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Additional Information

Study of ignition delay time and generalization of auto-ignition for PRFs in a RCEM by means of natural chemiluminescence

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Abstract

An investigation of the effects of contour conditions and fuel properties on ignition delay time under Homogeneous Charge Compression Ignition (HCCI) conditions is presented in this study. A parametric variation of initial temperature, intake pressure, compression ratio, oxygen concentration and equivalence ratio has been carried out for Primary Reference Fuels (PRFs) in a Rapid Compression Expansion Machine (RCEM) while applying the optical technique of natural chemiluminescence along with a photo-multiplier. Additionally, the ignition delay time has been calculated from the pressure rise rate and also corresponding numerical simulations with CHEMKIN have been done. The results show that the ignition delay times from the chemical kinetic mechanisms agree with the trends obtained from the experiments.

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Moreover, the same mechanism proved to yield consistent results for both fuels at a wide range of conditions. On the other hand, the results from natural chemiluminescence also showed agreement with the ignition delay from the pressure signals. A $310nm$ interference filter was used in order to detect the chemiluminescence of the OH^* radical. In fact, the maximum area and peak intensity of the chemiluminescence measured during the combustion showed that the process of auto-ignition is generalized in the whole chamber. Moreover, the correlation of peak intensity, maximum area and ignition delay time demonstrated that natural chemiluminescence can also be used to calculate ignition delay times under different operating conditions. Finally, the area of chemiluminescence was proved to be more dependant on the fuel and ignition delay time than on the operating conditions.

Keywords: RCEM, ignition delay, OH^* chemiluminescence, PRF

1. Introduction

New combustion modes, such as Homogeneous Charge Compression Ignition (HCCI), Premixed Charge Compression Ignition (PCCI) and others based on Low Temperature Combustion (LTC), have shown high potential for the simultaneous reduction of soot and NO_x [1, 2]. These modes show virtually zero emissions of soot and NO_x , but high emissions of unburned hydrocarbons (UHC) and carbon monoxide (CO), by avoiding the soot and NO_x formation peninsulas, which can be seen in equivalence ratio - temperature diagrams [3]. However, UHC and CO can be easily oxidized with the current post-treatment systems. Thus, the main challenge to implement these new combustion strategies in commercial reciprocating internal com-

12 bustion engines is the lack of control over the autoignition process and over
13 the heat release rate [4].

14 The ignition control is much more difficult under these conditions because
15 the autoignition is controlled by the chemical kinetics of the charge [5], which
16 can be modified by adjusting the engine operating parameters, such as the
17 Exhaust Gas Recirculation (EGR) rate and the inlet temperature. Therefore,
18 it is necessary to improve the knowledge about the autoignition event and
19 about the combustion process under low temperature conditions to properly
20 modify the operating conditions of the engine and control the heat release.

21 Natural chemiluminescence is a non-intrusive optical technique widely
22 used in combustion diagnosis [6], such optical techniques are powerful tools
23 to analyze not only the ignition of homogeneous mixtures, but also different
24 parameters of sprays and even exhaust emissions. Natural luminosity analysis
25 and spectroscopy have shown to be able to describe the different phases of
26 the combustion process under HCCI conditions [7].

27 Mancaruso and Vaglieco [8] performed chemiluminescence measurements
28 in a transparent engine fuelled with RME and diesel commercial fuel. They
29 found that OH^* is responsible of the NO formation in the chamber and,
30 therefore, of much of the NO_x emissions. The OH^* behaviour in chamber
31 was strictly correlated to formation-oxidation of NO_x -PM, demonstrating
32 that OH^* chemiluminescence can be used to study exhaust emissions.

33 Dubreuil et al. [9] studied the global effect of the EGR on the HCCI
34 combustion of n -heptane in a transparent single-cylinder diesel engine for
35 two EGR rates at a constant equivalence ratio by means of OH^* chemilu-
36 minescence. By observing cool and main flame emissions, the authors found

37 that the EGR delays and degrades the combustion phenomenon. They also
38 proved that the natural emissions of combustion are sufficiently sensitive to
39 yield combustion process analysis. Finally, the authors observed that the
40 increase of the EGR rate decreases the OH^* chemiluminescence, which is
41 linked to the reduction of the global combustion reactivity.

42 Liu et al. [10] have used natural chemiluminescence to study the inho-
43 mogeneities present in HCCI combustion under different injection strategies
44 and cooling fluid temperatures. Also, their results have been compared to
45 CFD calculations.

46 Anders et al. [11] have performed studies in a truck size engine modified
47 to have optical accesses and have applied chemiluminescence of OH^* and
48 CH to describe the combustion process. Results show that no luminosity is
49 emitted during the NTC zone and that the radiation of the high temperature
50 combustion is one order of magnitude larger than that of the cool flames.

51 Jin and Zheng [12] have elaborated a review where the description and
52 literature revision of the diverse optical techniques applied to HCCI combus-
53 tion can be found.

54 The motivation of this study is the analysis of the combustion under
55 HCCI conditions in order to extend the aforementioned studies. The ignition
56 delay behavior is now analysed under a wider range of conditions, extending
57 the studies of Dubreuil et al. not only to other fuels, but also to different
58 equivalence ratios, compression ratios and EGR rates. The importance of
59 studying the ignition delay time under HCCI conditions is justified by the
60 role of this parameter in controlling the heat release rate and the efficiency
61 under such conditions. Additionally, it contributes to the HCCI knowledge

62 using *iso*-octane which has similar ignition characteristics to gasoline. The
63 optical study contributes by quantifying the homogeneity and propagation of
64 the combustion and also by allowing the ignition delay time to be determined
65 by chemiluminescence. Finally, a chemical kinetic mechanism for *n*-heptane
66 and *iso*-octane blends is validated in a broad range of operating conditions.
67 This will allow the use of the mechanism in future works of research.

68 In this study autoignition and combustion are studied under HCCI con-
69 ditions in a Rapid Compression Expansion Machine (RCEM) by means of
70 natural chemiluminescence. The study has been performed with two different
71 surrogate fuels with reactivities typical of diesel fuel and gasoline: *n*-heptane
72 and *iso*-octane, respectively. Despite the fact that more sophisticated surro-
73 gate fuels for diesel and gasoline can be found in the literature, *n*-heptane
74 and *iso*-octane were chosen because extended and fully validated chemical
75 kinetic mechanisms are available for both. Moreover, *n*-heptane, *iso*-octane
76 and their blends are Primary Reference Fuels (PRF) employed to define the
77 octane reference scale and they are widely used in the literature as surrogates
78 of diesel and gasoline under engine conditions [13].

79 Ignition delay time and chemiluminescent intensity distribution will be
80 experimentally obtained under different conditions of pressure, temperature,
81 equivalence ratio and oxygen mass fraction. Experiments are reproduced
82 with the software of chemical simulation CHEMKIN. This software, which
83 is developed by Reaction Design (ANSYS), is consolidated in the world of
84 engineering investigations and the chemical kinetics mechanisms of several
85 hydrocarbons are perfectly defined to be used with it. Finally, the numerical
86 results are validated experimentally using a RCEM.

87 The structure of the paper is as follows: First, the experimental facilities
88 involved in the study are presented. Then, the methodological approach is
89 described, including the experimental methods, the chemical kinetic simu-
90 lations and the parametric study performed. Afterwards, the trends of the
91 ignition delay time are analyzed and the chemical kinetic mechanism is val-
92 idated by comparison with experimental results. Next, the combustion pro-
93 cess is studied from a point of view of natural chemiluminescence, comparing
94 the experimental results with chemical simulations. Finally, the conclusions
95 of this study are shown.

96 **2. Experimental tools**

97 *2.1. Rapid Compression Expansion Machine*

98 A RCEM is an experimental facility widely used in autoignition studies
99 due to its capability to reproduce engine conditions [14]. It can replicate
100 reasonably well the combustion process of reciprocating engines with fully
101 controlled initial and boundary conditions while avoiding the complexities
102 associated to engines [15].

