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Additional Information

1 **Long-term behavior of cracked fiber reinforced concrete under service**  
2 **conditions**

3  
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18  
19 **HIGHLIGHTS:**

20 Creep behavior of cracked fiber reinforced concrete is analyzed in service conditions.

21 Different stress levels and pre-cracking damages are considered.

22 Creep does not generate negative effects on the residual post-creep behavior.

23 Crack Opening Rate depends on both pre-cracking damage and applied stress level.

24 An expression to predict Crack Opening Rate values is proposed.

25

26 **LIST OF NOTATION:**

27 (Note: the order is not alphabetical to be more consistent with the test process and analysis)

|    |                                 |   |
|----|---------------------------------|---|
| 28 | CMOD                            | Crack Mouth Opening Displacement  |
| 29 | CMOD <sub>p</sub>               | maximum CMOD reached in the pre-cracking process  |
| 30 | CMOD <sub>pr</sub>              | residual CMOD after pre-cracking process  |
| 31 | CMOD <sub>ci</sub>              | instantaneous CMOD at loading in the creep period   |
| 32 | CMOD <sub>cd</sub> <sup>j</sup> | deferred CMOD after j days under sustained load in the creep period                                   |
| 33 | CMOD <sub>ct</sub> <sup>j</sup> | total CMOD after j days in the creep period (CMOD <sub>ci</sub> + CMOD <sub>cd</sub> <sup>j</sup> )   |
| 34 | CMOD <sub>cri</sub>             | instantaneous CMOD recovery after unloading   |
| 35 | CMOD <sub>crd</sub>             | deferred CMOD recovery after unloading  |
| 36 | CMOD <sub>o</sub> <sup>j</sup>  | absolut CMOD after j days, referred to origin of deformations   |
| 37 | COR <sup>j-k</sup>              | Crack Opening Rate between j and k days   |
| 38 | COR <sub>th</sub>               | theoretical Crack Opening Rate  |
| 39 | CTOD                            | Crack Tip Opening Displacement  |
| 40 | $f_L$                           | limit of proportionality  |
| 41 | $f_{R,1}$                       | residual flexural tensile strength corresponding to CMOD = 0.5 mm                                     |
| 42 | $f_{R,3}$                       | residual flexural tensile strength corresponding to CMOD = 2.5 mm                                     |
| 43 | $f_{R,p}$                       | residual flexural tensile strength at CMOD <sub>p</sub>   |
| 44 | $f_{R,c}$                       | applied stress during a creep period, in MPa  |
| 45 | $I_n$                           | nominal creep index, or nominal stress level as percentage of $f_{R,1}$                               |
| 46 | $I_c$                           | applied creep index, or applied stress level as percentage of $f_{R,1}$ ( $I_c = f_{R,c} / f_{R,1}$ ) |
| 47 | $I_{ce}$                        | effective creep index, or creep index normalized by fiber density                                     |
| 48 | $\delta_F$                      | fiber density at the fracture surface   |
| 49 | $k_p$                           | parameter that depends on the time period considered  |

50

51 **ABSTRACT**

52 This paper presents an experimental study on creep behavior of cracked Fiber Reinforced  
53 Concrete (FRC). To this end, a conventional strength concrete incorporating hooked-end steel  
54 fibers was used. In order to represent service conditions, specimens under low sustained loads  
55 with small pre-cracking damage (lower than 0.5 mm) were tested and analyzed. Taking as a  
56 reference the residual stress  $f_{R,1}$  determined according to EN 14651:2007, different stress  
57 levels for the creep tests were applied: 25%, 35% and 45% of  $f_{R,1}$ . For the pre-cracking  
58 damage, several initial crack openings were considered: 0.05, 0.1, 0.2, 0.3 and 0.5 mm. The  
59 specimens were tested under a four-point bending scheme for 180 days in controlled  
60 environmental conditions. The Crack Opening Rate (COR) was used as parameter to  
61 characterize the creep behavior of cracked FRC. After the creep tests, conventional flexural  
62 tests were carried out to evaluate the residual capacity and to survey the fiber density of each  
63 specimen. Both the stress level and the initial crack opening factors affected significantly the  
64 creep behavior. In all cases, the measured increases in crack openings resulted compatible  
65 with service conditions, and no negative effects on the residual post-creep behavior were  
66 observed. An expression to predict Crack Opening Rate values is proposed.

67

68 **KEYWORDS:** creep; fiber-reinforced concrete; steel fiber; bending; crack opening.

69

70

71 **1. INTRODUCTION**

72

73 Fiber Reinforced Concrete (FRC) is an enhanced composite material obtained by adding  
74 fibers into concrete. Different types of fibers can be used with the aim of improving the post-  
75 cracking concrete performance for a variety of structural applications. A main contribution

76 played by fibers consists in controlling cracks propagation, which lead to improvements in  
77 tensile and impact toughness, ductility and durability [1]. This contribution is activated from  
78 fiber pull-out after concrete cracking due to the fiber bridging mechanisms across the concrete  
79 crack surfaces [2,3]. Once the action of the fibers has begun, the stabilization of cracks to  
80 prevent excessive crack openings through time constitutes a crucial issue regarding the  
81 service life of FRC structures [2,4]. In fact, the crack width of FRC beams seems to stabilize  
82 after 10 months under sustained load [5], which is not well predicted from codes [6].  
83 Therefore, the post-cracking behavior due to the time-dependent fiber pull-out may be one of  
84 the significant sources of time-dependent deformation under sustained loads (creep in the  
85 following), along with the development of new cracks [7], which will constitute a complex  
86 phenomenon as properties of the concrete matrix change due to aging and the mechanical  
87 behavior is dependent on the duration of loading [8].

88

89 With the increased knowledge and progress in new design perspectives [3,9,10], the improved  
90 properties of FRC over conventional reinforced concrete with standard rebars are allowing an  
91 increased use. However, a proper knowledge on the long-term behavior of FRC has not yet  
92 been achieved as research and literature on this topic are still scarce [1,11,12].

93

94 In 2000 both ASTM and RILEM organizations offered test methods to determine the residual-  
95 behavior of FRC: the first version of ASTM C1399 [13], the early recommendations proposed  
96 by RILEM TC162-TDF [14] and the actual European Standard 14651:2005+A1:2007 [15],  
97 which has been adopted by several countries. This allowed a better characterization of the  
98 ability of the fibers to provide crack bridging forces, that is, the post-cracking behavior of  
99 FRC. However, these standards only allow gaining a better understanding of the instantaneous  
100 behavior of FRC. Regarding the long-term behavior of FRC, different phenomena may

101 contribute in the cracked state: creep of the concrete matrix, creep of the fibers material and  
102 time-dependent fiber slip by a pull-out process at the fiber/matrix interface; the extent to  
103 which each of these phenomena contributes to the global behavior (and to creep failure) is not  
104 yet clear [12].

105

106 In the absence of a standard test, the experimental methodology commonly applied in beams  
107 by researchers considers some aspects included in EN 14651 [15], mainly those focused on  
108 the specimen characteristics. Also, variations in load scheme (three-point bending tests  
109 (3PBT) or four-point bending tests (4PBT)), pre-cracking process, and load level and time  
110 period, among others factors, has been carried out. Concerning experimental research on  
111 flexural creep behavior, the literature found in international journals reports important  
112 contributions from research on steel FRC (SFRC) [2,4,16-20].

113

114 Some studies were carried out introducing large initial crack widths (0.5 mm to 2 mm in  
115 beams [2,16,18-20], and 3.5 mm in round panels [4]), and also high sustained load levels were  
116 applied in relation to cracking load (more than 90% in some cases [2], commonly about 50-  
117 70% [16,19], and lower than 50% in few cases [20]).

118

119 Stable responses without increased crack width were obtained in [2] for the following cases:  
120 (a) for a small Crack Mouth Opening Displacement (CMOD) of 0.2 mm regardless the stress  
121 level; and (b) for a CMOD of 0.5 mm when stress level ratio was under the 80% of the  
122 residual stress reached during the initial cracking process. By using multiple linear regression  
123 analysis to relate measured creep parameters to fiber geometry and content, concrete strength  
124 and composition and high sustained load level –ranging from 50% to 90% of the residual  
125 flexural tensile strength corresponding to a pre-crack of 0.5 mm CMOD–, it was found that

126 fiber content and fiber slenderness together with load ratio were the main factors influencing  
127 flexural creep behavior [16]. In addition to higher creep coefficients for greater initial crack  
128 openings, creep performance was found influenced by the fiber distribution [17]. From  
129 comparative studies, flexural creep behavior was found influenced by both the fiber type and  
130 the load level [4], in this case by using round panels made of shotcrete which were pre-  
131 cracked up to 3.5 mm. It was stated that the fiber type and environmental conditions can  
132 affect creep behavior and the residual response of SFRC [18]. Similar creep behavior was  
133 observed by using three- and four-point loading configurations [19].

134

135 With regard to the load level, it was found that 50% guarantees a durable system with a long-  
136 term load carrying capacity in the cracked state when macro-synthetic fibers are used [21].  
137 Also, it was detailed that for load levels below 50%, the measured flexural creep coefficients  
138 were similar to those of the unreinforced concrete loaded in compression [21]. The tensile  
139 creep of cracked SFRC also increases as the stress level increases [22]. Recently, in a  
140 comparative study including steel fibers and synthetic fibers which considered initial crack  
141 widths ranging from 0.2 to 2 mm and applied sustained stress level of  $0.45f_{R,1}$  for the creep  
142 test [20], it was concluded that for SFRC the Crack Opening Rate (COR) shows stabilized  
143 values after 30 days of creep tests regardless the initial CMOD. Also, it was found in [20] that  
144 COR values for synthetic macro-fibers after 180 days of creep test show more variability than  
145 those for steel fibers, and that stress level seems to be more significant than initial crack  
146 opening for COR values. For this reason, it was recommended in [20] to study the influence  
147 of initial CMOD in creep test with different stress levels. At this point, it is important to  
148 highlight that velocity stabilization means small continuous deferred deformations but not  
149 deferred deformations stop.

