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# The effect of nozzle geometry over ignition delay and flame lift-off of reacting direct-injection sprays for three different fuels

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## Abstract

<sup>1</sup> The influence of internal nozzle flow characteristics over ignition delay, and flame lift-off of reacting direct-injection sprays is studied experimentally for three fuels using two different nozzle geometries. This is a continuation of previous work by the authors, where, evaporative and non-evaporative, isothermal spray developments were studied experimentally for the same nozzle geometries and fuels. Current study reports the ignition delay through Schlieren technique, and flame lift-off length through OH\* chemiluminescence visualization. The nozzle geometries consist of a conical nozzle and a cylindrical nozzle with 8.6% larger outlet diameter when compared to the conical nozzle. The three fuels considered are n-heptane, n-dodecane and a three-component surrogate to better represent the physical and chemical properties of diesel fuel. Reacting spray is found to penetrate faster than non-reacting spray due to combustion induced acceleration after ignition. Higher oxygen concentration, and ambient temperature enhance the reactivity leading to higher spray tip penetration. Injection pressure does not affect

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the reactivity significantly and hence, influences spray penetration through momentum—similar to a non-reacting spray. Both ignition delay and lift-off length are found to be shortest and longest for n-dodecane and n-heptane, respectively, while the surrogate fuel falls in-between the two pure component fuels. Both ignition delay and lift-off length are found to decrease with increase in oxygen concentration, ambient temperature, and density. The cylindrical nozzle, in spite of shorter lift-off length is found to have longer ignition delay, when compared to the conical nozzle. This could be due to better atomization leading to larger spread angle and evaporative cooling from the cylindrical nozzle compared to a conical nozzle. The longer ignition delay also leads to leaner equivalence ratios at the time of ignition.

*Keywords:* Reactive spray development, surrogate fuels, lift-off length, vapor penetration, ignition delay

<b>Nomenclature</b>			
$\rho$	Ambient density	CFD	Computational fluid dynamics
$C_{p,liq}$	Liquid phase constant pressure specific heat capacity	ECN	Engine Combustion Network
$d_o$	Orifice nominal diameter	FOV	Field of view
$h_{vap}$	Specific enthalpy of vaporization	ID	Ignition delay
$P_r$	Rail pressure	LOL	Lift-off length
$T_a$	Ambient temperature	PAH	Polycyclic aromatic hydrocarbon
$T_b$	Boiling temperature	SoCF	Start of cool flames
$k0$	Cylindrical nozzle	SOI	Start of injection
$k15$	Conical nozzle	SSI	Second stage ignition

## 1. Introduction

Fuel injection, mixing, evaporation and combustion processes are the key to reduce pollutant formation and improve efficiency of direct injection diesel

1 engines [1]. To this end, engine research and development groups have been  
2 trying for decades to better understand and describe the fundamentals of  
3 these processes, including alternative fuels and combustion strategies [2–5].  
4 Fuel sprays, being primarily characterized by physically complex phenomena  
5 and intrinsically stochastic behavior, are remarkably challenging to compre-  
6 hend by engine and combustion researchers. Over the last three decades,  
7 experimental researchers have studied fuel sprays thoroughly in search for a  
8 better understanding of these phenomena and also for supporting data that  
9 enables validation of detailed numerical models [6]. Among all challenges  
10 presented by the physics of fuel sprays injected in-cylinder, the effect of noz-  
11 zle geometry on the formation, mixing and combustion of the diesel spray  
12 is still of interest to the research community and the automotive industry  
13 [7–10]. Even though it has been studied before, the full extent of the nozzle  
14 geometry effects over a wide span of operating conditions (including fuels)  
15 and response variables is not yet fully understood. For instance, Ganippa  
16 et al. [11] presented results claiming that nozzle flow characteristics have  
17 negligible influence over the spray formation and that momentum is the only  
18 controlling variable for mixing. Opposite to this study, several authors show  
19 that the flow inside the nozzle influences the near-nozzle region of the spray in  
20 terms of liquid-phase break-up, liquid length, and spray angle [12, 13]. Other  
21 studies also evidence the effects of nozzle flow characteristics over the macro-  
22 scopic spray [14, 15]. This contrast, along with the remaining uncertainty  
23 on the effect of nozzle geometry on entrainment, combustion, and pollutant  
24 formation, leaves room for fundamental questions on the subject.

25 These fundamental questions could be addressed from the information  
26 provided by computational fluid dynamic (CFD) models, which output a  
27 large amount of temporal and spatial data that the experimental approach  
28 is unable to acquire [6]. Current models still require high-fidelity experimen-  
29 tal data for validation and accurate bounding of the problem. Majority of  
30 current spray models employ initial and boundary conditions at the nozzle  
31 exit as an indirect coupling to the flow inside the nozzle [15, 16]. Such meth-  
32 ods often dampen or lose smaller scale nozzle flow characteristics, and also  
33 present numerical issues such as different time-step lengths for each model  
34 to be coupled. Hence, the computed spray development using the indirect  
35 coupling is mainly dictated by momentum, aerodynamics, and mixing. Re-  
36 cently, a few authors have published computational models that employ a  
37 full grid comprising the nozzle internal geometry and the spray [17–19]. It  
38 is important to point out that the work presented by Desantes et al. [19] and

1 Xue et al. [18] have benefited significantly by the considerable size and good  
2 quality of the Engine Combustion Network (ECN) open database and efforts  
3 (<http://www.sandia.gov/ecn/>, [20]), which allowed access to very high res-  
4 olution tomographies of the internal nozzle geometry, along with extensive  
5 experimental data from different institutions around the world. However,  
6 fuel properties were still out of the scope of the ECN and questions raised  
7 about the interaction between nozzle geometry and fuel characteristics over  
8 the reactive spray are still not fully understood.

9 CFD models require minimal uncertainties in physical and chemical fuel  
10 properties. The development of surrogate fuels is one way to achieve this  
11 while providing detailed chemical kinetic mechanisms [21–23] further reduced  
12 to computable sizes [22, 24] that can be employed in a fully reactive spray  
13 model. Surrogate fuels are often carefully tailored to mimic the behavior of  
14 real diesel fuel over the diagnostic being performed [22, 25, 26]. For some  
15 years, the surrogate of choice for diesel fuel has been a single-component  
16 species n-heptane. There have been two important reasons for this choice.  
17 First, n-heptane has a Cetane number of 56 that is reasonably close to the  
18 Cetane number of common diesel fuel, so its ignition is similar to that of  
19 diesel fuel which is convenient for ignition or heat release studies [15, 20,  
20 27]. In addition, a detailed kinetic reaction mechanism for n-heptane was  
21 published by Curran et al. [21] in 1998 with all of the detail required to carry  
22 out thorough combustion studies. Recently, it has become apparent that  
23 n-heptane is not sufficient as a diesel surrogate, for instance, Idicheria and  
24 Pickett [28] showed that the n-heptane flame produces considerably less soot  
25 than a #2 diesel flame at similar conditions, and the soot distribution within  
26 the flame was also found to be quite different. Therefore, richer surrogates  
27 containing aromatics and other species that are important components in  
28 diesel fuels must also be represented in the surrogate selected for this study.  
29 In the present paper, three surrogate fuels are employed, n-heptane as the  
30 classical diesel substitute, n-dodecane that has been widely accepted as a  
31 diesel substitute in recent years and it was decided as reference fuel for the  
32 ECN [20, 29] and finally a multi-component diesel surrogate consisting of n-  
33 tetradecane (0.5), n-decane (0.25) and  $\alpha$ -methylnaphthalene (0.25) is utilized  
34 [30, 31]. Numbers in parentheses represent mass fractions.

35 This study is a contribution to the current understanding on the effects  
36 of nozzle geometry and fuel type over combusting sprays, in terms of second  
37 stage ignition (SSI) delay and lift-off length. The study follows up on three  
38 previous works which analyze the effect of nozzle geometry over the liquid

1 iso-thermal non-evaporative spray formation [32], the effect of nozzle geom-  
2 etry combined with different fuels on the hydraulic performance and liquid  
3 isothermal non-evaporative spray formation [30], and the same nozzles and  
4 fuels on evaporative conditions [31]. In this work, all experiments were also  
5 performed for the same nozzle geometries (cylindrical and conical conver-  
6 gent) and fuel types. The experimental campaign consisted of characterizing  
7 the SSI delay through the Schlieren technique [20, 29, 33–36], and the lift-off  
8 length (LOL) through OH\* chemiluminescence visualization [29, 33, 34, 36–  
9 38], in high temperature/high pressure chamber conditions, covering a wide  
10 range of parametric variations that include temperature, density and oxygen  
11 concentration sweeps. With these experiments, two main goals are pursued:  
12 first, to evaluate the influence of nozzle flow characteristics over the basic  
13 combustion phenomena supporting experimental data for different fuels and  
14 second, an effort is made in obtaining a large database of quality data useful  
15 for CFD model validations with different fuels. State-of-the-art experimental  
16 techniques, facilities and equipment were employed in order to ensure highest  
17 quality of data acquired and reported. All experimental data presented in  
18 this paper is available for download at <http://www.cmt.upv.es/DD01.aspx>.

## 19 **2. Materials and methods**

### 20 *2.1. Hardware*

#### 21 *2.1.1. The high temperature and high pressure test rig*

22 All visualization experiments were performed in a constant pressure-flow  
23 test chamber, capable of mimicking the in-cylinder thermo-dynamic condi-  
24 tions of a diesel engine during the fuel injection. This test rig features the  
25 unique capability of obtaining nearly quiescent and, compared to other facili-  
26 ties such as constant volume chambers [39], steady thermodynamic conditions  
27 within the chamber. This is particularly useful for extensive experimental  
28 campaigns with parametric variations of thermodynamic test conditions. The  
29 steady, quiescent test conditions provide a high test repetition rate—also re-  
30 ducing the effective test time for a given set of test conditions—and enhance  
31 the shot-to-shot precision of the tests performed.