103 Different compression ratios can be reached in the RCEM by varying the
104 stroke and the clearance volume. Axial optical access is available [16] and the
105 compression velocity can be varied in order to simulate the effect of different
106 engine speeds. In a RCEM part of the expansion stroke of the piston can
107 be also analyzed and most of the engine parameters can be calculated, such
108 as the heat release rate or the combustion efficiency. In this facility both
109 homogeneous and heterogeneous (direct injection) mixtures can be tested, as
110 well as new combustion modes such as the dual fuel technology [17] or LTC

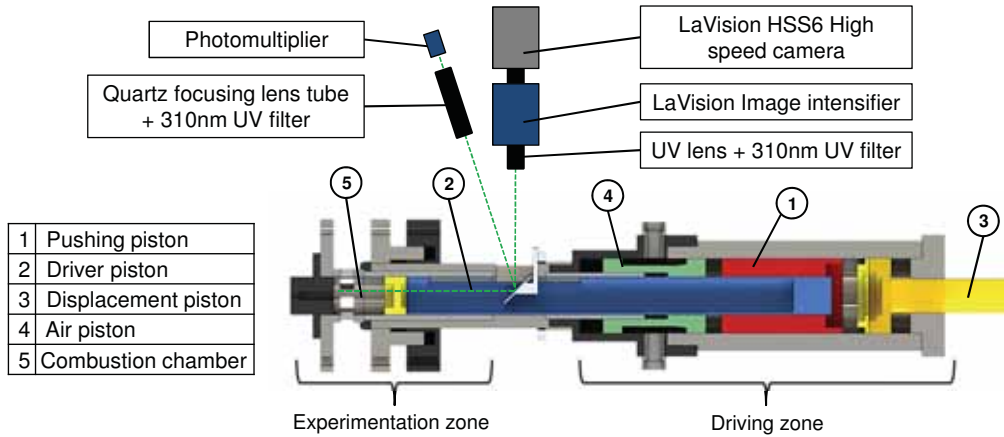


Figure 1: Rapid Compression Expansion Machine scheme.

111 [18].

112 A scheme of the RCEM is shown in Figure 1. The RCEM is pneumatically
 113 driven and its pistons are hydraulically coupled. As it can be seen, it can
 114 be divided in two different zones, the experimentation zone and the driving
 115 zone. The experimentation zone is composed by the combustion chamber,
 116 while the driving zone is composed by four different pistons. Piston 1, which
 117 is called pushing piston, is pneumatically driven and hydraulically coupled
 118 to piston 2, which is called driver piston and is directly connected with the
 119 combustion chamber. Piston 3 is hydraulically driven and it can be adjusted
 120 to select the compression stroke. Finally, piston 4 contains the compressed
 121 air that drives the machine.

122 The way in which the RCEM works is as follows: First, the oil is pressur-
 123 ized by the driving gas, which is compressed air. The driver piston does not
 124 move because it is perfectly coupled to piston 3, avoiding contact between
 125 the pushing oil and the piston base. Then, pressure is established behind

126 the driver piston by a bypass valve and it starts to advance at low veloc-
127 ity in a process called slow compression. It should be noted that when the
128 driver piston advances, the pushing piston must advance also in the oppo-
129 site direction, keeping constant the volume of oil. In fact, both pistons are
130 inertially balanced, leading to a process free of vibrations. When the driver
131 piston leaves the piston 3, it is suddenly accelerated and the rapid compres-
132 sion stroke starts. The driving air suffers an expansion process whereby its
133 pressure and, consequently, the pushing oil pressure, are reduced. The pis-
134 ton stops when the pressure in the combustion chamber is high enough to
135 compensate the pushing force and the inertia, defining TDC. Thereby, TDC
136 is highly dependent on the operation conditions of the RCEM, which is com-
137 pletely different for engines. Moreover, there is a certain maximum driving
138 pressure for each operating condition to avoid collision of the piston with
139 the cylinder head, since in the RCEM there is not any mechanism as the
140 rod-crank mechanism that fixes the maximum position of the piston. Once
141 the piston reaches TDC, the pressure in the combustion chamber is higher
142 than the pushing oil pressure and the expansion stroke starts. More details
143 on the operation principle of the RCEM can be found in [19].

144 The technical characteristics of the RCEM can be seen in Table 1. The
145 pushing piston and the driver piston are instrumented with two AMO LMK102
146 incremental position sensors ($0.01mm$ of resolution), which allow knowing
147 the absolute position of each piston and, therefore, the combustion chamber
148 volume. The combustion chamber is composed by three elements, the ex-
149 perimentation piston (mechanically connected to the driver piston), the liner
150 and the cylinder head. The experimentation piston consists of a steel-made

Bore	84	mm
Stroke	120 - 249	mm
Compression ratio	5 - 30	-
Maximum cylinder pressure	200	bar
Initial pressure	1 - 5	bar
Maximum heating temperature	473	K

Table 1: Technical characteristics of the RCEM.

151 piston with 84mm of bore and a quartz-made bowl with cylindrical shape,
152 50mm of bore and 2.2mm of depth, which allow the axial optical access. As
153 the bowl is flat, the chamber can be recorded without any image distortion.

154 Besides, the cylinder head and the cylinder liner have different heating
155 elements arranged in six separately controlled zones, which are responsible
156 for heating the cylinder walls and the experimentation piston. The wall tem-
157 perature is measured by a total of six K-type thermocouples, two located in
158 the cylinder head and four in the liner. Very good temperature homogeneity
159 has been observed [19], with a standard deviation of the gas temperature in
160 the order of 3K . It was found that the distribution of temperature is barely
161 affected by the gas in-flow due to its slow speed. An initial gas temperature
162 equal to the wall temperature is achieved due to the long duration of the
163 intake process.

164 The cylinder head is instrumented with a Kistler 7061B cooled piezo-
165 electric pressure sensor ($-80\text{pC}/\text{bar}$ of sensitivity), which is coupled to a
166 Kistler 5011 charge amplifier, and whereby the in-cylinder pressure is mea-

167 sured. Different piezo-resistive pressure sensors are available to control the
168 filling of the driving gas and of the combustion chamber (0.01bar of reso-
169 lution). The injection system is composed by a Siemens hollow cone piezo-
170 injector with a cone angle of 90° , which is centered in the cylinder head. Its
171 fuel delivery rate has been previously measured with an IAV injection rate
172 analyzer. The transient signals have been recorded at $100kHz$ with a PC-
173 based transient measurement recorder. The RCEM is filled from an external
174 tank that can be heated up to $373K$. The synthetic air is produced in the
175 tank by a filling based on partial pressures where N_2 , CO_2 and O_2 can be
176 used. The mixture is analyzed in a Horiba PG-250 portable gas analyzer in
177 order to know the exact composition and ensure the correct reproduction of
178 the experiments in CHEMKIN.

179 2.2. Optical setup

180 The area of autoignition and the luminous intensity were recorded by OH^*
181 chemiluminescence imaging. This technique records radiation at $310nm$,
182 which is controlled by the OH^* radical, a marker of the high temperature
183 combustion [20].

184 A schematic of the optical arrangement is shown in Figure 1. The camera
185 has been pointed directly at the mirror inside the machine, which due to its
186 45° tilt gives a direct view of the combustion chamber through the piston
187 window. A 12-bit LaVision HighSpeedStar 6 camera coupled to a LaVision
188 HighSpeed IRO intensifier equipped with a $100mm$ focal length $f = 2$ UV
189 objective (by Bernhard Halle Nachfolger GmbH) were used for image acqui-
190 sition; additionally, a $310nm$ interference filter (FWHM = $10nm$) was used
191 to eliminate any additional radiation outside the OH^* radical wavelength.

192 Because of the transient nature of the combustion, a acquisition frequency
193 of $30kHz$ has been chosen in order to capture the evolution of the radical
194 inside the combustion chamber. An exposure time of $33\mu s$ and a rectangular
195 image of 384×448 pixels allowed to see the whole effective window diameter
196 of $50mm$ while obtaining a pixel/ mm ratio of 6.89. The maximum expo-
197 sure time has been selected in order to use lower gain values and therefore
198 reducing image noise.

199 Additionally, a Hamamatsu H5784-03 photo-multiplier (PHM), spectrally
200 filtered at $310nm$, captured the spatially integrated light emission through
201 the piston window. It was placed at an angle due to the lack of space (Figure
202 1). The information from the photo-multiplier is complementary to the one
203 obtained by the camera, as its higher dynamic range allows to see peaks in
204 OH^* luminosity when the camera might be saturating. On the other hand,
205 the camera shows the distribution of the radical in a 2-dimensional image,
206 while the photo-multiplier only returns an integrated value for the whole
207 area.

