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Additional Information

Experimental and analytical study on vapor phase and liquid penetration for a high pressure diesel injector

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Abstract

In this study, a macroscopic characterization has been performed on a solenoid diesel injector (2200bar-8 hole nozzle) under various non-reacting but evaporative conditions. For vapor penetration a two pass Schlieren visualization set up was selected. A high speed camera was used to record high speed images of the injection event to analyze the transient evolution of the vapor phase of the spray. The transient liquid penetration of the spray has been measured via MIE-Scattering imaging technique using a high speed camera as well. Unsteady RANS based CFD Simulations have been performed to simulate the experimental conditions and correlation results are presented. Built-in models from commercial code StarCD have been used to model spray formation which includes submodels for turbulence, nozzle flow, break-up and fuel properties. A novel CAE process using an automation and optimization tool has been used to achieve robust model settings, and the final model prediction are compared with the experimental observation for the injector characterization with respect to the non-reacting spray penetration with change in ambient and injection conditions. The model correlates well with the sensitivities for temperature and injection pressures qualitatively however improvements required to capture the density effects mainly related to the mesh orientation, fixed time step size where further analysis required.

Keywords: Macroscopic characterization, Diesel spray, Lagrangian two-phase flow, Robust spray model, RANS

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Nomenclature

<i>RANS</i>	Reynolds Average Navier Stokes
<i>MPI2</i>	Modified Max-Planck-Institute
<i>RNG</i>	Renormalization Group
<i>MDO</i>	Multi-disciplinary Optimization
<i>SHERPA</i>	Simultaneous Hybrid Exploration that is Robust, Progressive and Adaptive
<i>CFD</i>	Computational fluid dynamics
<i>RMSE</i>	Root mean square error
<i>SOI</i>	Start of injection
<i>L/D</i>	Nozzle Length-to-Diameter Ratio
<i>Mie</i>	Mie-Scattering optical technique

1. Introduction

Nowadays, internal combustion engines continue to be an important alternative for energy transformation. The ever more demanding fuel consumption standards and the concerns about the environmental impacts of these engines have pushed the industry into the search of new strategies and technologies. This encourages new studies for improving engine performance and its emissions.

The injection process has been mentioned as an important player in order to improve emissions and engine performance.[1–6]. The spray formation includes complex and heterogeneous processes, majorly high-velocity jet flow, liquid droplet break-up, atomization, and evaporation of a dense liquid spray in a turbulent flow environment. The small temporal and spatial scales resulted from this process makes the diesel spray evolution a complicated problem.

To ensure a good mixture between the air and the fuel, the spray must penetrate into the combustion chamber and atomize. There are several parameters that help to characterize the diesel spray from a macroscopic point of view. The liquid length is an indicator of the evaporation capacity of the fuel and it is defined as the distance from the nozzle to the point where are found the ambient conditions necessary for evaporation. Mie scattering imaging technique is widely used by the engine community for the visualization of the fuel spray liquid phase. This technique consists in illuminating the fuel droplets with a light source and collecting the light scattered with a camera [7–9]. The vapor penetration largely determines both the mixing process and the probability of collision against the chamber walls. It depends essentially on the instantaneous momentum of the spray in the nozzle. The Schlieren technique is able to distinguish gradients in the reflective index of a transparent medium [10, 11], which allows clear identification of the vapor phase of the spray in evaporative conditions.

Since diesel combustion is predominantly a mixing-controlled reaction process, modeling the diesel spray formation process accurately is an essential prerequisite for modeling combustion events. The processes involved in the injection event are nonlinear and controlled

29 by multiphase, diffusion phenomena. Modeling the interaction between those complex
30 phenomena poses a huge challenge, and obtaining a unique model which can be robust
31 for a wide range of in-cylinder conditions during fuel injection event is important. For in-
32 dustrial application, it is important that the model is viable with computational time and
33 cost. A wide range of numerical models and sub-models exist in the literature [12–19] by
34 various research groups which are inherently different in many aspects. The database is
35 huge and detailed however still limited to the single-hole injector with moderate injection
36 pressures.

37 In this work, a diesel multi-hole common rail injector (2200 bar) has been modeled us-
38 ing Lagrangian two-phase flow spray model. Relevant turbulence, nozzle flow, break-up
39 models have been selected. Since the properties of diesel used in tests are unavailable var-
40 ious surrogate fuel properties have been applied. The sensitivities of model settings are
41 included in the study, observations are discussed in section 4. A novel CAE process using
42 an automation and optimization tool has been presented which was used to investigate a
43 range of model setting combinations to achieve robust spray model settings. The results
44 from simulations obtained with the optimized parameters, are compared with results from
45 visualization and characterization experiments carried out in this study.

46 **2. Model development**

47 *2.1. CFD Methodology*

48 The numerical simulations are performed using the commercial CFD tool Star-CD. The
49 turbulent flow field is resolved using the k- ϵ equation based Renormalization Group (RNG)
50 turbulence model as this is in-cylinder combustion best practice [20]. The Lagrangian
51 based two-phase flow model has been used to resolve spray formation. The nozzle inflow
52 models are used to capture the nozzle hole exit velocities, the two models considered
53 here are the Effective and MPI2 (modified Max-Planck-Institute) [21, 20] models from
54 Star-CD. The advantage of the MPI2 model is that it automatically determines whether
55 cavitation occurs inside the nozzle and distinguishes whether it reaches the nozzle exit
56 or ends inside the nozzle. For all simulations in this paper, the properties of n-dodecane
57 ($C_{12}H_{26}$) were used as a surrogate for diesel fuel, these were taken from the internal Star-
58 CD fuels library [22]. The properties of the surrogate can be seen in table 1.