32 A set of electrical resistors located inside the inlet pipe that leads into  
33 the chamber heat incoming gas. This arrangement is capable of reaching and  
34 maintaining a maximum temperature and pressure of 1000 K and 15 MPa  
35 respectively, in the test chamber. The chamber has three large optical access  
36 windows—128 mm in diameter—placed orthogonally, providing full optical

1 access to the injection event. A description, photo and schematic of the  
 2 installation can be found in works previously published by the authors [36,  
 3 40].

#### 4 *2.1.2. The fuel injection system*

5 A common-rail injection system consisting of a high pressure pump and a  
 6 conventional rail with an electronic pressure regulator is used [41, 42]. This  
 7 system can generate relatively high rail pressures of up to 220 MPa and main-  
 8 tain it at the set value while injecting fuel. The injector body temperature  
 9 is controlled using a special injector holder designed to have coolant flowing  
 10 in direct contact with the injector body. The temperature of the coolant  
 11 is adjusted in function of the discharge chamber gas temperature and den-  
 12 sity, to guarantee a constant sac inner wall temperature of approximately  
 13 110 °C [29, 43]. The injector’s return line was pressurized to 0.6 MPa as re-  
 14 quired by the injectors to work properly. The entire fuel injection system is  
 15 electronically controlled and all the settings are introduced digitally.

#### 16 *2.1.3. Nozzles*

17 All experiments were performed for two different nozzles, mounted on two  
 18 independent injector bodies. Table 1 summarizes the injectors utilized and  
 19 their nominal nozzle geometries. The injectors are piezo-electrically actuated  
 20 Bosch injectors. The two nozzles are micro-sac type single-hole nozzles, with  
 21 different conicity but equal hydro grinding (13.5% each) and nominal flow  
 22 rate (124 cm<sup>3</sup>/min/10 MPa each). Note that Table 1 includes reference sym-  
 23 bol and color columns which indicates the symbols and/or colors that will  
 24 be used to distinguish nozzles in the results section.

Table 1: Injector hardware utilized and nominal nozzle geometries.

Nozzle ref.	Nozz. type	$d_o$ [ $\mu m$ ]	k-factor	Ref. symbol	Ref. color
k0	micro-sac	151	0	◇	purple
k15	micro-sac	138	1.5	○	green

#### 25 *2.1.4. Fuels*

26 All experiments were also performed for three different fuels. The first  
 27 fuel selected is n-heptane. As stated in the section 1, n-heptane has long been

28 utilized as a diesel surrogate to mimic diesel fuels in ignition and/or heat re-  
 1 lease studies [21, 24, 27, 44, 45]. The second fuel selected is n-dodecane,  
 2 which features similar carbon content and boiling characteristics to those of  
 3 diesel fuels, so it is expected to better mimic the mixing behavior of diesel  
 4 fuels. This is one of the reasons n-dodecane was also selected as the primary  
 5 fuel of study for the main ECN campaign [20], and it has been extensively  
 6 characterized in the complete spectrum of experimental diagnostics and nu-  
 7 merical simulations performed by the group. However, n-dodecane is not  
 8 expected to be an adequate surrogate for ignition-related behavior, because  
 9 of its Cetane number (approx. 88). Last, a multi-component diesel surrogate  
 10 consisting of n-tetradecane (0.5), n-decane (0.25) and  $\alpha$ -methylnaphthalene  
 11 (0.25) is utilized. Numbers in parentheses represent mass fractions. This  
 12 surrogate—from this point forward simply referred to as “Surrogate”—is ex-  
 13 pected to better mimic the soot-related behavior of real diesel fuel due to  
 14 the PAH content and C/H ratio being closer to that of real diesel fuel. The  
 15 short ignition delays expected due to the large n-tetradecane and n-decane  
 16 contents (with Cetane numbers close to 96 and 77 respectively) are, at the  
 17 same time, delayed by the the  $\alpha$ -methylnaphthalene content. Fuel properties  
 18 relevant to this study are summarized in Table 2.

Table 2: Fuels utilized and their properties at 298 K and 101 kPa. Except for the Surrogate fuel, all properties were extracted from the NIST Chemistry WebBook [46]. For the Surrogate fuel, density, viscosity and surface tension were measured as per ASTM D1298, ASTM D445 and UNE EN 14370 respectively.

Property	Units	n-Heptane	n-Dodecane	Surrogate
Density	kg/m <sup>3</sup>	679.7	745.8	802.1
Viscosity	Pa s	5.59e-4	1.36e-3	1.61e-3
Surface tension	N/m	0.020	0.025	0.026
Boiling point	K	372	489	450 to 520
$C_{p,liq}$	J/kg/K	2234	2212	Tab. 3
$h_{vap}$	kJ/kg	359	358	Tab. 3
Ref. color	-	<b>cyan</b>	<b>blue</b>	<b>magenta</b>

19 The Surrogate distillation curve was presented by the authors in [31].  
 20 This Surrogate starts boiling near 450 K, which corresponds to the boiling  
 21 point of n-decane. On the other hand, it is completely evaporated near 520 K  
 22 which corresponds to the boiling point of n-tetradecane, which comprises half



Table 3: Components of the Surrogate fuel and their properties at 298 K and 101 kPa. All properties were extracted from the NIST Chemistry WebBook [46].

Property	Units	n-Tetradecane	n-Decane	$\alpha$ -methylnaphthalene
Boiling point	K	523	447	515
$C_{p,liq}$	J/kg/K	2208	2192	1578
$h_{vap}$	kJ/kg	361	361	415

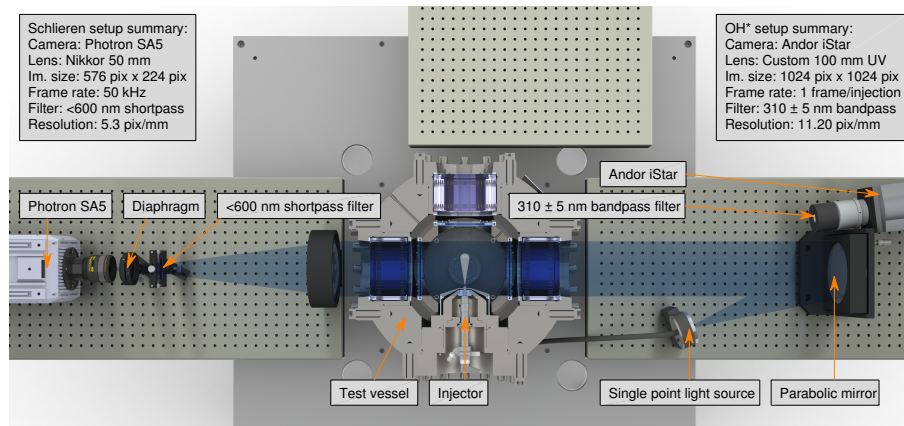


Figure 1: Plan view of the optical setup.

23 of the mass of the Surrogate fuel. Note also that n-heptane features a boiling  
 1 point considerably lower than the boiling range of the Surrogate, which at  
 2 the same time contains the boiling point of n-dodecane.

## 3 2.2. Reactive spray visualization

### 4 2.2.1. Optical setup

5 The optical setup, shown in Figure 1, consisted of two separate cameras  
 6 and optical arrangements for the visualization of the reactive spray develop-  
 7 ment and lift-off length. Note that both cameras recorded all injection events  
 8 simultaneously.

9 Schlieren imaging has been successfully employed several times to iden-  
 10 tify refractive index gradients in transparent mediums. For vaporizing diesel  
 11 sprays, this technique is able to capture the line-of-sight boundary between  
 12 vaporized fuel and ambient gases, as there is an appreciable difference in re-  
 13 fractive indices between these [6, 20, 47]. Since the rays of light are collimated

14 into a cylindrical beam, small deflections due to refractive index gradients  
1 are rendered in the image as shades. In this study, the vapor spray was visu-  
2 alized through a single-pass Schlieren setup [6, 31], which is often applied to  
3 axi-symmetrical single hole nozzles. Multi-hole injectors require a two-pass  
4 setup and a high temperature mirror as explained by Payri et al. [40, 48]. The  
5 final setup is very similar to the setups employed for the CMT experiments  
6 in [20, 33, 36, 49] and exactly the same to the setup utilized by Payri et al.  
7 [31] in their inert sprays study of the same nozzles and fuels. The camera  
8 was a Photron SA5, sampling images of  $576 \text{ pix} \times 224 \text{ pix}$  at 50 kHz with a  
9 spatial resolution of 5.3 pix/mm. This produced a field of view (FOV) along  
10 the spray axis of 108 mm, and considering window limits and nozzle location,  
11 the maximum penetration length measurable was 96 mm. The exposure time  
12 was set to 2.28  $\mu\text{s}$ .

13 The lift-off length (LOL), defined as the axial distance measure from the  
14 orifice outlet at which the flame stabilizes during steady state combustion,  
15 was measured capturing the signal from OH\* chemiluminescence following  
16 the ECN standard methodology [33, 36, 37]. An ICCD camera (Andor iStar)  
17 fitted with a custom 100 mm f/2.8 UV lens and a  $310 \text{ nm} \pm 5 \text{ nm}$  interferomet-  
18 ric filter was used to acquire the chemiluminescence signal. Since this signal  
19 is weak, the intensifier of the camera sensor was gated during the steady  
20 region of the injection—2.3 ms to 4.8 ms after start of energizing (SOE)—to  
21 obtain an on-chip time-averaged signal, minimizing the effect of local tur-  
22 bulent flame behavior. Note that this camera had to be inclined slightly  
23 off axis so not to block the collimated Schlieren beam. However, the angle  
24 is small at  $7^\circ$ , and the possible effects are accounted for by properly cor-  
25 recting the images. The camera sampled one image per injection event, of  
26  $1024 \text{ pix} \times 1024 \text{ pix}$  with a spatial resolution of 11.2 pix/mm. This produced  
27 a FOV along the spray axis of approximately 90 mm. Further details of the  
28 processing algorithm for the LOL estimations can be found in [33, 36].