208 The transient pressure, piston position and photo-multiplier signals along
209 with control and synchronization signals (i.e. camera triggers) have been
210 recorded at 100 kHz with a PC-based transient measurement recorder.

211 **3. Methodological approach**

212 *3.1. Rapid Compression Expansion Machine*

213 The desired stroke of the machine is selected and the RCEM is heated up
214 to the desired temperature. Then, the synthetic air-EGR mixture is prepared
215 in the mixing tank by a filling based on partial pressures. In this study, EGR

216 was considered as a combination of 20% CO_2 + 80% N_2 in volume, and it
217 is mixed with dry air until the amount of oxygen in the mixture is the one
218 desired by the user.

219 The combustion chamber is scavenged several times before the filling.
220 The fuel is injected into the combustion chamber at the start of the intake
221 process to avoid problems of stratification or other inhomogeneities. The
222 long duration of the process (approximately 40s), are enough to guarantee
223 a homogeneous environment in the chamber when the compression stroke
224 starts.

225 In order to ensure a representative ignition delay time measurement the
226 number of repetitions of each point has been selected so that the semi-
227 amplitude of the confidence interval with a level of confidence of 95% is
228 smaller than 1% of the mean ignition delay value.

229 In this work the autoignition of the mixture is considered to be produced
230 when the time derivative of the pressure signal (which will be referred as
231 pressure rise rate or, simply, pressure rise further on) reaches a maximum.
232 Thus, the ignition delay in the experimental facility is defined as the time
233 between the start of the rapid compression process and the instant in which
234 the maximum pressure rise is obtained, as can be seen in Figure 2. This way,
235 cool flames and high temperature ignition delay can be easily distinguished
236 in case of two-stage ignition.

237 Finally, the temperature profile is calculated for each experiment by ap-
238 plying the energy equation, since the pressure profile and the position of the
239 piston are known. The heat losses are characterized by a model based on the
240 Woschni correlation [21]. The calculation includes two additional models for

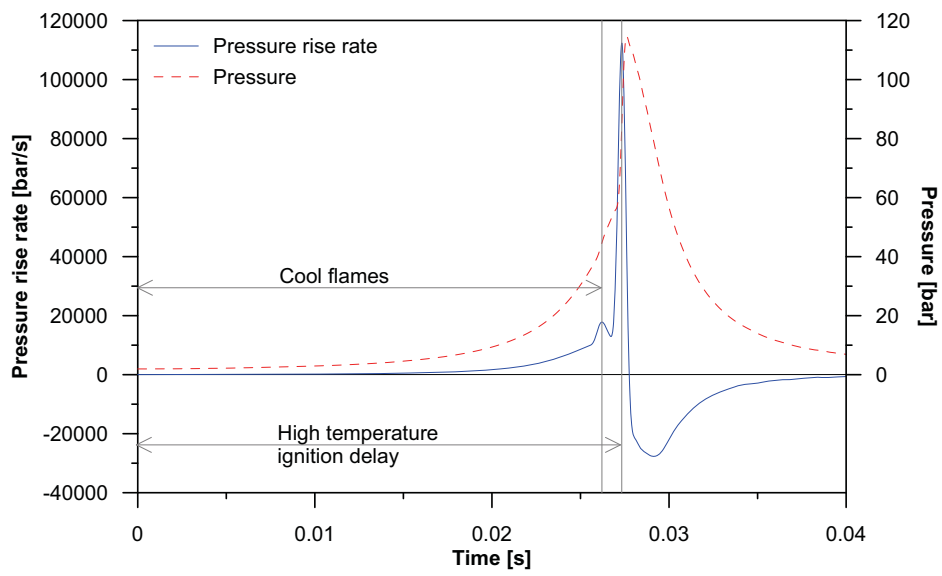


Figure 2: Ignition delay definition. The autoignition of the mixture is considered to be produced when the maximum pressure rise occurs. *n*-heptane at CR 17, T_i 358K, EGR 47% and Fr 0.58.

241 deformations and leaks, both of them explained in [22, 23].

242 3.2. OH^* chemiluminescence

243 The images have been processed by an in-house developed routine in
244 MATLAB. The processing algorithm starts by calculating the maximum pixel
245 intensity of each frame in order to determine the useful dynamic range of the
246 image sequence. Then, a background noise level is calculated by averaging
247 100 images where there is no presence of OH^* luminosity; also, the maximum
248 pixel intensity of all the images is determined. With this two values, the cut-
249 off intensity is calculated using the probable error in order to select which
250 frames are to be processed. A sample of the evolution of the pixel intensity,
251 the background level and the selected points are shown in Figure 3.

252 The area of the natural chemiluminescence is determined by using two
253 masks, one based on the window geometry and one on the luminosity. Figure
254 4-a shows a raw image straight from the camera, as it can be seen there is a
255 wide range of intensity values inside the chamber as well as reflections outside
256 the window. In order to discriminate the light reflected by the piston and
257 cylinder walls, a geometrical mask with the diameter of the effective window
258 is applied to all the images, leaving only the light that passes directly through
259 the window to be processed; a sample of that mask is shown in Figure 4-b.
260 Afterwards, a second mask based on the pixel intensity is applied. This mask
261 is determined using the values of the maximum pixel and the background
262 level previously calculated and a constant value to calculate a threshold.
263 Therefore, all the values that fall below this threshold are considered to
264 be part of the background, and the values above are accounted as OH^*
265 chemiluminescence; a sample of such mask is shown in Figure 4-c. Finally,

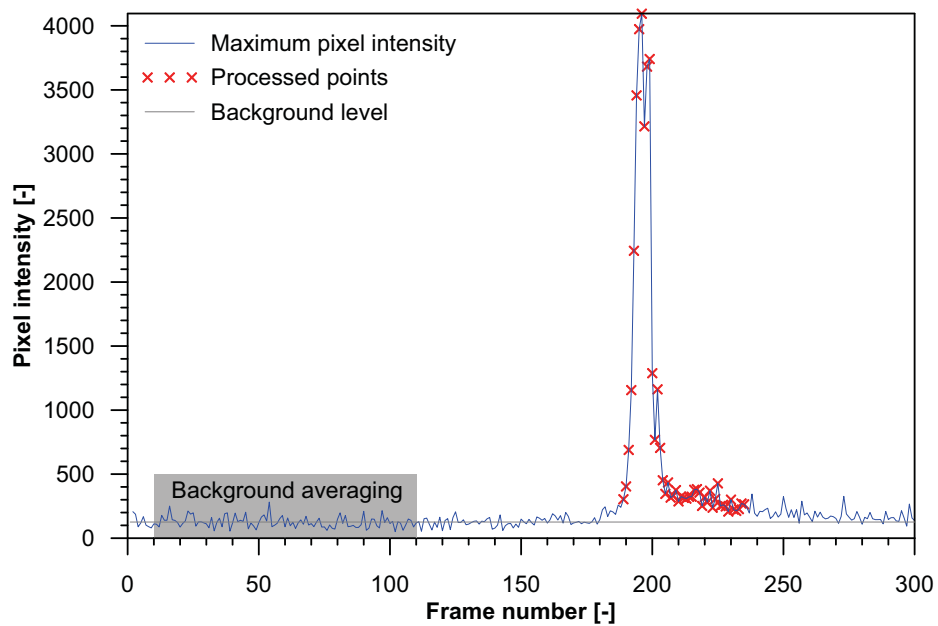


Figure 3: Evolution of maximum pixel intensity for frame sequence.

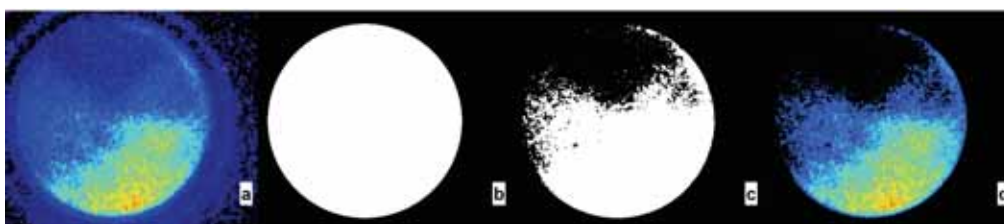


Figure 4: Processing sequence for raw image (a), geometrical mask (b), intensity mask (c) and final image (d).

266 the true intensity image excludes all the background light and the reflected
267 light, as shown in Figure 4-d. Once the final image is obtained, the OH^* area
268 is calculated as a percentage of the full window and also the accumulated
269 light intensity. Furthermore, since the images are time-resolved the instant
270 at which the largest area and peak of intensity take place are also determined.

