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Olmeda González, PC.; Martín Díaz, J.; Novella Rosa, R.; Carreño, R. (2015). An adapted heat transfer model for engines with tumble motion. *Applied Energy*. 158:190-202. doi:10.1016/j.apenergy.2015.08.051.



The final publication is available at

<http://dx.doi.org/10.1016/j.apenergy.2015.08.051>

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Additional Information

A new heat transfer model for engines with tumble motion

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Abstract

In the last years, a growing interest about increasing the engine efficiency has led to the development of new engine technologies. The accurate determination of the heat transfer across the combustion chamber walls is highly relevant to perform a valid thermal balance while evaluating the potential of new engine concepts. Several works dealing with heat transfer correlations that consider the swirl motion are found in the literature; however, there is a lack of works dealing with heat transfer correlations which take into account the effect of the tumble movement. In this work, a new heat transfer model accounting for the tumble motion is presented. A two stroke HSDI Diesel engine with high tumble and no swirl is used to perform the theoretical study, the model development and its final calibration. Initially, a theoretical analysis of the gas movement phenomena is carried out based on CFD results and then, a model is developed and calibrated based on a skip-fire testing technique. Finally, a sensitivity study focused on evaluating the model robustness is performed. The results confirm an average RMSE reduction of 70% with respect to the Woschni model, being this consistent improvement qualitatively evidenced in the instantaneous heat transfer evolution.

Keywords: Combustion Diagnosis, Engine Heat Transfer, Tumble

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Nomenclature

α	Crank Angle	[$^{\circ}$]
c_m	Piston mean speed	[m/s]
c_u	Tangential vortex speed	[m/s]
D	Cylinder diameter	[m]
ρ	Density	[kg/m^3]
$\Delta\alpha$	Angular duration	[$^{\circ}$]
γ	Adiabatic exponent	[-]
h	Heat transfer coefficient	[J/kg]
η_{tr}	Trapping ratio	[-]
k	Conductivity	[W/mK]
\dot{m}	Mass flow rate	[kg/s]
p	In-cylinder pressure	[bar]
Q	Accumulated heat transfer	[kW]
\dot{Q}	Heat transfer rate	[$J/^{\circ}$]
R	Specific gas constant	[J/kgK]
RoHR	Rate of Heat Release	[$J/^{\circ}$]
S	Engine stroke	[m]
T	Temperature	[K], [$^{\circ}C$]
V	Volume	[m^3]
v	Velocity	[m/s]
μ	Dynamic viscosity	[$Pa\cdot s$]

Abbreviations

ATDC	After Top Dead Centre
BDC	Bottom Dead Centre
BTDC	before Top Dead Centre
CFD	Computational Fluid Dynamics
CI	Compression Ignition
EGR	Exhaust Gas Recirculation
HCCI	Homogeneous Charge Compression Ignition
HSDI	High Speed Direct Injection
HT	Heat Transfer
ICE	Internal Combustion Engine
IGR	Internal Gas Recirculation
IVC	Intake Valve Closing
IVO	Intake Valve Opening
PCCI	Premixed Charge Compression Ignition
SI	Spark Ignition
TDC	Top Dead Centre
TR	Tumble Ratio

1. Introduction

The global awareness towards the greenhouse gases emissions has led to a more stringent ICE emissions legislation, thus focusing the automotive researchers and manufacturers attention on the development of cleaner and more efficient powertrains. In the last years, the efforts have been mainly focused on the reduction of the NO_x and soot emissions by means of different injection strategies [1], high pressure fuel injection systems [2], multiple injections [3], high boost pressure [4], exhaust gases recirculation (EGR) [5], variable valve timing [6], high swirl [7, 8] and tumble ratios [9], new clean fuels [10, 11] or after treatment systems [12]. Nowadays, there is an increasing interest towards the optimization of the fuel consumption, and hence the reduction of the CO_2 emissions [13]. To comply with the upcoming requirements, new combustion concepts such as HCCI [14] and PCCI [15], and new automotive engine concepts such as downsizing [16, 17], and two-stroke engines [18] are in the centre of the research. The air management is a common factor in these works, since it is a key issue to improve the air-fuel mixing process and achieve faster burning rates [19], and therefore modern ICE are designed to generate high vorticity and turbulence in the combustion chamber.

There are several methodologies aimed at the evaluation of specific engine technologies or operation strategies, being some of the most widespread the combustion diagnosis based on in-cylinder pressure evolution [20] and the energy balances [21]. On the one hand, the combustion diagnosis evaluates the combustion performance by analysing the rate of heat released (RoHR), which is obtained by solving the first law of thermodynamics and the equation of state [22]. On the other hand, the energy balance determines the chemical fuel energy distribution in the different engine subsystems, being the most important terms the brake power, the exhaust gases energy and the heat transfer from the combustion chamber [19]. Thus, it is evident that both methodologies requires the accurate determination of the heat transfer (HT) across the combustion chamber walls, which can be carried out either by experimental or modelling methodologies. The main drawback for the experimental determination is the necessity of a special instrumentation, usually not available in a standard test bench [21]. Therefore, the HT modelling is interesting as an alternative approach, whose complexity and time consuming depends on the specific application.

The aforementioned applications demand HT models with low computational effort (with time consumption of minutes or even seconds). Thus, a lot of proposals dealing with the HT estimation in ICE can be found in the literature for both, compression ignition (CI) and spark ignition (SI) engines. Some of the most widespread correlations for HT coefficient calculation are those based on the well-known Woschni [23], Annand [24] and Hohenberg [25] works. To make these models suitable for a specific application, they must be adjusted based on different techniques such as CFD modelling [26], experimental measurements [27, 28], or thermodynamic assumptions [29]. In general, these models correlate the HT coefficient with the thermodynamic state of the gases (pressure and temperature), the engine geometry (bore and stroke) and the gas movement (gas velocity and flow pattern), being this last parameter the most

35 difficult to assess due to the unsteady nature of the gas motion, since it is a turbulent 3D flow with no clear symmetry
36 of the gas properties along the chamber.

37
38 The main rotative macro structures that can be found in ICE are the swirl and tumble, being differentiated by
39 their rotary axis (swirl rotates in the cylinder axis and tumble in the diametrical axis). Both are generated during
40 the intake process and evolve in the compression-expansion stroke thanks to the engine geometry (particularly ports
41 and combustion chamber configuration). The swirl movement is prompted by chamber configurations consisting of a
42 shallow bowl engraved into the piston crown [19], whilst the tumble movement is enhanced by pentroof combustion
43 chambers [30, 31]. The swirl characteristics have been widely studied, and most of the HT correlations include a term
44 accounting for its contribution to the characteristic gas velocity used to calculate the HT coefficient [23, 26]. How-
45 ever, there is a lack of HT correlations dealing with the effect of the tumble characteristics on the HT coefficient. The
46 tumble is usually evaluated by means of experimental techniques such as LDV and PIV [32, 33], involving several
47 engine modifications, or CFD modelling [34, 35] leading to a long computational time. Alternatively, there are some
48 quasi-dimensional models [30, 36], which are faster than the CFD models, but require a detailed knowledge of the
49 fluid motion pattern and also of the engine ports and combustion chamber geometry. This specific information is not
50 usually available for combustion diagnosis, and hence leads to a lack of generality of the models for be applied to in
51 different engines.

52
53 In this work, a generally applicable HT model that includes the gas velocity evolution produced by a high tumble
54 movement is proposed. For the model development, a research CI two-stroke HSDI engine with high tumble motion
55 and no swirl was experimentally characterized. The fluid motion analysis and model development was carried out
56 on the basis of CFD simulations, whereas the calibration of the model was carried out in a specific experimental
57 installation, specially developed to perform skip-fire tests.

58 **2. Methodology**

59 The objective of this work is the determination of a semi-empirical HT model, which takes into account the tumble
60 flow motion pattern in the combustion chamber. The model was developed following three main stages as shown in
61 Figure 1. In the next paragraph, the definition of each step are explained:

- 62 – Theoretical analysis: a well-grounding model requires a first deep understanding of the phenomena involved,
63 and thus a comprehensive bibliography review regarding the tumble generation and dissipation processes, and
64 the influence of the geometrical and operating parameters was carried out. In order to analyse the instantaneous
65 gas evolution in the specific engine under investigation, CFD simulations of the complete skip-fire cycle were
66 used. The results are discussed in detail with the aim of identifying the operating parameters that must be taken
67 into account by the model.

