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Additional Information

Experimental assessment for instantaneous temperature and heat flux measurements under Diesel motored engine conditions

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Abstract

The main goal of this work is to validate an innovative experimental facility and to establish a methodology to evaluate the influence of some of the engine parameters on local engine heat transfer behaviour under motored steady-state conditions. Instantaneous temperature measurements have been performed in order to estimate heat fluxes on a modified Diesel single cylinder combustion chamber. This study was divided into two main parts. The first one was the design and setting on of an experimental bench to reproduce Diesel conditions and perform local-instantaneous temperature measurements along the walls of the combustion chamber by means of fast response thermocouples. The second one was the development of a procedure for temperature signal treatment and local heat flux calculation based on one-dimensional Fourier analysis. A thermodynamic diagnosis model has been employed to characterise the modified engine with the new designed chamber. As a result of the measured data coherent findings have been obtained

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in order to understand local behaviour of heat transfer in an internal combustion engine, and the influence of engine parameters on local instantaneous temperature and heat flux, have been analysed.

Keywords: Diesel engine, transient heat flux, local wall temperature, fast response thermocouples.

Nomenclature

Latin:

A_n, B_n	Fourier coefficients	(-)
C_{w1}	Woschni coefficient	(-)
n	Harmonic number	(-)
N	Number of harmonics	(-)
\dot{q}	Heat flux	W/m ²
T_w	Wall temperature	K
T_x	In-depth temperature	K
T_m	Time average value of wall temperature	K
T_i	Temperature at different locations ($i = 1$ to 5)	K
x	Distance from in-depth and tip thermocouples	mm

Abbreviations:

BDC	Bottom dead centre
CAD	Crack angle degrees
CFD	Computational fluid dynamics
emf	Thermocouple electromotive force
EVC	Exhaust valve closing
IVC	Inlet valve closing

LHR Low heat rejection
NIST National institute of standards
RoHR Rate of heat release
rpm Revolutions per minute
TDC Top Dead Centre

Greek symbols:

α Thermal diffusivity m^2/s
 ω Angular speed rad/s
 μ micro (-)

1. Introduction

From the beginning of Diesel engine development, the importance of understanding the heat transfer through the combustion chamber walls has been recognised [1–3]. Although this phenomenon is difficult to analyse because of its complexity, it that involves transient three dimensional behaviour, rapid temperature swings, piston motion, cooling passages, and turbulent reactive fluid dynamics, among others [4], it cannot be ignored.

Most of the studies performed in this area have been motivated by the influence of heat transfer on engine performance. Higher heat transfer rates to the combustion chamber walls will lead to a reduction of the mass average in cylinder temperature and pressure, and therefore the work per cycle transferred to the piston will be lower. As a consequence, specific power, engine fuel consumption and efficiency will be affected by the magnitude of engine heat transfer [5]. Concerning this, some investigations have been focused on the development of adiabatic engines, covering the combustion chamber walls with ceramic coatings [6] or manufacturing them with materials with extremely high thermal resistance [7], so that the heat losses are reduced and recovered to be partially converted into useful work. These engines are commonly known as Low Heat Rejection (LHR) engines [8].

The important role that the preservation of the atmospheric environment plays in the society nowadays is well-known. Heat transfer knowledge is important in this suit as changes in gas temperature due to in-cylinder heat losses to the walls strongly affect emissions formation and after burning processes [9]. Additionally, in recent years, emissions regulations have become more restrictive while high-efficiency is still being demanded [10]. Some im-

26 improvements in cooling system design and strategies have been also considered
27 in order to reduce emissions [11–14].

28 Finally it is also recognised the influence that engine heat transfer esti-
29 mations have on the successful development of combustion diagnosis models,
30 widely used by engine manufacturers during the early stages of engine design,
31 in order to determine thermal loads and stresses [15, 16].

32 All these reasons have motivated a large amount of studies in the last
33 decades aiming of a deeper understanding of engine heat transfer. Some of
34 them are based on simple resistive models [18]. Some others are focused on
35 the development of correlations to estimate heat transfer rates based on the
36 theory of forced convection in turbulent flows [19], which can be time and
37 locally averaged [20], instantaneous but spatially averaged [3, 21, 22], and
38 those allowing for changes not only with time but also with location [23, 24].
39 In recent years, with the increasing use of computational fluid dynamics
40 software (CFD) studies of temporal and spatial variation of the heat transfer
41 coefficient inside the combustion chamber had also been reported [26–28].

42 Although all the theoretical work described above is valuable, experimen-
43 tal data is needed in order to validate the results, in the case of correlations,
44 and to determine the initial and boundary condition values in the case of
45 CFD codes for more accurate calculations [29, 30]. With this purpose sig-
46 nificant efforts have been devoted to develop increasingly precise, fast, and
47 robust sensors [31–33]. Most of the experimental studies in this field make
48 use of high-frequency thermocouples, because of its low cost and fast response
49 [34–36]. If temperature is measured on the piston the signal is transmitted
50 by means of telemetry systems [37].

51 The main goal of this paper is to set up and validate an innovative ex-
52 perimental facility, and the corresponding methodology, to perform reliable
53 studies on local Diesel engine heat transfer behaviour. In order to accomplish
54 this, a well controlled bench was built that reproduces in-cylinder Diesel con-
55 ditions, and a robust measurement system was set up, including fast response
56 thermocouples for instantaneous temperature measurement at the same time
57 that a protocol of measurement was established.

58 A compact procedure was written as a routine into a commercial code in
59 order to perform temperature signal filtering and local heat flux calculations
60 through the walls based on one-dimensional heat conduction theory with
61 Fourier analysis. Additionally a thermodynamic diagnosis model was used
62 to characterise and process the in-cylinder pressure signal.

63 This paper has been organised as follow: the next section presents a
64 detailed description of the experimental facility, including the special re-
65 quirements to measure instantaneous heat fluxes through the walls of the
66 modified combustion chamber and the data acquisition system. In section
67 three, the methodology is described, and temperature signal processing tools
68 and mathematical methods for local heat flux calculation are described to-
69 gether with a brief description of the method used to characterise the engine
70 with the thermodynamic model. Then, the experimental results obtained
71 are discussed in the fourth section. Finally, some conclusions of the findings
72 obtained throughout the whole development of the work are pointed out.