271 An important factor to keep in mind while applying two-dimensional
272 imaging on a 3D phenomenon is the following. Since the light detected is an
273 integrated value of the whole volume and not a single first plane acquisition,
274 the maximum local intensity per volume may not be accurately detected by
275 the optical techniques applied. The intensity gradients could have had an
276 effect on the threshold of the images to obtain the combustion area, so a
277 high-intensity single point could have been ignored. Nevertheless, since the
278 combustion of homogeneous mixtures is being studied the existence of high-
279 intensity single points is very unlikely. Furthermore, for the calculation of
280 the total combustion area this single non-detected point should represent a
281 very small deviation.

282 3.3. *CHEMKIN and chemical kinetic mechanisms*

283 As previously mentioned, CHEMKIN is the software used to obtain the
284 different ignition delays and critical concentrations. The version used is
285 CHEMKIN-PRO. Curran's kinetic mechanism is used for *n*-heptane and
286 *iso*-octane [24, 25]. This mechanism consists of 1034 species and 4238 re-
287 actions, and includes the chemical kinetics of the two hydrocarbons used in
288 this investigation. Its validity has been checked in several articles [13, 26] by
289 comparison with experimental results.

290 The model used to obtain ignition delay times under variable conditions

291 is a reciprocating internal combustion engine operating with homogeneous
292 charge (IC-engine, closed 0-D reactors from CHEMKIN). The volume profile
293 as well as the heat loss profile are imposed in order to reproduce the RCEM
294 conditions. The piston starts at bottom dead center (BDC) and a complete
295 cycle of the RCEM is simulated. The autoignition is considered to be pro-
296 duced when the time derivative of the pressure signal reaches a maximum.
297 This is the same criterion as the used in the experiments and, therefore, it
298 allows comparing the simulated results directly with the experimental ones.
299 Moreover, the OH^* , CO and CO_2 concentration profiles are obtained and
300 an analysis of their reaction rates has been performed in order to compare
301 the simulations with the results obtained from the photo-multiplier and from
302 the high speed camera. This way, the ignition delay referred to a maximum
303 concentration can be directly compared with the ignition delay referred to
304 natural chemiluminescence. The maximum time step for CHEMKIN simu-
305 lations has been set as 10^{-5} s which is the experimental resolution of the
306 pressure and photo-multiplier signals.

307 *3.4. Parametric study performed*

308 The performed experimental study was as follows:

- 309 • Fuel: *n*-heptane and *iso*-octane.
- 310 • Initial temperature (T_i): 358K (only for n-heptane), 383K, 408K,
311 433K and 458K.
- 312 • Initial pressure (P_i): 0.14MPa and 0.17MPa.
- 313 • Compression stroke: 249mm.

- 314 • Compression ratio (CR): 15 and 17.
- 315 • Oxygen mass fraction (Y_{O_2}): 0.23 (0% EGR), 0.147 (30% EGR), 0.126
316 (40% EGR) and 0.105 (50% EGR).
- 317 • Equivalence ratio (Fr): from 0.3 to 0.8 depending on the fuel and on
318 the oxygen mass fraction.

319 The maximum equivalence ratio is limited by the working oxygen mass
320 fraction in order to avoid extremely violent combustions. The equivalence
321 ratio of 0.4 has been chosen as base point in order to have the possibility to
322 try leaner and richer mixtures without damaging the facility. The performed
323 parametric study can be seen in Table 2. Finally, the temperature of the
324 combustion chamber is always above the boiling point of the fuel, therefore
325 ensuring that the fuel is in vapour phase before the beginning of the cycle.

		T_i [K]				
		358	383	408	433	458
Fr [-]	0.3	40		0, 30, 40, 50		40
	0.4	0, 30, 40, 50	40, 50	0, 30, 40, 50	40, 50	0, 30, 40, 50
	0.5	40	40	40, 50	40	40
	0.6	40		40, 50		40
	0.7			40, 50		
	0.8			40, 50		

Table 2: Parametric study performed. EGR percent for different initial temperatures and equivalence ratios. Blue.- exclusively for n-heptane. Red.- exclusively for iso-octane.

326 4. Results and discussion

327 In this section, the trends of the ignition delay time, including the Neg-
328 ative Temperature Coefficient (NTC) behavior, are analyzed. This phe-
329 nomenon is referred to the loss of reactivity when the temperature is increased
330 due to the promotion of formation of olefins by the alkyl radicals, which com-
331 petes with the formation of chain-carriers, retarding the ignition. Therefore,
332 the ignition delay time increases when the temperature is increased in a cer-
333 tain range (NTC zone).

334 The validity of the chemical kinetic mechanism is checked by comparing
335 them directly with the experimental results. Then, the combustion process
336 is studied from a point of view of natural chemiluminescence. The general-
337 ization of the auto-ignition (percent of the combustion chamber ignited) is
338 analyzed. Finally, the luminous intensity is studied, comparing the exper-
339 imental results from the photo-multiplier and from the high speed camera
340 with the chemical simulations.

341 4.1. Study of the ignition delay time and validation of the chemical mecha- 342 nism

343 Ignition delay times obtained solving the *n*-heptane and *iso*-octane de-
344 tailed chemical kinetic mechanism are compared with the experimental re-
345 sults as a method to validate the mechanism in the desired range.

346 As can be seen in Figure 5, simulations reproduce with high accuracy
347 not only the trends, but also the values of the experimental results. Ignition
348 delay is defined in both cases as the time between the start of the rapid
349 compression process and the instant at which the maximum pressure rise