- 68 – Model development: starting from the CFD results, a semi-empirical model able to reproduce the gas velocity
69 dissipation was developed. The model must be suitable for combustion diagnosis, and therefore it considers
70 the main mechanisms involved in the tumble generation (geometry and operating conditions), but is still simple
71 and keeps low computational cost in terms of power and time. The input parameters required, consist mainly of
72 those mean values generally acquired in a test bench.
- 73 – Model calibration: this stage was carried out based on skip-fire measurements, which consist of skipping the
74 injection of one cycle, thus obtaining a motoring measurement with the air management of a conventional
75 combustion. The calibration focuses on the determination of the fitting constant values, with the objective of
76 reducing the difference between the experimental HT (obtained through the application of the first law in the
77 combustion chamber) and the modelled HT. The skip-fire tests are used instead of motoring measurements to
78 reproduce a realistic intake process, being this specially important to ensure the scavenging process in two-
79 stroke engines [37, 38], and also to have the same thermal properties (temperatures and gas composition) in the
80 chamber than those of a conventional combustion operation, being this critical for the HT process.

81

82 Once the model is calibrated, it is compared with a model widely accepted as a reference by the scientific
83 community. For this purpose, the Woschni HT model [23, 39], which has been widely evaluated [40] and
84 used as a start point in several works [26, 27] has been selected. The objective is to evaluate the effect of the
85 tumble gas velocity evolution on HT. Finally, a sensitivity study is carried out with the objective of evaluating
86 its robustness against the effects of possible experimental and calibration uncertainties, being this information
87 important when the model is transferred to different engines.

88 **3. Experimental and Theoretical tools**

89 A research single-cylinder engine, with high tumble induction and no swirl was selected to perform the CFD
90 simulations and the experimental tests. Initially, the engine architecture and the test cell characteristics are described.
91 Subsequently, the details of the CFD simulation, along with the model validation are discussed.

92 *3.1. Engine architecture and hardware*

93 The experimental activities were performed on a single-cylinder, research version of a 2-stroke HSDI diesel engine
94 prototype. The main engine geometrical characteristics are presented in Table 1. The cylinder head and combustion
95 chamber geometry, shown in Figure 2, were completely adapted to ensure a suitable in-cylinder flow pattern, in order
96 to optimize the scavenging of burnt gases and to reduce the short-circuit losses of fresh air going directly from the
97 intake into the exhaust ports. The cylinder head geometry presents a staged roof for baffling the flow of air between
98 the intake and exhaust valves, forcing the air to follow the path of the cylinder wall toward the bottom of the cylinder.
99 This geometry provides the best compromise between scavenging efficiency, acceptable permeability, and convenient

100 combustion chamber geometry [18].

101

102 The engine is equipped with an hydraulic cam-driven variable valve timing system, allowing a flexibility of 30°
103 on both intake and exhaust valve timings independently from the mechanical cam timing. The optimum camshaft
104 configuration, presented in Table1, was experimentally defined by testing different opening durations and maximum
105 lifts in a medium load and medium speed operating point. The trapping ratio η_{tr} was experimentally determined by
106 means of a tracer gas method [41, 42]. It consist on injecting a controlled quantity of CH_4 (tracer gas) along with the
107 intake flow, and then measure the CH_4 concentration at the intake and exhaust manifolds. It is assumed that the CH_4 is
108 homogeneous mixed with the intake air, and the mass trapped in the cylinder was completely burn during combustion,
109 therefore the CH_4 ratio at intake and exhaust is an indicator of the short-circuited air mass. Thus, an accurate estima-
110 tion of the short-circuited mass was performed by means of mass balances at the intake, exhaust and in-cylinder gases.

111

112 The total trapped mass in the cylinder is a key issue to accurately obtain the IVC thermodynamic conditions,
113 required for both CFD calculations and experimental HT determination. It is given by the addition of the trapped air
114 mass after the IVC, and the internal gas recirculation (IGR), defined as the fraction of residual gases retained from the
115 previous combustion cycle in the total trapped mass in the cylinder. The effective air/fuel ratio is calculated from the
116 mass of usable air available in the trapped charge, excluding the air in the IGR since it is not really usable, divided
117 by the injected fuel mass. The IGR ratio and the total trapped mass are estimated using simplified thermodynamic
118 calculations. This estimation is based on an enthalpy balance, where the enthalpy of the total trapped mass at the
119 IVC equals to the enthalpy of the residual mass plus the enthalpy of the intake delivered trapped mass (fresh air plus
120 external EGR), both estimated also at the IVC.

121 3.2. Test cell characteristics

122 The engine was assembled into a fully instrumented test cell, whose scheme is shown in Figure 3. The engine is
123 fed with the compressed air provided by an external compressor, which operation was set to simulate boost conditions.
124 A throttle valve located at the exhaust line after the exhaust settling chamber is used to simulate the back-pressure
125 that would be produced by the turbine. The experimental cell also includes a low pressure EGR system, designed to
126 provide arbitrary levels of cooled EGR even at very high intake boost pressures. Water and oil cooling circuits are
127 also independent from the engine, and temperatures are strictly controlled and monitored during all the experimental
128 tests. The fuel consumption of the engine is measured with a gravimetric dynamic fuel meter.

129

130 Pollutant emissions from the engine are sampled in the exhaust line, close to the exhaust settling chamber and
131 routed to a state-of-the-art gas analyser by a heated pipe to ensure gas temperatures above 150 °C. Measurements of
132 CO_2 at intake and exhaust (to determine the EGR rate) are performed for all the tests, and a dedicated CH_4 analyser is
133 used to trace both intake and exhaust CH_4 concentrations when performing η_{tr} measurements. Most relevant averaged

134 test cell and engine parameters, pollutant emissions, and high frequency instantaneous signals are recorded using a
135 dedicated data acquisition system. The high frequency signals are acquired using an oscilloscope synchronized with
136 an optical angular encoder with a resolution of 0.2° . The cylinder pressure is measured using a piezoelectric sensor,
137 while a piezo-resistive pressure sensor is placed at the cylinder liner near to the BDC, with the objective of reference
138 the piezoelectric sensor differential pressure signal. The combustion diagnosis and the determination of relevant pa-
139 rameters such as: indicated mean effective pressure (imep), combustion phasing, maximum cylinder pressure, rate of
140 heat release (RoHR), etc. are calculated from the in-cylinder pressure by means of an in-house combustion diagnosis
141 software named CALMEC [22, 43].

142

143 To perform the skip-fire test, a special electronic device consisting of an electronic controlling box and a dummy
144 injector was implemented. Its operation consist on deviating a injection command pulse to the dummy injector ev-
145 ery determined number of injections; therefore, there is no injection nor combustion in the cycle, thus obtaining a
146 motoring test with the same thermal, intake and scavenging process of a nominal combustion cycle. The main draw-
147 back of the skip-fire is that once a combustion is skipped, the following cycles are faulty, which results in abnormal
148 combustion or even in misfire. This issue is solved by waiting until the engine operation returns to normality, which
149 occurs about 5 cycles after the skip-fire cycle, as can be seen in Figure 4. In this work, 15 cycles (14 combustion and
150 1 motoring) were measured to ensure complete engine stabilization. Since 20 cycles are required in order to have a
151 confident averaged cycle [20], the measurement of 300 cycles is required.

152

153 The experiments performed consist on a combination of engine speed and $\Delta p = p_{in} - p_{out}$ sweeps, being p_{in} and
154 p_{out} the inlet and outlet pressures. These parameters were varied owing to their major influence on the intake and
155 scavenging processes, i.e. low speed and Δp implies longer intake time but also lower air flow rate at intake, on the
156 counterpart, high speed and Δp has the opposite effect. The combination of the effects of these parameters are related
157 with the scavenging efficiency of the exhaust gases and the short-circuit. Therefore, the engine speeds tested are 1200,
158 1500 and 1800 rpm and Δp are 300, 400 and 500 *mbar*. The complete experimental matrix is presented in Table 2.
159 It provides a complete data base of the engine operating conditions, which allows performing a further calibration of
160 the model that optimise the results in a wide operating range.

161 3.3. CFD model

162 The computational model was built in the CONVERGE CFD platform. Full coupled open and closed cycle com-
163 putations using the full intake/exhaust and cylinder geometries were carried out since the combustion chamber is
164 non-symmetric. The computational domain at the intake valve closing (IVC) angle is shown in the left plot from Fig-
165 ure 2. The CFD code uses a structured Cartesian grid with base cell size of 3 mm. Three additional grid refinements
166 linked to flow velocity and temperature were performed by means of an adaptive mesh refinement as well as a fixed
167 three level refinement within the spray region.

168

169 The injection rate profile was generated from the experimental database available after the injector characteriza-
170 tion (mass flow rate and spray momentum flux) performed in dedicated test rigs. The diesel injection is simulated by
171 the standard Droplet Discrete Model. Diesel fuel physical properties are defined using Diesel 2 as surrogate. Spray
172 atomization and break-up are simulated by means of the KH-RT model. Turbulent flow is modelled by means of the
173 RNG $k - \varepsilon$ model with wall-functions in order to account for wall heat transfer. Concerning combustion modelling, a
174 direct integration of detailed chemistry approach was used by means of the CONVERGE code and the SAGE solver.
175 Finally, the chemical properties of Diesel fuel are defined using n-heptane as surrogate.