73 **2. Experimental facility**

74 A fully equipped experimental cell has been designed with the aim of
75 reproducing the thermodynamic conditions in a single cylinder Diesel engine,
76 in order to evaluate the instantaneous heat fluxes through the walls. Special
77 care has been taken not only in the configuration and control of the variables
78 but also on the selection of the measurement equipment. A single cylinder,
79 water cooled, two valve, direct injection Diesel research engine was chosen
80 to perform these studies as this type of engines permit better control and
81 management. Constructive characteristics and additional technical data of
82 the engine are given in Table 1.

83 *2.1. Engine Modifications*

84 Carrying out instantaneous wall temperature measurements at different
85 locations on the combustion chamber of an engine is complex due to pis-
86 ton motion and geometric characteristics, which include cooling passages
87 throughout engine block and head. Thus, some modifications were made to
88 the original engine. The original bowl was filled, obtaining a flat piston. Be-
89 tween the head and the cylinder block an extension of steel F114 was placed,
90 in whose centre a combustion chamber was machined. A simplified sketch of
91 the final combustion chamber configuration is shown in Figure 1.

92 For the design of this feature some considerations were taken into account.
93 On the one hand it must be capable of reproducing in-cylinder conditions,
94 by keeping as possible the compression ratio, allowing the opening and clos-
95 ing of intake and exhaust valves and enclosing the injector sleeve. On the
96 other hand, the fitting of heat flux sensors must be easier than in a conven-

97 tional combustion chamber, while permitting measurements on different wall
98 locations of the chamber.

99 Subsequent to an evaluation of different designs and an optimisation pro-
100 cess, the solution that better fulfilled all the main requirements was selected.
101 Figure 2 shows the chosen solution and the final appearance of the in-house
102 built elongation piece. It can be observed that the extension piece can be
103 instrumented with heat flux sensors at different locations, one near the intake
104 valve zone, one near the exhaust valve zone, and the other three in differ-
105 ent positions along the central zone of the combustion chamber walls, where
106 different temperature swings are expected, as a consequence of chamber ge-
107 ometry that produces dissimilar local conditions. A hole to place a pressure
108 transducer was also machined in the middle zone of the combustion chamber.

109 As a consequence of engine modifications and experimental requirements,
110 some changes had to be done on engine lubrication and cooling systems. First
111 of all both systems must be external and split up as the piece between the
112 head and the cylinder block does not allow the pass of any of the two liquids
113 between them. It is also necessary to set, control and regulate initial wall
114 temperatures in all tests and thus coolant circuit with temperature, pressure,
115 and flow rate control was installed.

116 *2.2. Test bench*

117 A sketch of the experimental facility, data acquisition, and monitoring
118 system is given in Figure 3. The engine was set to be motored under steady-
119 state conditions connecting the crankshaft to an asynchronous dynamometer
120 (Schenk LI 145) and torque was controlled and registered with a torque
121 flange. In-cylinder pressure was measured with a piezoelectric transducer

122 (Kistler 6055Bsp) connected to an amplifier (Kistler 5011) and registered
123 with 0.5 crank angle resolution. Synchronisation between pressure signal
124 and instantaneous crankshaft angle was performed by an optical encoder
125 (AVL 1203).

126 Air was supplied to the engine by means of a screw compressor, which
127 could set the inlet pressure from 1 to 2.5 bar. Once the air is compressed,
128 passes through a liquid-gas heat exchanger, dryer and finally a filter. After-
129 wards air mass flow measurement was simultaneously performed via two flow
130 meters: a hot wire anemometer (Sensyflow Sensycon Siemens) and a volu-
131 metric mass flow meter downstream located (ELSTER 75034124). These two
132 measurement methods were used for various reasons. The hot wire anemome-
133 ter allows proper measurements of low mass flows when the single cylinder
134 engine is operating at low speed, whereas, the volumetric meter was used in
135 tests with higher mass flow values. Additionally this permits to control the
136 proper behaviour of the hot wire meter and to benefit measurement accu-
137 racy and avoid uncertainties in the cases where both systems were able to
138 measure. Finally, the air passes through a settling chamber to prevent pulses
139 generated by the intake process and a second heater before entering into the
140 engine.

141 Air temperature and pressure were measured at different locations along
142 the intake circuit with K type thermocouples and glow-plug piezoresistive
143 pressure transducers. Air temperature was controlled via a PID with mea-
144 surement performed a few millimetres upstream of the intake valve. This
145 ensured that the intake temperature was as close as possible to the set value
146 at the very entrance of the engine. Air pressure was also controlled via PID

147 but it was measured in the settling chamber in order to avoid signal fluctu-
148 ations.

149 Exhaust path consisted of a filter installed upstream of the settling cham-
150 ber, where temperature and pressure were measured. The settling chamber
151 was used, as it was say before, to reduce pressure fluctuations that can alter
152 the measurement accuracy. A valve was installed at the end of the exhaust
153 circuit to control back pressure. Finally blow-by mass flow was measured
154 with a diaphragm-type flow meter (AVL 442) placed in a small tank con-
155 nected to the oil pan.

156 As it can be seen several variables need to be measured and controlled.
157 Short term response parameters need to be measured and registered with a
158 very high sampling frequency (instantaneous measurements) while others can
159 be measured and controlled with a longer response (in-depth temperature),
160 or even average values (mean variables).

161 *2.3. Instantaneous temperature measurements*

162 With the aim of measuring in cylinder wall temperature variations, fast
163 response, E-type, eroding thermocouples were employed as its response time
164 is less than $10 \mu\text{s}$ [38] making them very suitable for in cylinder applica-
165 tions. In order to measure the steady term of in-depth temperature, E-type
166 thermocouples were also chosen. The reason of the choice was to obtain the
167 maximum voltage change per degree.