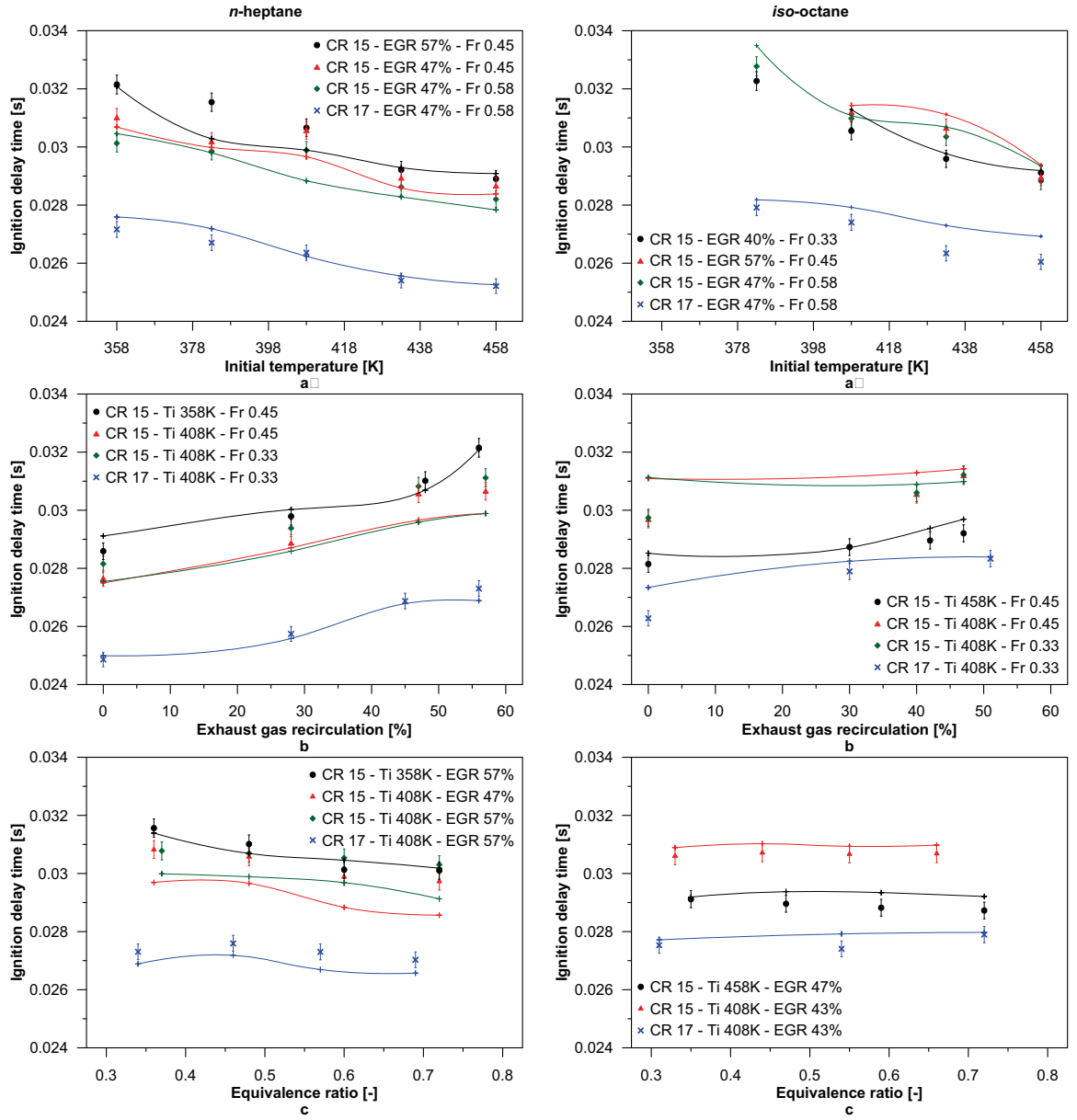


Figure 5: Ignition delay versus initial temperature (a), EGR rate (b) and equivalence ratio (c) for both fuels and under different operating conditions. Markers - RCEM, Lines - CHEMKIN.

350 OCCURS.

351 As expected, the following trends in the ignition delay were experimen-
352 tally observed (markers in Figure 5):

353 • Ignition delay decreases when the initial temperature is increased. For
354 *n*-heptane, this is the case except in the negative temperature coef-
355 ficient zone, NTC. In the parametric study performed, and if other
356 parameters such as pressure, equivalence ratio and oxygen fraction are
357 obviated, the ranges of the NTC zone are the following:

358 – *iso*-Octane: initial temperature between 408K and 433K.

359 – *n*-Heptane: initial temperature between 383K and 408K.

360 Moreover, it can be seen that the NTC zone becomes less pronounced
361 if the pressure is increased. In the same way, it can be seen that the
362 NTC behaviour becomes less evident if the EGR rate is increased in
363 the explored range. This is caused because a higher EGR rate leads
364 to lower temperatures, leading to an ignition outside the NTC zone.
365 However, the NTC behaviour becomes clearer if the oxygen proportion
366 is reduced at constant ignition temperature [27]. Besides, the NTC zone
367 becomes less apparent if the equivalence ratio is increased. Finally and
368 as expected, the NTC zone of the *n*-heptane is more distinct than the
369 NTC zone of the *iso*-octane.

370 • Ignition delay decreases when the compression ratio is increased, since
371 higher temperatures are reached in the cycle.

- 372 • Ignition delay increases when the EGR rate is increased. Lower amount
373 of oxidizer implies minor reactivity. Moreover, higher EGR rates im-
374 plies higher amount of CO_2 , which implies higher heat capacity of the
375 mixture and lower temperatures reached in the cycle.
- 376 • Ignition delay decreases when the equivalence ratio is increased in the
377 explored range. Because the reaction paths at low temperatures are
378 dependant on radical species formed directly from the fuel, the richer
379 mixtures have lower ignition delays than the leaner ones. Whereas this
380 trend is clear for *n*-heptane, the ignition delay seems to be much more
381 independent of the equivalence ratio for *iso*-octane.

382 The percentage deviation in ignition delay (ϵ), was calculated in order to
383 compare more easily experimental and simulation results. This deviation is
384 defined as follows:

$$\epsilon = \frac{t_{i,ICE} - t_{i,RCEM}}{t_{i,RCEM}} \times 100 \quad (1)$$

385 where t_i represents the time of ignition (ignition delay under variable con-
386 ditions). The subscript *ICE* represents a data obtained from a chemical
387 simulation with CHEMKIN using a closed 0-D IC-engine reactor. Finally,
388 the subscript *RCEM* represents a data obtained experimentally from the
389 RCEM.

390 The ignition delay deviation is shown in Figure 6 for all cases and both
391 fuels. The results show that simulations are able to reproduce the experi-
392 mental ignition delays with quite good accuracy. The average of the errors in
393 absolute value, $\bar{\epsilon} = \sum |\epsilon| / n$, has been calculated for each fuel. In fact, the

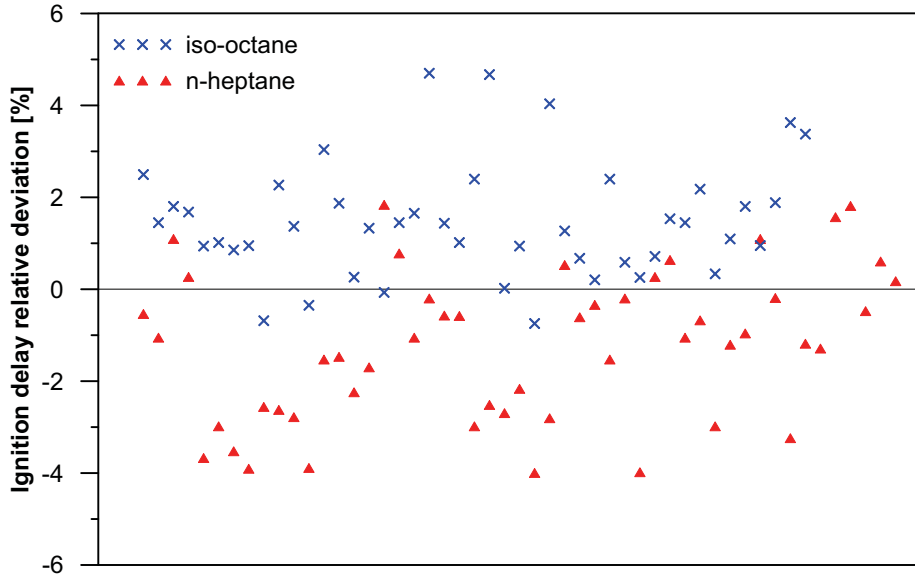


Figure 6: Percentage deviation in ignition delay.

394 confidence interval for the mean deviation with a confidence level of 95% is
 395 equal to $[1.33, 1.98]$ % for *n*-heptane and to $[1.21, 1.89]$ % for *iso*-
 396 octane. Ignition delay errors are caused partly by the chemical kinetic mechanism used
 397 and partly by the uncertainties in the calculation of the effective volume and
 398 the heat losses of the RCEM.

399 In Figure 6 the chemical kinetic mechanism under-predicts the ignition
 400 delay for *n*-heptane whereas it over-predicts the ignition delay times for *iso*-
 401 octane. This fact is independent of the physical models used to characterize
 402 the heat losses and the deformations, since they are the same for both fu-
 403 els. Moreover, the temperature is always high enough to keep the fuel in
 404 vapor phase, therefore these discrepancies are not due to a physical evapo-

405 ration time. The chemical mechanism that was used has been designed from
406 the detailed chemical kinetics of *n*-heptane [24] and *iso*-octane [25]. More-
407 over, the mechanism of *iso*-octane is based on that of *n*-heptane, where in
408 order to reproduce the very low reactivity observed experimentally at low
409 temperatures (600-770 *K*) the rates of alkyl-peroxy radical isomerization
410 and peroxy-alkylhydroperoxy radical isomerization have been decreased by
411 a factor of three (relative to the *n*-heptane). The reason why the isomer-
412 ization rates of *iso*-octane are slower than those of *n*-heptane is not clear
413 (other pathways could occur at low temperatures). The authors think that
414 this reduction of isomerization rates is the cause why the mechanism under-
415 predicts the ignition delay times for *n*-heptane, and over-predicts them for
416 *iso*-octane.

417 4.2. Generalization and intensity of the auto-ignition

418 The radiation profiles are compared to the oxidation rates of *CO* to *CO*₂
419 and also to the *OH*^{*} concentration obtained from CHEMKIN (Figure 7).
420 The radiation emitted at 310nm could have two possible origins. On one
421 hand it could be due to the natural chemiluminescence of the *OH*^{*} radical,
422 and on the other hand, it could also come from the *CO* continuum (oxidation
423 of *CO* to *CO*₂).