### 29 *2.2.2. Schlieren image processing*

30 Each image is processed using an algorithm that detects the spray bound-  
31 ary and computes its associated properties. The program—similar to what  
32 the authors employed in their previous study of the same nozzles and fu-  
33 els at inert ambient conditions [31]—consists of two extensively used ap-  
34 proaches for the processing of these type of images. Two binarized im-  
35 ages are obtained from two different criteria and then merged to maxi-  
36 mize sensitivity. The first algorithm was originally developed at Sandia Na-

37 tional Laboratories (SNL) and is available for download on the ECN website  
1 (<http://www.sandia.gov/ecn/>). The routine detects temporal changes in  
2 pixel-wise intensities by taking the temporal derivative of a series of images.  
3 This produces a 2D map where pixels with higher intensity represent pixels  
4 that are changing their digital values in time. The temporal nature of this  
5 algorithm makes it robust to variations between optical setups, and makes it  
6 very strong for transparent spray images, for example, of very dilute sprays,  
7 low ambient density conditions, light fuels, etc. On the other hand, it does  
8 not work properly for spray images with relatively constant intensity levels,  
9 for example: images of non-evaporative sprays, diaphragm-cut Schlieren va-  
10 por sprays (which are often very dark), combustion-saturated sprays, etc. In  
11 these situations, spray tip penetration is still captured correctly while the full  
12 spray boundary is not. Therefore, an additional intensity-sensitive algorithm  
13 was adapted, explained in detail by Payri et al. [40], enhanced with the dy-  
14 namic background correction detailed by Benajes et al. [33] and Payri et al.  
15 [36]. The two binary maps obtained from each algorithm are combined into  
16 a single binary image from which the contour is extracted. This approach  
17 maximizes sensitivity since it takes advantage of the robustness of the SNL  
18 algorithm for the spray tip region—and dilute regions or sprays—but at the  
19 same time allows for good contour detection in the near nozzle region, where  
20 the liquid core often generates a very dark image.

21 The algorithm then extracts macroscopic characteristics from the de-  
22 tected contours. Spray tip penetration is calculated as the distance between  
23 the outlet orifice and the furthest point in the detected boundary. The esti-  
24 mation of SSI delay comes from the signal obtained by computing the sum  
25 of the pixel-wise intensities within this boundary (from this point forward  
26 referred to as *total intensity*). This summation is done over the inverted  
27 spray image, so an increase in the total value indicates a darker and/or  
28 larger spray. The resulting signal, and its derivative in time (from this point  
29 forward referred to as *total intensity increment*), present unique-consistent  
30 features that allow for reliable estimation of the SSI, as thoroughly detailed  
31 by Benajes et al. [33], Payri et al. [36].

### 32 2.2.3. Test plan

33 The test plan is presented in Table 4, it is centered on ECN Spray A  
34 boundary conditions [20], with parametric variations around these. Since  
35 the time available for experiments was limited, note that the test matrix  
36 does not comprise every combination of the studied variables, but sweeps

Table 4: Spray visualization test plan, centered on ECN Spray A boundary conditions [20].

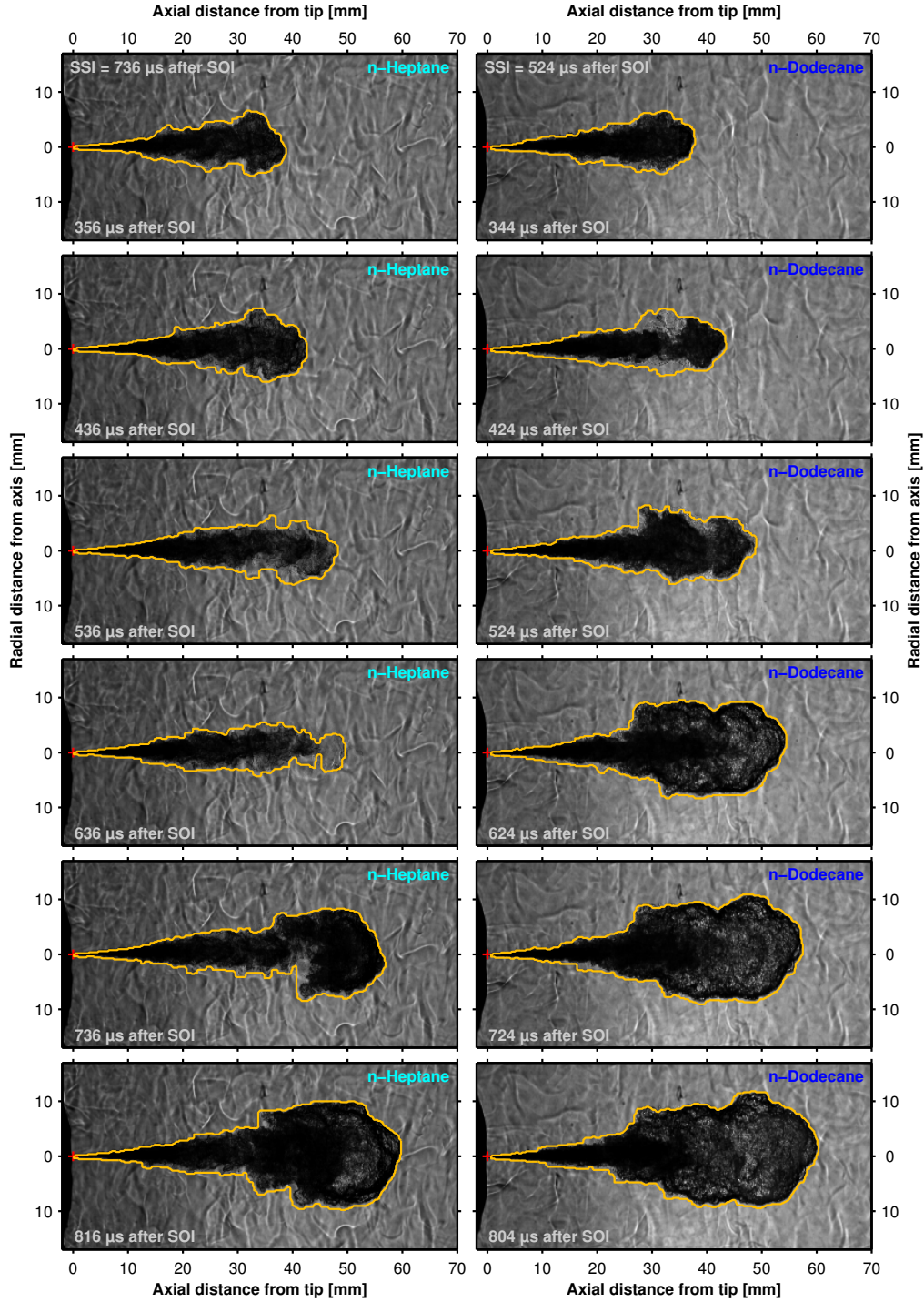
Parameter	Value-Type	Units
Ambient density	22.8	kg/m <sup>3</sup>
Ambient temp.	800, 900, 970	K
Rail pressure	60, 90, 150, 200	MPa
Oxygen conc.	21	%
No. of points	12/nozzle/fuel	
Ambient density	22.8	kg/m <sup>3</sup>
Ambient temp.	800, 900, 970	K
Rail pressure	60, 90, 150	MPa
Oxygen conc.	15	%
No. of points	9/nozzle/fuel	
Ambient density	15.2, 30.4	kg/m <sup>3</sup>
Ambient temp.	900	K
Rail pressure	60, 90, 150	MPa
Oxygen conc.	21	%
No. of points	6/nozzle/fuel	
Total points	27/nozzle/fuel	

37 of certain resolutions depending on the interest of each variable. Table 4 is  
1 sub-divided into these three groups of points for easier visualization of the  
2 test plan. For all conditions the energizing time was fixed at 2.5 ms. All  
3 test points were performed for the two nozzles and three fuels, comprising a  
4 total of 162 test points in the high temperature/high pressure test rig. Note  
5 that all experimental results presented in this manuscript are available for  
6 download at: <http://www.cmt.upv.es/DD01.aspx>.

### 7 **3. Results and discussion**

#### 8 *3.1. Reactive spray development*

9 Figure 2 presents a sequence of Schlieren images of two independent in-  
10 jection events for two different fuels. This sequence demonstrates the typical  
11 behavior of reactive diesel-type sprays: liquid fuel is injected into a hot am-  
12 bient gas, the jet velocity and ambient density shear and atomize the liquid  
13 core, the spray entrains the surrounding hot gas which transfers energy to  
14 the liquid fuel and, downstream, liquid fuel eventually evaporates completely  
15 [31]. Note that a similar figure was presented by Payri et al. [31] for inert  
16 sprays, showing simultaneous contours for the liquid and vapor phases. Once  
17 the reactive spray reaches ignitable fuel mixture fractions, ignition and high  
18 temperature combustion take place. The reactive spray continues to pen-  
19 etrate, still exchanging momentum with the ambient gas and progressively  
20 slowing down. At the same time, the flame stabilizes in the upstream region  
21 at the LOL (see the last images shown for the n-dodecane spray). In these  
22 Schlieren images, the first stage of the ignition process is appreciable as a  
23 brief disappearance of the spray, followed by a sudden expansion and dark-  
24 ening, which corresponds to the second stage ignition (SSI, [50]) as explained  
25 by Benajes et al. [33] and Payri et al. [36, 49]. In the particular case pre-  
26 sented in Figure 2, note how the n-dodecane spray (right column) starts the  
27 SSI earlier than the n-heptane spray—524  $\mu\text{s}$  and 736  $\mu\text{s}$  respectively—which  
28 is expected, since n-dodecane is a heavier *n*-alkane with longer chain, mak-  
29 ing it more reactive. This difference in ignition delay (ID) causes differences  
30 in the corresponding spray tip penetrations, making the n-dodecane spray  
31 penetrate further. These findings will be analyzed in detail in the following  
32 sections.