150

151 Therefore, this paper focuses on the study of the long-term behavior of cracked steel FRC  
152 specimens in suitable service conditions for structural elements. The significance of the initial  
153 crack width and the load levels as the main variables affecting the long-term performance are  
154 analyzed with the aim of defining load levels safe enough to keep service conditions along the  
155 life of the structure.

156

157

## 158 **2. EXPERIMENTAL PROGRAM**

159

### 160 **2.1. Materials**

161

162 Two different batches (“a” and “b”) with the same FRC matrix were cast. The concrete mix  
163 design shown in Table 1 was adopted since it is a usual concrete in FRC structures. Hooked-  
164 end steel fibers 50-mm long and 1-mm diameter were used with a dosage of 50 kg/m<sup>3</sup>. It was  
165 expected to obtain a residual performance classified “c” by the *fib* Model Code [23], what  
166 means a ratio  $f_{R,3}/f_{R,1}$  between 0.9 and 1.1.

167

168

**Table 1.** Mix design of the FRC.

| Component                                   | Content (kg/m <sup>3</sup> ) |
|---|------------------------------|
| Portland Cement (type CPF40)                | 410                          |
| Water                                       | 170                          |
| Natural siliceous sand                      | 935                          |
| Granitic crushed stone (12 mm maximum size) | 825                          |
| Superplasticizer                            | 3.4                          |
| Steel fibers                                | 50                           |

169

170



## 171 **2.2. Specimens**

172

173 The experimental program was developed to study the influence of the damage level (initial  
174 crack opening) and the stress level on the flexural creep behavior of cracked FRC. A total of  
175 18 prismatic specimens of 150x150x600 mm were cast. In addition, three Ø150x300 mm  
176 cylindrical specimens were cast for each batch in order to perform characterization tests in  
177 compression. All specimens were demolded and cured for 28 days in moist chamber ( $20 \pm 3$   
178 °C, relative humidity 95%). The average concrete compressive strength at 28 days was 41.7  
179 and 39.2 MPa for batches “a” and “b”, respectively.

180

181 All prismatic specimens were prepared as proposed in EN 14651 [15], with a 25 mm depth  
182 notch sawed at midspan. At 28 days, two specimens (one from each batch) were tested for  
183 characterization according to EN 14651 [15] until a CMOD of 3.5 mm, and the others were  
184 used for creep tests. The CMOD was controlled by means of a clip gauge placed at both sides  
185 of the notch. The stress at the Limit Of Proportionality (LOP,  $f_L$ ) and the residual strengths at  
186 CMOD of 0.5 and 2.5 mm ( $f_{R,1}$  and  $f_{R,3}$ ) were obtained.

187

## 188 **2.3. Testing details**

189 Based on an experimental test method to characterize flexural creep behavior of pre-cracked  
190 FRC specimens [9], which has been used in recent studies [24, 25], the complete flexural  
191 creep test consisted of three main stages (Fig. 1): "pre-cracking", "creep period" and "rupture  
192 after creep (or post-creep)". In the first stage, specimens were pre-cracked using a 3PBT  
193 configuration as proposed in EN 14651 [15] up to the desired  $CMOD_p$  (point B in Fig. 1, after  
194 point A -limit of proportionality- is reached) and then unloaded (point C). In the second stage,  
195 the cracked specimens are located into the creep frames and loaded (from point D to E/E'/E'')

196 in a 4PBT configuration. The load, that was kept constant over time (creep period, from point  
197 E to F, or E' to F'; or E'' to F''), was selected in order to reach a stress in the cracked section  
198 as close as possible to the nominal creep index  $I_n$ , which was established as a percentage of  
199 the residual flexural strength at a CMOD of 0.5 mm ( $f_{R,1}$ ). During the creep period, the crack  
200 opening evolution was registered. Once the creep period was finished, the specimens were  
201 unloaded (from points F/F'/F'' to G), and afterwards were tested again in bending until failure  
202 (points H, I and J). Applied stresses were computed from forces considering a linear  
203 distribution along the cross section of specimens according to EN 14651:2007 [15] and taking  
204 into account the actual loading configuration (4PBT in the creep period and 3PBT in pre-  
205 cracking and post-creep periods).

206

207 **Fig. 1.** Main stages of the cracked FRC creep test methodology.

208

209 The creep behavior was analyzed in specimens having different initial damage level  
210 (CMOD<sub>p</sub>), all of them with suitable values compatible with service conditions. The selected  
211 CMOD<sub>p</sub> were 0.05, 0.1, 0.2, 0.3 and 0.5 mm. Regarding the sustained load, low stress levels  
212 were applied to specimens: 25%, 35% and 45% of  $f_{R,1}$ .

213

214 As the pre-cracking stage ended in most specimens before a 0.5 mm crack width, it was not  
215 possible to refer the applied stress to the specific  $f_{R,1}$  of each specimen. Therefore, it was  
216 adopted as a general reference the average of  $f_{R,1}$  values obtained from the three specimens  
217 pre-cracked up to 0.5 mm and the two specimens that were tested up to 3.5 mm used for FRC  
218 characterization.

219

220 The creep specimens were identified based on the nominal  $CMOD_p$  achieved in the pre-  
221 cracking stage (0.05, 0.1, 0.2, 0.3 and 0.5 mm) followed by the letter of the FRC batch (“a” or  
222 “b”) and the stress level (25, 35 or 45, as % of  $f_{R,1}$ ) applied during the creep period. By way of  
223 example, 0.3b-35 corresponds to a specimen from concrete bath “b”, with a pre-crack 0.3 mm  
224 width, loaded at  $0.35 f_{R,1}$  during the creep period.

225

226 After curing in the moist chamber for 28 days, the prismatic specimens remained in the  
227 laboratory environment (20 °C, relative humidity 70%) during an additional time span of one  
228 month. Pre-cracking tests in bending were performed during the first two weeks after curing.  
229 Once the desired pre-crack level was reached, the stress corresponding to the  $CMOD_p$  was  
230 recorded and was identified as  $f_{R,p}$ , and then the specimens were unloaded and the residual  
231 pre-crack value ( $CMOD_{pr}$ ) was recorded. After unloading, the cracked specimens were turned  
232 90° and placed on plywood boards in order to avoid any influence on the crack opening  
233 during transport or waiting times. Then, with the aim of reducing thermal and drying  
234 shrinkage effects during creep measurements, the specimens were stored in the creep test  
235 room where the temperature was kept constant at  $22 \pm 3$  °C.

236

237 After one week, the specimens were placed in the frames and the sustained creep loads were  
238 applied. The applied stress during the creep period was identified as  $f_{R,c}$ . Using a 4PBT  
239 configuration, a multiple specimen’s setup in column was adopted for the creep period. Fig. 2  
240 shows the load configuration inside the creep frame as well as details of the dial gauges used  
241 for the measurements of crack evolution. The specimens with highest residual strength were  
242 placed at the bottom of the column in order to minimize differences in stress levels due to  
243 weight of the upper specimens, in this way, the applied stresses were similar.

244

245 The CMOD values for each specimen were registered during 180 days by means of a dial  
246 gauge placed at one side of the specimen. Additionally, with the aim of improving the  
247 accuracy of instantaneous deformations during the loading as well as to record the creep  
248 variations during the first days, one Linear Variable Differential Transformer (LVDT) was  
249 located at one side of each specimen measuring the Crack Tip Opening Displacement  
250 (CTOD) as seen in Fig. 2. Since the notch depth was 25 mm, the CTOD were corrected by  
251 applying a factor of 1.2 to compare with those of the CMOD. This correction factor was  
252 obtained as the ratio of the distance of both transducers referred to the top of the specimen  
253 ( $150/125 = 1.2$ ).

254

255 **Fig. 2.** Applied load and measurement devices during of creep period: (a) load configuration  
256 scheme; (b) specimen's setup in column; (c) transducer location at rear side; and (d)  
257 transducer location at front side.

258

259 Based on the creep test results, the Crack Opening Rate (COR) was calculated to explore the  
260 conditions for crack stability. This parameter evaluates the deferred crack opening rate in a  
261 certain period of time by means of equation (1):

262

$$263 \text{COR}^{j-k} = ( \text{CMOD}_{\text{ct}}^k - \text{CMOD}_{\text{ct}}^j ) / ( k - j ) \quad (1)$$

264

265 where  $j$  and  $k$  are the age in days after loading (within the creep period), and  $\text{CMOD}_{\text{ct}}^j$  and  
266  $\text{CMOD}_{\text{ct}}^k$  are the total crack openings at  $j$  and  $k$  times in the creep period, respectively.

267

268 After 180 days under sustained loads (creep period), the specimens were unloaded and kept in  
269 the frames while the CMOD recovery was measured during 30 days. Afterward, the

270 specimens were removed from the creep frames, and then tested until failure using a 3PBT  
 271 configuration following the general guidelines of EN 14651 standard. Finally, the number of  
 272 fibers at the fracture surface was counted and the density of fibers ( $\delta_F$ ) was calculated.

273

274

### 275 3. TEST RESULTS

276

#### 277 3.1. Characterization and pre-cracking tests results

278

279 Fig. 3 shows the curves of the flexural tests for each specimen in pre-cracking stage (cases (a)  
 280 to (e)) as well as the complete tests performed on two prismatic specimens in accordance to  
 281 EN 14651 (case (f)). Table 2 summarizes the obtained results for both characterization and  
 282 pre-cracking tests. In case of the specimens where the pre-cracking stage ended before 0.5  
 283 mm for CMOD, the stress corresponding to  $CMOD_p$  ( $f_{R,p}$ ) is detailed instead of  $f_{R,1}$ .