176

177 The set-up and validation of the CFD model was performed at the reference case (see Figure 5), operating with
178 the conventional Diesel combustion concept. The quality of the model was evaluated by comparing its combustion
179 and emissions results with those obtained experimentally in the engine as presented in Table 3. Figure 5 shows the
180 comparison between the CFD and experimental cylinder pressure and RoHR profiles. The CFD model performance
181 is considered as suitable for being used along the evaluation of the gas flow motion within the chamber as well as the
182 HT from the gas to the combustion chamber walls.

183

184 In order to increase the weight of the tumble motion on the HT, the original engine configuration was adapted
185 for the CFD calculation: aluminium was set as the piston material because its higher conductivity than steel, and the
186 intake settings were defined to enhance the tumble generation, achieving higher tumble ratios and subsequently higher
187 HT than the original configuration.

188 4. Theoretical analysis

189 The theoretical analysis starts with the description of the reference HT model, with the objective of identifying
190 how it considers the instantaneous gas velocity to calculate the HT evolution. Then, the main tumble generation
191 process according to the literature is presented, followed by the specific CFD simulation of the engine used for the
192 model development.

193 4.1. Reference heat transfer model

194 The HT to the combustion chamber walls in motoring or skip-fire tests (or during the compression stroke and after
195 the end of combustion) is essentially governed by convection, even though gas radiation to the walls also occurs, its
196 weight is negligible at these conditions in comparison with convection [24]. On the contrary, the radiation gain rele-
197 vance in combustion measurements since the formation of soot particles at high temperatures increases the radiation
198 emission. There is no general agreement with respect to the fraction of the HT that is transferred by radiation, Morel
199 and Keribar [44] obtained values ranging from 4% to 20%, whereas Heywood [19] states that this fraction can be

200 higher than 20%. An accurate radiative model requires the calculation of the soot formed in the spray [45]; therefore,
 201 it is usual to consider a HT coefficient that accounts for convection and radiation together.

202

203 The convective HT is determined by means of the Newton's law of cooling $Q = hA(T_g - T_w)$, where h is the heat
 204 transfer coefficient, A is the heat transfer surface area (i.e. the instantaneous combustion chamber area), and T and T_w
 205 are the spatially averaged gas and walls temperatures (i.e. piston, piston head and cylinder liner temperatures). The A
 206 value is geometrically defined, being its instantaneous value obtained from the kinematic analysis of the mechanism.
 207 T_g is estimated from the in-cylinder pressure, the ideal gas law and the trapped mass, meanwhile T_w requires specific
 208 models to their determination [20]. In this work, the trapped mass is determined by means of the mass balance de-
 209 scribed in Section 3.1, and T_w by means of a conductance lumped model [46, 47].

210

211 One of the most scientifically accepted expressions for the h determination is the Woschni correlation [23, 39].
 212 Woschni developed his model based on a dimensional analysis, where the relation between the Nusselt, Reynolds and
 213 Prandtl numbers (N_u , R_e and P_r respectively) is defined as:

$$N_u = a R_e^b P_r^c \quad (1)$$

214 being a , b and c constants. N_u and R_e are expressed in terms of the engine geometry and gas properties as $N_u = \frac{hD}{k_g}$
 215 and $R_e = \frac{\rho_g D v_g}{\mu_g}$, where D is the engine bore, k_g , ρ_g , v_g and μ_g are the conductivity, the density, the gas velocity and
 216 the dynamic viscosity of the gas respectively. k_g and μ_g depends on gas temperature according to $k_g \propto T_g^{0.75}$ and
 217 $\mu_g \propto T_g^{0.62}$. P_r of the air has a value about 0.7 in the working temperature range within the combustion chamber, thus
 218 P_r^c is usually considered constant. Rearranging in Equation 1, Equation 2 is obtained:

$$h = C D^{-0.2} p^{0.8} T_g^{-0.55} v_g^{0.8} \quad (2)$$

219 where C is a constant value, and the gas velocity is determined as:

$$v_g = C_{w1} c_m + C_{w2} c_u + C_2 \frac{V_d T_{IVC}}{V_{IVC} p_{IVC}} (p - p_0) \quad (3)$$

220 being C_{w1} , C_{w2} and C_2 constant values, c_m the mean piston speed, V_d the displaced volume, T_{IVC} , V_{IVC} , p_{IVC} the
 221 temperature, volume and pressure at IVC respectively, p is the instantaneous pressure and p_0 is the motoring pressure,
 222 obtained by assuming a polytropic evolution (note that in motoring conditions $p \equiv p_0$) and c_u is the tangential speed
 223 generated by the swirl vortex, which is defined as $c_u = \pi D N_T$, where swirl vortex frequency N_T is experimentally
 224 determined.

225

226 In the original Woschni equation, the c_u value was considered as constant. The reference model used in this work
 227 accounts for the impact of the instantaneous evolution of the swirl over c_u by means of a trigonometric function [26],

228 which simulates the vortex acceleration due to the upward piston movement.

229 *4.2. Tumble formation, evolution and dissipation*

230 The determination of the specific generation and dissipation timing as well as the mechanisms that enhance or
231 reduce the vortex formation requires dedicated experimental techniques [32, 33] or CFD simulations [34, 35]. More-
232 over, the tumble characteristics depend on the engine geometry and the operating conditions; however, there is a
233 general agreement regarding the main process. The vortex formation starts few crank angle degrees after the IVO,
234 when the high speed air goes into the chamber. Due to the combustion chamber geometry, the air is forced to sweep
235 the wall and tries to form a small vortex; however, at this stage the piston is in the proximities of the TDC and the
236 small size of the combustion chamber is not enough to stand for the vortex formation. In the early intake process
237 (scavenging process in two-stroke engines), both intake and exhaust valves are simultaneously opened, thus some air
238 is short-circuited as shown in Figure 6a, which also goes in detriment of the tumble generation. Figure 6b shows how
239 during the intake stroke a small vortex is generated and continuously accelerated thanks to the angular moment added
240 by the incoming air, this process lasts until the IVC, as shown in Figure 6c. The completely developed vortex is then
241 accelerated along the compression stroke as a result of the angular momentum conservation (since the vortex radius
242 decreases). At some stage of the compression stroke, the tumble is completely dissipated by the effect of friction
243 with the wall and the increasingly smaller vortex radius, which forces the formation of turbulent micro-structures as
244 depicted in Figure 6d.

245
246 Regardless the specific application, the gas movement inside the chamber has a major importance in the HT
247 coefficient calculation since the high rotational speed increases the HT before the TDC. Once the tumble dissipates,
248 the resulting high turbulence still contributes in a smaller extent to the HT, extending the gas velocity influence several
249 degrees after the TDC. It is interesting to highlight that, contrary to the tumble, the swirl is considered to be accelerated
250 in the compression stroke, and decelerated similarly in the expansion stroke, resulting in a symmetrical effect on the
251 HT. Thus, the most general swirl models cannot accurately determine the effect of the gas velocity on the HT for
252 engines with tumble, being necessary a model that considers the instantaneous evolution of the spatially averaged gas
253 velocity, taking into account the previous discussed characteristics. This objective is achieved in this work, based on
254 simplifications of the more complex 3D phenomena described in the next section.

255 *4.3. CFD results analysis*

256 The main features of the tumble motion can be identified in Figure 7, where the tumble ratio (TR) in the X-Y plane
257 (defined in Figure 6a) is sketched. Results obtained confirms how the TR in the Z-Y plane and the swirl ratio (SR)
258 in the X-Z plane are negligible. Figure 7 shows that the tumble formation begins in the piston upward stroke about
259 175° BTDC, reaching its maximum around 130° BTDC (before the IVC). It means that the tumble dissipation starts
260 in the compression stroke, which is in accordance with that found in the literature [32]. The tumble is completely

261 dissipated close to TDC, however, the TR value is close the zero several degrees before the TDC as observed in the
 262 detail presented in Figure 7.