13

Figure 2: Time sequence of Schlieren images of two injection events of n-heptane and n-dodecane sprays. Images have been trimmed both in the radial and axial directions from their original size, for better fit in this figure. The contours detected are plotted to scale over the original Schlieren images. In this case, the nozzle is *k15*, rail pressure is 150 MPa, ambient density is  $15.2 \text{ kg/m}^3$ , ambient temperature is 900 K and the oxygen concentration is 21 %.

### 33 3.2. Reactive spray tip penetration

1 The effect of reactivity on spray tip penetration for three fuels is presented  
2 in Figures 3 and 4. Mixture reactivity is controlled by the oxygen concen-  
3 tration in the chamber, ambient temperature and fuel properties. In Figure  
4 3, 0% oxygen concentration corresponds to the non-reacting spray studied  
5 previously by the authors [31] while 21% oxygen concentration corresponds  
6 to highest reactivity spray, in terms of oxygen concentration. All three sprays  
7 penetrate at the same rate up to a certain time, after which, the penetration  
8 curve of the spray with higher reactivity (or oxygen concentration) starts to  
9 deviate more and penetrate faster. Once ignition takes place the spray tip  
10 accelerates due to the sudden expansion, which results in a faster spray tip  
11 penetration rate in comparison to the inert case. The faster ignition with  
12 higher oxygen concentration translates to higher spray tip penetration, as  
13 Pastor et al. [51] observed for a set of fuels with different reactivities. Figure  
14 3 shows that, for all three fuels, higher oxygen concentration leads to earlier  
15 spray acceleration or deviation from the non-reacting case.

16 A similar situation can occur for different ambient temperatures. It is  
17 known that ambient temperature is not a determinant variable for evapo-  
18 rating, non-reactive spray development if the ambient density is matched  
19 [31, 40]. Under reactive conditions however, ambient temperature plays an  
20 important role in all the chemical reactions prior to the SSI, and this could  
21 result in different spray penetration rates, as Figure 4 illustrates. Higher am-  
22 bient temperatures lead to higher reactivity, shorter ignition delays and thus  
23 higher spray tip penetration. The ignition delay does not change significantly  
24 above 900 K and hence, when the temperature changes from 900 K to 970 K  
25 the spray penetration does not deviate as much as when the temperature is  
26 changed from 800 K to 900 K [23, 24].

27 Figure 5 presents the effect of injection pressure on reactive spray tip  
28 penetrations. The effect of the injection pressure for non-reacting sprays  
29 is clear from the literature [30, 31]. Increasing injection pressure increases  
30 spray momentum and thus, the spray tip penetration rate. As will be shown  
31 in section 3.4.2, injection pressure does not significantly influence the spray  
32 reactivity and ignition delay. Hence, the effect of rail pressure for reacting  
33 sprays is very similar to that of a non-reacting spray. The three different fuels  
34 show appreciable difference in reactivity and ignition delay and hence, a no-  
35 ticeable effect on spray penetration can be observed. The n-dodecane sprays  
36 penetrate faster earlier, followed by the Surrogate sprays, and n-heptane.

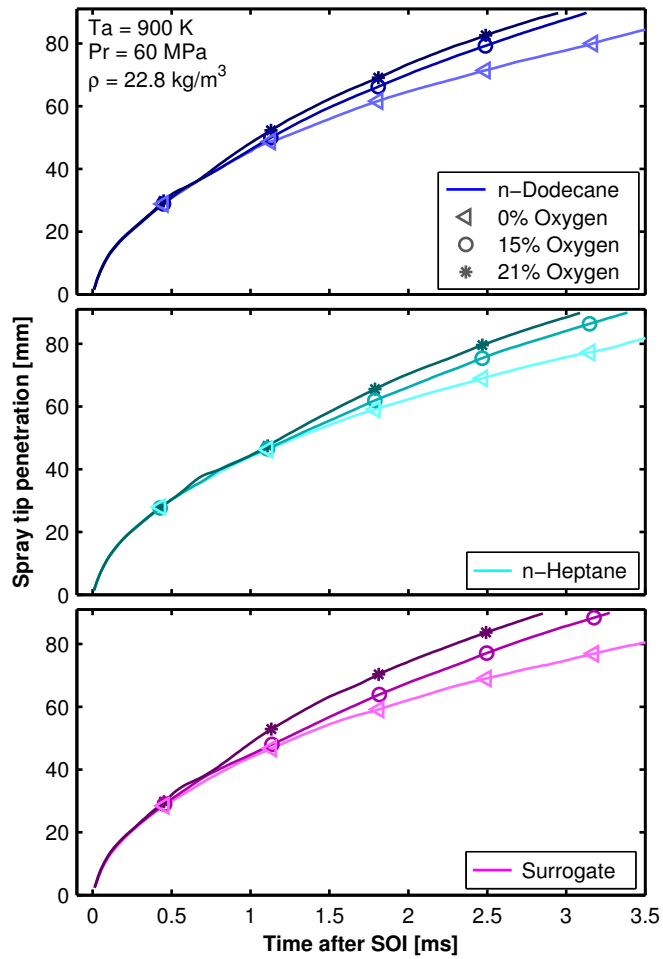


Figure 3: Spray tip penetration for all fuels at different oxygen concentrations. In this case, the nozzle is *k15*, rail pressure is 60 MPa, ambient density is  $22.8 \text{ kg/m}^3$  and ambient temperature is 900 K. Note that inert spray results correspond to a previous study from the authors [31].



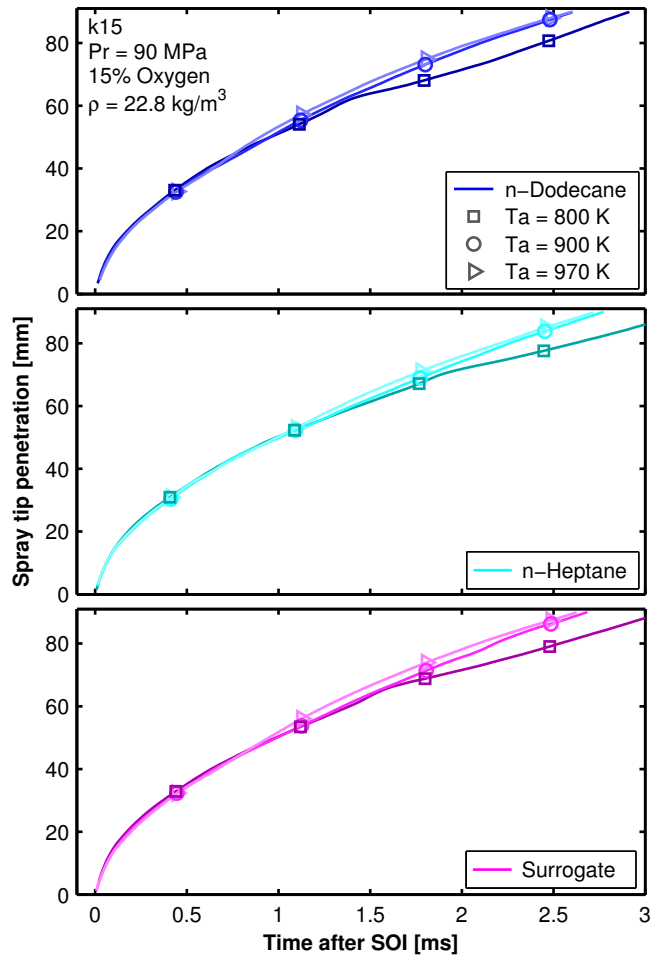


Figure 4: Spray tip penetration for all fuels at different ambient temperatures. In this case, the nozzle is *k15*, rail pressure is 90 MPa, ambient density is  $22.8 \text{ kg/m}^3$  and oxygen concentration is 15%.

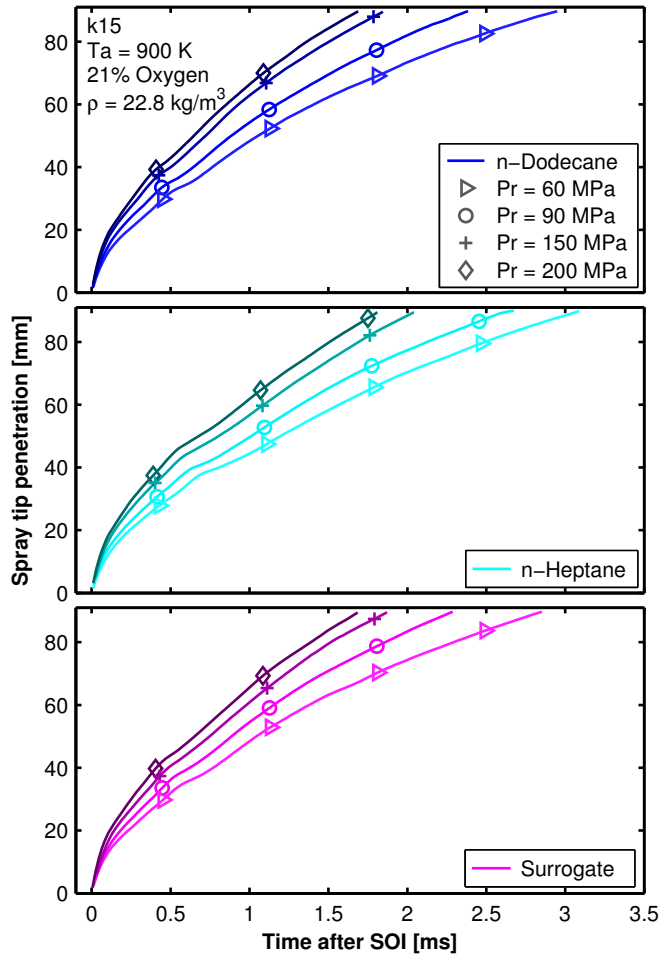


Figure 5: Spray tip penetration for all fuels at different rail pressures. In this case, the nozzle is *k15*, ambient temperature is 900 K, ambient density is 22.8 kg/m<sup>3</sup> and oxygen concentration is 21%.