284

285 **Fig. 3.** Three point bending test curves up to different CMOD: (a) 0.05 mm, (b) 0.1 mm,  
 286 (c) 0.2 mm, (d) 0.3 mm, (e) 0.5 mm, (f) 3.5 mm.

287

288 **Table 2.** Characterization and pre-cracking flexural test results.

| Specimen | $f_L$<br>(MPa) | $f_{R,p}$<br>(MPa) |      |     |     | $f_{R,1}$ | $f_{R,3}$ | $CMOD_p$<br>( $\mu m$ ) | $CMOD_{pr}$<br>( $\mu m$ ) |
|----------|----------------|--------------------|------|-----|-----|-----------|-----------|-------------------------|----------------------------|
|          |                | CMOD (mm)          |      |     |     |           |           |                         |                            |
|          |                | 0.05               | 0.1  | 0.2 | 0.3 | 0.5       | 2.5       |                         |                            |
| 0.05a-25 | 4.24           | 4.78               | -    | -   | -   | -         | -         | 52                      | 28                         |
| 0.05b-35 | 3.90           | 4.79               | -    | -   | -   | -         | -         | 51                      | 15                         |
| 0.05a-45 | 3.58           | 5.31               | -    | -   | -   | -         | -         | 56                      | 21                         |
| 0.1a-25  | 4.00           | 4.70               | 4.06 | -   | -   | -         | -         | 106                     | 61                         |
| 0.1b-35  | 3.54           | 4.80               | 5.15 | -   | -   | -         | -         | 102                     | 40                         |
| 0.1b-45  | 4.66           | 4.28               | 4.37 | -   | -   | -         | -         | 100                     | 45                         |

|  |      |      |      |      |      |      |      |     |     |
|--|------|------|------|------|------|------|------|-----|-----|
| 0.2b-25                                    | 3.96 | 4.94 | 4.73 | -    | -    | -    | -    | 137 | *   |
| 0.2b-35                                    | 3.88 | 5.19 | 5.27 | 5.02 | -    | -    | -    | 205 | 113 |
| 0.2a-45                                    | 3.72 | 5.09 | 4.54 | 4.39 | -    | -    | -    | 205 | 114 |
| 0.2b-45                                    | 4.16 | 5.32 | 5.20 | 4.97 | -    | -    | -    | 195 | 112 |
| 0.3a-25                                    | 4.20 | 5.19 | 4.95 | 4.92 | 4.94 | -    | -    | 300 | 191 |
| 0.3b-35                                    | 3.25 | 4.73 | 4.63 | 4.38 | 4.25 | -    | -    | 306 | 180 |
| 0.3b-45                                    | 4.27 | 5.23 | 5.24 | 5.46 | 5.75 | -    | -    | 308 | 195 |
| 0.5a-25                                    | 3.99 | 5.38 | 5.58 | 5.62 | 5.99 | 6.44 | -    | 501 | 347 |
| 0.5a-35                                    | 4.41 | 5.08 | 4.93 | 4.59 | 4.51 | 4.42 | -    | 512 | 340 |
| 0.5b-45                                    | 5.35 | 5.68 | 5.68 | 5.55 | 5.58 | 5.85 | -    | 503 | 341 |
| 3.5a                                       | 3.74 | 4.14 | 4.47 | 4.27 | 4.34 | 4.56 | 4.53 |     |     |
| 3.5b                                       | 3.60 | 4.79 | 5.23 | 5.34 | 5.35 | 5.70 | 5.86 |     |     |
| Average                                    | 4.03 | 4.97 | 4.94 | 4.96 | 5.09 | 5.39 | 5.20 |     |     |
| CoV (%)                                    | 12   | 8    | 9    | 10   | 13   | 16   | 18   |     |     |
| *: not measured, test unexpectedly stopped |      |      |      |      |      |      |      |     |     |

289

290 As shown in Table 2, the average  $f_{R,1}$  obtained from the two specimens used for  
291 characterization test and the three specimens pre-cracked up to 0.5 mm was 5.39 MPa. This  
292 average residual strength was the reference stress value that served to obtain the sustained  
293 load to be applied for each specimen in the creep period stage.

294

### 295 3.2. Distribution of specimens in the frames

296

297 The specimens (with different  $CMOD_p$ ) were located in six frames according to the nominal  
298 stress levels: 5 specimens at 0.25 of reference  $f_{R,1}$  (1.35 MPa), 5 specimens at 0.35 of  
299 reference  $f_{R,1}$  (1.89 MPa) and 6 specimens at 0.45 of reference  $f_{R,1}$  (2.43 MPa). Table 3 shows  
300 the distribution and the relative position for each specimen inside the corresponding creep  
301 frame: top, middle or bottom. The indicated load corresponded to that applied on the top  
302 specimen of the column; in the case of middle and bottom specimens, the weight of the upper  
303 specimens was considered to obtain the applied stress (presented in Table 4). As a result, the  
304 applied stresses ( $f_{R,c}$ ) for each stress level group ranged from 1.30 to 1.38 MPa (24.1 to

305 25.6%), from 1.82 to 1.95 MPa (33.7 to 36.1%), and from 2.27 to 2.54 MPa (42.0 to 47.0%)  
 306 respectively, and were not exactly the desired stress level values in all cases.

307

308 **Table 3.** Location and relative position for each specimen in the creep frames.

| Frame N°                | 1        | 2        | 3        | 4        | 5        | 6        |
|-------------------------|----------|----------|----------|----------|----------|----------|
| Position Top            | 0.05a-25 | 0.30b-35 | 0.20b-25 | 0.05b-35 | 0.05a-45 | 0.30b-45 |
| Middle                  | --       | --       | 0.30a-25 | 0.50a-35 | 0.10b-45 | 0.50b-45 |
| Bottom                  | 0.10a-25 | 0.20b-35 | 0.50a-25 | 0.10b-35 | 0.20b-45 | 0.20a-45 |
| $I_n$ (% of $f_{R,1}$ ) | 25       | 35       | 25       | 35       | 45       | 45       |
| 4PBT load (kN)          | 6.3      | 8.9      | 6.3      | 8.9      | 11.3     | 10.6     |

309

310 The creep period stage was developed in a room with controlled temperature ( $22 \pm 3$  °C). Fig.  
 311 4 shows the logs of temperature and relative humidity inside the room during the 6 months  
 312 under sustained load and one additional month where the  $CMOD_{crd}$  measurements were  
 313 continued while the specimens remained unloaded. The experiences started in winter and, as  
 314 it can be seen, during the first months the relative humidity was near 50%, whereas at later  
 315 ages the humidity increased up to 80%.

316

317 **Fig. 4.** Environmental conditions during the development of the creep period.

318

319

### 320 **3.3. Measurement corrections due to shrinkage**

321

322 As the  $CMOD$  was determined by dial gauges (see Fig. 2), the measures were referred to two  
 323 points located on both sides of the crack with a base length close to 150 mm; in certain cases,  
 324 as multiple specimens arrangement was used, it was necessary to increase the base length in  
 325 order to place the dial gauges. Due to the low humidity in the creep room, it was expected that  
 326 concrete undergo drying shrinkage. Therefore, the dial measurements included both the creep

327 crack opening increase and the length reduction due to concrete shrinkage. It must be noted  
 328 that many studies in the literature were performed using high stress levels and wider initial  
 329 cracks, and the former effect is preeminent. However, the second effect is not negligible in  
 330 case of specimens working at a low stress level and with a small initial crack opening, as  
 331 performed in this study.

332

333 To evaluate the shrinkage effect, additional series of unloaded prismatic specimens with the  
 334 same concrete mix design were cast, instrumented and conserved in the same conditions than  
 335 the loaded specimens. The drying shrinkage measurements for periods between 30 to 60 days,  
 336 60 to 90 days and 90 to 180 days were 47, 32 and 60 microstrains, respectively.

337

### 338 3.4. CMOD evolution in time

339

340 Table 4 summarizes the main results for the CMOD evolution under loading. The  
 341 instantaneous ( $CMOD_{ci}$ ) and total measurements at different ages ( $CMOD_{ct}^j$ ) up to 180 days  
 342 are given. In addition, the instantaneous CMOD recovered after unloading ( $CMOD_{cri}$ ) is  
 343 presented. It must be noted that in many cases, mainly for very thin pre-cracks and for low  
 344 sustained loads, a CMOD reduction was found at later ages, which can be explained  
 345 considering the dial base length reductions due to concrete shrinkage as indicated in section  
 346 3.3. When comparing  $CMOD_{ci}$  (at loading) and  $CMOD_{cri}$  (at unloading), a good agreement  
 347 can be observed; thus, it can be inferred that the application of long-term loads did not  
 348 produce a major damage in the FRC internal structure.