263

264 Higher TR boost the HT from the gases to the walls by increasing the value of the last term in Equation 2. In
 265 Figure 8, the HT obtained in the CFD simulations (\dot{Q}_{CFD}) is presented, along with a HT calculated accounting only
 266 for the piston speed term (\dot{Q}_{c_m}), but neither swirl nor tumble. \dot{Q}_{c_m} is calculated by adjusting the constant C_{w1} of
 267 Equation 3 in the expansion stroke, keeping $c_u = 0$. This step was performed to assess the effect of the vortex (swirl
 268 or tumble) on the HT, by determining the spatially averaged gas velocity. The CFD spatially averaged gas velocity
 269 (v_{CFD}) presented in Figure 8 is the velocity generated by the tumble vortex. It is calculated from a HT coefficient h'
 270 obtained from the difference between \dot{Q}_{CFD} and \dot{Q}_{c_m} as presented in Equation 4, then v_{CFD} is obtained by combining
 271 Equations 4 and 2 as presented in Equation 5. It is interesting to notice how the v_{CFD} presented in Figure 8 becomes
 272 zero near to 30° ATDC, although the tumble has been completely dissipated. This can be explained by the turbulence
 273 generated as result of the tumble destruction, which has an important effect in the HT coefficient since it occurs in the
 274 proximities of the TDC, where the in-cylinder pressure and the temperature difference between gas and walls are the
 275 highest.

$$h' = \frac{\dot{Q}_{CFD} - \dot{Q}_{c_m}}{A} \quad (4)$$

$$v_{CFD} = \sqrt[0.8]{\frac{h'}{C D^2 p^{0.8} T^{-0.55}}} \quad (5)$$

276 The results obtained from the CFD model regarding the description of the shape of v_{CFD} , allows the development
 277 of a HT model that accounts for the instantaneous tumble velocity characteristics. The HT model proposed in this
 278 work is detailed described in the following section.

279 5. Model development

280 The last term in Equation 2 corresponding to the gas velocity v_g , is replaced by a tumble-generated gas velocity
 281 v_t , which is assumed to be proportional to a characteristic gas mean velocity \bar{v}_m during the intake process, a tumble
 282 gas velocity dissipation function f_w and the mean piston speed c_m as follows:

$$v_t = C_{t1} \bar{v}_m f_w + C_{t2} c_m + C_2 \frac{V_d T_{IVC}}{V_{IVC} p_{IVC}} (p - p_0) \quad (6)$$

283 where C_{t1} and C_{t2} are proportionality constants affecting f_w and c_m respectively. Starting from the results obtained
 284 for the gas velocity in the CFD simulations, it was found that an exponential Wiebe-like function (Equation 7) suitably
 285 follows the trend observed for the tumble gas velocity.

$$f_w(\alpha) = \exp^{a\left(\frac{\alpha-\alpha_0}{\alpha_f-\alpha_0}\right)^m} \quad (7)$$

where f_w is a S -shape function, with values between 0 and 1, m is a fitting constant used to adjust the shape, α is the crank angle, α_0 and α_f are angles related to the begin and the end of the gas velocity dissipation process, and $a = -6.907$ is a constant value, adjusted to ensure a v_t dissipation of 99.9% at $\alpha = \alpha_f$.

Considering the parameters that can affect the vortex formation, a mean gas velocity during the intake process \bar{v}_m was defined in Equation 8. It takes into account the air mass going into the cylinder \dot{m}_a , the intake process duration $\Delta\alpha_{IVO-IVC}$, the mean air density during the intake process $\bar{\rho}_a$, the trapping efficiency η_{tr} and the effective intake valve area A_{eff} .

$$\bar{v}_m = \frac{360 \eta_{tr} \dot{m}_a}{A_{eff} \bar{\rho}_a \Delta\alpha_{IVO-IVC}} \quad (8)$$

where $\bar{\rho}_a$ was calculated considering a mean gas in-cylinder temperature $\bar{T}_{cyl,int}$, estimated assuming an adiabatic process between the intake and the cylinder as follows:

$$\bar{T}_{cyl,int} = \bar{T}_{int} \left(\frac{\bar{p}_{cyl,int}}{\bar{p}_{int}} \right)^{\frac{\gamma-1}{\gamma}} \quad (9)$$

where \bar{T}_{int} and \bar{p}_{int} are the mean temperature and pressure at the intake settling chamber, $\bar{p}_{cyl,int}$ is the mean in-cylinder pressure during the intake process and γ is the adiabatic gas constant. Finally, the instantaneous gas velocity v_t is defined in Equation 10, which was determined by combining Equation 6, Equation 7 and Equation 8.

$$v_t(\alpha) = C_{t1} \bar{v}_m \exp^{-6.907\left(\frac{\alpha-\alpha_0}{\alpha_f-\alpha_0}\right)^m} + C_{t2} c_m + C_2 \frac{V_d T_{IVC}}{V_{IVC} p_{IVC}} (p - p_0) \quad (10)$$

6. Model calibration and sensitivity analysis

6.1. Model calibration

The calibration of C_{t1} and C_{t2} was carried out on the basis of the RoHR error reduction in a set of skip-fire tests, following an adjustment methodology presented in a previous work [48]. The use of skip-fire cycles avoids the combustion uncertainties, therefore the term in the right of the Equation 10, related with the pressure increment due to combustion equals to zero. Moreover, the criteria of RoHR error reduction in skip-fire test used to adjust C_{t1} and C_{t2} is equivalent to the reduction in the difference between the experimental and the modelled HT.

The model calibration consisted of the determination of the function shape parameters (i.e. α_0 , α_1 and m) and the proportionality constants (i.e. C_{t1} and C_{t2}). In order to assure the model stability and generality, a constant ratio $r_t = C_{t1}/C_{t2}$ was defined. This ratio is defined since the increase of variables to be adjusted could result in undesired

310 behaviour and non-convergence. After the calibration process, the final adjusted values are included in Table 4.

311

312 The CFD gas velocity and the model gas velocity are compared in Figure 9 along with the resulting heat transfer
313 for an operating point at 1500 rpm and Δp of 450 mbar. As can be seen, the CFD and the proposed model gas ve-
314 locities are in good agreement, being remarkable how the model follows the CFD gas velocity trend. However, there
315 are some differences between the gas velocity obtained from the CFD and the model, mainly due to the adjustment
316 criteria which takes into account the RoHR instead of the gas velocity itself. This criteria was selected because the
317 final objective is to accurately calculate the HT, therefore the HT adjustment observed in Figure 9 is better than that of
318 the gas velocity, which indicates the model potential for considering the tumble phenomena and finally for estimating
319 the HT.

320

321 The comparison between the experimental HT (\dot{Q}_n), the HT calculated using the reference Woschni model (\dot{Q}_w),
322 whose constants in Equation 3 were calibrated for the engine tested, and the HT estimated using the model developed
323 in this work (\dot{Q}_t) is presented in Figure 10. For the sake of simplicity, Figure 10 presents the medium Δp operation
324 points of each engine speed. It is evident how \dot{Q}_t provides a better fit with \dot{Q}_n than \dot{Q}_w , and some qualitative and
325 quantitative remarks can be underlined:

326

– Low discrepancy is observed between the instantaneous evolution of \dot{Q}_n and \dot{Q}_t , which indicates that in the
327 studied cases, v_t accounts well for the spatially averaged gas velocity. The good agreement between the max-
328 imum \dot{Q}_n and \dot{Q}_t observed in Figure 10 indicates that \bar{v}_m can retain the information regarding the operating
329 condition variations. Moreover, the C_{t1} value close to 1 indicates that \bar{v}_m is also representative of the maximum
330 gas velocity in the combustion chamber.

331

– In motoring tests, the swirl-model used in Equation 3 assumes that the vortex is accelerated in the compression
332 stroke, and since no friction between gas and walls is considered, the vortex is symmetrically decelerated in
333 the expansion stroke. Therefore, \dot{Q}_w is almost symmetric with respect to the TDC, with its maximum value
334 few degrees BTDC. The slight asymmetry of \dot{Q}_w is caused by the effect of the higher pressure and temperature
335 BTDC in the HT estimation (see Equation 2). In the case of \dot{Q}_t , the maximum value is reached several degrees
336 BTDC which is more reasonable, considering the mechanism of the tumble dissipation explained in Section
337 4.2.

338

– The RMSE of the complete experimental matrix is included in Table 5. It is observed how the uncertainty of
339 \dot{Q}_t is remarkably lower than that of \dot{Q}_w in all the cases, showing an average improvement of about 70%. This
340 corroborates the better performance of the proposed model in all the operating range. In the particular case of
341 1800 rpm and Δp of 500 mbar, the uncertainty reduction of \dot{Q}_t is lower than in the rest of the operating points
342 (about 40%), this is probably due to the higher experimental uncertainty at this operating condition, since this
343 point shows the highest noise to signal ratio.

344 6.2. Sensitivity analysis

345 In the engine tested in the present work, the skip-fire tests were available but in a standard test cell the equipment to
346 perform such measurements is not usual, thus only motoring test are available. As previously discussed, the motoring
347 tests have the drawback that they are performed with different thermodynamic conditions than combustion test, which
348 can result in a lower quality of the HT adjustment. Thus, a sensitivity study was performed to evaluate the effect of
349 the model parameters on the HT prediction quality.