59

Table 1: n-Dodecane properties @ 298.15 K & 101325 Pa

Properties	Value	Units
Molecular weight	170	kg/mol
Critical temperature	658.65	K
Critical pressure	1.835×10^6	Pa
Boiling temperature	489.48	K
Density	745.76	kg/m ³
Molecular viscosity	0.00137563	kg/ms
Surface tension coefficient	0.0248679	N/m

60 The injected liquid with high velocity starts to break-up into smaller droplets, the
61 process comprises of primary breakup (i.e. atomization) and secondary breakup. A
62 range of built-in sub-models is available with-in Star-CD to model this phenomenon.
63 Atomization models differ in the way droplet size distribution and initial velocities are
64 calculated. The difference between the droplet break-up models is the correlations that are
65 used to estimate the time scale of the break-up process and the stable droplet diameter.
66 The Huh atomization model and Reitz -Diwakar droplet break-up models have been used
67 [20]. The Huh model calculates the spray cone angle during simulation so this is not
68 required as an input.

69 The heat and mass transfer process is modeled using Ranz-Marshall correlation [23] to
70 capture the evaporation process. The drag process and turbulence dispersion are modeled
71 using stand correlation [20]. The inter-droplet collisions are not modeled, as the RNG
72 turbulence model does not take this into account [20]. The droplet-wall interaction is not
73 significant in this bomb case setup, however, the Bai model has been selected to consider
74 any such process [22].

75 *2.2. Computation grid and boundary conditions*

76 The 3D computational domain used to represent the spray chamber fluid volume has
77 been created as shown in Fig.1. The boundary with the injector is defined an adiabatic
78 wall whereas the other boundaries are defined as pressure-outlets. The dimensions of
79 the cuboid are maintained to enclose the non-reacting spray from the multiple injector
80 holes whilst minimizing the influence of the boundary wall on the spray. The location
81 of injector hole is defined at the center of the domain, at a certain depth below the wall
82 surface for the same reason. All the dimensions and characteristics of the computational
83 domain can be found in [22].

84 The coordinate system seen in Fig.1 represents each of the 8 holes of the injectors,
85 and was used to set the injection locations within the domain based on the geometrical
86 specification of the injector. A uniform grid with cell size 0.8 mm has been selected as
87 these are the settings used within full combustion models within the JLR 3D-thermofluids

88 diesel combustion group, for which the tuned spray model parameters are required [22].

89 2.3. Automation and optimization: HEEDS

90 In this section, HEEDS is briefly explained. For this study the HEEDS MDO (Multi-
91 Disciplinary Optimization) software was utilized in two manners, firstly to carry out a
92 DoE (design of experiment study) into a set of tuning factors to investigate the effects of
93 each on the penetration, and secondly to provide an automated workflow and optimization
94 methodology to target a CFD solution which matches the experimental observations for
95 a specific operating condition. SHERPA is the main algorithm used for optimization;
96 the basic mechanism is that algorithm uses the results from numerical simulations to
97 adapt to a new search path after each run and thus the number of evaluations required to
98 arrive to given target performance can be quite different from run to run. The detailed
99 mechanisms of algorithm is intellectual property of the Red Cedar technologies, however
100 the high level description of the algorithm can be found in[24]. The SHERPA algorithm
101 automatically applies the appropriate optimization algorithms for the problem based on
102 what it has learnt about the design space in the previous results. The design space is
103 navigated as the optimization algorithm performs real CAE analyses, rather than an
104 approximated surrogate model. Without the need to generate a surrogate model, the
105 number of analysis runs required is reduced, saving time and resources. Other advantages
106 of using this method include the fact that the user does not need to understand the design
107 space to select an appropriate algorithm prior to starting to the optimization, nor does
108 the user need to have any expertise in optimization applications.

109 Within the HEEDS GUI an automatic process is setup, instructing the software which
110 step is the following to execute for the optimization. First, the input parameters to be
111 varied, for the spray characterization, were selected and tagged for editing in Star-CD
112 model example input files. Similarly, the responses are created and tagged in example
113 output files, the penetrations for liquid and vapor are written out to a database file using
114 an user sub-routine. The main responses in this study are liquid penetration and vapor
115 penetration. Experimental data of liquid and vapor penetration is read into HEEDS as
116 target curves (Y_t) and the simulation output of liquid and vapor penetration were tagged
117 as design curves (Y_d). HEEDS using the equation below for the Root Mean Squared Error
118 (RMSE) to find by targeting the minimum value for each response from the entire design
119 space.

$$RMSE = \sqrt{\frac{1}{n} \sum_{i=1}^n (Y_d + Y_t)^2} \quad (1)$$

120
121 Finally a study is setup in which the variables and responses to be taken into account in
122 the optimization are set, as well as targets (which can be weighted based on importance)
123 and constraints for the solver. For the DoE, the low/high values for each input factor were
124 given and no constraints or objectives were needed. For the calibration run no constraints
125 were set, and a standard SHERPA parameter optimization with weighted curves were

126 used, and the objectives were set to minimize the “Liquid” and “Vapor” responses, i.e.
127 to minimize RMSE. The details of variables and their levels for respective studied are
128 presented in detail in table 3 and 5.

129 **3. Experimental set up**

130 *3.1. The high pressure and high temperature test rig and fuel injection system*

131 The feeding system for the fuel consists of a conventional common rail configuration that
132 contains a high pressure pump and a rail with a pressure regulator. This system allows
133 injections at high pressures up to 2200 bar. The measures were carried out in a test ves-
134 sel classified as a constant-pressure flow (CPF) facility (fig.2), where the thermodynamic
135 conditions of the engine combustion chamber can be reproduced [25]. The gas is stored
136 by volumetric compressors in high pressure reservoirs and flows continuously through the
137 test chamber. To keep the gas in the test section at the desired temperature, 30kW elec-
138 trical heaters were placed upstream of the chamber. The control system is a closed loop
139 PID that adjusts both the pressure in the chamber and the power of the heaters to obtain
140 the test conditions required for the experiments.