349

350 **Table 4.** CMOD ( $\mu m$ ) evolution through time (creep period).

| Specimen | $f_{R,c}$<br>(MPa) | Under loading |                  |                  |                  |                  | Unloading         |              |
|----------|--------------------|---------------|------------------|------------------|------------------|------------------|-------------------|--------------|
|          |                    | $CMOD_{ci}$   | $CMOD_{ct}^{14}$ | $CMOD_{ct}^{30}$ | $CMOD_{ct}^{60}$ | $CMOD_{ct}^{90}$ | $CMOD_{ct}^{180}$ | $CMOD_{cri}$ |
|          |                    |               |                  |                  |                  |                  |                   |              |



|          |      |    |     |     |     |     |     |    |
|----------|------|----|-----|-----|-----|-----|-----|----|
| 0.05a-25 | 1.30 | 21 | 32  | 34  | 36  | 36  | 27  | 19 |
| 0.05b-35 | 1.82 | 17 | 36  | 37  | 37  | 35  | 27  | 19 |
| 0.05a-45 | 2.27 | 25 | 32  | 36  | 37  | 35  | 27  | 21 |
| 0.1a-25  | 1.35 | 31 | 46  | 46  | 49  | 49  | 46  | 26 |
| 0.1b-35  | 1.89 | 22 | 32  | 39  | 38  | 36  | 32  | 19 |
| 0.1b-45  | 2.34 | 25 | 52  | 57  | 65  | 69  | 69  | 29 |
| 0.2b-25  | 1.31 | 35 | 50  | 58  | 59  | 64  | 60  | 35 |
| 0.2b-35  | 1.88 | 30 | 54  | 54  | 56  | 57  | 57  | 26 |
| 0.2a-45  | 2.34 | 49 | 72  | 82  | 87  | 92  | 91  | 45 |
| 0.2b-45  | 2.54 | 65 | 100 | 110 | 120 | 130 | 132 | 64 |
| 0.3a-25  | 1.35 | 32 | 43  | 49  | 50  | 52  | 50  | 29 |
| 0.3b-35  | 1.82 | 98 | 121 | 135 | 143 | 153 | 161 | 74 |
| 0.3b-45  | 2.34 | 51 | 70  | 76  | 80  | 80  | 78  | 48 |
| 0.5a-25  | 1.38 | 32 | 75  | 78  | 75  | 74  | 69  | 25 |
| 0.5a-35  | 1.95 | 90 | 123 | 135 | 148 | 149 | 154 | 85 |
| 0.5b-45  | 2.46 | 71 | 97  | 107 | 118 | 125 | 125 | 70 |

351

352 Fig. 5 plots the CMOD deformation curves for all the specimens arranged by the initial  
353 damage level from 0.05 to 0.5 mm, where the correction by the effect of the drying shrinkage  
354 explained in section 3.3 is already updated. Each graph includes the specimens loaded at the  
355 same nominal stress level (25, 35 or 45% of  $f_{R,1}$ ). The instantaneous  $CMOD_{ci}$  due to loading  
356 stage, deferred  $CMOD_{cd}^j$  during 180 days, together with the instantaneous recovery  $CMOD_{cri}$   
357 and the deferred recovery  $CMOD_{crd}$  during 30 days are shown in Fig. 5. It can be observed  
358 that for the specimens with lowest initial crack opening (0.05 mm) there is not a big influence  
359 of the stress levels on the deferred behavior. When increasing the initial damage level, the  
360 specimens develop more creep but with similar trend curves: the wider the initial crack, the  
361 higher the effect of the stress level on creep behavior (both in instantaneous and deferred  
362 CMOD). The scatter in  $CMOD_{ci}$  values during loading is not negligible and may be caused by  
363 slight differences in initial stiffness and/or in the number of fibers crossing the cracked  
364 section.

365

366

**Fig. 5.** CMOD evolution during the creep period.

367

### 368 3.5 Post-creep bending test

369

370 Fig. 6 shows the complete stress-CMOD curves including the three stages of a complete  
371 flexural creep test: pre-cracking, creep period and post-creep bending test. Following similar  
372 criteria than in Fig. 5, each graph includes the specimens (having different  $CMOD_p$  values)  
373 submitted to the same sustained load (25, 35 and 45% of  $f_{R,1}$ ) and additional plots are included  
374 showing the enlarged area corresponding to the first 600 microns. As a part of the post-creep  
375 stage, two loading–unloading cycles up to the same stress applied during the creep period  
376 were done. Each plot also includes the stress-CMOD curves of reference specimens 3.5a and  
377 3.5b (aimed for FRC characterization) for comparison. It can be seen that there was not found  
378 significant loss in residual capacity by sustained loads. Moreover, it is possible that some  
379 improvements in residual capacity can be enhanced by the matrix strength evolution and the  
380 subsequent effects on fiber bond strength.

381

382 **Fig. 6.** Stress-CMOD curves including pre-cracking, creep period and post-creep stages at  
383 different stress levels  $I_n$ : (a) 25%, (b) 35% and (c) 45%.

384

385 Once the post-creep bending stage tests were performed, the number of fibers was surveyed on  
386 the fracture surface. Table 5 arranges the residual stress  $f_{R,3}$  as well as the fiber density ( $\delta_F$ ).  
387 As expected, a clear relationship appears when comparing the individual results of residual  
388 strength  $f_{R,3}$  and fiber density. Fig. 7 shows this relationship distinguishing the specimens  
389 exposed to sustained loads and the two specimens used for FRC characterization. Even for a  
390 same fiber density, the specimens that remained under sustained loads show higher residual  
391 capacity.

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**Fig. 7.** Residual stress  $f_{R,3}$  vs. fiber density  $\delta_F$  at the fracture surface.

As already exposed, the load applied in the creep period was determined based on the nominal creep index  $I_n$  referred to the average  $f_{R,1}$ , which was obtained from the characterization tests and the specimens pre-cracked up to 0.5 mm. Due to the relative position in the frames, the load varies for each specimen depending on their position. Thus, the actual creep index  $I_c$  for each specimen is defined as the ratio  $f_{R,c} / f_{R,1}$ . In addition, considering that there is a strong reliance between residual FRC strength and fiber content, the fiber density parameter was used to normalize the applied stress level ( $I_c$ ) during the creep period of each specimen, into an effective stress level (or effective creep index,  $I_{ce}$ ) obtained by multiplying the actual creep index  $I_c$  by the ratio between the average fiber density for all specimens (0.50) and the individual fiber density in each specimen. Table 5 details the  $f_{R,c}$  of each specimen, in absolute and relative to  $f_{R,1}$  terms, and the resulting effective values for the applied load stress level during the creep period.

**Table 5.** Post-creep bending test results and effective stress level obtained.

| Specimen | Post-creep test results |   | Creep period stress level |              |              |                 |
|----------|-------------------------|---|---------------------------|--------------|--------------|-----------------|
|          | $f_{R,3}$<br>(MPa)      | $\delta_F$<br>(fibers/cm <sup>2</sup> ) | $f_{R,c}$<br>(MPa)        | $I_n$<br>(%) | $I_c$<br>(%) | $I_{ce}$<br>(%) |
| 0.05a-25 | 5.11                    | 0.45                                    | 1.30                      | 25           | 24.1         | 26.8            |
| 0.05b-35 | 6.05                    | 0.45                                    | 1.82                      | 35           | 33.7         | 37.5            |
| 0.05a-45 | 8.37                    | 0.60                                    | 2.27                      | 45           | 42.1         | 35.1            |
| 0.1a-25  | 4.98                    | 0.47                                    | 1.35                      | 25           | 25.0         | 26.7            |
| 0.1b-35  | 7.10                    | 0.50                                    | 1.89                      | 35           | 35.0         | 35.1            |
| 0.1b-45  | 4.89                    | 0.38                                    | 2.34                      | 45           | 43.4         | 57.1            |
| 0.2b-25  | 5.53                    | 0.45                                    | 1.31                      | 25           | 24.3         | 27.0            |
| 0.2b-35  | 7.59                    | 0.49                                    | 1.88                      | 35           | 34.9         | 35.6            |
| 0.2a-45  | 6.62                    | 0.55                                    | 2.34                      | 45           | 43.4         | 39.5            |
| 0.2b-45  | 6.11                    | 0.41                                    | 2.54                      | 45           | 47.1         | 57.5            |
| 0.3a-25  | 5.52                    | 0.42                                    | 1.35                      | 25           | 25.0         | 29.8            |
| 0.3b-35  | 4.46                    | 0.42                                    | 1.82                      | 35           | 33.7         | 40.2            |
| 0.3b-45  | 8.46                    | 0.59                                    | 2.34                      | 45           | 43.4         | 36.8            |

|         |      |      |      |    |      |      |
|---------|------|------|------|----|------|------|
| 0.5a-25 | 8.22 | 0.66 | 1.38 | 25 | 25.6 | 19.4 |
| 0.5a-35 | 5.44 | 0.52 | 1.95 | 35 | 36.2 | 34.8 |
| 0.5b-45 | 8.05 | 0.66 | 2.46 | 45 | 45.6 | 34.6 |

409

410

#### 411 4. DISCUSSION

412

413 Table 6 presents the Crack Opening Rate (COR) calculated for different time periods: 30 - 60  
414 days ( $COR^{30-60}$ ), 60 - 90 days ( $COR^{60-90}$ ) and 90 - 180 days ( $COR^{90-180}$ ). The effective stress  
415 level of each specimen, as defined in the previous section, is also included as reference. Fig. 8  
416 shows the individual and average values of COR for the different periods for each nominal  
417 stress level considered. As expected, the COR has a clear tendency to be reduced through the  
418 time. The scatter of the COR values for the first months is attributed to differences in both  
419 stress levels and initial crack openings, as concluded in previous researches [2,20]. The stress  
420 level and initial damage level have a significant influence on the early age creep behavior.  
421 After the first months under sustained load, the scatter is reduced and there is an important  
422 decrease in COR. However, clear velocity stabilization is not usually observed until 90 or 150  
423 days, as observed in Fig. 8 from the average COR values.