350

351 The first term of Equation 6 indicates that the parameters affecting the tumble gas velocity model are the cal-
352 ibration constants (C_{t1} and C_{t2}), the characteristic gas velocity (\bar{v}_m) and the exponential function (f_w) parameters.
353 According to Equation 8 the product $\eta_{tr}\dot{m}_a$ is the only uncertainty since A_{eff} and $\Delta\alpha_{IVO-IVC}$ are defined by the en-
354 gine geometry, meanwhile the trapped mass is subject of experimental uncertainty, and additionally η_{tr} is not usually
355 measured. In Equation 7, the constant a mathematically set, therefore only the exponent m and the angles α_0 and α_f
356 can vary. It is important to remark that these three parameters cannot be adjusted simultaneously, since a different
357 coefficient is related to different α_0 and α_f , thus in this study the exponent was kept constant and the angles were
358 varied. In Equation 6 it is possible to observe that C_{t1} has almost the same effect as \bar{v}_m , being the unique difference its
359 effect on C_{t2} due to the r_t ratio.

360

361 The sensitivity study is carried out to determine the parameter variation that produces a given uncertainty in the
362 HT computation. Each parameter is swept to produce maximum variations of $\pm 20\%$ in the accumulated HT, with
363 steps of $\pm 5\%$. The results presented in Figure 11 indicates how the term $\eta_{tr}\dot{m}_a$ has a linear effect, and a variation of
364 1.4% in $\eta_{tr}\dot{m}_a$ results in a variation of 1% in Q_t . α_f shows also a linear trend, and a variation of 0.5° produces an
365 uncertainty of 1% in Q_t , so the model is very sensitive to this parameter and it should be carefully calibrated. Finally
366 α_0 has a different behaviour depending on its value: advancing α_0 about 1.5° results in a reduction of Q_t about 1%
367 (being this trend linear), however delaying α_0 leads to a non-linear trend, thus a variation of 14° produces the maximum
368 Q_t uncertainty of 5%, however a higher variation of α_0 produces lower Q_t uncertainty. This behaviour is explained by
369 the instantaneous evolution of \dot{Q}_t which is discussed later.

370

371 To analyse the instantaneous effect of each parameter on \dot{Q}_t , a reference well-adjusted test at 1200 rpm and Δp 400
372 mbar is compared with the HT resulting by considering the uncertainties of the parameters analysed in the previous
373 paragraph. The results include in Figure 12 correspond to variations of $\pm 10\%$ on Q_t except in the case of delaying α_0 ,
374 where the uncertainty in Q_t is 5%. \dot{Q}_w is also presented in Figure 12, with the objective of highlight how in spite of
375 the parameters variation, the qualitative fit of the proposed model is always better than the reference Woschni model.
376 Some remarkable effects on the instantaneous HT profile are:

377 – Positive or negative variations of $\eta_{tr}\dot{m}_a$ affects directly the maximum HT reached, being the instantaneous

378 difference more evident between 30° BTDC and 20° ATDC.

- 379 – Advancing α_0 has a moderate effect on the HT shape, while its effect on the maximum HT is higher. It is inter-
380 esting to notice how advancing α_0 results in slightly more HT in the early compression stroke. It is explained
381 by the higher tumble at this stage; however, at the beginning of compression, the pressure and temperature of
382 the chamber are both low, thus the global effect on the HT is small. The tumble dissipation starts earlier, which
383 results in a lower HT in the proximities of the TDC, which is not compensated by the small HT increase at the
384 beginning of the compression stroke so the accumulated HT decreases.
- 385 – Delaying α_0 has a major effect on the HT shape, reducing significantly the HT at the beginning of the compres-
386 sion stroke since no tumble is considered until α_0 , and increasing the HT close to the TDC. This HT increment
387 close to TDC is compensated by the reduction at the beginning of compression, thus the mean HT uncertainty
388 is constant (about 5%) as can be seen in the Figure 11. In spite of this moderate uncertainty, the main issue
389 of delaying α_0 is the deformation of the \dot{Q}_t , which results in higher HT peaks but abnormally low HT in the
390 compression stroke.
- 391 – The effect of α_f is observable between the maximum HT peak and the end of the tumble as can be seen in
392 Figure 12. Delaying α_f results in a longer dissipation process, and hence higher HT. Consequently, advancing
393 α_f has the contrary trend. The shape of \dot{Q}_t is moderately affected by the variation of this parameter, but due to
394 the cumulative effect on the mean HT, an accurate determination of α_f is important.

395 From this analysis, it can be concluded that the model is robust enough to be generally applied for engines with
396 tumble, and it is robust against uncertainties of the parameters in a reasonable ranges, still performing better than the
397 classical Woschni model, even including the effect of swirl, in terms of the instantaneous evolution and the mean HT
398 determination.

399 7. Conclusions

400 A detailed HT model taking into account the gas velocity evolution caused by a high tumble movement is de-
401 veloped and validated. The model was developed in a research CI two-stroke engine with high tumble ratio and no
402 swirl. The theoretical analysis was carried out by means of CFD simulations, whereas the calibration of the model
403 was carried out by means of skip-fire tests. The most relevant conclusions are:

- 404 – The model presents a reduction of about 70% on the RMSE when comparing with the reference Woschni model.
405 This improvement is observed for all the experimental conditions. The model also provides a better agreement
406 with the experimental instantaneous HT, since it was observed that HT profile was well followed at different
407 operating conditions.

- 408 – From the sensitivity analysis, the quantitative and qualitative effect of varying some parameters related with the
409 gas velocity model were evaluated. The results show how variations of $\eta_{tr}\dot{m}_a$ about $\pm 1.4\%$, advancing α_0 1.5°
410 and varying $\alpha_f \pm 0.5^\circ$ leads to 1% of uncertainty in Q_t .
- 411 – Retarding α_0 generates an uncertainty of about 5% in Q_t due to a compensation effect between the HT in the
412 beginning of compression (low HT) and in the proximities of the TDC (high HT). However, the impact on the
413 mean HT computation is moderate, while it has a remarkable effect in the instantaneous HT evolution.

414 It has been confirmed how the developed HT model is robust to be applied in engines with tumble movement.
415 The model accounts for the spatially averaged gas velocity due to high tumble motion, and it allows to accurately
416 determining its influence on the HT to the combustion chamber walls, which is critical to improve the quality of the
417 combustion diagnosis models, widely used in the field of research and development.

418 **8. Acknowledgements**

419 The support of the Spanish Ministry of Economy and Competitiveness (TRA2013-41348-R) is greatly acknowl-
420 edged.

421

422 The authors would like to thank RENAULT SAS for all the technical support provided to perform the research
423 activities.

424

425 The authors want also to express their gratitude to CONVERGENT SCIENCE Inc. and IGNITE3D Engineering
426 GmbH for their kind support for performing the CFD calculations using CONVERGE software.

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9. Figures

Figure 1. Procedure Scheme

Figure 2. Sketch of the cylinder head designed for the 2-stroke engine architecture (Patent Renault FR2931880)

Figure 3. Engine test cell scheme

Figure 4. Skip-fire pressure

Figure 5. CFD validation

Figure 6. Main process of tumble evolution

Figure 7. Tumble ratio evolution in a perpendicular plane of the cylinder axis

Figure 8. CFD HT and tumble gas velocity

Figure 9. Tumble gas velocity model

Figure 10. Heat transfer comparison ($\Delta p = 400\text{mbar}$)