141 This test rig allows a maximum ambient temperature of 1000 K and a maximum pres-
142 sure of 150 bar. The gas at high pressure and temperature continuously flow through
143 the chamber at 0.3m/s. The test rig has three large windows (128 mm diameter) that
144 give full optical access, and the big chamber diameter (200 mm) minimizes the spray-wall
145 interaction. In this study, the vessel has been filled with nitrogen to guarantee the evap-
146 orative but non-reacting conditions sought.

147

148 *3.2. Optical set up for vapor penetration (Schlieren-based)*

149 The Schlieren imaging technique was used to identify the spray vapor phase boundaries
150 at evaporative conditions. The technique is based on the change of refraction of parallel
151 light rays that pass through non-homogeneous fluids checking density variations [26]. The
152 refractive index gradient into the region of interest will cause the deviation of some rays.
153 Using a spherical lens to collect the beam, only parallel rays will converge to the focus
154 point of the lens. Then parallel rays can be identified and trimmed using a diaphragm at
155 the focus point, obtaining a shadowgraphic image. For this test a double-pass Schlieren
156 configuration using a high temperature mirror has been used. The set up for this con-
157 figuration can be seen in fig.2. The main difference of the optical arrangement for a
158 double-pass Schlieren setup, is the fact that the rays are passing two times through the
159 test section, being reflected by the mirror placed right behind the test section. Since the
160 light is reflected by the mirror toward the same direction it is coming from, a beam-splitter
161 is required to complete the setup and reflect the image to the camera [26, 27, 11]

162 *3.3. Optical set up for liquid penetration (Mie-scattering-based)*

163 The MIE scattering optic technique was used to identify the spray liquid phase boundaries.
164 It consists in illuminating the spray with a light source (continuous or pulsed) so the
165 scattered light could be collected by a fast camera.

166 In this work, the sprays have been illuminated by the front window with two continuous
167 Xe-arc lamps and the light scattered backward was collected by a high speed CMOS
168 camera (Phantom v12) aligned with the injector axis. The size of the images were 768
169 x 712 pixel with a spatial resolution of 5.41 pixel/mm. The acquisition rate was 24 kfps
170 The set up for this configuration can be seen in fig. 3.

171 *3.4. Image Processing*

172 Each image recorded is first divided into eight sectors, one for each outlet orifice and thus
173 one spray. In this way, each spray is processed separately by applying mask to isolate
174 the spray of interest. The algorithm used for the processing is described in [11, 28]. The
175 image is inverted in order to have the spray as the high luminosity area and the threshold
176 is calculated as the 3% of that image's dynamic range [11, 29]. Consequently, small areas
177 that come from background noise are ruled out and finally the spray contour is "cleaned"
178 free of small noise fluctuations through a pixel connectivity evaluation. This last step
179 could be seen as a contour smoothing. Liquid penetration (Mie scattering) and vapor
180 penetration (Schlieren) are calculated by detecting the pixel on the contour that is the
181 furthest from the outlet orifice; the penetration is then calculated as the axial distance
182 from the injector outlet to the furthest point [27].

183 *3.4.1. For Mie-Scattering*

184 The steps followed in the image processing for the Mie-scattering study are summarized
185 below [11]:

- 186 1. The image acquired right before the start of injection is arithmetically subtracted
187 from the spray images, in order to remove reflections and background artifacts.
- 188 2. In order to analyze each spray individually, the image has been divided in 8 sectors.
- 189 3. The contour of each spray is obtained using a variable threshold (ths). The threshold
190 is calculated as the 3 % of the dynamic range of the sector.
- 191 4. Applying the threshold the image is binarized. The connectivity algorithms are
192 employed to distinguish between the spray and the artifacts due to sensor noise
193 [11, 28].
- 194 5. The spray boundary is finally obtained as the contour of this area.

195 *3.4.2. For Schlieren*

196 For the image segmentation, it was used the same approach followed in Mie scattering
197 tests: the image was separated in sectors to process each spray separately, the background
198 subtraction was applied and a black and white image was obtained using a scaled threshold

199 as in Mie scattering. However, the images obtained in Schlieren are characterized by
 200 features that required to modify the processing routine used for the Mie images. Some
 201 considerations about this kind of tests are [11]:

- 202 1. The spray appears darker than the background, therefore an inversion of the image
 203 and of the background is convenient.
- 204 2. The head of the three bolts holding the mirror appears as dark spots in the im-
 205 age. Therefore, the sprays interfering with this bolts will not be processed to avoid
 206 erroneous measurements.
- 207 3. The temperature/density gradient related to the turbulent flow appears in the back-
 208 ground and caused fluctuations over a wide range of counts level. Connectivity al-
 209 gorithms have been modified to obtain accurate spray boundaries.

210

211 3.5. Test Plan

212 A wide range of conditions have been explored in order to study the spray penetration
 213 and are summarized in table 2.

Table 2: Experimental test program.

Parameter	Value-type	Units
Fuel	Diesel	-
Energizing time	1500	μs
Intercoolant temp.	363	K
Ambient gas density	20, 25, 30	kg/m^3
Ambient gas temp.	600, 800, 900	K
Injection pressure	1100, 1500, 1800, 2200	bar
Oxygen concentration	0	%

214 4. Results

215 4.1. DoE Investigation

216 The DoE investigation results are presented in this section. For this study, it was decided
 217 that a 2 level full factorial investigation would be carried out, meaning that 256 designs
 218 needed investigating, a task which could not have been done by hand, due to the time
 219 it would have taken to set up and post process each simulation. The factors included 2
 220 nozzle flow models, Reitz-Diwakar break-up model factors Te-Strip (refers to the empirical
 221 coefficient Cs2 of characteristic time scale of stripping break-up regime [20, 30] and We-
 222 Bag (refers to the empirical coefficient Cb1 to determine the stable droplet size for Bag

223 break-up [20, 30], along with other injector nozzle parameters with potential geometrical
 224 uncertainties. For the DoE, the low and high values for the investigation were set in
 225 HEEDS as shown in table 3.