424 The oxidation of *CO* and the accumulation of *OH*^{*} occur simultaneously.
425 Therefore, a priori, it is not possible to decide without a spectrograph if the
426 natural chemiluminescence at 310 nm belongs to *OH*^{*} or if it is out-shined
427 by the *CO* continuum. However, it should be noted that the *time of life* of
428 the luminous intensity is very different in case of belonging to *CO* continuum
429 or to *OH*^{*}, as can be seen in Figure 7, where the *time of life* of the *OH*^{*}

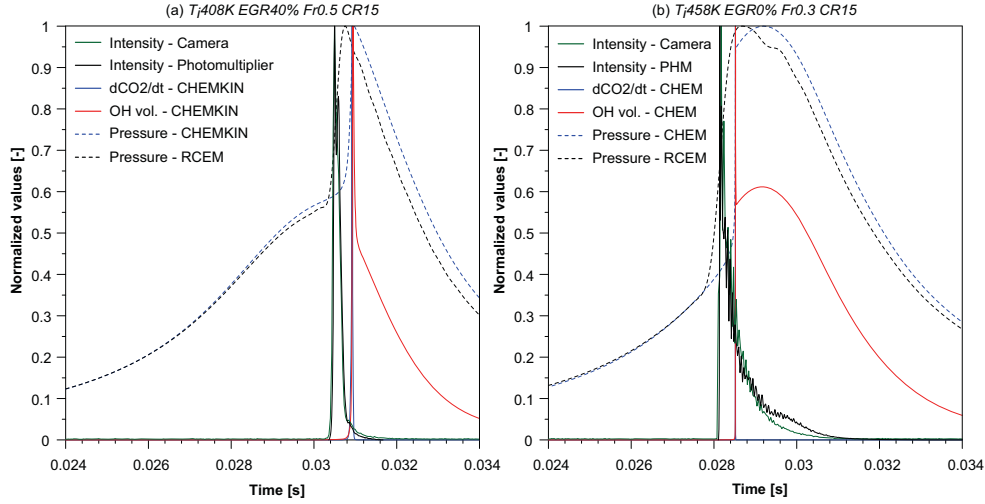


Figure 7: Normalized evolution of the oxidation rate of CO , OH^* molar fraction, natural chemiluminescence intensity from camera and photomultiplier, and simulated and experimental pressures for two different cases, based on chemiluminescence *time of life*. Left.- Short *time of life*. Right.- Long *time of life*.

430 chemiluminescence is 0.65 *ms* longer than the CO continuum luminosity.
 431 Thus, it is possible to use this parameter as a criterion to determine the
 432 source of the radiation.

433 Finally, from Figure 7 it can be seen that the profiles obtained from the
 434 photomultiplier and by integrating the luminosity detected by the camera
 435 are almost identical. This is a logical results as both detection methods are
 436 measuring the same parameter in parallel.

437 The generalization of the auto-ignition is defined as the percentage of
 438 the combustion chamber that has ignited; and since the auto-ignition is a
 439 transient event it can only be seen for a brief moment. Coincidentally the

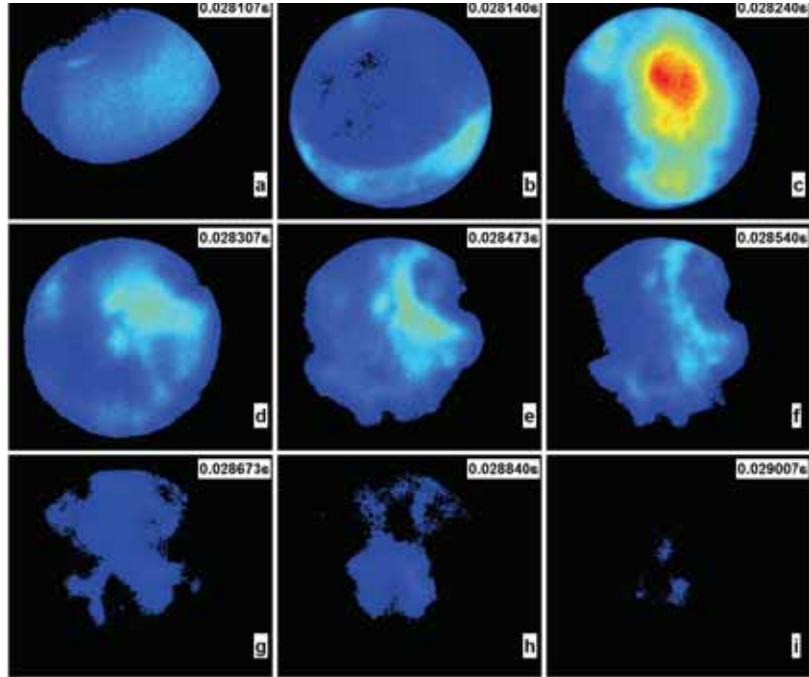


Figure 8: Evolution of chemiluminescence inside chamber for *iso*-octane at CR 15, T_i 458K, EGR 0% and Fr 0.3.

440 instant at which the largest area takes place is also the moment at which
 441 the radiation of the natural chemiluminescence is most intense. Moreover,
 442 both points also happen at the peak of the pressure rise curve; therefore, the
 443 mixture ignites abruptly causing a rise in the pressure while the radiation is
 444 maximum and present throughout the whole combustion chamber.

445 Figures 8 and 9 show two sequences of images where the evolution of
 446 the natural chemiluminescence over time can be seen in terms of intensity
 447 and area. Additionally, it can be seen that the ignition starts at the top of
 448 the combustion chamber, which occurred in most of the cases. This could be

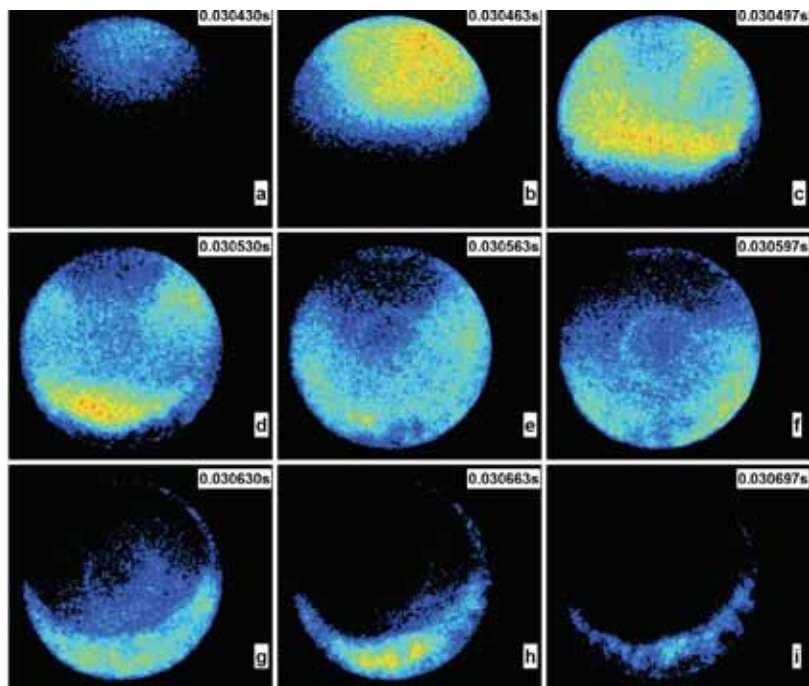


Figure 9: Evolution of chemiluminescence inside chamber for *iso*-octane at CR 15, T_i 408K, EGR 40% and Fr 0.5.

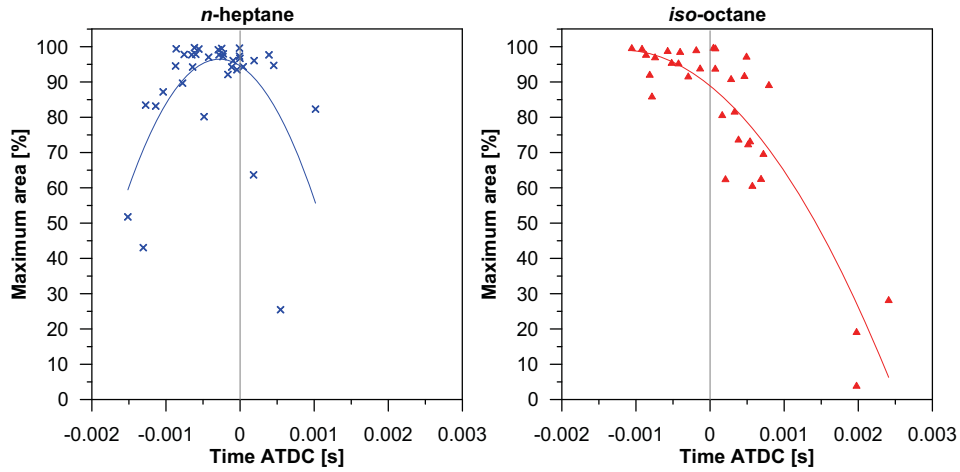


Figure 10: Maximum illuminated area versus time of largest area for both fuels at all conditions.