168 The special heat flux probe used in these studies was built by NANMAC.
169 It consists on a fast response eroding thermocouple on the tip of the probe
170 and a common E-type in-depth thermocouple placed at a distance of 5 mil-
171 limetres. Figure 4 shows a schematic draw of the probe. The two wires of

172 dissimilar metals are flattened into thin ribbons. Then both metals are elec-
173 trically insulated from each other with a very thin sheet of dielectric material
174 forming a sandwich. Afterwards the excess of material is machined off and
175 the tip surface is polished by an abrasive action [33]. This process forms
176 microscopic welded measuring junctions, joining one ribbon to the other. Fi-
177 nally, surface and in-depth thermocouples are placed in a cylindrical house
178 constructed with the same material as that of the wall and located in the
179 combustion chamber.

180 The voltage obtained from the five fast response thermocouples was inde-
181 pendently amplified and afterwards registered. Before signal amplification,
182 all wires pass through an isothermal block, ensuring that the reference tem-
183 perature is the same for the five probes. A platinum resistance was used to
184 read the junction temperature which is stored for every test. Once all the
185 signals were recorded, the emf corresponding to the junction temperature
186 was calculated by means of the E type NIST polynomial equation and cor-
187 rections were performed to each voltage via software. In depth temperature
188 was measure with a device that has internal compensation, and does not
189 need amplification.

190 Each flux sensor was flush mounted on the walls of the combustion cham-
191 ber by means of a conical compression fitting that fixed it and avoided any
192 gas loss from the combustion chamber. For better understanding during re-
193 sult discussion probes were numerated (Figure 2(a)). The one near the intake
194 valve has been called T_2 , and the one near the exhaust valve is called T_5 . The
195 other three are in the middle zone of the chamber: T_1 is nearby the intake
196 zone, T_4 nearby the exhaust zone and T_3 is most centrally situated on the

197 combustion chamber.

198 **3. Methodology and processing tools**

199 The methodology established and followed in order to evaluate the in-
200 fluence of engine operation parameters on the instantaneous heat transfer,
201 combines experimental procedures and two analysis tools: an in-cylinder
202 thermodynamic model based on signal pressure to characterise the new con-
203 figuration (CALMEC [39]) and the calculation of local heat fluxes from in-
204 stantaneous wall temperature measurements and Fourier analysis. Figure 5
205 shows a simplified flowchart of the methodology.

206 *3.1. Engine measurement methodology*

207 With the aim of performing parametric studies of the operating condi-
208 tions, a matrix of 40 experiments was performed under motored steady-state
209 conditions. It is important to remark that these studies were used also to
210 characterise the modified engine, so that an extensive sweep was tested. Five
211 levels of engine speed variation were consider (1000, 1500, 2000, 2500, 3000
212 rpm), intending to cover all the engine operation range. The effect of inlet
213 pressure was assessed by considering four levels of variation; the selection of
214 the values was done from the minimum value corresponding to natural aspi-
215 ration up to the maximum value that the compressor can reach by steps of
216 0.5 bar (1, 1.5, 2, 2.5). Finally, in order to estimate the possible fluctuation
217 introduced to the heat flux due to the inner wall temperature, two temper-
218 atures were consider in the cooling and lubricating systems, 60°C and 70°C.
219 The test matrix is summarised in Table 2.

220 The protocol established in order to measure each point of the matrix was
221 to motor the engine to the corresponding speed, until all the parameters were
222 stabilised (intake air temperature, engine block temperature, and intake and
223 exhaust pressure). During this period, steady state temperature was contin-
224 uously measured until a constant value was obtained. Once all parameters
225 were stable, instantaneous data were recorded on an angular base for twenty
226 five thermodynamic cycles, and all mean variables were recorded simultane-
227 ously. Each point was measured three times in order to avoid uncertainties
228 on the data.

229 *3.2. Thermodynamic diagnosis model*

230 CALMEC is an in-house diagnosis model firstly developed to calculate
231 the rate of heat release (RoHR) and the temporal evolution of in-cylinder
232 thermodynamic conditions from the instantaneous pressure signal. Despite
233 the main objective of the model is to work with combustion test, there are
234 some variables that determine the precision of the final results that can be
235 obtained only from motored tests. This process is called engine characterisa-
236 tion and is necessary to perform it, if any modification of the engine can lead
237 to a new compression ratio, a different deformation coefficient or different
238 thermodynamic delay.

239 The procedure performed to engine characterisation is based on an it-
240 erative adjustment of the heat losses by two methods: in one the first law
241 is solved, assuming that RoHR is zero and rejecting the term involving fuel
242 energy as no combustion is performed. The other one is related to the model
243 proposed by Woschni adjusting C_{w1} taking pressure as the reference value to
244 conclude the iteration. It is to be noticed that even if pressure is the param-

245 eter selected as a criterion for convergence, all the other parameters involved
246 in the calculation are simultaneously adjusted (mass flow rate, blow-by coeffi-
247 cient, compression ratio, thermodynamic delay, and deformation coefficient)

248 3.3. Instantaneous heat flux calculation

249 A time periodic heat conduction model is used to estimate the local heat
250 transfer, assuming that in the normal direction of the combustion chamber
251 wall this heat transfer is strictly one dimensional, that material properties
252 remain constant during an engine cycle, and that the internal and external
253 temperatures are uniform at the start of the test. These assumptions are
254 justified by the fact that the transient temperature penetrates only a small
255 distance from the combustion chamber surface. The transient heat flux cal-
256 culation can be performed by solving the unsteady heat conduction equation.

$$\frac{\partial T}{\partial t} = \alpha \frac{\partial^2 T}{\partial x^2} \quad (1)$$

257 where x is the distance between the surface and the in-depth thermocouple
258 (5mm), and α is the thermal diffusivity of the material.

259 Boundary conditions for equation (1) are derived from the experimental
260 temperature data and can be written as:

$$T(0, t) = T_w(t) \quad (2)$$

261 and

$$T(x, t) = T_x = \text{constant} \quad (3)$$

262 As the phenomenon can be assumed to be periodic and the thermal prop-
263 erties remain constant, the surface temperature $T_w(t)$, can be expressed by

264 Fourier analysis [4] as:

$$T_w = T_m + \sum_{n=1}^N [A_n \cos(n\omega t) + B_n \sin((n\omega t)] \quad (4)$$

265 where ω is the angular frequency, T_m is the time average value of T_w , A_n and
266 B_n are the Fourier coefficients, n the harmonic number and N is the total
267 number of harmonics.