Table 3: Range of parameters used in HEEDS DoE Investigation

Factor	Model parameter	Low	High
A	L/D	4	8
B	Cd	0.7	0.8
C	Contraction ratio	0.5	0.6
D	Te-Strip	9	20
E	Injection Temperature	330	353
F	Injection Parcels	$1e + 07$	$2e + 07$
G	We-bag	3.6	8.4
H	Nozzle Model	1 – Effective Model	3 – MPI2 Model

226 From an interaction point of view the DoE results show us that Te-Strip and nozzle
 227 model have the greatest interactions of all the factors for both the liquid and vapor. On
 228 the other hand, the number of injection parcels has very little effect on the results nor
 229 does it interact with any of the other factors, similar trends were seen with the injection
 230 temperature and contraction ratios.

231 From the main effects plots in Fig 4a and 4b, it is clear that the importance of each
 232 factor for the liquid and vapor penetrations are not equal. Te-Strip has the largest effect
 233 on both, followed by the nozzle model. The nozzle interaction for liquid and vapor is
 234 reversed, in other words the user must choose between a better match to experiment for
 235 liquid or for vapor but cannot do both at the same time.

236 In running this DoE, the injector hole diameter was fixed, however, the effective nozzle
 237 model requires the diameter to be reduced as it does not take account of cavitation inside
 238 the nozzle hole. An additional study was carried out using the ‘best’ settings from the
 239 DoE results with the effective nozzle model whilst sweeping the diameter from geometrical
 240 (0.000144m) down to minus 10% of geometrical (0.00013m). The results of the sub-study
 241 showed that the effective nozzle can provide a better result than the DoE for Vapor RMSE,
 242 whilst simultaneously preserving (or improving) the RMSE for Liquid (compared to the
 243 DoE when optimized for vapor). In other words, the Effective nozzle sweep provides a
 244 better trade-off, maintaining the best results from the DoE for both liquid and vapor.
 245 The final best settings taken forward to the CFD comparison to experimental have been
 246 listed in the table 4.

Table 4: "Best" settings taken from study.

Model parameter	Best setting
L/D	9
Cd	0.8
Contraction ratio	0.55
Te-Strip	20
Injection Temperature	353
Injection parcels	1e+07
We-Bag	8.4
Nozzle model	Effective model
Hole diameter	0.00013

247 4.2. Experimental to CFD Comparison

248 4.2.1. Effect of ambient density on vapor and liquid penetration

249 It has been reported before that for both liquid and vapor penetrations, ambient
 250 density is a crucial parameter [31–33]. The figures 5a and 5b reflect this behavior and
 251 show that an increase in density causes a slower spray penetration. A higher density of
 252 the entrained gas requires more kinetic energy to achieve the momentum transfer, and for
 253 this reason the spray penetrates slower.

254 The CFD spray model with the stated best settings captures the trend of density effect on
 255 the spray penetration qualitatively, although vapor penetration over-predicts and liquid
 256 penetration under-predicts.

257 Further analysis is required to improve the predictions quantitatively, mainly improving
 258 the overshoot in the liquid penetration before steady spray is achieved. It is believed this
 259 could be due to the fact the cell size and time-step size chosen is slightly too large to
 260 correctly capture the initial spray penetration phase.

261 4.2.2. Effect of Injection pressure on vapor and liquid penetration

262 As previously exposed in the table 2, two injection pressures were studied during the
 263 experimental stage. fig.6 illustrates the influence of the injection pressure. As expected,
 264 an increase in injection pressure produces an increase in vapor penetration rates, whilst
 265 having relatively little impact on the liquid penetrations.

266 The CFD model reproduces the injection pressure effects, for vapor the penetration rates
 267 are increased for the 2200bar injection, and the liquid steady spray for the 2200bar is
 268 slightly lower than for the 1100bar injection. However the CFD over estimates the differ-
 269 ence between the steady sprays for both injection pressures.

270 Once again the liquids steady sprays are under-predicted whilst the vapor spray penetra-
 271 tion trends are over-predicted.

272 *4.2.3. Effect of ambient temperature on vapor and liquid penetration*

273 For this study two ambient temperatures were selected (fig.7), and the effects on va-
274 por and liquid length penetrations were verified. Naturally, the ambient temperature does
275 not affect vapor penetration significantly, since the key parameters determining the pen-
276 etration were kept constant in the comparison (injection pressure and ambient density).
277 However, the figure shows a subtle consistent trend, indicating a slight decrease in vapor
278 spray penetration with increasing temperature. This behavior has been reported before
279 [34], and it is maybe caused by the reduction in droplet size due to the evaporation at
280 higher temperatures, which may facilitates the momentum transfer from the fuel to the
281 surrounding air.

282 For liquid penetration as expected, curves for each temperature overlap in the first tran-
283 sient part since the density is the same in the two cases. However, the liquid length
284 penetration stabilizes at different values as a result of the evaporation fuel at higher
285 temperatures. The subtle effect on vapor penetration and significant effect on liquid
286 penetration due to change in ambient temperature have been well captured by the CFD
287 model, which suggests the model sensitivity vs. the change in temperature is good enough
288 to be used in diesel-like in-cylinder conditions.