Figure 11. Sensitivity analysis

Figure 12. HT sensitivity at 1200 rpm and $\Delta p = 400\text{mbar}$

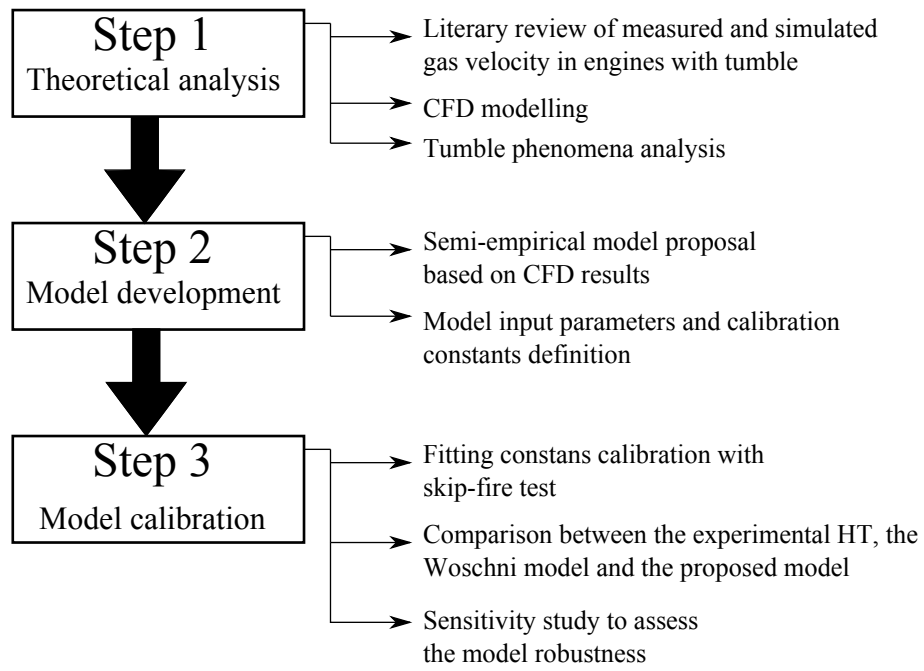


Figure 1: Procedure Scheme

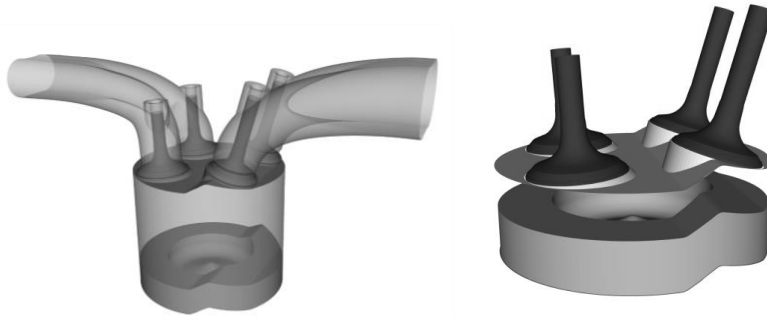


Figure 2: Sketch of the cylinder head designed for the 2-stroke engine architecture (Patent Renault FR2931880)