268 In order to determine, A_n and B_n , Fast Fourier Transform (FFT) was
269 applied to the measured surface temperature. It is important to adequately
270 select the value of the number of harmonics N since if it is too small, the
271 periodic temperature contour will not match the measured value accurately,
272 and if it is too large, the signal will be disturbed with noise and this could lead
273 to unphysical fluctuations in the profile [35]. In this work N was chosen by
274 applying a low pass filter based on a statistical method [40] to each location
275 temperature signal.

276 Once A_n and B_n are determined, the instantaneous heat flux is calculated
277 using the Fourier's law. Total heat flux will consist of a time-independent
278 steady term and a time dependent transient term.

$$\dot{q} = \frac{k}{x}(T_m - T_x) + k \sum_{n=1}^N \phi_n [(A_n + B_n) \cos(n\omega t) - (A_n - B_n) \sin(n\omega t)] \quad (5)$$

279 where $\phi_n = \sqrt{(n\omega/2\alpha)}$ and \dot{q} is total heat flux in W/m^2 .

280 4. Results and discussion

281 In this section the results obtained from the tests indicated in the method-
282 ology will be presented and analysed. The effect of each of the parameters

283 evaluated is independently presented as subsections as well as the validation
284 of the modified engine.

285 *4.1. Engine validation*

286 All the tests performed in order to evaluate the intake and engine pa-
287 rameters were used to characterise the experimental facility. With the aim
288 of validating the experimental facility, two sets of motored test were carried
289 out. The first set was performed with the original configuration whit the
290 bowl-in-piston combustion chamber, and the second with the modified com-
291 bustion chamber and flat piston configuration. Intake temperature and pres-
292 sure were set to be the same for the two groups of tests, as well as the exhaust
293 pressure and oil and water temperature. Then the characterisation proce-
294 dure of CALMEC was used to process the pressure signals and to calculate
295 compression ratios and thermodynamic delays among other thermodynamic
296 parameters.

297 Figure 6 presents a comparison between the original and modified pressure
298 signal for a motored test with 25°C and 2 bar of intake temperature and
299 pressure respectively and engine speed of 2000 rpm. It can be observed that
300 the maximum pressure was maintained (less than 0.6 bar of difference can be
301 observed). This information is also confirmed by the results obtained from
302 CALMEC, which are more representative as are obtained from the entire
303 set of tests; the given value of compression ratio for the original engine was
304 14.5:1 while for the modified engine was 13.8:1. Looking at the shape of both
305 curves during the compression stroke it can be observed that the modified
306 engine has a higher polytrophic coefficient. This fact can explain that, even if
307 there exists a difference of compression ratio, the original maximum pressure

308 it still being close to the modified one. This leads to higher thermodynamic
309 delays as can be seen on the right top corner of Figure 6, which presents
310 a detail around TDC [41]. The modified engine reaches its maximum peak
311 value almost 1 degree before the original engine. This can be interpreted
312 as the engine has become more adiabatic due to the fact that heat transfer
313 surfaces were slightly modified [42].

314 *4.2. Local behaviour*

315 Figure 7(a) presents the measured surface temperature for three of the five
316 thermocouples installed in the combustion chamber walls. Thermocouples T_3
317 and T_4 were not able to operate satisfactorily over the majority of the tests
318 and for that reason its signal is not presented. The mean value of the steady
319 state in-depth temperature is also presented for the three probes. The data
320 were taken with the engine motored at 2000 rpm, with intake pressure of 2.5
321 bar and intake temperature of 25°C.

322 The peak surface temperature at location 1 is about 11K above its mini-
323 mum surface value, for location 2 is about 16K, and 19K for location 5. Even
324 if the temperature swings in the last two locations are similar, lower temper-
325 atures along the whole cycle are observed in location 2. This is reasonable, as
326 probe 2 is located near the inlet zone and therefore lower wall temperatures
327 should be expected in comparison with location 5, that is situated on the
328 exhaust zone where higher values are reached. Observing carefully the tem-
329 poral variation of the wall temperature signal, a little fluctuation produced
330 by the opening of exhaust valve can be noticed, which is due to the changes
331 induced in the fluid motion, and is more evident in the sensor that is near
332 the exhaust valve. The one that is in the middle zone (T_2) does not present

333 any noticeable change during valve opening and closing.

334 Changes in thermodynamic conditions when the clean charge is admitted
335 can be also observed. Specifically, after EVC the wall temperature decreases
336 until it reaches its minimum value for T_1 and T_5 probes near the IVC, and
337 with a small delay in the case of T_2 . It may be noticed that the wall surface
338 temperature peaks occur about 10 degrees after TDC for the three thermo-
339 couples, while the maximum gas temperature is expected at TDC or even
340 a few degrees before. This observation may be explained by the relative
341 temperatures of the gas and wall. Gas temperature starts decreasing near
342 TDC but it still hotter than the surfaces of the walls, thus, heat will continue
343 flowing from the gas to the walls, and the surface temperature will continue
344 rising until the rate of heat conduction into the inner wall material exceeds
345 that from the gas to the wall [43].

346 In Figure 7 (b) the heat transfer through the wall-gas interface, computed
347 from the surface temperature variations described above, are shown. Due to
348 the fact that the highest heat transfer rates to the walls take place during the
349 compression an expansion strokes, the data shown has been limited to these
350 strokes. The peak heat transfer reaches a level of 0.29 MW/m^2 in the exhaust
351 zone, while in the middle zone of the chamber it is 34% lower, reaching a
352 level of 0.19 MW/m^2 .