289 *4.2.4. Effect of density and temperature on maximum liquid length penetration*

290 In fig.8 two different trends for the maximum liquid penetration can be seen. The first
291 trend describes a decrease of the maximum penetration with increased temperature, as
292 it has been explained in the previous section. The second trend seen in the experimental
293 data is a linear decrease of the maximum penetration with increased density.
294 The CFD model captures the sensitivity due to change in temperature very well. How-
295 ever, it is less capable when it comes to the sensitivity due to the change in density,
296 initially showing a decrease between 20 kg/m^3 and 25 kg/m^3 , but then rising again at
297 30 kg/m^3 . It is thought this could be due to the mesh not being spray oriented, hence
298 causing discrepancies in the spray momentum predictions. Further analysis is needed to
299 understand the spray orientation effects.

300 *4.3. Model calibration study*

301 A further study was carried out using the HEEDS optimization features to calibrate a
302 model to a specific operating condition to obtain a new set of tuned settings [22]. To
303 do so the experimental data was offset so that both had $\text{SOI} = 0$ as this is what the
304 CFD is set to. For this study the number of factors for tuning was reduced, using the
305 results of the DoE to remove those with little effect on the results. In total 50 models
306 were run. The table 5 outlines which factors were used and the ranges set for each, and
307 the best settings found from the calibration. All other factors were either set to best
308 practice or those found in the DoE. The deviation in the initial liquid penetration could
309 pertain to the Eulerian - Lagrangian approach, where the initial velocity (estimated using
310 nozzle flow sub-model) and time-step lead to droplet parcels with higher momentum which

311 penetrates farther. Velocity profile from detailed nozzle in-flow simulation and variable
 312 time-step could improve the results. A further detailed analysis has been carried out to
 313 check the model related uncertainties to understand this phenomenon

Table 5: Factors and ranges used for calibration

Parameter	Minimum	Baseline	Maximum	Interval	Best Settings
L/D	4	8	8	0.05	4.65
Cd	0.61	0.85	0.9	0.01	0.77
Te-Strip	2	13	20	0.1	19.7
We-bag	3.6	8.4	8.4	0.1	5
Hole Diameter	0.00013	0.000144	0.000144	0.000001	0.000144

314 A direct comparison of the liquid and vapor penetrations between experimental, DoE
 315 tune, CFD results, and the calibrated model settings CFD results has been plotted in fig.9.
 316 We can see that the calibrated settings give a better steady state liquid penetration match
 317 than the DoE settings, however the initial rise in penetration is somewhat delayed. For
 318 vapor, the difference is less extreme, though the calibrated model does appear to better
 319 match the shape of the experimental curve than the DoE settings. It was observed that
 320 the SOI for the experimental data may have been moved too far during the pre-processing.
 321 The measured SOI of the experimental data is extrapolated from the measured data and
 322 so there will be a margin of error. The largest difference in factor settings between the
 323 DoE best settings and the HEEDS calibration settings are for L/D (8 to 4.65) and We-
 324 Bag (8.4 to 5). The injector hole diameter is also different. The HEEDS optimization
 325 has selected the geometrical value, whereas we would normally expect to have to decrease
 326 this to account for cavitation.

327 Overall the main differences in the two settings have a greater effect on the liquid
 328 penetration than vapor penetration. The increase in hole diameter for the calibrated
 329 from DoE settings results in a reduction of droplet velocities at nozzle exit and hence
 330 lower break-up and evaporation, hence deeper penetrations. Further work is required
 331 to understand how further optimization should be run in terms of numbers of designs
 332 requested, weighting of targets, and looking into the experimental uncertainties related
 333 to vapor and liquid measurements.

334 5. Conclusions

335 A solenoid 8-hole 2200 bar diesel injector has been characterized from a macroscopic point
 336 of view by means of Mie-Scattering and Schlieren optical techniques. The effects of ambi-
 337 ent temperature, injection pressure and ambient density were studied in a constant flow
 338 high pressure and high temperature test rig. The facility emulates in-chamber conditions
 339 at the time of injection by means of pressurized and heated gas, to a maximum pressure

340 and temperature of 150 bar and 1000 K respectively.

341 As expected, injection pressure affects the vapor penetration but not so much liquid pen-
342 etration. A negligible decrease in vapor penetration has been found by increasing tem-
343 perature, this is maybe caused by the reduction in droplet size at higher temperatures,
344 which facilitates the momentum transfer. Both liquid and vapor penetration decrease with
345 an increase in the ambient density. This is due to higher momentum transfer at higher
346 densities.

347 On the other hand, 3D CFD simulations were performed using the built-in sub-models
348 from commercial software StarCD. The spray model DoE investigation showed sensitivity
349 and interaction of various factors, where nozzle flow model and break-up model factors
350 have significant effect compared to other parameters. A robust model setting has been
351 used to run different conditions, and the model results were correlated with experimental
352 data.

353 In summary, the CFD model is capable of capturing trends in the penetrations as per
354 the experimental data regardless of whether or not the model has been calibrated to
355 those conditions. Overall, the steady state liquid penetrations are under predicted by the
356 model and vapour over-predicts. This is because HEEDS exploration obtains a common
357 model setting that fits best for both liquid and vapor penetration. Extending further this
358 exploration study might result in obtaining combination of breakup model parameter
359 might achieve better match with experiment results. It has also been noted during the
360 work carried out, that the vapour penetration is less sensitive to changes in the factors
361 investigated than the liquid length. The model reproduces the key effects that are ob-
362 served in experimentation, such as injection pressure dominance on the vapour length and
363 temperature influence on the liquid length, however the sensitivity with density change
364 is marginal captured. Further model calibration and understanding the mesh orientation
365 effects is needed to improve the predictability. To conclude, the CFD results suggest that
366 when time or data is not available to calibrate the model, the DoE settings could be used
367 as a baseline as they appear fairly robust.