### 37 3.3. Lift-off length

1 Before presenting results obtained it is interesting to review the param-  
2 eters affecting LOL as known from the literature. These are fuel composition,  
3 ambient temperature, ambient density, ambient oxygen concentration, injec-  
4 tion pressure and nozzle geometry [33, 36, 37, 51]. Because of the large data  
5 base of the present investigation, select cases will be presented to illustrate  
6 the effect of each variable studied.

#### 7 3.3.1. OH\* chemiluminescence signals

8 Figure 6 depicts a comparison of the reactive sprays produced by the two  
9 nozzles at particular test conditions. The top and middle parts of Figure  
10 6 show ensemble-average OH\* chemiluminescence images while the bottom  
11 part plots column-wise intensity maximums of the images. The results at  
12 these test conditions show that nozzle *k0* produces a spray with shorter LOL  
13 when compared to nozzle *k15*, even though its diameter is slightly larger [15,  
14 37]. Signals presented at the bottom part of Figure 6 illustrate very similar  
15 behaviors between the two nozzles in terms of flame shape and intensity  
16 levels. The flame produced by nozzle *k15* shows slightly higher intensity  
17 levels downstream which, qualitatively, could be attributed to a more fuel-  
18 rich flame, and thus, soot. Payri et al. [30, 31, 32] observed that, when  
19 compared to the cylindrical nozzle *k0*, the conical nozzle *k15* features smaller  
20 spreading angles, so it would be expected for this nozzle to produce a more  
21 soot-promoting flame.

22 Figure 7 depicts a comparison of the reactive sprays produced by the  
23 three fuels at particular test conditions. Contrary to the comparison shown  
24 in Figure 6, images and profiles depicted in Figure 7 do show significant  
25 differences. Images are normalized to the dynamic range of the brighter of  
26 the three images, so that they can be visually compared. Note how the  
27 OH\* chemiluminescence image is brighter and the intensity profile at the  
28 bottom plot shows larger values downstream for the Surrogate flame. Qual-  
29 itatively, the Surrogate fuel produces a flame with more soot in comparison  
30 to n-dodecane and n-heptane due to the heavier components and, especially,  
31 aromatic content. The LOL results obtained are in agreement with the ex-  
32 pected trends discussed in section 1: shorter LOL for n-dodecane, followed  
33 by Surrogate fuel and finally higher LOL for n-heptane.

34 These particular cases presented in detail show only a small sample of the  
35 behaviors observed for the whole test matrix. The trends observed in the full  
36 test matrix, however, were very consistent as shown in the following section.

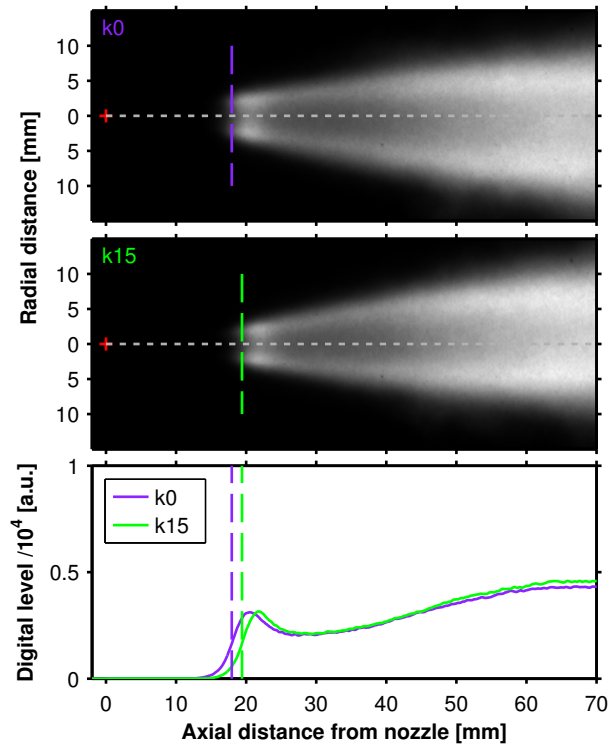


Figure 6: Ensemble-average  $\text{OH}^*$  chemiluminescence images of the flames produced by the two nozzles at particular test conditions. The intensity profiles shown in the bottom plot depict the column-wise intensity maximum of each image. In this case the fuel is n-heptane, rail pressure is 150 MPa, ambient density is  $22.8 \text{ kg/m}^3$ , oxygen concentration is 15% and ambient temperature is 970 K.

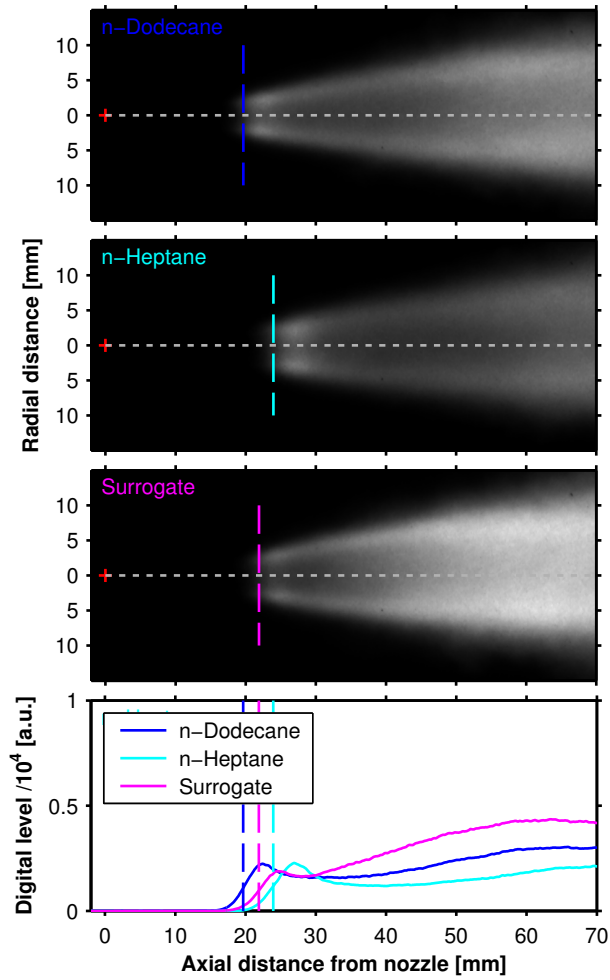


Figure 7: Ensemble-average OH\* chemiluminescence images of the flames produced by the three fuels at particular test conditions. The intensity profiles shown in the bottom plot depict the column-wise intensity maximum of each image. In this case the nozzle is *k15*, rail pressure is 150 MPa, ambient density is 22.8 kg/m<sup>3</sup>, oxygen concentration is 15% and ambient temperature is 900 K.

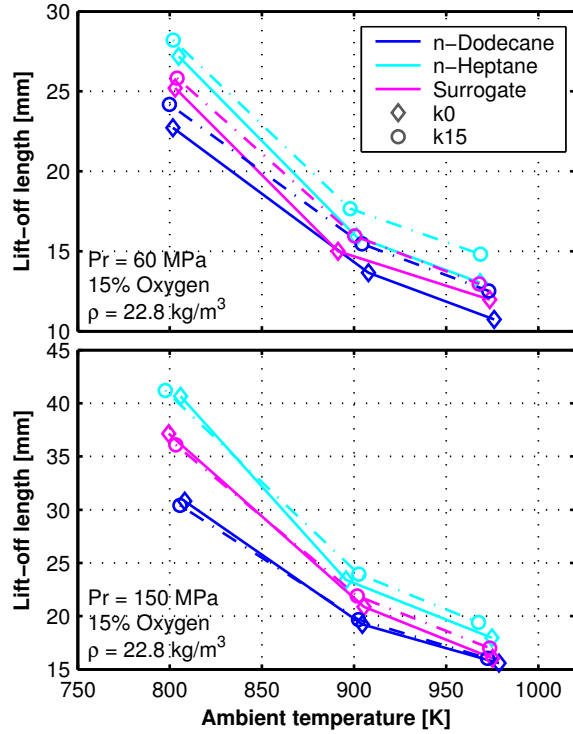


Figure 8: Lift-off length as a function of ambient temperature for all nozzles and fuels at an ambient density of  $22.8 \text{ kg/m}^3$ , an oxygen concentration of 15% and rail pressures of 60 MPa (top) and 150 MPa (bottom).

### 37 3.3.2. Parametric variations

1 This section presents LOL results for a larger window of test conditions  
 2 and parametric variations. Figure 8 presents stabilized LOLs for all nozzles  
 3 and fuels in a subset of the whole dataset, a sweep of ambient temperature  
 4 at high and low rail pressures. First, it is easily noticeable how fuels are  
 5 stratified in terms of LOL. n-Heptane consistently showed the longest LOLs,  
 6 followed by the Surrogate fuel. Pickett et al. [38] and later Pastor et al.  
 7 [51] observed that LOL was mainly controlled by ignition delay time, rather  
 8 than flame velocity. Since there is very little reason to suspect significantly  
 9 different flame velocities among these fuels [52, 53], the observations in these  
 10 study concur with that conclusion: regarding fuel properties, LOL is mainly  
 11 determined by the reactivity of the fuel.

12 On another line, note in Figure 8 how LOL increases considerably with

13 rail pressure. This is also explained by the relationship between LOL and  
1 ignition delay: for larger injection pressures—thus, larger injection velocities  
2 [30]—fuel travels a longer distance for that given ignition delay time.

3 Figure 8 also shows the effect of nozzle geometry over LOL. Under vir-  
4 tually all conditions tested, the cylindrical  $k0$  produced shorter LOLs when  
5 compared to the conical nozzle  $k15$ . All previous studies of these exact noz-  
6 zles showed that differences in the development of the sprays produced by  
7 the two nozzles are reduced as rail pressure is increased [30–32]. Even though  
8 differences in hydraulic characteristics indeed increase with rail pressure [30],  
9 the higher injection and entrainment velocities induced by higher rail pres-  
10 sure suppress the small effects of nozzle geometry over the development of  
11 turbulent profiles in the spray and momentum exchange between the spray  
12 and the ambient gas [30–32]. This trend is also present in the LOL results,  
13 where the difference in LOL is reduced with increasing rail pressure. This  
14 proves consistency in the behavior of these nozzles in terms of isothermal  
15 liquid spray formation [30, 32], evaporative inert spray formation [31] and  
16 LOL stabilization of reactive sprays presented in this study.