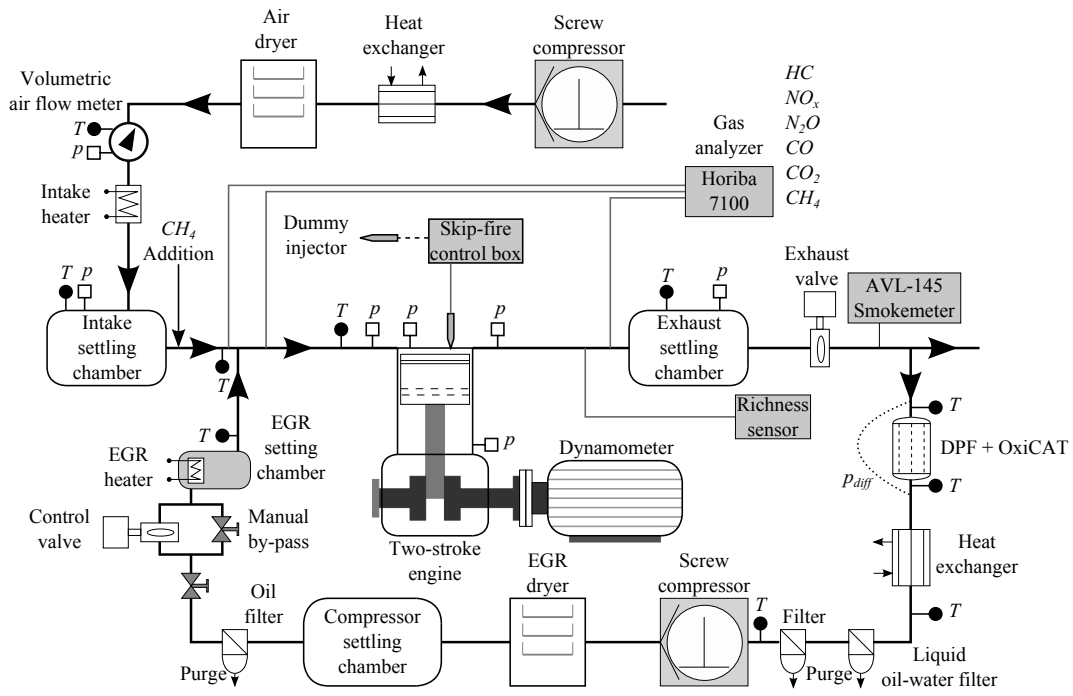


Figure 3: Engine test cell scheme

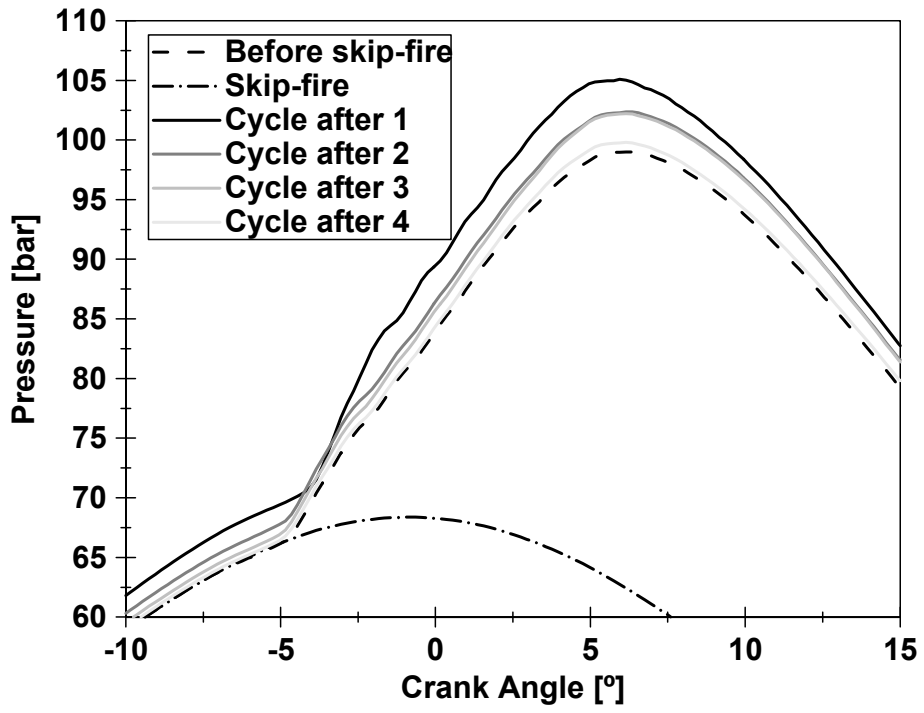


Figure 4: Skip-fire pressure

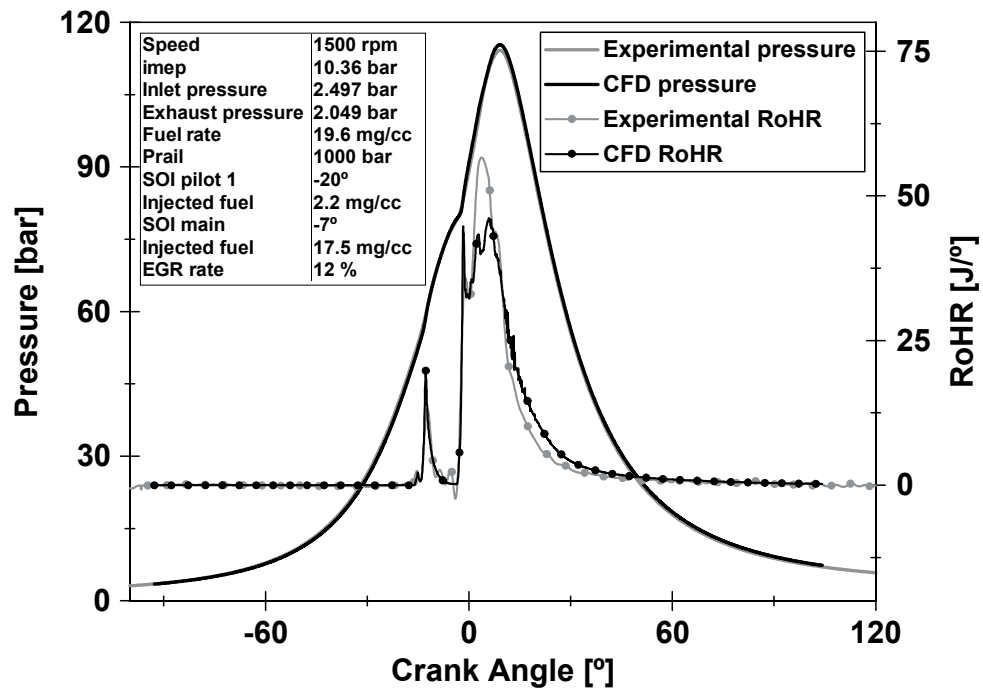


Figure 5: CFD validation

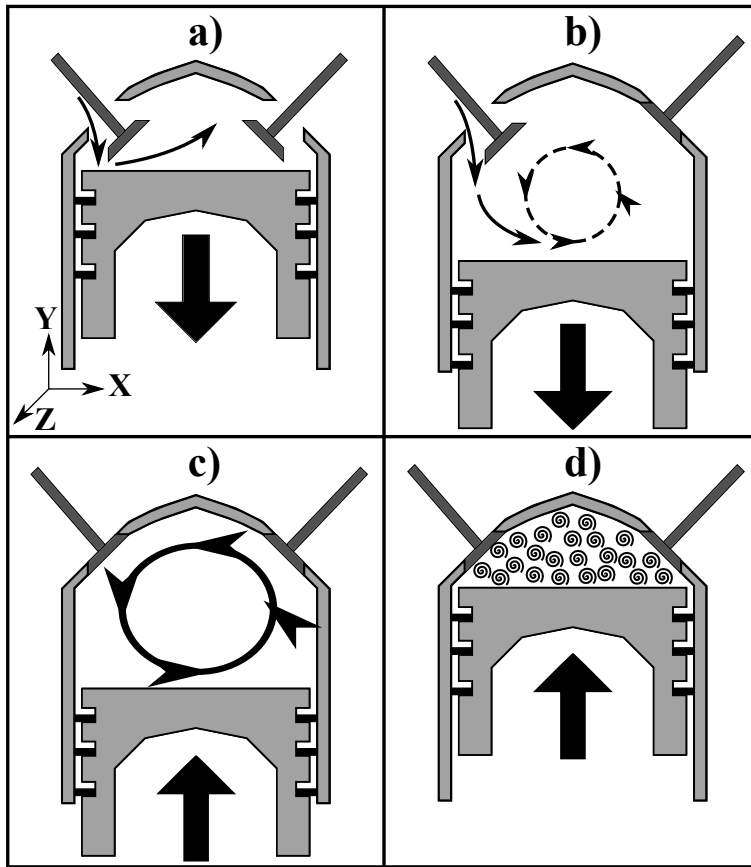


Figure 6: Main process of tumble evolution

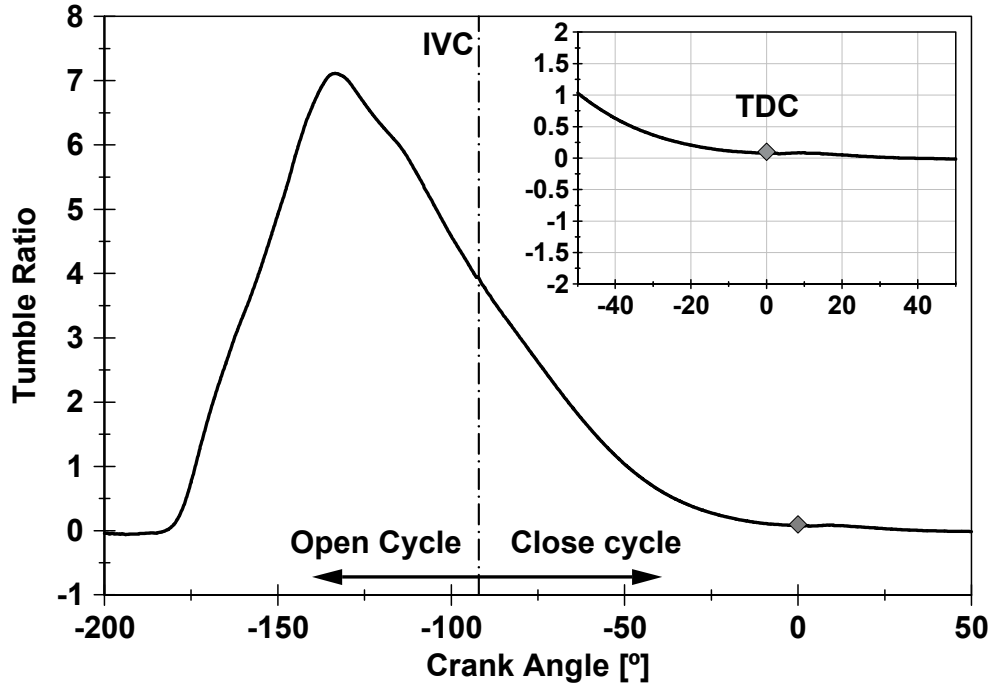


Figure 7: Tumble ratio evolution in a perpendicular plane of the cylinder axis

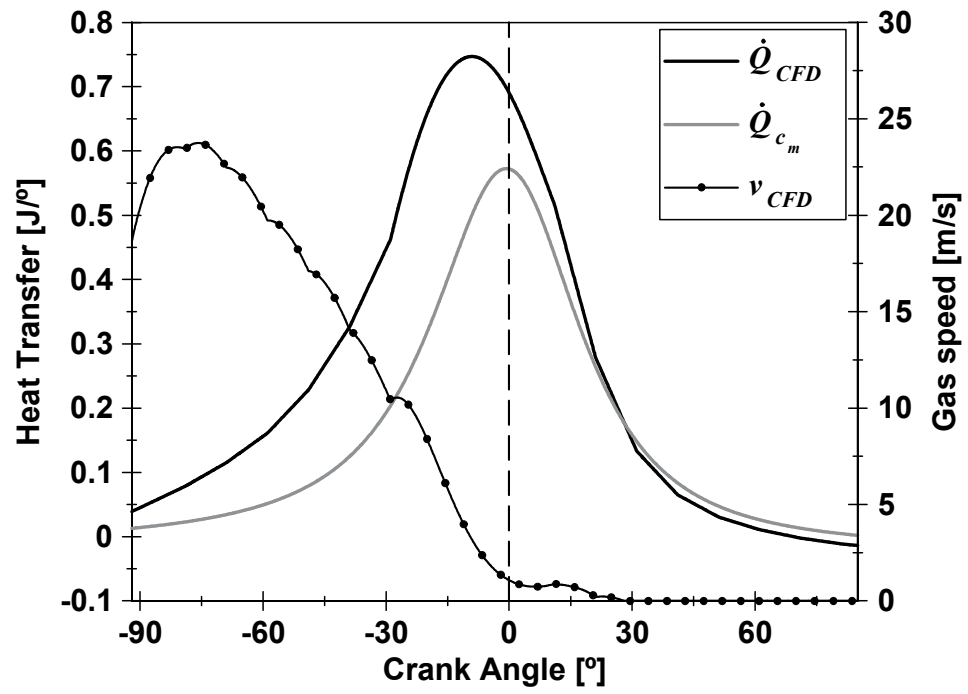


Figure 8: CFD HT and tumble gas velocity

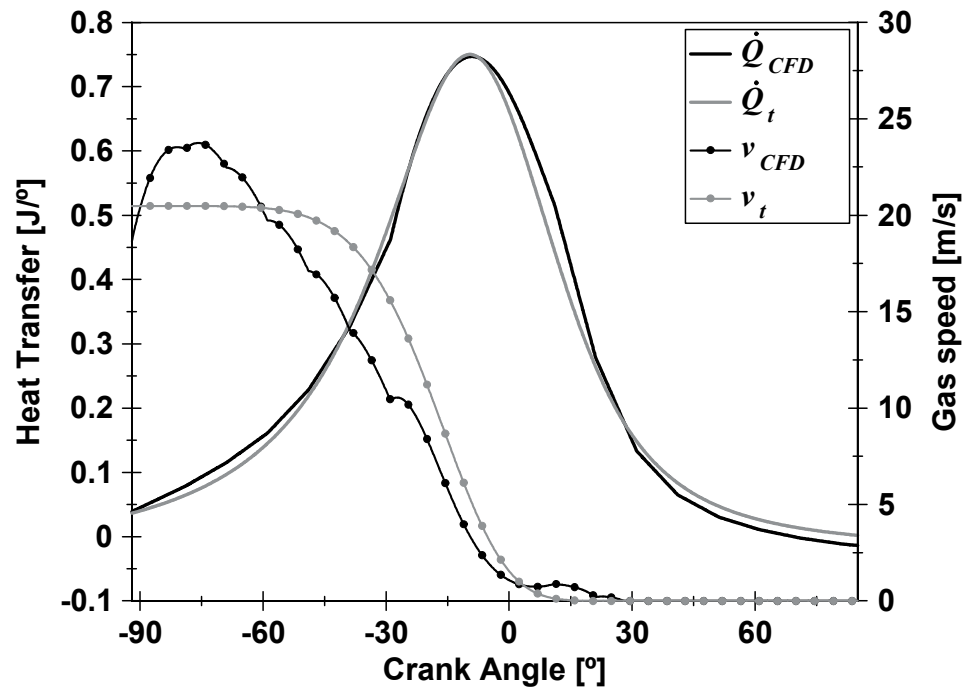


Figure 9: Tumble gas velocity model

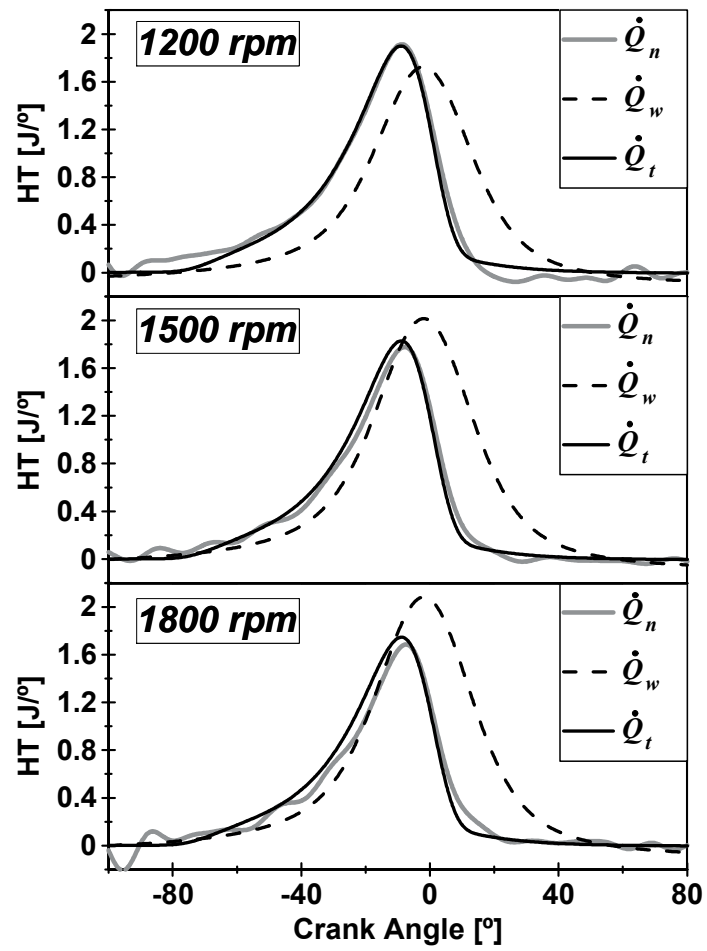


Figure 10: Heat transfer comparison ($\Delta p = 400\text{mbar}$)

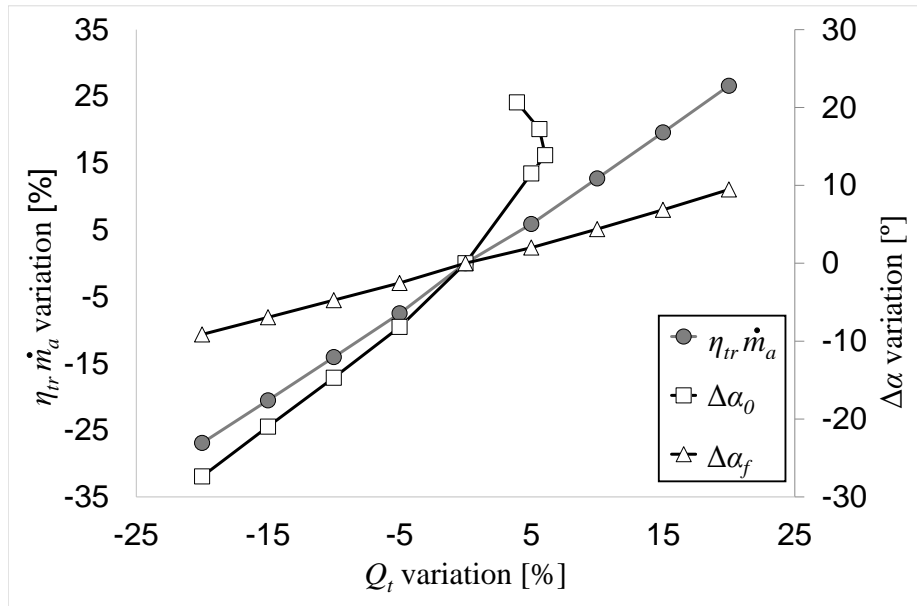


Figure 11: Sensitivity analysis

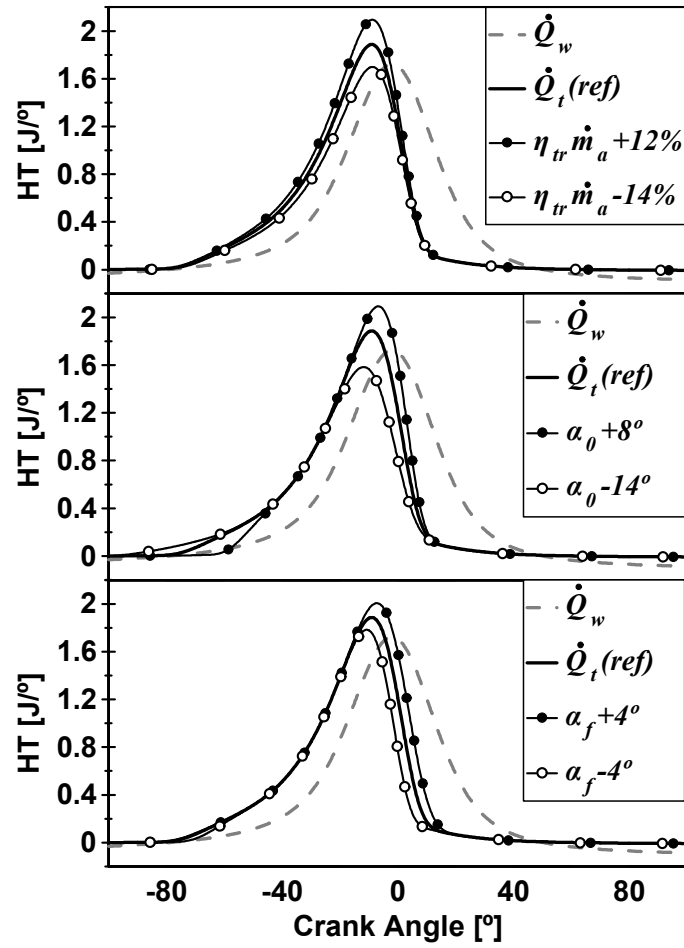


Figure 12: HT sensitivity at 1200 rpm and $\Delta p = 400\text{mbar}$

10. Tables

Table 1. Main engine geometrical characteristics

Table 2. Measured operational points

Table 3. CFD emissions validation

Table 4. Model adjusted parameters

Table 5. RMSE of the HT computation by using the Woschni model and the proposed model

Table 1: Main engine geometrical characteristics

Displacement	365 cm ³
Bore	76 mm