368 **Acknowledgments**

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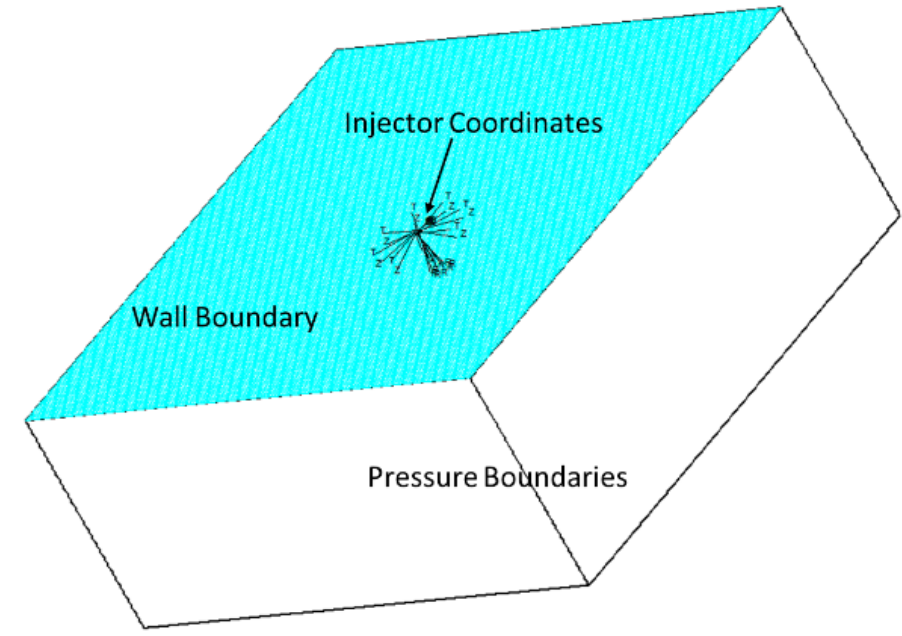
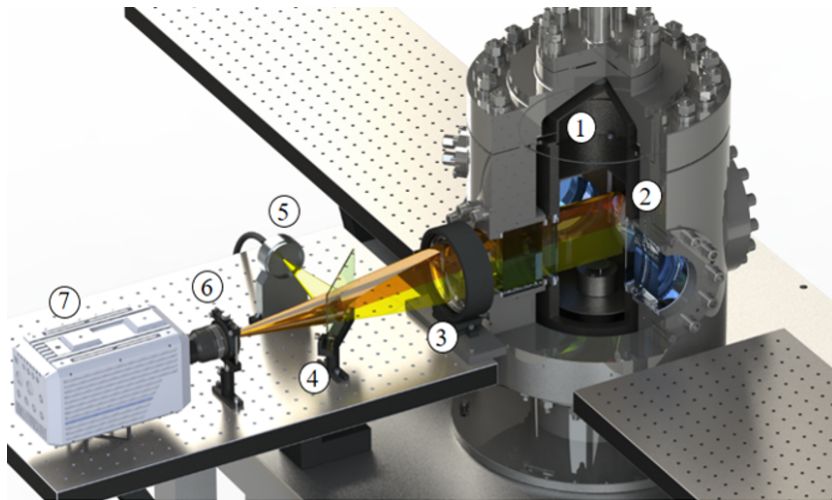
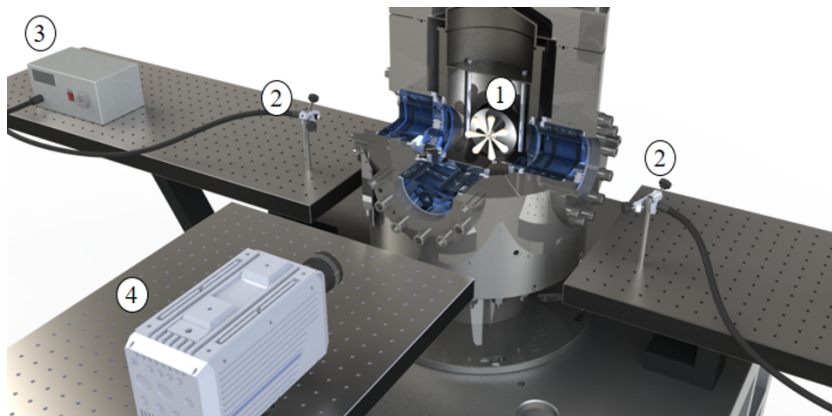


Figure 1: Domain Configuration



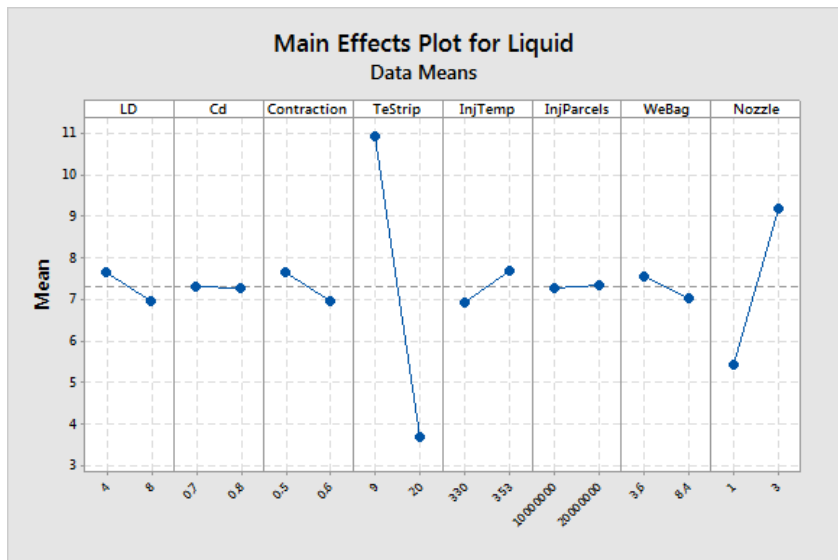
- | | |
|------------------|----------------|
| 1. Test rig | 5. Light font |
| 2. Mirror | 6. Diaphragm |
| 3. Convex Lens | 7. Fast camera |
| 4. Beam splitter | |