17 Note that the hydraulic characterization of these nozzles, presented by  
18 Payri et al. [30], showed that the cylindrical nozzle  $k0$  features larger outlet  
19 flow velocities in comparison to the conical nozzle  $k15$ , due to the area con-  
20 traction. Higher velocity should also translate into larger LOL, but in this  
21 case the cylindrical nozzle geometry also produces larger spreading angles  
22 and fluctuations [30–32] which dominate over the higher velocity to produce  
23 shorter LOL after all.

24 Moreover, Figure 9 shows a subset of the the LOL results as a function  
25 of ambient temperature, for all fuels, in this case illustrating the effect of  
26 oxygen concentration. Note the large differences in LOL due to the oxygen  
27 concentrations. Even though laminar flame velocities are indeed strongly  
28 affected by equivalence ratio [52, 53], the different LOL values observed here  
29 for the two oxygen concentrations are attributed to the corresponding ignition  
30 delay times.

31 Figure 10 depicts a subset of the LOL results as a function of ambient  
32 density, for all nozzles and fuels. Note that the effect of ambient density is  
33 very straight forward: higher density implies that more oxygen is available to  
34 oxidize the fuel and thus, shorter ignition delays are expected, which reduce  
35 the corresponding LOL [37, 38].

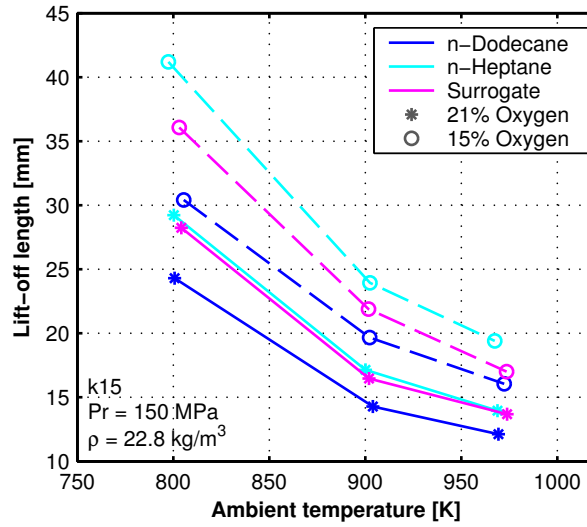


Figure 9: Lift-off length as a function of ambient temperature for all fuels and the two oxygen concentrations tested. In this case, the nozzle is *k15*, ambient density is  $22.8 \text{ kg/m}^3$  and rail pressure is 150 MPa.

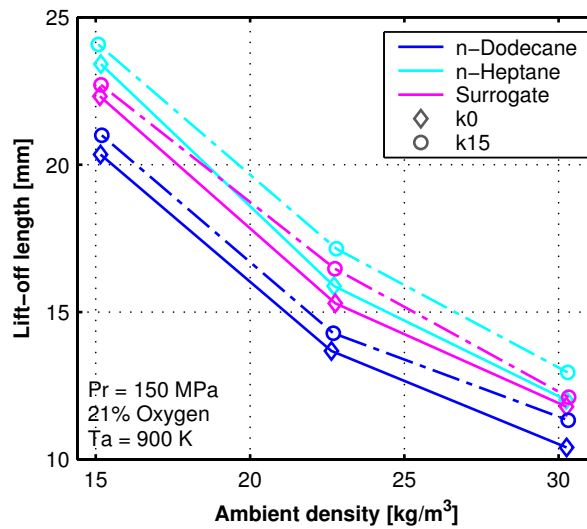


Figure 10: Lift-off length as a function of ambient density for all nozzles and fuels at a rail pressure of 150 MPa, an oxygen concentration of 21% and an ambient temperature of 900 K.



### 36 3.4. Ignition delay

1 Analogous to the LOL results section, it is also interesting to review the  
2 parameters affecting ID as known from the literature. These are fuel compo-  
3 sition, ambient temperature, ambient density, ambient oxygen concentration,  
4 injection pressure and nozzle geometry [33, 36, 37, 51]. Because of the large  
5 data base of the present investigation, select cases will be presented to illus-  
6 trate the effect of each variable studied.

#### 7 3.4.1. Tracer signals

8 Figures 11 and 12 present the time evolution of the tracers signals involved  
9 in the SSI detection. The top part of each figure shows the result of the  
10 pixel-wise intensity sum within the spray boundary (total intensity signal),  
11 while the bottom part of each figure shows its derivative (total intensity  
12 increment). As Figure 2 illustrates, at the first onset of chemical reactions,  
13 also known as start of cool flames (SoCF), the spray becomes transparent  
14 and the slope of the total intensity values changes suddenly. This transparent  
15 phase may not occur in some test conditions where ignition delays are very  
16 short, but the rest of the process develops in a very defined fashion. After this  
17 first stage, the spray tip appears again in the image and the total intensity  
18 values increase rapidly (in the inverted image, thus, darken in the actual  
19 image) to then steadily keep increasing with the spray growth. Benajes  
20 et al. [33], Payri et al. [36, 49] demonstrated that it is possible to correlate  
21 this rapid increase in total intensity values to the SSI. At the SSI, the total  
22 intensity signal slope reaches a maximum, as a result of the high temperature  
23 combustion, which causes rapid spray volume expansion but also changes  
24 refractive indexes inside the spray, darkening the spray or in fact making it  
25 brighter in the inverted image.

26 Benajes et al. [33], Payri et al. [36] studied in detail the effects of ambient  
27 temperature, oxygen concentration, density and injection pressure over these  
28 tracer signals, and the results found in this study follow closely the trends  
29 reported in those studies [33, 36]. Therefore, this section will focus only on  
30 the effects of nozzle geometry and fuel.

31 Figure 11 depicts a comparison of these tracer signals, both total intensity  
32 (top) and total intensity increment (bottom), produced by the two nozzles at  
33 particular test conditions. Payri et al. [30, 31, 32] demonstrated that sprays  
34 produced by the cylindrical nozzle *k0* features larger spreading angles in  
35 comparison to those produced by the conical nozzle *k15*. This is translated  
36 into larger line-of-sight area which, at the same time, increases the total

37 intensity value and its increments in time. Figure 11-bottom shows two  
1 maximums which correspond to the two SSI timings of the sprays produced  
2 by the two nozzles. Note that the spray produced by the conical nozzle  
3 *k15* ignites before the spray produced by nozzle *k0*, while the latter shows a  
4 higher maximum which can be associated with a more pre-mixed combustion  
5 [33, 36]. Even though this is a single example at particular test conditions,  
6 this trend was found to be consistent throughout the complete test matrix,  
7 as will be discussed later in the paper.

8 Fuels, on the other hand, do not alter the vapor spray spreading angle  
9 and penetration significantly [31], so the line-of-sight spray area is similar  
10 between fuels, which implies similar total intensity signals before ignition,  
11 as depicted by Figure 12. In these and virtually all conditions tested, n-  
12 dodecane sprays ignite the earliest, followed by Surrogate sprays and last,  
13 n-heptane sprays, as will be discussed next. As explained also for Figure 11,  
14 longer ignition delays imply a more pre-mixed combustion which produces  
15 a sharper slope of the total intensity signal and thus, a larger maximum in  
16 the total intensity increment signal. After ignition is complete and diffusion  
17 takes over, the three signals behave similarly, as shown in both the top and  
18 bottom parts of Figure 12.

### 19 3.4.2. Parametric variations

20 A larger set of data is presented in Figure 13, with the top and bottom  
21 parts showing sub-sets of injections at 60 MPa and 150 MPa respectively. As  
22 expected from the literature, increasing ambient temperature increases the  
23 reactivity of the ambient gas which decreases IDs [24, 26, 33, 36, 38, 49–51].  
24 Also, as Payri et al. [10, 36] observed, IDs decrease with increasing rail pres-  
25 sure, which is the result of faster liquid break-up and mixture preparation.

26 Moreover, Figure 13 shows that, in all cases, n-dodecane sprays feature  
27 the shortest SSI delays, followed by Surrogate sprays and last, n-heptane.  
28 Also as stated before, the conical nozzle *k15* produces sprays with slightly  
29 shorter SSI delays when compared to the cylindrical nozzle *k0*, this will be  
30 discussed further at the end of this section.

31 Figures 14 and 15 illustrate the effect of ambient oxygen concentration  
32 and ambient density over the SSI delay. Results are in good agreement with  
33 the trends found in the literature, increasing oxygen concentration and am-  
34 bient density decreases IDs [33, 36, 38]. At these relatively low ambient  
35 temperatures, chain-branching reactions are highly dependent on fuel reac-  
36 tivity and oxygen availability for the formation of radicals [51]. These figures

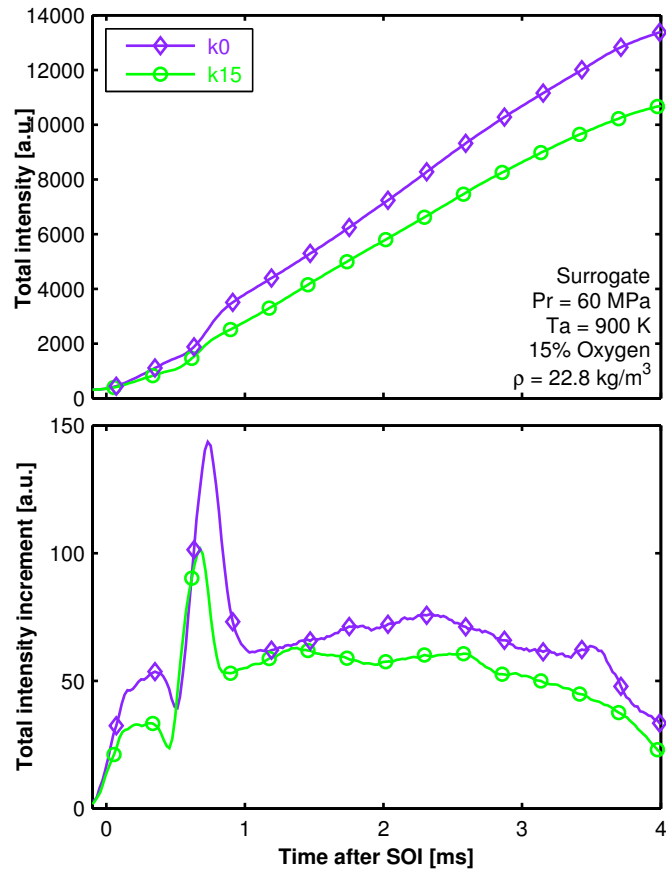


Figure 11: Total spray intensity (top) and intensity increment (bottom) tracer signals for the two nozzles at particular test condition. In this case the fuel is the Surrogate, rail pressure is 60 MPa, ambient density is  $22.8 \text{ kg/m}^3$ , ambient temperature is 900 K and oxygen concentration is 15%.