449 due to non-homogeneous temperature in the chamber walls; nevertheless, the
 450 ignition starts and is very quickly generalized, which is consistent with Figure
 451 7. The luminosity is then exhausted first at the top, as the combustion began
 452 there and the last section of the window to show any chemiluminescence is,
 453 of course, at the bottom. The observed flame expansion velocity is too fast
 454 for a real flame front propagation, which supports the argument that it is an
 455 autoignition process.

456 As a final point of analysis using the area occupied by natural chemilumi-
 457 nescence, Figure 10 shows the evolution of the maximum area as a function
 458 of the instant after TDC at which the ignition takes place. The maximum
 459 area is highly dependant on the instant at which the ignition takes place,
 460 because if even though all the initial conditions are favourable for a full-area

461 auto-ignition, if the ignition occurs far away of TDC, the thermodynamic
462 conditions in the combustion chamber are not optimal for the propagation
463 of the auto-ignition front, leading to slower ignitions and areas below 100% .
464 With this in mind, the trends presented in Figure 10 back up the hypotheses
465 of area vs time relationship, as we can see that for *n*-heptane which has ig-
466 nition delay times close to TDC most values are very high and close to 100%
467 (Figure 10-left), and the values that either happen too soon or too late have
468 much smaller areas. For *iso*-octane the trend is also similar, however the
469 values are not. Since for this second fuel most ignition delay times happen
470 after TDC, hence the fit points downwards instead of being balanced on the
471 center (Figure 10-right).

472 The maximum luminous intensity is analyzed as an estimator of the com-
473 bustion intensity. As can be seen in Figure 11, the maximum intensity has
474 a potential behaviour with the maximum temperature reached in the cycle.
475 Moreover, the luminous intensity is higher in case of *iso*-octane for the same
476 temperature. Figure 12 shows the trends of the combustion intensity with
477 the initial temperature (a), the EGR rate (b) and the equivalence ratio (c)
478 for both fuels and under different operating conditions. As expected, the
479 higher the initial temperature, the lower the EGR rate or the higher the
480 equivalence ratio, the higher the maximum OH^* luminous intensity. How-
481 ever, a higher reactivity of the mixture does not imply a higher combustion
482 intensity (changes of trend in Figure 12). If ignition occurs far away of TDC,
483 the maximum thermodynamic conditions reached can be lower than the ones
484 obtained with a less reactive mixture that ignites near TDC, leading to lower
485 luminous intensity.

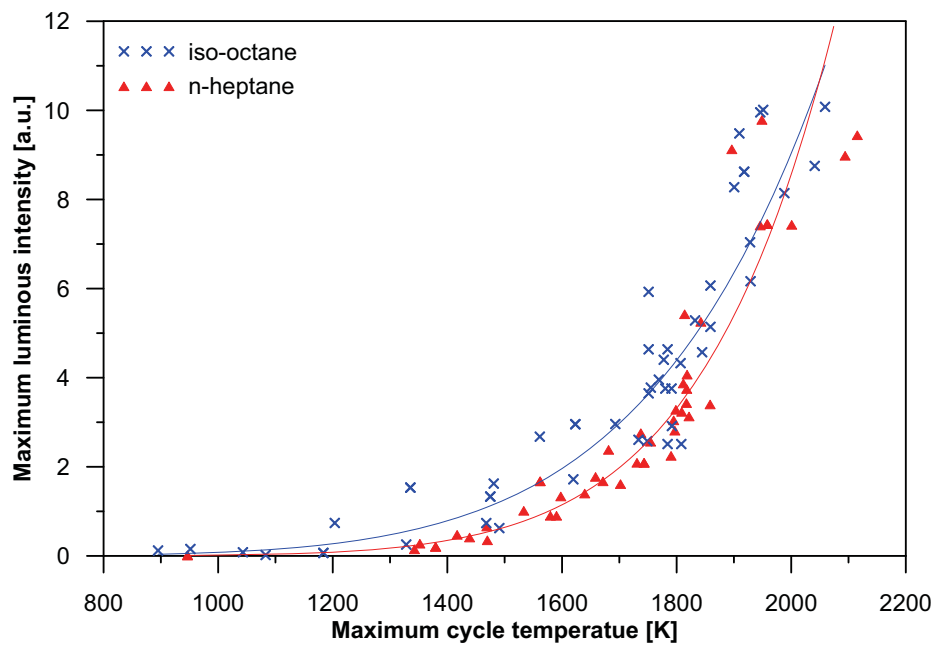


Figure 11: Maximum luminous intensity from the photo-multiplier versus maximum temperature reached in the cycle.

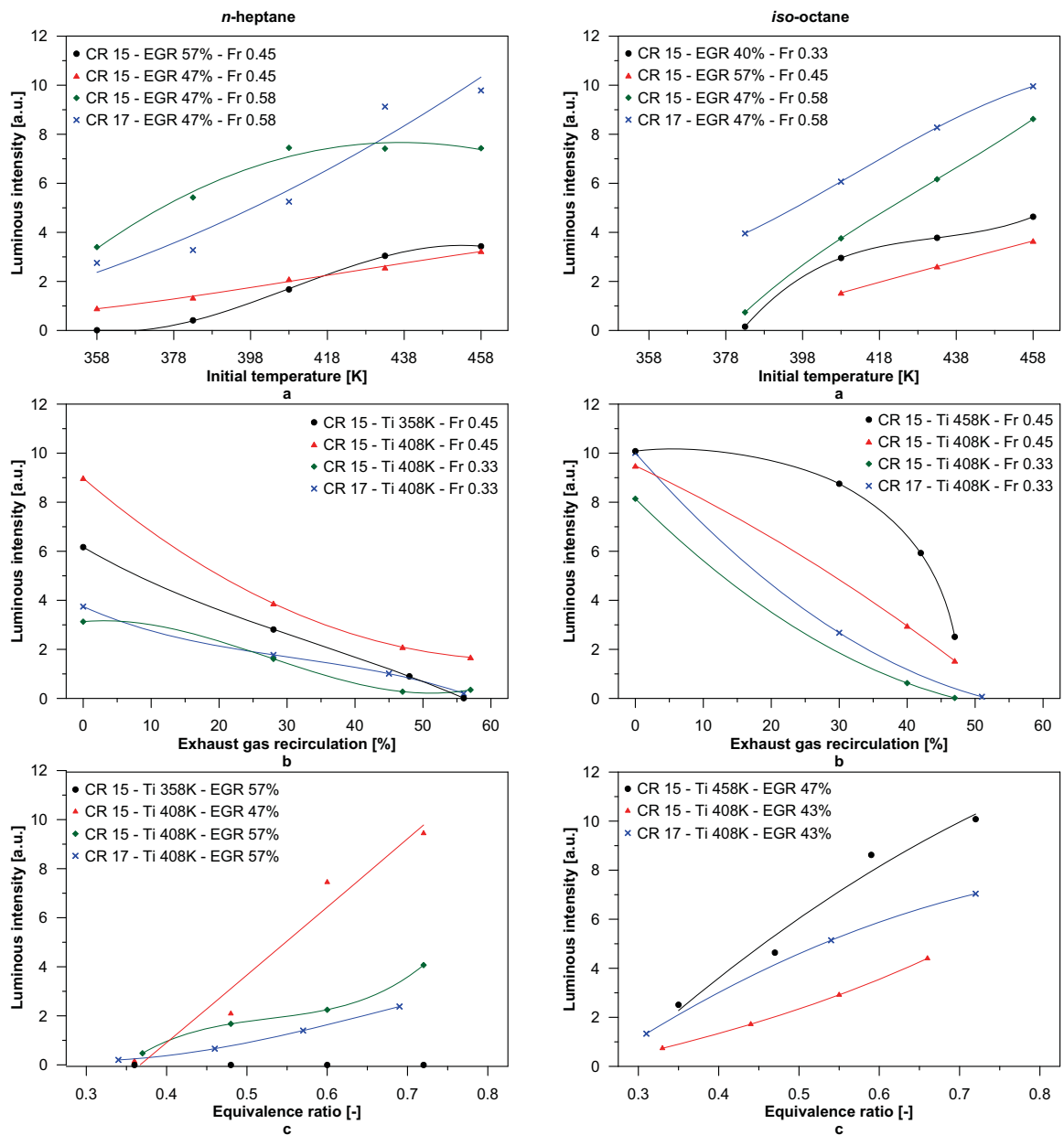


Figure 12: Maximum luminous intensity from the photo-multiplier versus initial temperature (a), EGR rate (b) and equivalence ratio (c) for both fuels and under different operating conditions. Markers - Experiments, Lines - Trends.