424

425

**Table 6.** Crack Opening Rate (COR).

| Specimen | $I_{ce}$ (%) | COR ( $\mu\text{m}/\text{year}$ ) |            |             |
|----------|--------------|-----------------------------------|------------|-------------|
|          |              | 30-60 days                        | 60-90 days | 90-180 days |
| 0.05a-25 | 26.8         | 152                               | 76         | 17          |
| 0.05b-35 | 37.5         | 121                               | 52         | 19          |
| 0.05a-45 | 35.1         | 133                               | 58         | 19          |
| 0.1a-25  | 26.7         | 123                               | 59         | 25          |
| 0.1b-35  | 35.1         | 74                                | 34         | 22          |
| 0.1b-45  | 57.1         | 184                               | 101        | 37          |
| 0.2b-25  | 27.0         | 139                               | 143        | 37          |
| 0.2b-35  | 35.6         | 133                               | 76         | 46          |
| 0.2a-45  | 39.5         | 147                               | 120        | 35          |

|         |      |     |     |    |
|---------|------|-----|-----|----|
| 0.2b-45 | 57.5 | 202 | 180 | 47 |
| 0.3a-25 | 29.8 | 99  | 83  | 29 |
| 0.3b-35 | 40.2 | 178 | 180 | 71 |
| 0.3b-45 | 36.8 | 170 | 83  | 41 |
| 0.5a-25 | 19.4 | 79  | 52  | 24 |
| 0.5a-35 | 34.8 | 245 | 71  | 57 |
| 0.5b-45 | 34.6 | 220 | 138 | 37 |

426

427

**Fig. 8.** Crack Opening Rate (COR) evolution through time.

428

429 This general trend can be also observed in Fig. 9, which represents the COR values based on  
 430 the effective stress level  $I_{ce}$  showing clear tendencies: for the whole of values for a same time  
 431 period, the higher the effective stresses, the higher the COR values.

432

433

**Fig. 9.** Crack Opening Rate (COR) vs. effective creep index  $I_{ce}$ .

434

435 Fig. 10 shows the COR values at different time periods vs. the total CMOD after  $j$  days  
 436 —referred to origin of deformations ( $CMOD_o^j$ )— at the beginning of each time period  $j$   
 437 ( $CMOD_{pr} + CMOD_{ci} + CMOD_{cd}^j$ ), being  $j$  the first day (30, 60 or 90) of the corresponding  
 438 time period considered. The analyzed time lapses were 30-60, 60-90 and 90-180 days,  
 439 avoiding the first ages in order to better focus the study in the tendency to creep stabilization.  
 440 As observed for all time lapses considered, the COR values tend to increase as total  $CMOD_o$   
 441 increases.

442

443

**Fig. 10.** COR vs.  $CMOD_o^j$  at the beginning of the analyzed time period.

444

445 As the study shows that the stress level, the time under loading, and the initial crack width  
 446 affected COR results, an equation for prediction purposes is proposed as a first approach:

447

448 
$$\text{COR}_{\text{th}} = I_{\text{ce}} \cdot \text{CMOD}_o^j \cdot k_p \quad (2)$$

449

450 where  $\text{COR}_{\text{th}}$  is the theoretical Crack Opening Rate (in  $\mu\text{m}/\text{year}$ ),  $I_{\text{ce}}$  is the effective creep  
451 index (applied stress in % of  $f_{R,1}$ , normalized by fiber density),  $\text{CMOD}_o^j$  is the total crack  
452 opening at the beginning of the analyzed period (in  $\mu\text{m}$ ) and  $k_p$  is a parameter that depends on  
453 the time period considered.

454

455 The  $k_p$  values, obtained from an approach based on the Least Squares between theoretical and  
456 experimental COR, were 1.65, 1.03 and 0.39 for the periods 30-60 days, 60-90 days and 90-  
457 180 days, respectively. For these periods, and for the different nominal stress levels  
458 considered, Fig. 11 depicts the curves of estimated COR for some total crack opening values.  
459 In all cases, for a same time period, COR values tend to increase as nominal stress level  
460 increases. Regarding the prediction of the measured COR values, Fig. 12 plots the comparison  
461 between estimated COR and experimental COR results, differentiating each analyzed time  
462 period. As it can be observed, the adjustment of the prediction improves for high COR values  
463 and for the last/long time period.

464

465 **Fig. 11.**  $\text{COR}_{\text{th}}$  vs.  $\text{CMOD}_o^j$ , for each analyzed time period and nominal stress level

466 considered.

467

468 **Fig. 12.** Comparison between estimated COR and experimental COR results.

469

470 The experimental results show that the previous damage of the concrete at a certain moment  
471 (regardless of the load history that caused it) with the applied load levels are two main factors  
472 influencing the creep behavior of cracked FRC. The proposed approach is quite simple as it

473 assumes a linear dependence on both the initial CMOD and the applied stress. More complex  
474 approaches could be proposed even if the variability is relatively high. For the COR at long  
475 term (90-180 days) and at high COR values, the variability is reduced. Therefore, this  
476 approach is useful to evaluate the studied phenomenon and to predict experimental results in a  
477 sufficiently acceptable manner according to the actual state of knowledge.

478

## 479 **5. CONCLUSIONS**

480

481 With the aim of analyzing acceptable conditions for structures in service, the creep behavior  
482 of cracked steel FRC specimens by applying different stress levels and initial crack openings  
483 significantly lower than those traditionally used in the literature was studied. The main  
484 conclusions from this study are as follows:

- 485 • The creep results obtained for initial cracks lower than 0.5 mm and stress levels lower  
486 than 45% of  $f_{R,1}$  indicate that the crack openings tend to stabilize and then long service life  
487 can be expected.
- 488 • The crack opening rate (COR) appears as a good tool to characterize the creep behavior of  
489 cracked steel FRC. The COR values drastically decrease after the first months under  
490 loading, though it is recommended the use of this parameter for times under loading  
491 exceeding 3 months.
- 492 • In a similar way than the significant dependency between fiber density and residual FRC  
493 strength, it has been verified that the fiber density strongly affect the creep results and  
494 contributes to increase their variability.
- 495 • It has been confirmed that for the studied stress levels, no matter the initial crack width,  
496 creep phenomenon does not generate negative effects on the residual capacity of FRC.

497 • The creep behavior is mainly affected by the stress level and also by the previous damage  
498 in the concrete. An expression to evaluate this influence is proposed.

499

500 Furthermore, under the load levels applied and for the initial openings of selected cracks, the  
501 deferred deformations in the cracked FRC were really low, and make possible to infer a great  
502 stability for years. However, as this study has been carried out on a steel FRC (Class 5c  
503 according to *fib* MC2010 [23]), it should be explored the response of FRC with other type of  
504 fibers materials or shape and also FRCs with softening post-peak behavior.

505

506

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508

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## 520 **REFERENCES**

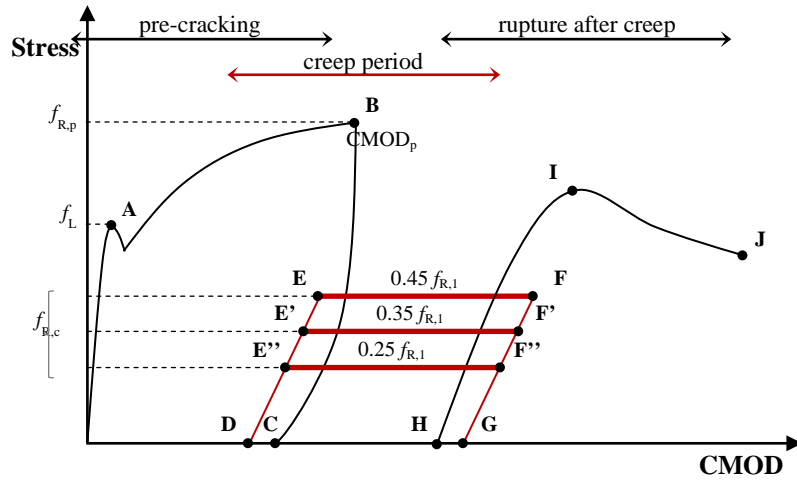
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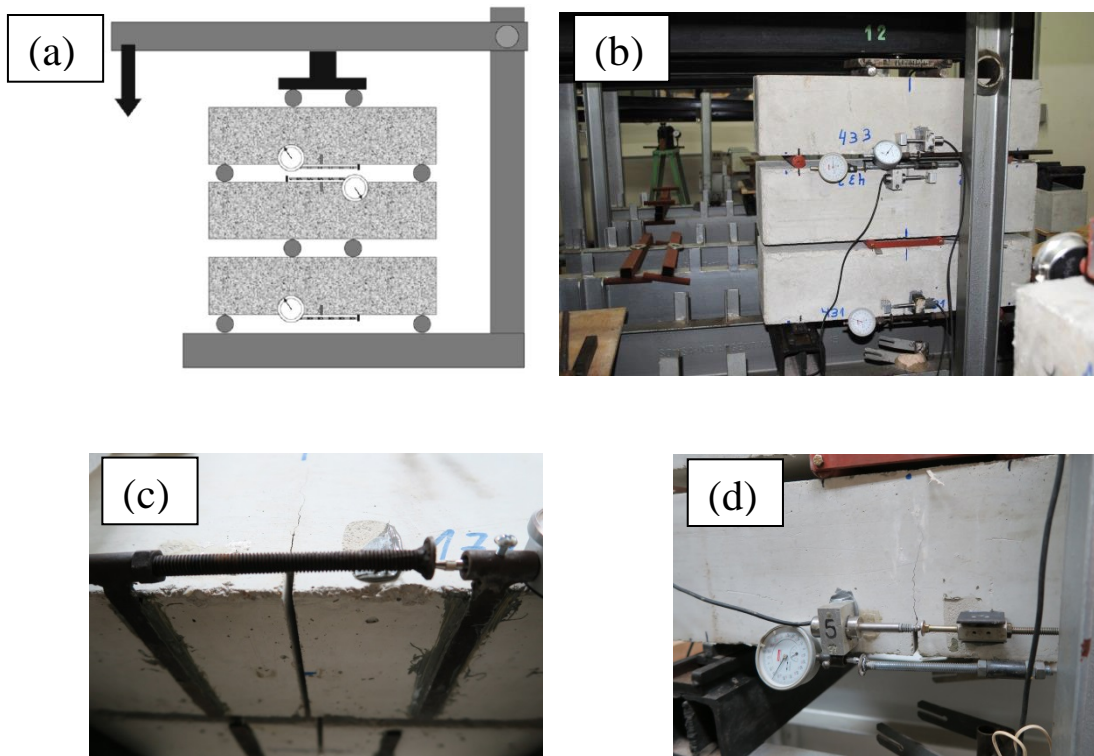
- 522 [1] G. Zhao, M. Di Prisco, L. Vandewalle, Experimental investigation on uniaxial tensile  
523 creep behavior of cracked steel fiber reinforced concrete, *Mater. Struct.* 48 (2015)  
524 3173–3185.
- 525 [2] R.L. Zerbino, B.E. Barragán, Long-term behavior of cracked steel fiber-reinforced  
526 concrete beams under sustained loading, *ACI Mat. J.* 109(2) (2012) 215–224.
- 527 [3] N. Buratti, C. Mazzotti, M. Savoia, Post-cracking behaviour of steel and macro-  
528 synthetic fibre-reinforced concretes, *Constr. Build. Mat.* 25 (2011) 2713–2722.
- 529 [4] E.S. Bernard, Influence of fiber type on creep deformation of cracked fiber-reinforced  
530 shotcrete panels, *ACI Mat. J.* 107(5) (2010) 474–480.
- 531 [5] E. Vasanelli, F. Micelli, M.A. Aiello, G. Plizzari, Long term behavior of FRC flexural  
532 beams under sustained load, *Eng. Struct.* 56 (2013) 1858–1867.
- 533 [6] E. Vasanelli, F. Micelli, M.A. Aiello, G. Plizzari, Crack width prediction of FRC beams  
534 in short and long term bending condition, *Mater. Struct.* 47 (2014) 39–54.
- 535 [7] W.P. Boshoff, V. Mechtcherine, G.P.A.G. van Zijl, Characterising the time-dependant  
536 behaviour on the single fibre level of SHCC: Part 1: Mechanism of fibre pull-out creep,  
537 *Cem. Concr. Res.* 39 (2009) 779–786.
- 538 [8] P.D. Nieuwoudt, W.P. Boshoff, Time-dependent pull-out behaviour of hooked-end steel  
539 fibres in concrete, *Cem. Concr. Comp.* 79 (2017) 133–147.
- 540 [9] M. Di Prisco, G. Plizzari, L. Vandewalle, Fiber reinforced concrete: new design  
541 perspectives, *Mater. Struct.* 42(9) (2009) 1261–1281.
- 542 [10] J.C. Walraven, High performance fiber reinforced concrete: progress in knowledge and  
543 design codes, *Mater. Struct.* 42 (2009) 1247–1260.
- 544 [11] S.E. Arango, P. Serna, J.R. Martí-Vargas, E. García-Taengua, A test method to  
545 characterize flexural creep behaviour of pre-cracked FRC Specimens, *Exp. Mech.* 52(8)  
546 (2012) 1067–1078.