Figure 2: Optical setup for double-pass Schlieren

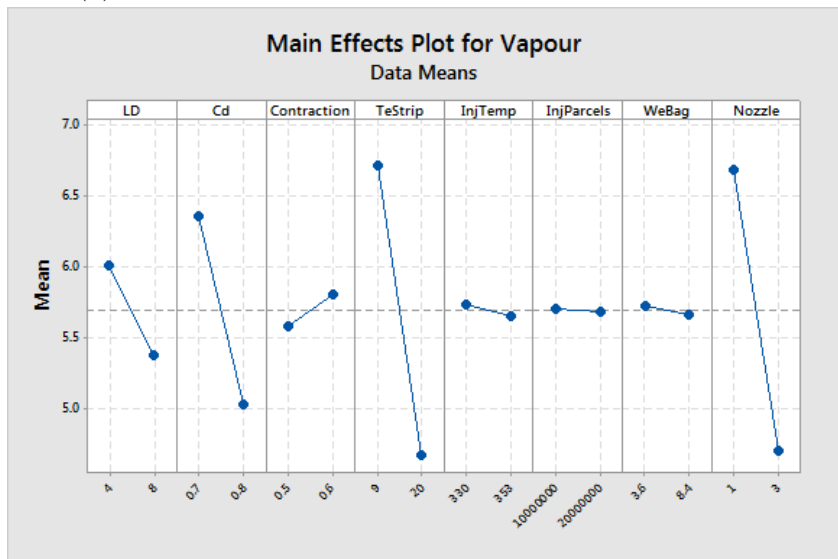


- | | |
|-----------------|----------------|
| 1. Test rig | 3. Light font |
| 2. Light source | 4. Fast camera |

Figure 3: Optical setup for MIE-Scattering

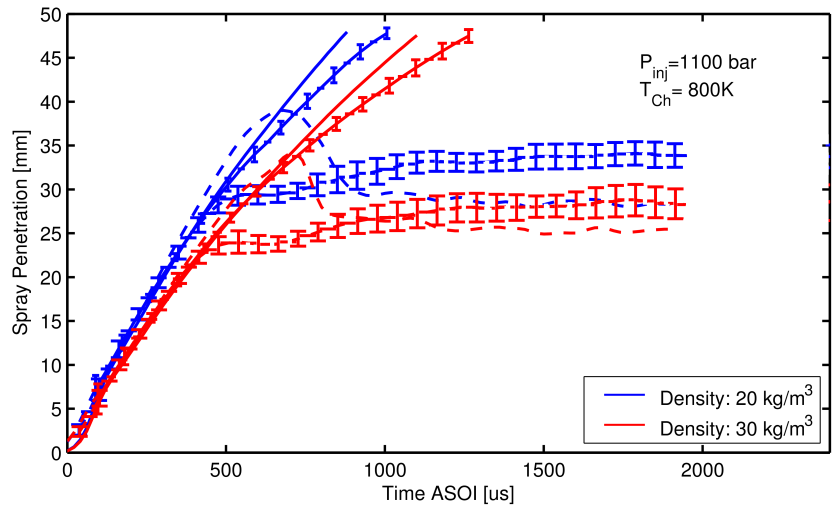


(a) Main Effects Plot for Liquid

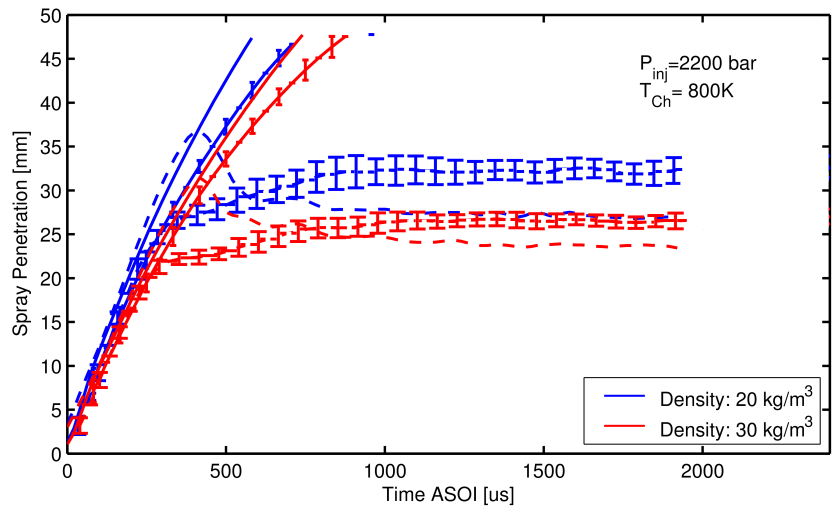


(b) Main Effects Plot for Vapor

Figure 4: Parameters effects.