486 4.3. Comparison between experimental and simulated results

487 An additional ignition delay time is defined referred to a maximum lu-
488 minous intensity. This ignition delay is obtained experimentally from the
489 photo-multipliers and from the high speed camera and by simulation from
490 CHEMKIN. Despite the fact that the luminous intensity could proceed from
491 two different sources, OH^* or the CO continuum, the maximum radiation
492 emitted by both occurs at the same instant. The percentage deviation in
493 the ignition delay (ε), was calculated in order to compare more easily exper-
494 imental and simulation results. This deviation is defined as follows:

$$\varepsilon = \frac{t_{il,ICE} - t_{il,x}}{t_{il,x}} \times 100 \quad (2)$$

495 where t_{il} represents the ignition delay time referred to the light. The subscript
496 ICE represents a data obtained from a chemical simulation with CHEMKIN
497 using a closed 0-D IC-engine reactor. Finally, the subscript x represents one
498 of the experimental methods, photo-multiplier or high speed camera.

499 The average of the errors in absolute value ($\bar{\varepsilon} = \sum | \varepsilon | / n$), has been
500 calculated for each fuel, as well as its confidence interval with a confidence
501 level of 95%. The following results are obtained:

- 502 • Mean relative deviation ($\bar{\varepsilon}$) between simulations and photo-multiplier:
503 [0.531 1.686] % for *n*-heptane and [1.652 2.930] % for *iso*-octane.
- 504 • Mean relative deviation ($\bar{\varepsilon}$) between simulations and high speed camera:
505 [0.581 1.725] % for *n*-heptane and [1.602 2.910] % for *iso*-octane.

506 Of course, the times obtained from the experimental methods should
507 be virtually the same, since both methods measure the same parameter in

508 parallel. In fact, the confidence interval with a confidence level of 95% of
509 the mean relative deviation between these two methods is equal to [0.069
510 0.116] % for *n*-heptane and to [0.112 0.223] % for *iso*-octane. The existence
511 of certain deviation between both experimental method is justified by the
512 different acquisition frequency.

513 The chemical kinetic mechanism is able to predict with high accuracy
514 the time at which the OH^* is accumulated (high temperature stage of the
515 autoignition process). Figure 7 shows the time evolution of the normalized
516 OH^* intensities from the photo-multiplier and from the high speed camera,
517 as well as the normalized OH^* molar fraction from CHEMKIN, for a certain
518 case. Both experimental methods show similar OH^* profiles. However, the
519 *time of life* of the OH^* obtained by CHEMKIN is longer than the obtained
520 experimentally. This is caused because the *time of life* cannot be completely
521 measured, since the luminous intensity is too low as the temperature de-
522 creases. Moreover, the intensity of the OH^* accumulated during the cool
523 flames cannot be seen in the experimental results. The OH^* intensity is
524 directly related with the amount of accumulated OH^* and with the ther-
525 modynamic conditions in the combustion chamber. The higher the reached
526 temperature and the higher the concentration of OH^* , the higher its lumi-
527 nous intensity. The combination of low temperature and small concentration
528 of OH^* is the reason why cool flames are not detected by the photo-multiplier
529 nor by the camera. It can be seen that a short *time of life* of the luminous
530 intensity is directly related with the CO continuum light emission.

531 **5. Conclusions**

532 In this work a study of ignition delay time and generalization of auto-
533 ignition for PRFs in a RCEM by means of natural chemiluminescence has
534 been performed. The trends of the ignition delay, including the NTC be-
535 haviour, were analyzed. The validity of the chemical kinetic mechanism used
536 was checked by comparing its predictions directly with the experimental re-
537 sults. Then, the combustion process was studied from a point of view of
538 $OH^* - CO$ chemiluminescence. The generalization of the auto-ignition (per-
539 cent of the combustion chamber ignited) was related with the time of ignition.
540 Finally, the luminous intensity was studied, comparing the experimental re-
541 sults from the photo-multiplier and from the high speed camera with the
542 chemical simulations.

543 The following conclusions can be deduced from this study:

- 544 • The chemical kinetic mechanism used in this work can reproduce not
545 only the trends, but also the values of the experimental ignition de-
546 lays. Not only the ignition delay referred to the maximum pressure
547 rise can be numerically obtained with high accuracy, but also the ig-
548 nition delay referred to the maximum radiation. The mean relative
549 deviation between simulations and experiments is lower than 2.93% in
550 all cases. The results show that simulations under-predict the ignition
551 delay for *n*-heptane whereas the over-predict the ignition delay times
552 for *iso*-octane.
- 553 • Cool flames can not be seen by OH^* chemiluminescence by keeping
554 constant the gain during all the combustion process because both the

555 OH^* concentration and the temperature are too low. OH^* luminous
556 intensity starts when the CO starts oxidizing into CO_2 , since before
557 all the generated OH^* is consumed by the generation of CO . The
558 maximum intensity appears very close to maximum oxidation rate of
559 the CO to CO_2 , which coincides with the time of maximum pressure
560 rise rate. This makes it difficult to separate one from the other by just
561 measuring luminosity. Nevertheless, it has been seen that the *time of*
562 *life* of the OH^* is much longer than that of the CO . So, the *time of*
563 *life* is a good indicator that the luminosity seen corresponds to OH^* .

- 564 • The maximum area occupied by natural chemiluminescence is strongly
565 dependent on where ignition occurs. For ignitions far away of TDC, the
566 thermodynamic conditions in the combustion chamber are not optimal
567 for the propagation of the auto-ignition front, leading to slower ignitions
568 and areas below 100%.
- 569 • The maximum luminous intensity has a potential behavior with the
570 maximum temperature reached in the cycle, being higher in case of
571 using *iso*-octane for the same temperature. The higher the initial tem-
572 perature, the lower the EGR rate or the higher the equivalence ratio,
573 the higher the maximum luminous intensity. However, a higher reac-
574 tivity of the mixture does not imply a higher combustion intensity. If
575 ignition occurs far away from TDC, the maximum thermodynamic con-
576 ditions reached can be lower than the ones obtained with a less reactive
577 mixture that ignites near TDC, leading to lower luminous intensity.

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586 **Notation**

<i>BDC</i>	Bottom Dead Center
<i>CR</i>	Compression Ratio
<i>EGR</i>	Exhaust Gas Recirculation
<i>Fr</i>	Working equivalence ratio
<i>FWHM</i>	Full Width at Half Maximum
<i>HCCI</i>	Homogeneous Charge Compression Ignition
587 <i>ICE</i>	Referred to data obtained from CHEMKIN using the in- ternal combustion engine reactor
<i>LTC</i>	Low Temperature Combustion
<i>NTC</i>	Negative Temperature Coefficient
P_i	Initial pressure
<i>PCCI</i>	Premixed Charge Compression Ignition
<i>PHM</i>	Photo-multiplier

<i>PRF</i>	Primary Reference Fuels
<i>RCEM</i>	Rapid Compression Expansion Machine
<i>SI</i>	Spark Ignition
T_i	Initial temperature
<i>TDC</i>	Top Dead Center
t_i	Ignition delay time
<i>UHC</i>	Unburned hydrocarbons
Y_{O_2}	Oxygen mass fraction
ϵ	Percentage deviation in ignition delay referred to the maximum pressure rise between experimental and simulation results
$\bar{\epsilon}$	Mean relative deviation in ignition delay referred to the maximum pressure rise between experimental and simulation results
ϵ	Percentage deviation in ignition delay referred to the peak of OH^* or maximum oxidation rate of CO between experimental and simulation results
$\bar{\epsilon}$	Mean relative deviation in ignition delay referred to the peak of OH^* or maximum oxidation rate of CO between experimental and simulation results

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