353 This confirms the importance of the knowledge on local heat fluxes in
354 the design stages as, the exit zones are much more thermally loaded than
355 others. The probe located in the inlet zone has its maximum in 0.23 MW/m^2 .
356 The curves of heat flux display asymmetry with respect to TDC, in that
357 they decay slower during the expansion stroke than during the compression

358 stroke. This phenomenon can be originated in the gradually decreasing of
359 intensity of the in-cylinder motions caused by wall friction and turbulence
360 decay [44]. In-cylinder pressure is also depicted in Figure 7(b). The peak
361 heat flux occurred slightly before TDC for the three locations when the air
362 temperature and pressure were highest, as expected.

363 *4.3. Intake pressure influence*

364 The increase of intake pressure produced a significant increase on peak
365 heat fluxes at the three locations, as can be seen in Figure 8(a-c) where the
366 tests at constant engine speed (1500 rpm) and intake temperature (25 °C) are
367 shown. This is due to two reasons: first, the air temperature and pressure
368 increase producing higher wall temperatures and, secondly the swirl also
369 increases [2]. It was detected in the three locations that not only the surface
370 temperature increases but in-depth temperatures also rise; for example, in
371 the case of location 5 changes of almost 50 degrees were reached between 1
372 and 2.5 bar.

373 In Figure 8(b) it is notorious that the peak rise of heat flux is not linear
374 with the rise of the inlet pressure. An increase from 1 to 1.5 bar produces
375 a rise of heat flux peak of only 0.02 MW/m² whereas the change between
376 2 and 2.5 bar of inlet pressure produces a change five times larger in the
377 maximum heat flux of 0.1 MW/m². In the case of Figure 8(a) this tendency
378 is also observed. This can be explained if the mass flow rate and pressure are
379 analysed. For the same two cases, the increment of 0.5 bar in inlet pressure
380 will produce an increase of 27% in the air mass flow for the first case, while the
381 same increase of pressure in the second case just produces an increment of 2%
382 in the mass flow. Additionally, the peak of pressure signal (Figure 8(d)) rises

383 linearly with the increment of inlet pressure. Since, under motored condition,
384 gas temperature only depends on these two parameters, an abrupt change is
385 to be expected both in the gas and the wall temperatures.

386 At the beginning of the compression stroke in Figure 8(a) and (c), the
387 heat fluxes corresponding to lower inlet pressure were higher than the upper
388 ones. This can be due to the fact that inlet temperature was kept constant
389 and, therefore, since the mass was increasing with inlet pressure, more energy
390 was needed to rise the charge temperature and hence less heat was transferred
391 to the walls.

392 *4.4. Wall temperature influence*

393 Figure 9(a) shows the instantaneous surface temperature and heat flux
394 variation with coolant temperature for the three probes. The combustion
395 chamber wall temperature was varied by controlling the coolant and oil tem-
396 peratures to 60°C and 70°C, while all other parameters were maintained
397 constant to 2500 rpm, 25°C of inlet temperature, and 2 bar of inlet pres-
398 sure. The surface temperatures decrease linearly as the coolant temperature
399 decreases at the three locations. The zone with higher surface temperature
400 it still being located near the exhaust valve, while the lower temperature
401 is near the inlet. Differences of 160°C between the coolant and the hottest
402 spot of the walls are reached, this can seem quite high, but is normal as the
403 elongation piece it is not directly cooled.

404 Figure 9(b) shows the heat flux of the three probes for the two coolant
405 temperatures. As soon as the coolant temperature decreases, the local heat
406 flux decreases as well. This might seem counter-intuitive, but it may be due
407 to the fact that the gas temperature is the main driver in the transient heat

408 transfer process. While in zones 1 and 5 the variation of the fluxes when the
409 coolant temperature is risen, is more notorious, in zone 2 is almost negligible,
410 which is an indication of the uneven behaviour of the thermal boundary layer
411 in combustion engines. At the peak of heat fluxes a small delay can be noticed
412 for the hotter cooling temperatures. This might be due to the time that the
413 surface temperature will take to equal the gas temperature, longer with lower
414 wall values.

415 *4.5. Engine speed influence*

416 Inside the cylinder, heat transfer under motored conditions changes with
417 engine speed because of several reasons: with an increase in engine speed, the
418 gas motion increases and thus the convection coefficient; but as engine speed
419 rises, the engine also becomes more adiabatic as there is less time to exchange
420 energy. Engine speed also affects volumetric efficiency, producing changes in
421 pressure, temperature and mass values in the IVC and consequently in the
422 evolution of heat transfer during the entire cycle.

423 Figure 10(a) shows the temperature swings variation on location 1 with
424 different engine speeds for five motored tests in which the inlet density was
425 kept constant to 1.75 kg/m^3 . The other two locations are not plotted because
426 their behaviours are similar. The increment of engine speed produces a rise in
427 both wall temperatures, in depth and surface. This is reasonable as from 1000
428 until 2000 rpm the maximum peak of pressure (Figure 10(c)) rises gradually
429 as the engine speed increases, while the mass remains almost constant. In the
430 case of 2500 and 3000 rpm, though, the maximum pressure decreases down
431 to the value of the 2000 and 1000 rpm test, respectively, the total mass in
432 the cylinder is almost 20% and 40% lower and, as a result, the temperature

433 in the case of 2500, almost remain constant and in the case of 3000 suddenly
434 increases.

435 Figure 10(b) shows the heat flux through the walls calculated for the five
436 engine speeds, and the tendency is to increase with the speed, as expected,
437 owing to the turbulence and velocities are higher. The big difference between
438 the two

439 **5. Summary and Conclusions**

440 An original experimental approach to study of the gas-wall local heat
441 transfer on the walls of the combustion chamber of a Diesel engine was pre-
442 sented. The experimental device was designed and constructed in order to
443 recreate the main features of a real engine. The first set of instantaneous tem-
444 perature and heat transfer measurements indicate that the data obtained is
445 consistent and coherent with previous works performed by other researchers.
446 This means that not only the experimental facility had been validated but
447 also that the methodology and processing tools are set up to estimate heat
448 losses in fired conditions.

449 Parametric studies were performed in order to evaluate the effect of the
450 most important engine parameters. These should be useful in upcoming
451 studies aiming at proposing and verifying new heat transfer models based on
452 a combination of experimental and numerical computational work based on
453 fluid dynamic simulations. From the studies presented on independent vari-
454 ations of inlet pressure, engine speed, and coolant temperature the following
455 conclusions may be drawn:

- 456 • The heat transfer through the combustion chamber is strongly affected

457 by the air consumption.