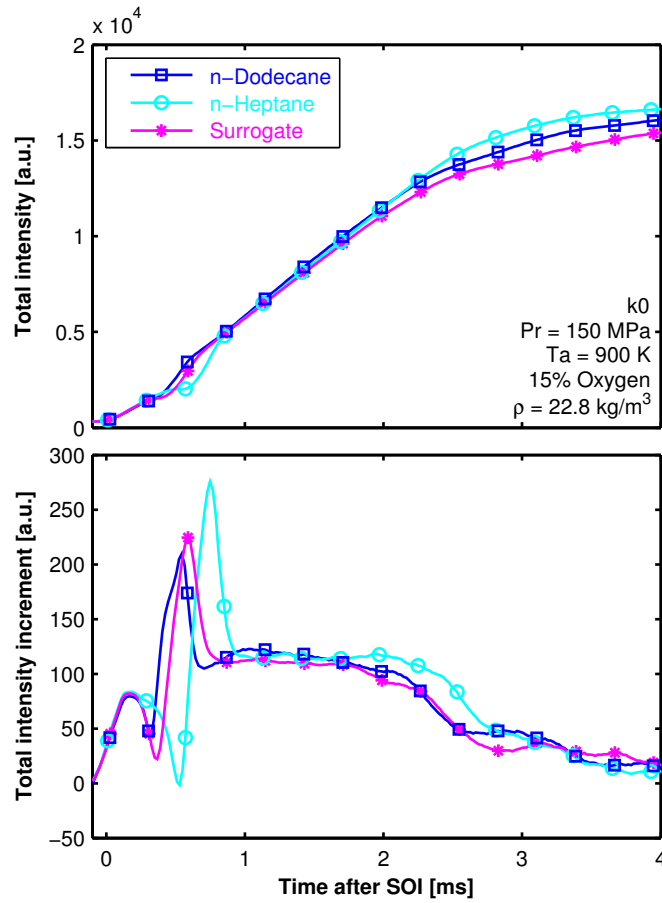


Figure 12: Total spray intensity (top) and intensity increment (bottom) tracer signals for the three fuels at particular test conditions. In this case the nozzle is  $k0$ , rail pressure is 150 MPa, ambient density is 22.8 kg/m<sup>3</sup>, ambient temperature is 900 K and oxygen concentration is 15 %.

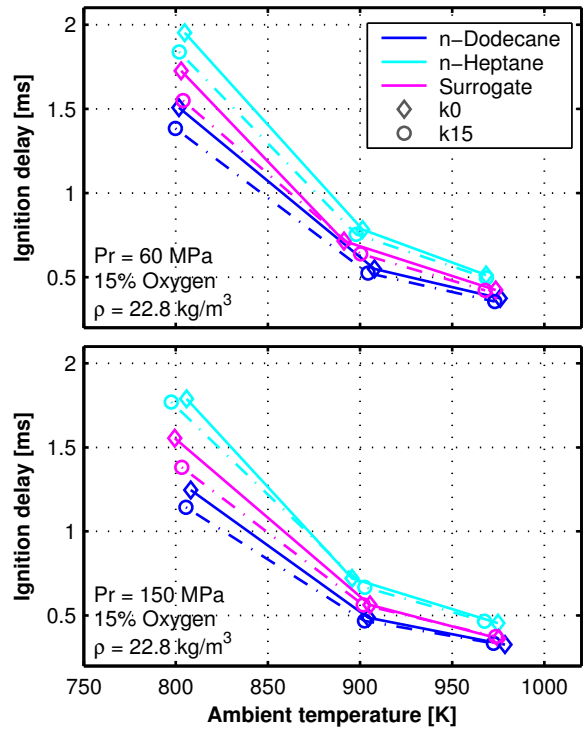


Figure 13: Ignition delay after SOI as a function of ambient temperature for all nozzles and fuels at an ambient density of  $22.8 \text{ kg/m}^3$ , an oxygen concentration of 15% and rail pressures of 60 MPa (top) and 150 MPa (bottom).

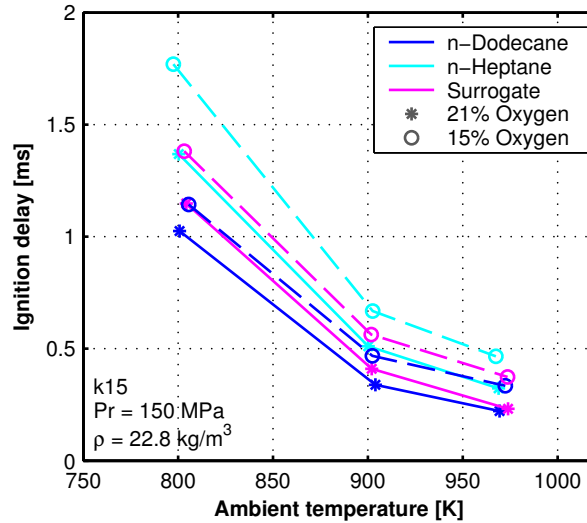


Figure 14: Ignition delay after SOI as a function of ambient temperature for all fuels and the two oxygen concentrations tested. In this case, the nozzle is *k15*, ambient density is  $22.8 \text{ kg/m}^3$  and rail pressure is 150 MPa.

37 also show that the different fuels respond consistently, in terms of ignition  
 1 performance, to variations in ambient conditions, as is also the case for the  
 2 nozzle geometry.

### 3 3.4.3. Further discussion on the effect of nozzle geometry over the SSI

4 The effect of nozzle geometry over the ignition performance of the fuel  
 5 sprays was found to be very consistent throughout the full test matrix, as  
 6 Figures 13, 14 and 15 illustrate. The conical nozzle *k15* produces sprays that,  
 7 in average, feature 5.1% shorter SSI delays in comparison to those produced  
 8 by the cylindrical nozzle *k0*. It is important to point out that similar results  
 9 were previously reported by Kong and Bae [2] and Payri et al. [10] from  
 10 their studies in optically accessible engines, both of which found that conical  
 11 nozzles produced shorter ignition delays in comparison to cylindrical nozzles,  
 12 but these results contradict the numerical predictions reported by Som et al.  
 13 [15]. This contrast, along the little information found in the literature on  
 14 the extent of the effect of nozzle geometry over ignition performance of diesel  
 15 sprays, leave room for further discussions on the subject, and additional  
 16 analyses—both numerical but also experimental—should be performed to  
 17 arrive at solid conclusions.

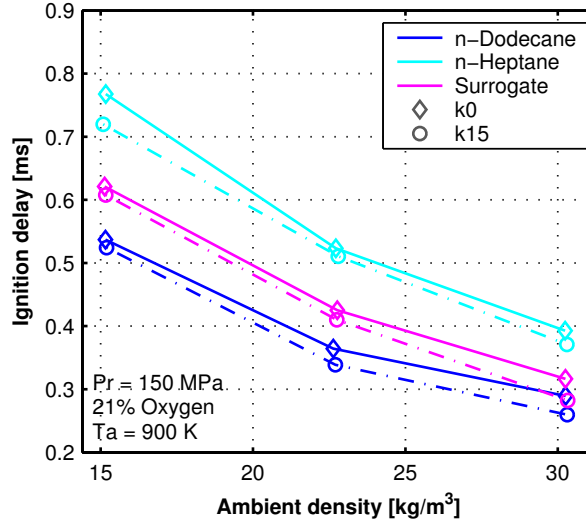


Figure 15: Ignition delay as a function of ambient density for all nozzles and fuels at a rail pressure of 150 MPa, an oxygen concentration of 21 % and an ambient temperature of 900 K.