- 547 [12] N. Buratti, C. Mazzotti, Experimental tests on the effect of temperature on the long-term  
548 behaviour of macrosynthetic Fibre Reinforced Concretes, *Constr. Build. Mat.* 95 (2015)  
549 133–142.
- 550 [13] ASTM C1399, Standard test method for obtaining average residual-strength of fiber-  
551 reinforced concrete, American Society for Testing and Materials, West Conshohocken,  
552 PA, 2015.
- 553 [14] RILEM TC 162-TDF, Test and design methods for steel fiber-reinforced concrete,  $\sigma$ - $\epsilon$   
554 design method, *Mater. Struct.* 36(262) (2003) 560–567.
- 555 [15] European Committee for Standardization, European Standard EN 14651:2007: Test  
556 method for metallic fibered concrete - Measuring the flexural tensile strength (limit of  
557 proportionality (LOP), residual), Brussels, 2007.
- 558 [16] E. García-Taengua, S. Arango, J.R. Martí-Vargas, P. Serna, Flexural creep of steel fiber  
559 reinforced concrete in the cracked state, *Constr. Build. Mat.* 65 (2014) 321–329.
- 560 [17] A. Abrishambaf, J.A.O. Barros, V.M.C.F. Cunha, Time-dependent flexural behaviour of  
561 cracked steel fibre reinforced self-compacting concrete panels, *Cem. Concr. Res.* 72  
562 (2015) 21–36.
- 563 [18] P. Serna Ros, J.R. Martí-Vargas, J.R., M.E. Bossio, R. Zerbino, Creep and residual  
564 properties of cracked macro-synthetic fibre reinforced concretes, *Mag. Concr. Res.*  
565 68(4) (2016) 197–207.
- 566 [19] R. Zerbino, D.H. Monetti, G. Giaccio, Creep behaviour of cracked steel and macro-  
567 synthetic fibre reinforced concrete, *Mater. Struct.* 49 (2016) 3397–3410.
- 568 [20] A. Llano-Torre, P. Serna, R. Zerbino, J.R. Martí-Vargas, Effect of initial crack opening  
569 on flexural creep behavior of FRC specimens, RILEM Proceedings PRO 116 - 9th  
570 RILEM International Symposium on Fiber Reinforced Concrete: The Modern  
571 Landscape (BEFIB 2016), Vancouver, 2016, pp. 117–126.

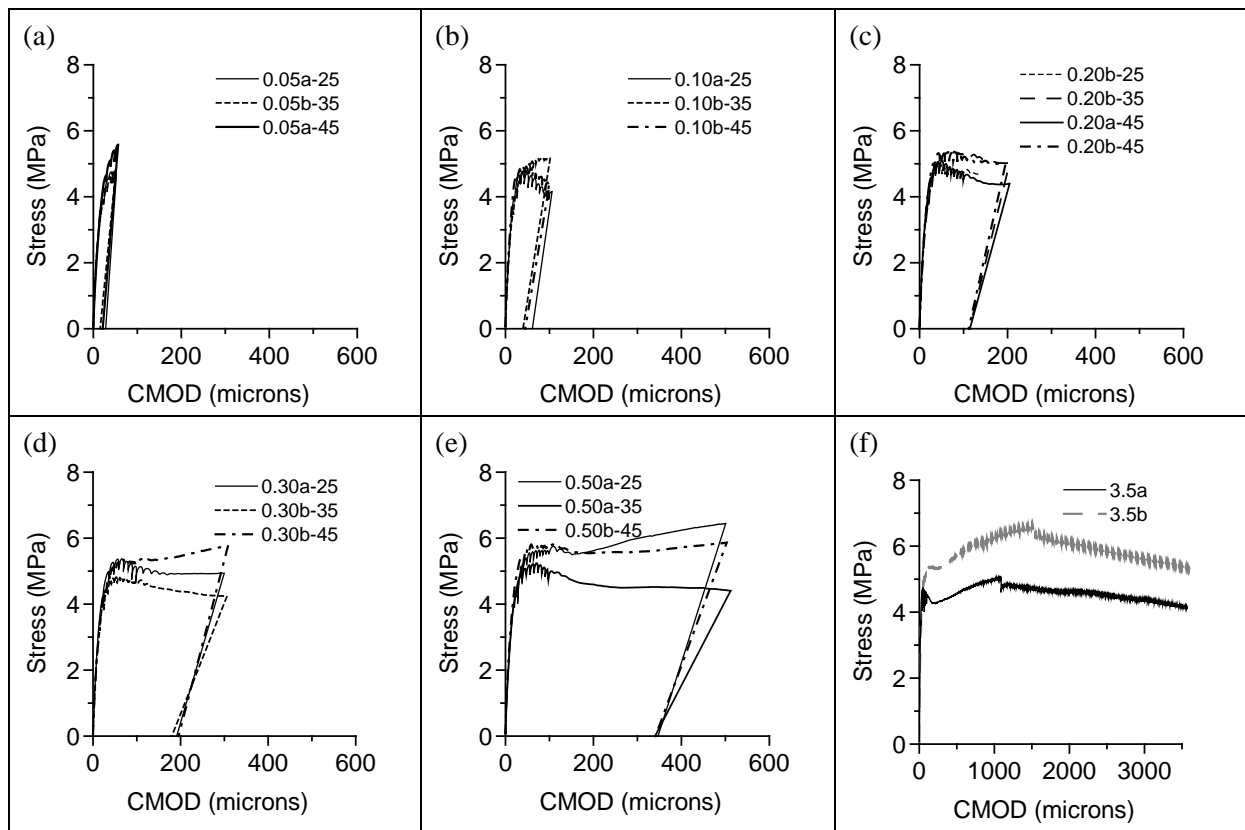
- 572 [21] U. Gossla, K.A. Rieder, Zeitabhängiges Verhalten von Makrokunststofffaserbeton und  
573 dessen Einfluss auf die Bemessung von Industriefußböden, Beton und Stahlbetonbau  
574 104 (2009) 76–87.
- 575 [22] P.D. Nieuwoudt, A.J. Babafemi, W.P. Boshoff, The response of cracked steel fibre  
576 reinforced concrete under various sustained stress levels on both the macro and single  
577 fibre level, Constr. Build. Mat. 156 (2017) 828–843.
- 578 [23] Fédération Internationale du Béton, fib Model Code for Concrete Structures 2010. Ernst  
579 & Sohn, Berlin, 2013, pp. 74–150.
- 580 [24] A. Llano-Torre, P. Serna, S.H.P. Cavalaro, “International Round Robin Test on creep  
581 behavior of FRC supported by the RILEM TC 261-CCF”, RILEM Proceedings PRO  
582 116 - 9th RILEM International Symposium on Fiber Reinforced Concrete: The Modern  
583 Landscape (BEFIB 2016), Vancouver, 2016, pp. 127–140.
- 584 [25] Llano-Torre A., Arango S.E., García-Taengua E., Martí-Vargas J.R., Serna P. (2017)  
585 Influence of Fibre Reinforcement on the Long-Term Behaviour of Cracked Concrete.  
586 In: Serna P., Llano-Torre A., Cavalaro S. (eds) Creep Behaviour in Cracked Sections of  
587 Fibre Reinforced Concrete. RILEM Bookseries, vol 14. Springer, Dordrecht.