458 • The transient heat transfer is strongly affected by the gas temperature
459 rather than by the inner wall temperatures.

460 • In order to evaluate engine speed influence it is not enough to keep the
461 inlet density constant, but it is also necessary to take into account the
462 volumetric efficiency effects.

463 • Engine heat transfer is not uniform in the different walls of the combus-
464 tion chamber due to the irregular behaviour of the thermal boundary
465 layer.

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601

602

603 **List of figures**

604

605 Fig. 1. Sketch of the final combustion chamber configuration (not to scale).

606 Fig. 2. Elongation piece with combustion chamber, heat flux sensors and
607 pressure transducer housing: (a) Final design, (b) Final constructed piece.

608 Fig. 3. Schematic diagram of the experimental facility and data acquisition
609 system.

610 Fig. 4. Heat flux probe design.

611 Fig. 5. Flowchart of the methodology applied to perform engine heat trans-
612 fer analysis.

613 Fig. 6. In-cylinder pressures of original engine configuration (black) and of
614 modified configuration (gray) for a motored cycle at 2000 rpm and 25°C of
615 inlet temperature and 2 bar of inlet pressure.

616 Fig. 7. (a) Local temperature swings and (b) local heat fluxes and pressure
617 measured at 2000rpm engine speed and 2.5 bar of inlet pressure and 25°C of
618 inlet temperature.

619 Fig. 8. (a-c) Local heat fluxes measured at different locations at 1500 rpm
620 and 25°C of inlet temperature (d) and in-cylinder pressure with different in-
621 let pressure.

622 Fig. 9. (a) Temperature profiles and (b) heat fluxes at three locations with
623 60°C (full line) and 70°C (dash line) at 2500 rpm, 2 bar, and 25°C of inlet
624 pressure and temperature.

625 Fig. 10. (a) Temperature swings, (b) heat fluxes, and (c) in-cylinder pressure
626 signal for five different speeds on location 1.

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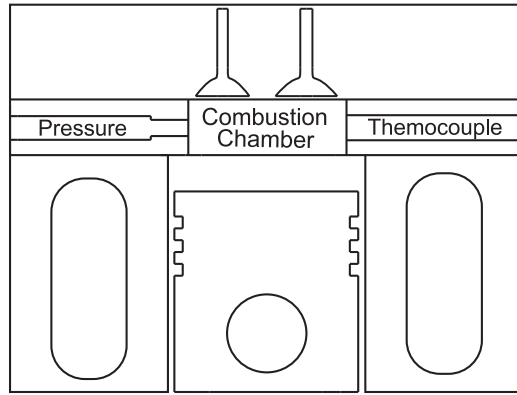


Figure 1: Sketch of the final combustion chamber configuration (not to scale)

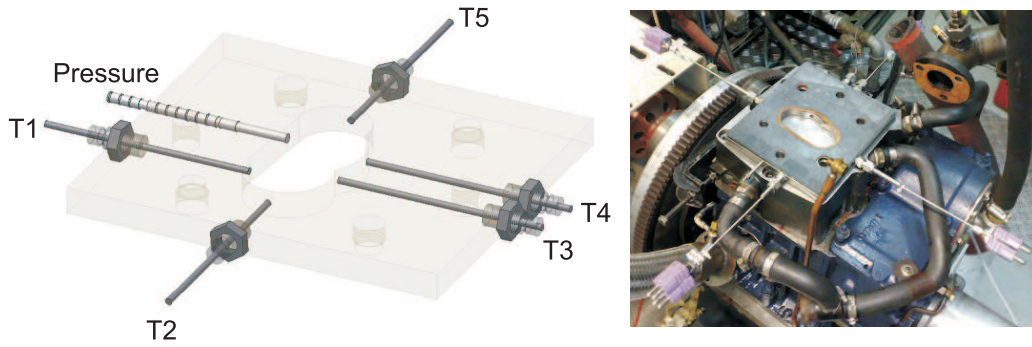


Figure 2: Elongation piece with combustion chamber, heat flux sensors and pressure transducer housing: (a) Final design, (b) Final constructed piece.

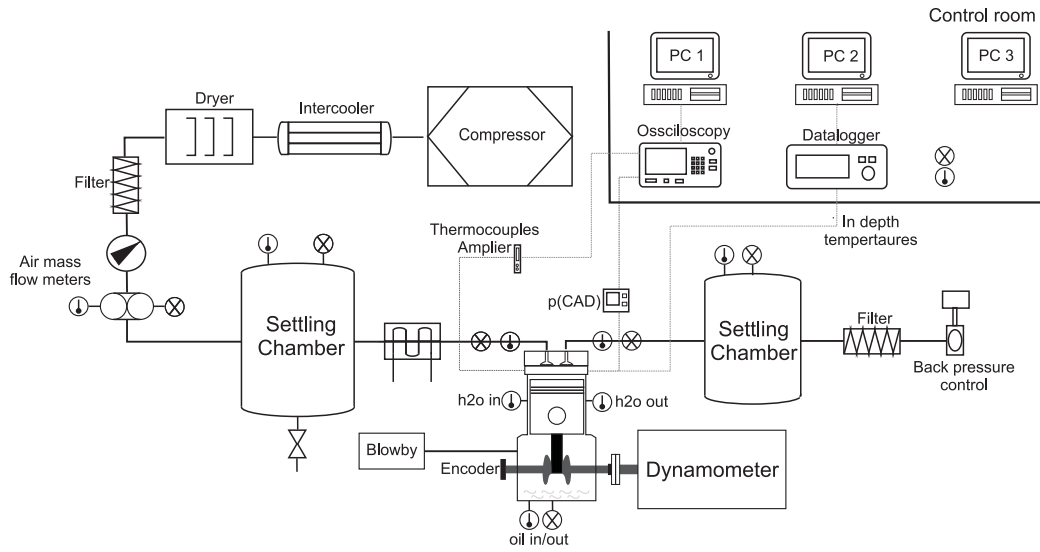


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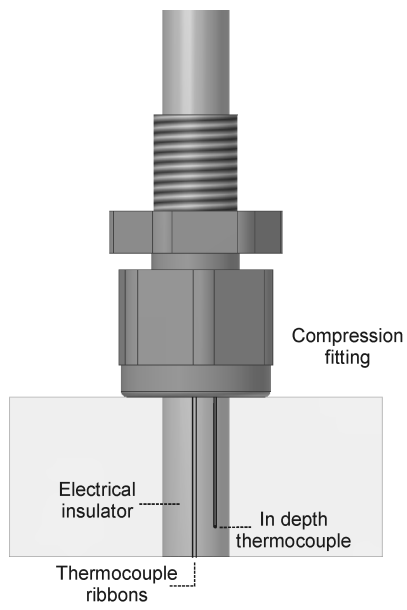