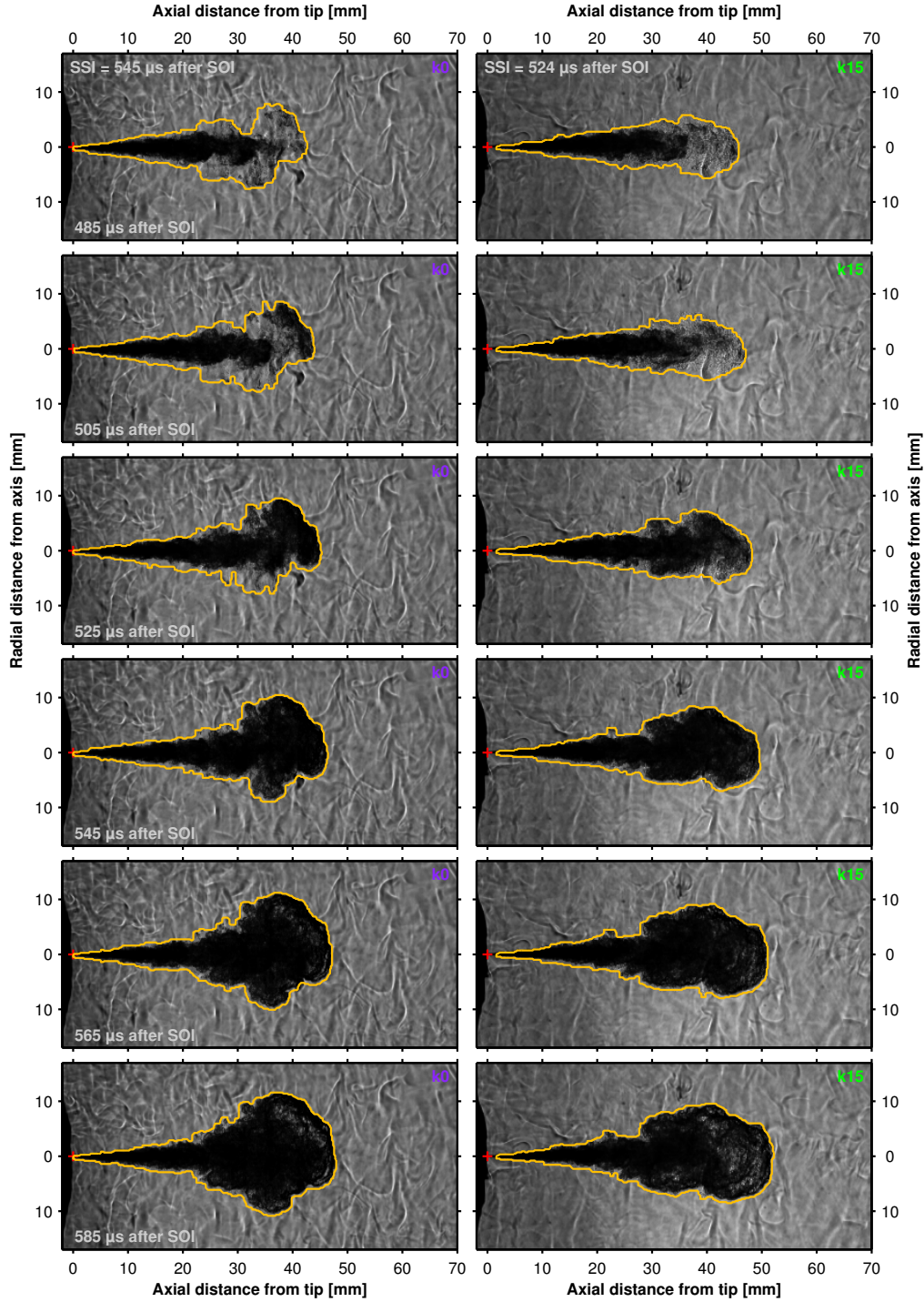
18 In the case of this study, as the test matrix is so large and the trend  
 1 between nozzles is so consistent, hypotheses should be discussed. The be-  
 2 havior observed might be unexpected at first, since the cylindrical nozzle  $k0$   
 3 features stronger turbulent velocity profiles at the outlet orifice that produce  
 4 larger spreading angles and spray boundary fluctuations [30–32], which conse-  
 5 quently lead to shorter liquid lengths [31]. Shorter liquid lengths can mislead  
 6 one to expect shorter SSI delays because of the often associated faster liq-  
 7 uid breakup and mixture preparation. Nevertheless, in these mixing-limited  
 8 sprays, liquid length is strongly dependent on spreading angle, which the  
 9 authors believe is the dominant parameter for the shorter liquid lengths fea-  
 10 tured by the cylindrical nozzle  $k0$ . In their studies, Kong and Bae [2] and  
 11 Payri et al. [10] attribute the shorter ignition delays found for their conical  
 12 nozzles—in comparison to their cylindrical nozzles—to better atomization  
 13 and liquid breakup, due to the thinner liquid core produced by the smaller  
 14 nozzle diameter of the conical nozzles. This could also be the case for the  
 15 present study. On the same lines, even if the two nozzles in this study were  
 16 assumed to produce sprays with similar break-up/vaporization times or even  
 17 shorter for the cylindrical nozzle, at the moment either spray reached va-  
 18 porized ignitable mixtures the local equivalence ratio at the ignition location

19 of the spray produced by the cylindrical nozzle *k0* would be expected to be  
1 lower than that of the spray produced by nozzle *k15*, because of its signifi-  
2 cantly larger spread volume [31, 54] at virtually similar injected mass [30].  
3 As it is largely known, the reaction paths at low temperatures are dependent  
4 on radical species formed directly from the fuel, so richer mixtures oxidize  
5 faster [21, 55].

6 Figure 16 presents a sequence of Schlieren images showing particular ig-  
7 nition events for the sprays produced by the two nozzles at equivalent test  
8 conditions. Each row is labeled with the corresponding elapsed time after  
9 SOI, and the detected contours are shown to illustrate the spray line-of-sight  
10 area. The spray produced by the cylindrical nozzle *k0* (left side of Figure  
11 16) ignites later than the spray produced by the conical nozzle *k15* (right  
12 side of Figure 16), the difference in this case is small at 21  $\mu$ s but still appre-  
13 ciable in the images. It can be seen in this figure that, at the corresponding  
14 times of SSI for each nozzle (indicated at the top of each column), the spray  
15 produced by the cylindrical nozzle *k0* has spread considerably more, as was  
16 expected from the behavior of their inert vapor sprays [31]. The complete  
17 SSI delay time is a composition of the liquid break-up, fuel vaporization and  
18 chemical kinetic mechanisms. When comparing nozzles in this study, SSI  
19 delay results lead to think that the chemical kinetics are the dominant factor  
20 to the final outcome of the SSI delay behavior, due to the differences in local  
21 equivalence ratios between the sprays produced by the two nozzles. In cases  
22 with larger differences in ignition delay between nozzles, the spray produced  
23 by the cylindrical nozzle will have penetrated and spread further into the  
24 ambient gas, probably igniting at even lower equivalence ratios. This is also  
25 observable in both the top and bottom parts of Figure 11, since the total in-  
26 tensity signal is also a measurement of the line-of-sight spray area detected,  
27 and lines that corresponds to the cylindrical nozzle *k0* stays above lines that  
28 corresponds to the conical nozzle *k15*. This trend between nozzles regarding  
29 total intensity signals was consistent along the full test matrix. Moreover,  
30 Figure 6 also shows larger intensity profiles downstream of the LOL for the  
31 conical nozzle *k15* that could, qualitatively, be associated to a more sooting  
32 flame, also indicating richer equivalence ratios near the LOL region.

33 Finally, even though soot formation is out of the scope of this publication,  
34 the authors point out that a further study should be carried out to analyze  
35 soot formation for the three fuels. In particular, the Surrogate fuel is of inter-  
36 est, since it is conceptualized to better mimic the soot-related behavior of real  
37 diesel fuel, as its PAH content should increase soot formation in comparison





32  
 Figure 16: Time sequence of Schlieren images of two injection events of the sprays produced by  $k0$  and  $k15$  nozzles. Images have been trimmed both in the radial and axial directions from their original size, for better fit in this figure. The contours detected are plotted to scale over the original Schlieren images. In this case, fuel is n-dodecane, rail pressure is 150 MPa, ambient density is  $15.2 \text{ kg/m}^3$ , ambient temperature is 900 K and the oxygen concentration is 21 %.

38 to pure  $n$ -alkanes. However, it is also possible that the SSI delay induced  
1 by the  $\alpha$ -methylnaphthalene content is large enough so to delay ignition to  
2 a point where the local equivalence ratio is very low, which would render a  
3 less-sooting flame: a similar situation what was found for the secondary fuel  
4 of the ECN, which is a mixture of n-dodecane and m-xylene [36, 56].

#### 5 4. Conclusions

6 This study is a continuation of previous work by the authors. In previous  
7 publications, the authors studied the internal flow characteristics, isothermal  
8 liquid spray development, and evaporative inert spray development for the  
9 same nozzle geometries and fuels. Current study reports the ignition delay  
10 through Schlieren technique, and flame lift-off length through OH\* chemi-  
11 luminescence visualization, aiming to enhance the size and quality of the  
12 database already published. The nozzle geometries consist of a conical noz-  
13 zle and a cylindrical nozzle with 8.6 % larger outlet diameter when compared  
14 to the conical nozzle. Among the three fuels, two are pure components—  
15 n-heptane and n-dodecane—while the third consists of a three-component  
16 surrogate to better represent the physical and chemical properties of diesel  
17 fuel.

18 Reacting spray is found to penetrate faster than non-reacting spray due to  
19 combustion induced acceleration after ignition. Higher oxygen concentration,  
20 and ambient temperature enhance the reactivity leading to higher spray tip  
21 penetration. Injection pressure does not affect the reactivity significantly  
22 and hence, influences spray penetration through momentum—similar to a  
23 non-reacting spray.

24 Both ignition delay and lift-off length are found to be shortest and longest  
25 for n-dodecane and n-heptane, respectively, while the surrogate fuel falls  
26 in-between the two pure component fuels. Both ignition delay and lift-off  
27 length are found to decrease with increase in oxygen concentration, ambient  
28 temperature, and density, in agreement with previous works found in the  
29 literature. The conical nozzle, in spite of longer lift-off length is found to  
30 have shorter ignition delay, when compared to the cylindrical nozzle. This  
31 could be due to smaller liquid vain that breaks-up and vaporizes quicker to  
32 form a reactive mixture faster than the droplets from cylindrical nozzle, but  
33 could also be to the fact that the spray produced by the cylindrical nozzle  
34 spreads considerably more, which reduces the local equivalence ratio at the  
35 time of ignition. This trend between nozzles was found to be in agreement

36 to previous experimental studies [2, 10] but in contradiction to numerical  
1 predictions [15]. Hence, further analysis for a more in-depth understanding  
2 of this mechanics involved in this process should be considered.

3 The experimental findings from this work on the macroscopic spray be-  
4 havior, and the large database obtained (available for download at: <http://www.cmt.upv.es/DD01.aspx>), could be used to validate CFD models that  
5 could help the community understand the fundamental driving mechanisms  
6 behind these observations.  
7

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12 67679-c2-1-R. Additionally, the employed nozzles and Diesel surrogate were  
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