(a)



(b)

Figure 5: Influence of ambient density at 1100 (a) and 2200 bar (b). Experiments (lines with error bars), CFD (lines w/o error bars), vapor (line), liquid (Dash line)

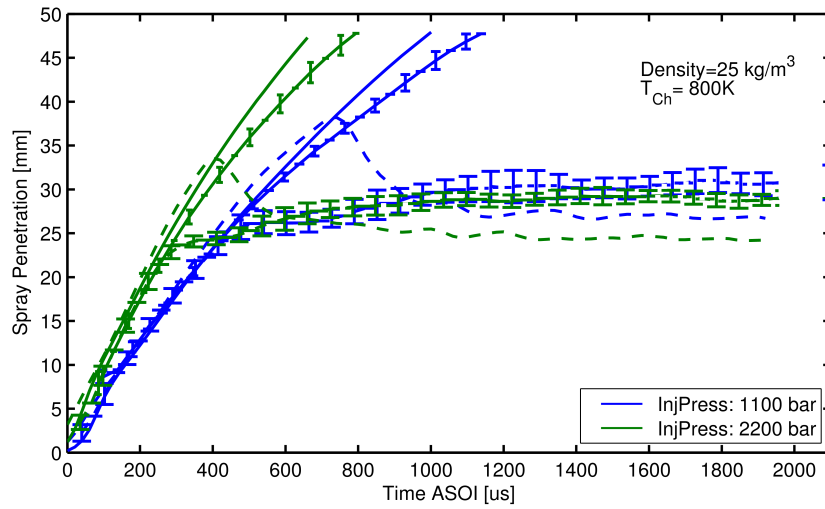


Figure 6: Influence of injection pressure. Experiments (lines with error bars), CFD (lines w/o error bars), vapor (line), liquid (Dash line)

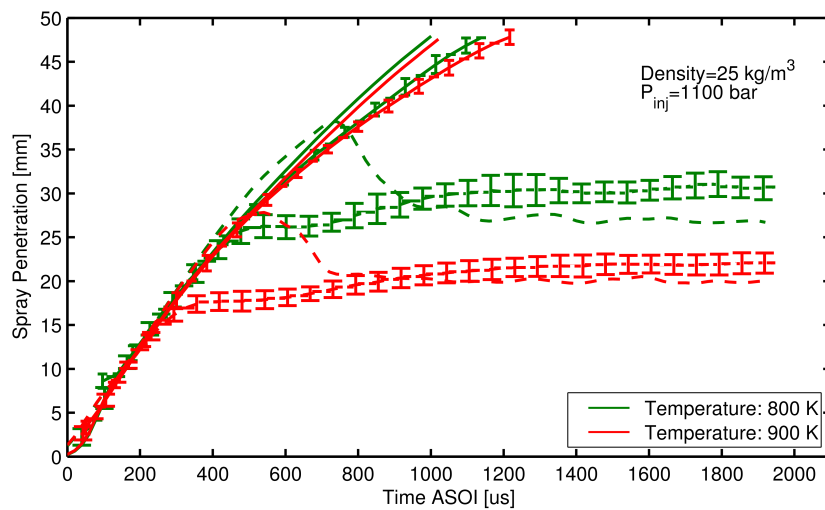


Figure 7: Effect of ambient temperature. Experiments (lines with error bars), CFD (lines w/o error bars), vapor (line), liquid (Dash line)

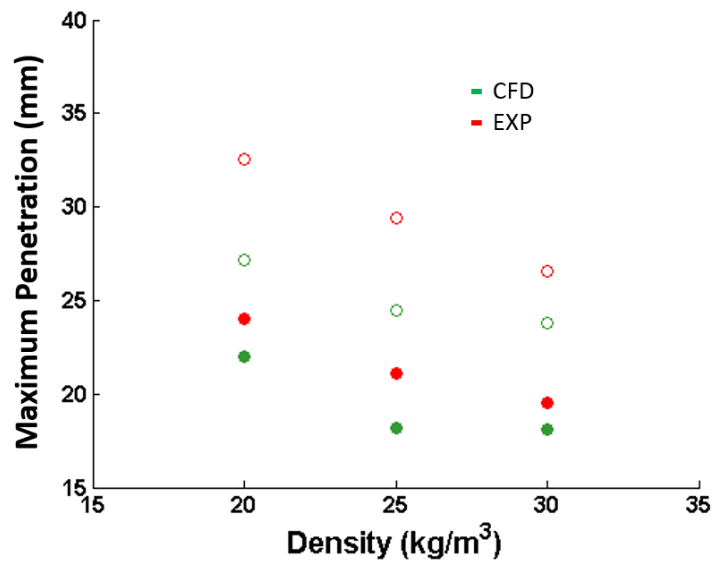


Figure 8: Effect of density and temperature on liquid length $P_{inj}=2200\text{bar}$. $T=800\text{K}$ (Markers unfilled), $T=900\text{K}$ (Markers filled)

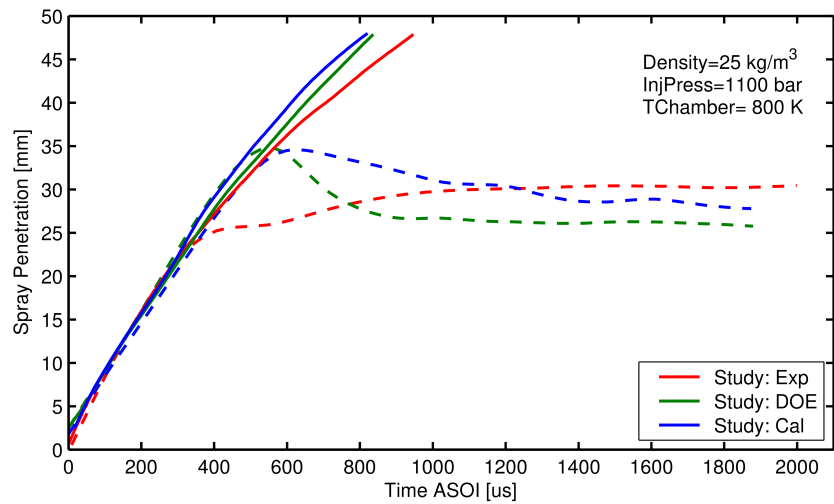


Figure 9: Calibration vs experiments. Vapor (Solid line), liquid (Dash line)