Figure 4: Heat flux probe design.

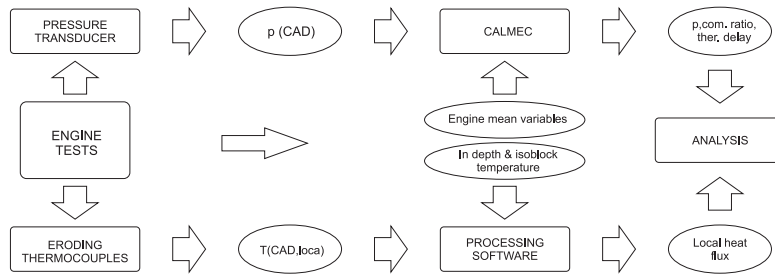


Figure 5: Flowchart of the methodology applied to perform engine heat transfer analysis.

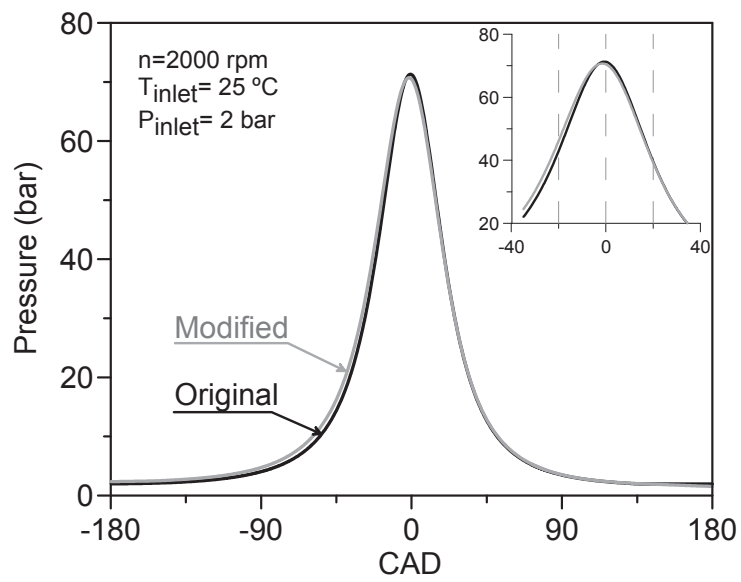


Figure 6: In-cylinder pressures of original engine configuration (black) and of modified configuration (gray) for a motored cycle at 2000 rpm and 25°C of inlet temperature and 2 bar of inlet pressure.

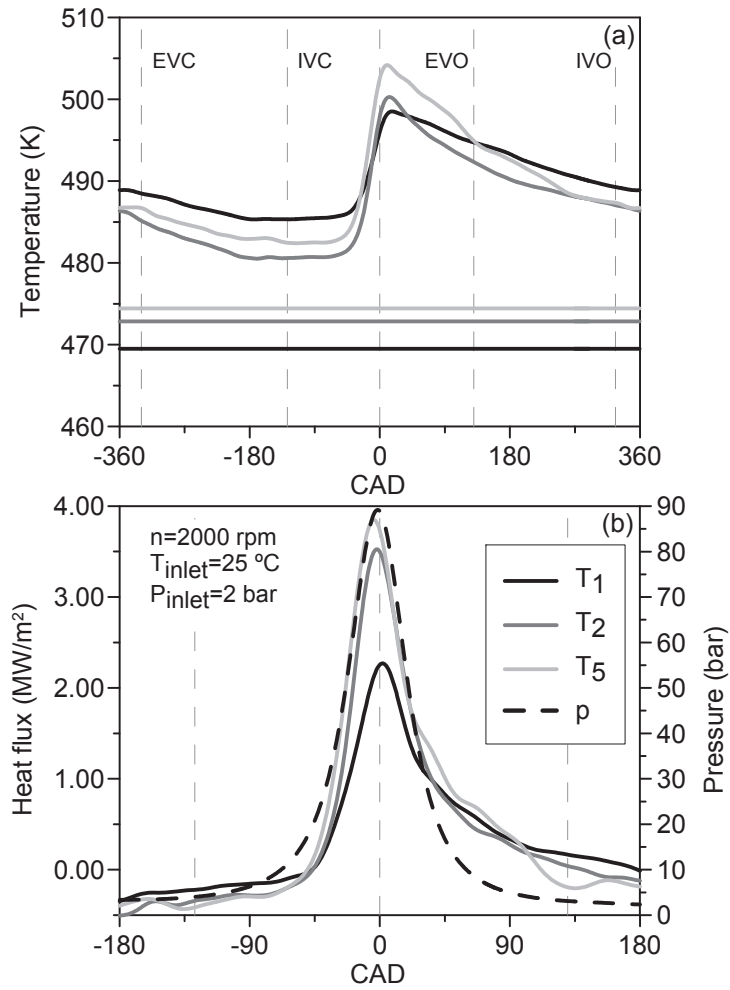


Figure 7: (a) Local temperature swings and (b) local heat fluxes and pressure measured at 2000rpm engine speed and 2.5 bar of inlet pressure and 25°C of inlet temperature.