Stroke	80.5 mm
Connecting rod length	133.75 mm
Geometric CR	17.8
Number of valves/cylinder	4
Type of scavenge	Poppet valves with scavenge loop
Intake camshaft profile	Duration: 80°/max. lift: 6 mm
Exhaust camshaft definition	Duration : 90°/max. lift: 8.5 mm

Table 2: Measured operational points

Speed [rpm]	Δp [mbar]	$imep$ [bar]	Air mass flow rate [g/s]	η_{ir} [%]
1200	300	3.7	7.4	69
1200	400	5.6	10.1	72
1200	500	7.5	12.2	84
1500	300	3.7	9.6	78
1500	400	5.6	11.0	79
1500	500	7.5	13.4	79
1800	300	3.7	8.7	80
1800	400	5.6	11.4	84
1800	500	7.5	14.7	85

Table 3: CFD emissions validation

	CFD	Experiment
<i>CO</i>	2,2017 mg/s	3,7500 mg/s
soot	0,0565 mg/s	0,0670 mg/s
<i>HC</i>	0,0032 mg/s	0,3000 mg/s
<i>NO_x</i>	5,4780 mg/s	5,9000 mg/s

Table 4: Model adjusted parameters

Parameter	Value
m	6
a	-6.907
α_0	34.15° BTDC
α_f	14.9° ATDC
C_{i1}	0.804
C_{i2}	1.293
r_t	0.622

Table 5: RMSE of the HT computation by using the Woschni model and the proposed model

Speed (rpm)	Δp (mbar)	Q_w RMSE ($J/^\circ$)	Q_t RMSE ($J/^\circ$)	Improvement (%)
1200	300	0.30	0.09	70
1200	400	0.33	0.08	77
1200	500	0.38	0.10	74
1500	300	0.31	0.08	75
1500	400	0.30	0.07	76
1500	500	0.31	0.12	61
1800	300	0.28	0.05	82
1800	400	0.28	0.08	72
1800	500	0.31	0.18	41