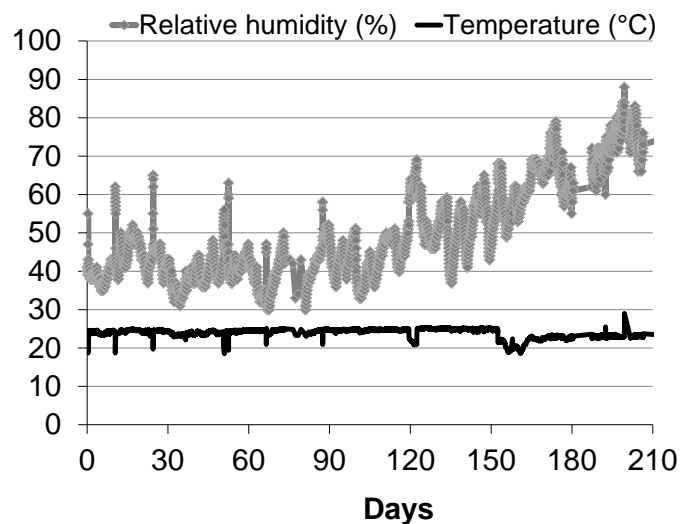
**Fig. 1.** Main stages of the cracked FRC creep test methodology.



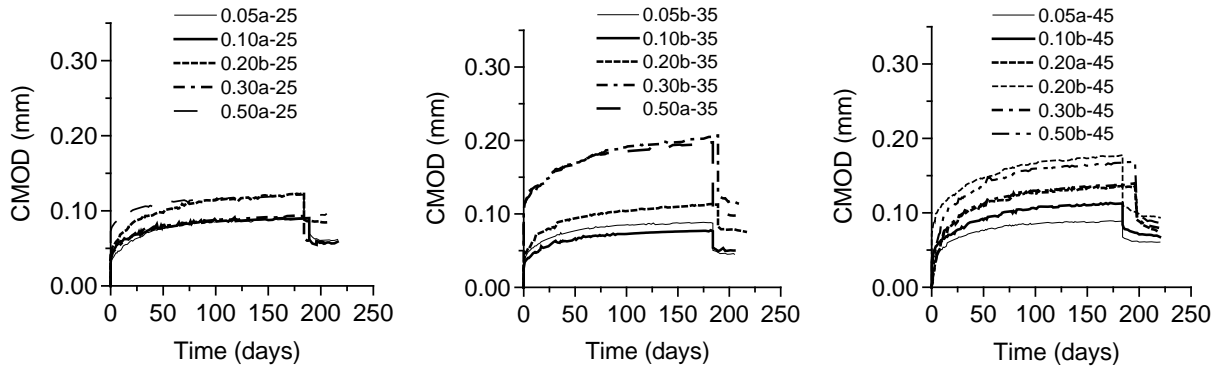
**Fig. 2.** Applied load and measurement devices during of creep period: (a) load configuration scheme; (b) specimen's setup in column; (c) transducer location at rear side; and (d) transducer location at front side.



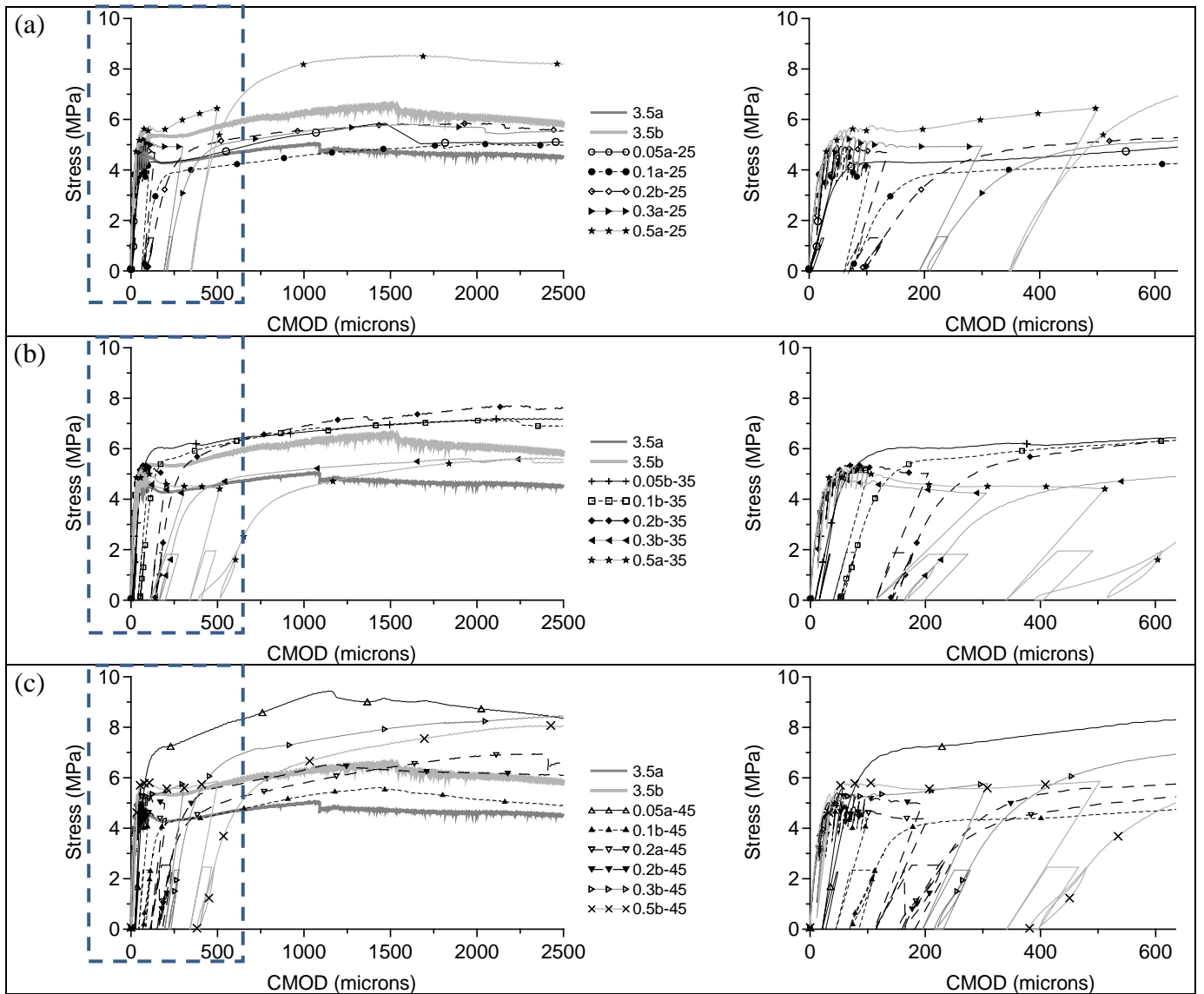
**Fig. 3.** Three point bending test curves up to different CMOD: (a) 0.05 mm, (b) 0.1 mm, (c) 0.2 mm, (d) 0.3 mm, (e) 0.5 mm, (f) 3.5 mm.



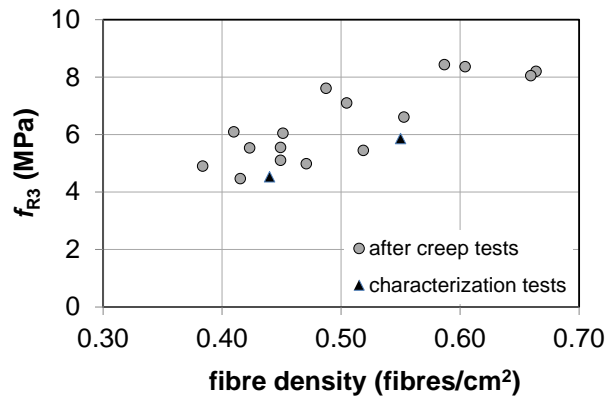
**Fig. 4.** Environmental conditions during the development of the creep period.



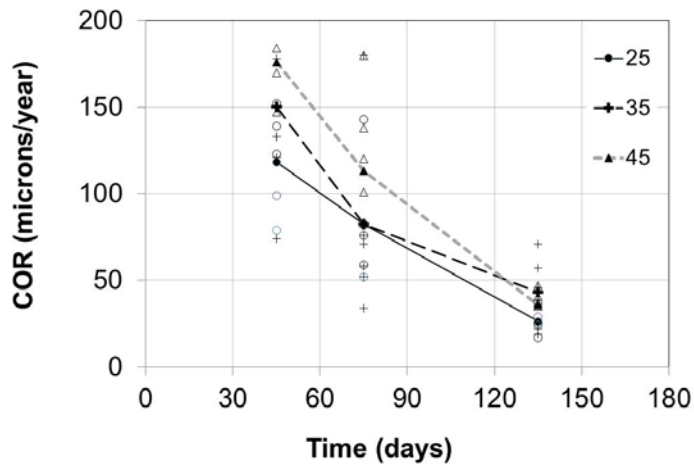
**Fig. 5.** CMOD evolution during the creep period.