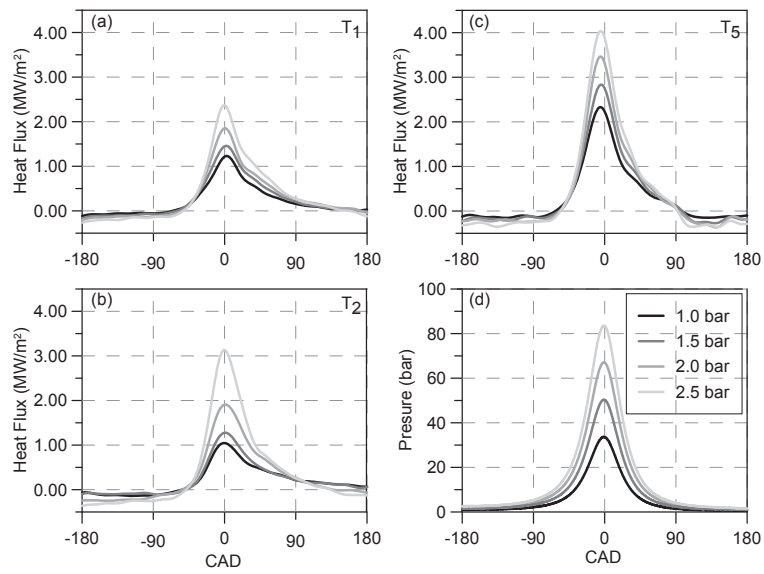


Figure 8: (a-c) Local heat fluxes measured at different locations at 1500 rpm and 25°C of inlet temperature (d) and in-cylinder pressure with different inlet pressure.

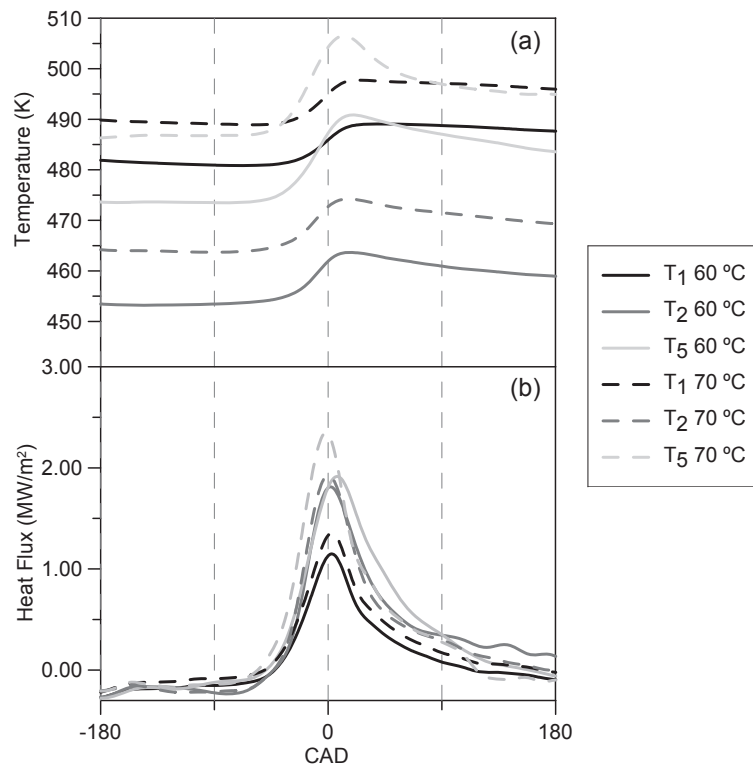


Figure 9: (a) Temperature profiles and (b) heat fluxes at three locations with 60°C (full line) and 70°C (dash line) at 2500 rpm, 2 bar, and 25°C of inlet pressure and temperature.

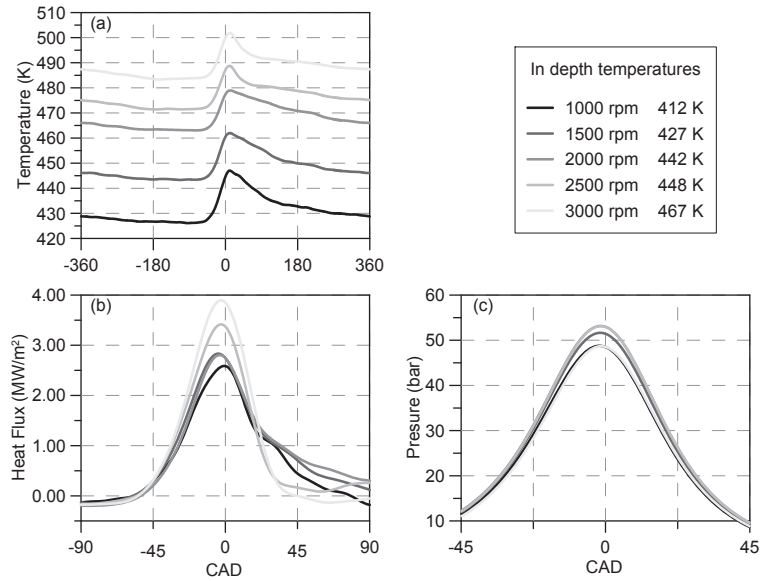


Figure 10: (a) Temperature swings, (b) heat fluxes, and (c) in-cylinder pressure signal for five different speeds on location 1.

628 **List of tables**

629

630 Table 1. Engine geometric characteristics.

631 Table 2. Motoring tests.

632

Table 1: Engine geometric characteristics.

Engine type	Single cylinder, four-stroke, water cooled	
Displacement	(l)	0.573
Bore/Stroke	(mm)	90/90
Original compression ratio		14.5 : 1
Modified compression ratio		13.8 : 1
Maximum speed	(rpm)	4000
Inlet/Exhaust valve diameter	(mm)	41.7/34.712
Inlet/Exhaust valve opening	(CAD)	70 bTDC/42.5 bBDC
Inlet/Exhaust valve closing	(CAD)	40 aTDC/72.5 aBDC

633

634

Table 2: Motoring tests.

Engine speed (rpm)	Intake pressure (bar)	Cooling water temperature (°C)
1000		
1500	1	
2000	1.5	60
2500	2	70
3000	2.5	