to develop a 0D thermodynamic model that takes into account the heat transfer, the blow-by leakage, the fuel injection and engine deformations. The objective of this study seems to be rather straightforward. I recommend the authors to have a careful consideration on the points as described below.

1. The author claim the effect of blow-by leakage is considerable in small DI diesel engines (0.35L of displacement), but not important in normal operating conditions. However, the experimental study was carried out in a diesel engine with 2.0L of displacement. What is the point of necessity in considering the blow-by leakage?

Extending the information stated in the section "Experimental study: adjustment and validation", the authors would like to underline that the experimental study was carried out in a 4 cylinder high speed direct injection diesel engine with 2.0 liters of total displacement, therefore, the displacement per cylinder is 0.5 liters. Thus the considerations performed in the paper regarding with the importance of blow-by effects on performance in small DI diesel engines is completely applicable in the case of study.

On the other hand, the necessity in considering the blow-by leakage in the 0D model developed is proved with a parametrical study of sensitivity included in section 3.2. As it is shown in the new figure 7, one of the most relevant parameters to predict accurately the Imep in a HSDI is the blow-by. More information related to this topic is answered in the following question.

- 2. How about the validation of results if the effect of heat transfer, the blow-by leakage, the fuel injection and engine deformations are not individually considered in the 0D thermodynamic model? Which is the most important influence? I think the author could further discuss it to show the importance of considering these sub-models.

 Taking into account the reviewer comments a short sensitivity study has been included to illustrate the effects of the submodels in comparison with usual engine parameter and operating conditions variations (SOI and compression ratio). This study includes a new table (Table 4) and a new figure (Fig 7) and some explanations in the last paragraphs of section 3.2 Validation of results.
- 3. The legends of Fig. 1~3 only showed the experimental curve and modeled curve. However, it is not easy to distinguish which curve is related to variation of pressure or rate of heat transfer.

According to the reviewer comment, some arrows have been included in Figs. 1-3 to make easier the differentiation between pressure traces and rate of heat release evolutions. In the authors' opinion the information is now clear, anyway, the editor or reviewer can suggest any additional change if necessary.

4. The description of Fig. 4~6 was not found in manuscript.

There was a mistake in the text at line 400. In the previous version the text stated: "To illustrate the general behavior of the model, Figs. 1, 2 and 3 show the maximum in-cylinder pressure..." and it should have stated "To illustrate the general behavior of the model, Figs. 4, 5 and 6 show the maximum in-cylinder pressure..." The correction has been done.





5. I recommend the author to replace some references with recent literature because most references in manuscript were published before year 2007.

Considering the reviewer advice the references have been updated, including several papers published after year 2007:

- 1. Wang X, Huang Z, Zhang W, Kuti OA, Nishida K. Effects of ultra-high injection pressure and micro-hole nozzle on flame structure and soot formation of impinging diesel spray. Appl Ener 2011;88:1620-1628.
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Reviewer #5:

Thanks for your helpful comments. You suggestions have brought a lot of light to achieve a more complete paper. Let's give you an answer to your different comments

Comments:

1. There seems many studies in this area as the references listed in the manuscript. I would like to author to clearly address what the major new scientific contributions of this paper compared to previous studies. I also feel author shall update the research literature to give the most recent results in this area.

In the introduction, the following sentence has been changed accordingly to this comment: "The objective of the present work is to develop a 0D single-zone thermodynamic model covering all the previous statements (the simulation of the rate of heat release law, heat transfer and gas properties sub-models as well as blow-by leakage and the fuel injection contributions to the diesel engine cycles) that have been partially covered by previous researchers." The references has been updated (see answer to question 5).

2. A comparison of the method in this paper to other performance evaluation of a engine shall be discussed. In particular, it needs further address how this method can be applied in the real application?

The reviewer consideration is well addressed; however, the authors would like to underline that in sub-section 3.2 the model presented in the paper is validated versus experimental measurements derived from an in-cylinder pressure sensor. Specifically, in figures 1, 2 and 3 is possible to prove how is the temporal evolution of in-cylinder pressure and heat release for three very different engine operating conditions. Thus, as is stated in the paper, the agreement between measured and predicted cylinder traces is good and therefore, in a first stage, the applicability of the developed model in real engine conditions is shown.

Continuing with the answer and with the aim of better illustrating the general and real behavior of the model, in figures 4, 5 and 6 are shown the maximum in-cylinder pressure, the pressure at exhaust valve opening and the indicated mean effective pressure for all operating points used for both the model adjustment and model validation. Once again, considering these parameters, the accuracy of the model in real conditions is proved.