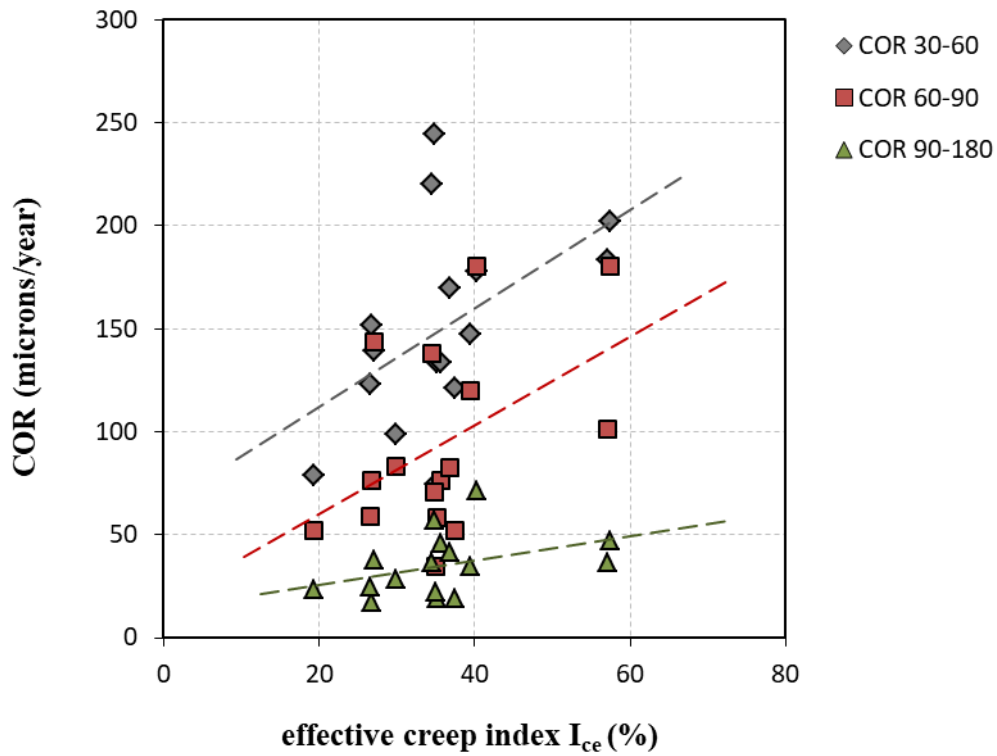
**Fig. 6.** Stress-CMOD curves including pre-cracking, creep period and post-creep stages at different stress levels  $I_n$ : (a) 25%, (b) 35% and (c) 45%.



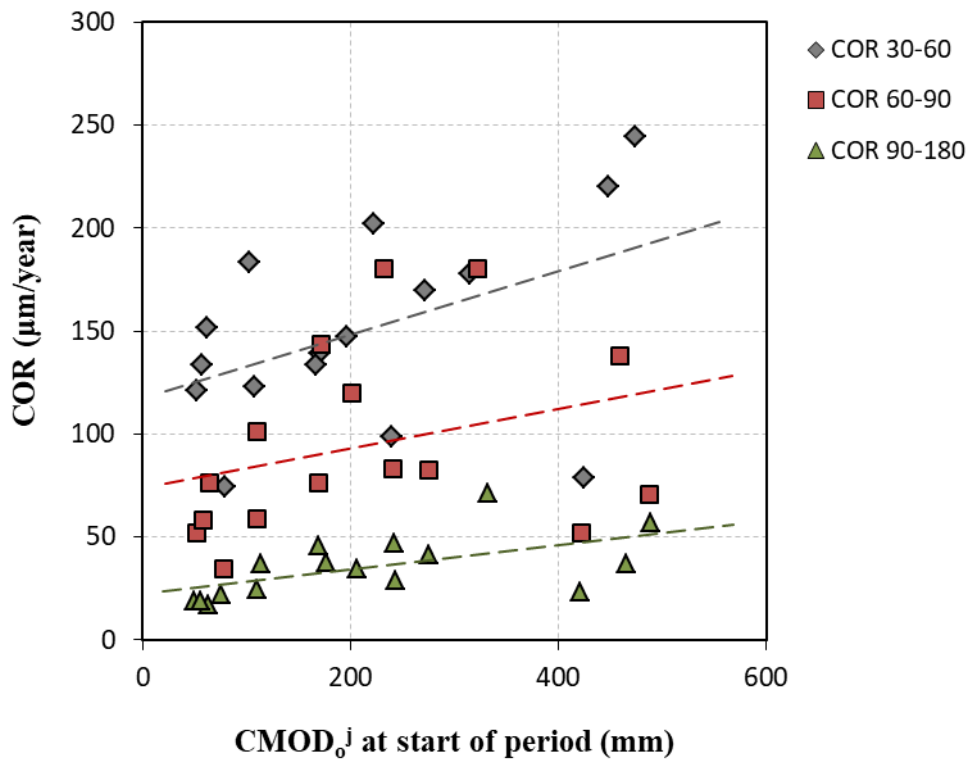
**Figure 7.** Residual stress  $f_{R,3}$  vs. fiber density  $\delta_F$  at the fracture surface.



**Fig. 8.** Crack Opening Rate (COR) evolution through time.

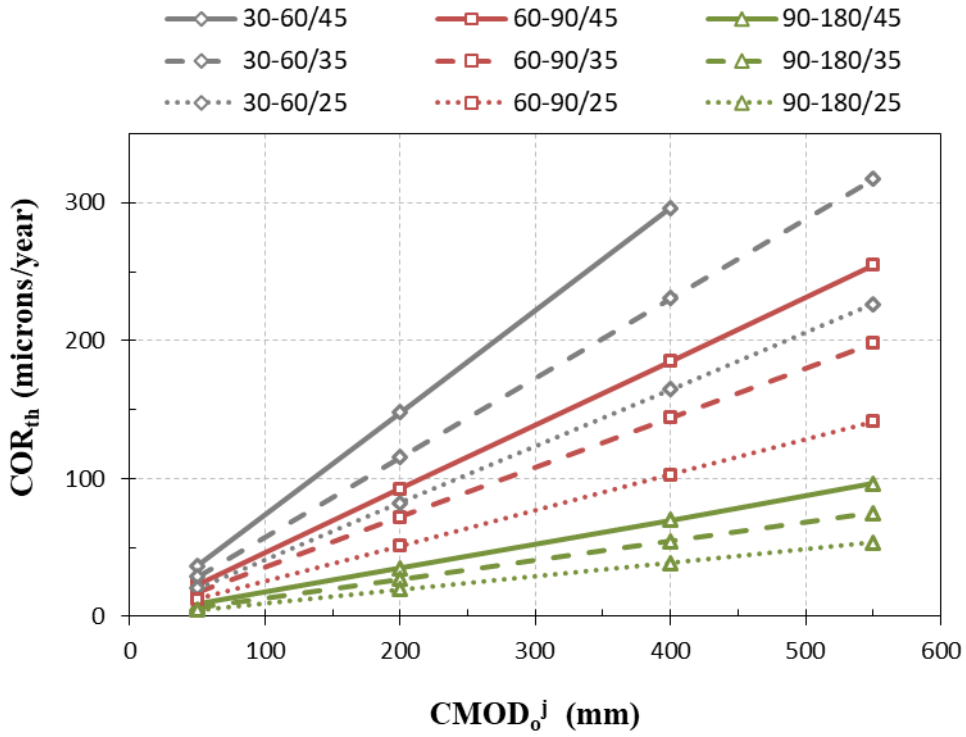


**Fig. 9.** Crack Opening Rate (COR) vs. effective creep index  $I_{ce}$ .

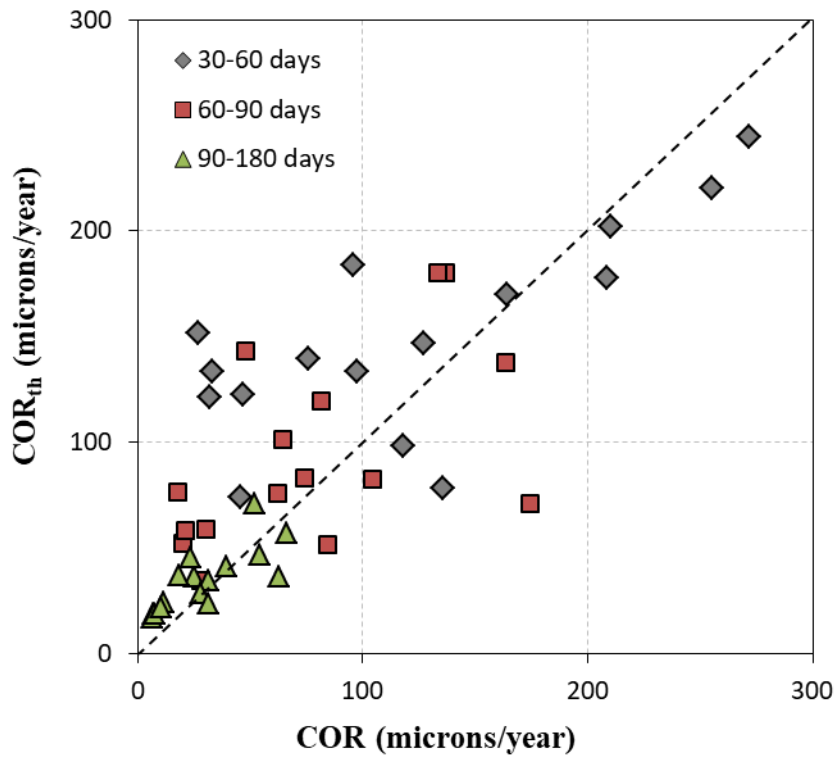


**Fig. 10.** COR vs.  $CMOD_o^j$  at the beginning of the analyzed time period.





**Fig. 11.**  $COR_{th}$  vs.  $CMOD_o^j$ , for each analyzed time period and nominal stress level considered.



**Fig. 12.** Comparison between estimated COR and experimental COR results.