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Additional Information

Model reduction based on sparse identification techniques for induction machines: towards the real time and accuracy-guaranteed simulation of faulty induction machines

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Abstract

The development of condition monitoring (CM) systems of induction machines (sIMs) is essential for the industry because the early fault detection helps engineers to optimise maintenance plans. However, the use of several IMs to test and validate the fault diagnosis methods developed requires also the use of costly test benches that, anyway, often face limitations in the range of faults and operating conditions to be tested. To avoid it, the use of accurate models such as those based on finite element method (FEM) would reduce the major drawbacks of test benches but their inability to execute FEM models in real time largely reduces their application in the development of on-line continuous monitoring systems. To alleviate this problem a hybrid FEM-analytical model has been proposed. It uses an analytical model that can be run in real-time in a hardware in the loop (HIL) system, after its parameters have been computed through FEM simulations. In this way, the proposed model provides high accuracy but at the cost of long simulation times and high computational costs (both computing power and memory resources) to compute the IM parameters. This work aims at reducing these drawbacks. In particular, a model based on sparse identification techniques is proposed. The method balances model complexity and accuracy by selecting a sparse model that reduces the number of FEM simulations to accurately compute the coupling parameters of an induction machine model with different fault severity degrees. Particularly, the proposed methodology has been applied to develop models with abnormal eccentricity levels as this fault is related to development of mechanical faults that produce most of IM breakdowns.

Keywords: Fault diagnosis, Induction machines, Model order reduction, Sparse identification

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1. Introduction

The CM of IMs has been a subject of eager interest over the last years due to the costly downtimes that an unexpected breakdown can cause [1]. The main sources of failures, about the 40-50% in large IMs, are related to mechanical faults leading to eccentricity [2] with catastrophic consequences [3]. Therefore, the early detection of the eccentricity fault in IMs would be crucial to adjust maintenance plans and ensure the continuity of the industry operation.

During the latest years the detection of the eccentricity fault in IMs has gathered great efforts from the scientific community. In fact, it has been proposed the acquisition and analysis of a wide variety of physical variables of the IM such as the magnetic stray flux [4, 5] or vibrations [6] among others. Nevertheless, their use has several drawbacks. On the one hand, their acquisition requires costly sensors, which are also difficult to install on the IM working in the industry. On the other hand, it is not possible to detect all types of faults through the analysis of these magnitudes [7]. Alternatively, the analysis of the stator current has been widely used for the CM of IMs as it is a magnitude conveying relevant information about the machine condition. It is well-known that each type of fault induces or amplifies a family of harmonic components in the stator current, whose frequencies have already been demonstrated theoretically and validated experimentally. Moreover, it has low requirements on hardware and software for its acquisition and analysis.

The eccentricity fault in an IM can appear in three different forms [8] as shown in Figure 1: static, dynamic or mixed eccentricity. In the case of static and dynamic eccentricity, the rotor symmetry axis is shifted from the stator centre. In the case of static eccentricity, the rotor rotates around its symmetry axis whereas in the case of dynamic eccentricity not. This leads to different configurations of the air-gap width. In the first case, static eccentricity, there are fixed angular position where the air-gap width is minimum and maximum respectively. On the contrary, in case of dynamic eccentricity, the position of the minimum and maximum air-gap widths vary as the rotor rotates. The main frequencies due to these type of eccentricities are derived from the general equation to detect the so called principal slot harmonic (PSH) or rotor slot harmonic (RSH) [8, 9]:

$$f_h = \left[(kR \pm n_d) \frac{1-s}{p} \pm \nu \right] f_1 \quad (1)$$

where k is any positive integer, R are the number of rotor slots, $n_d = 0$ for static eccentricity or a positive integer for dynamic eccentricity, s is the slip, p is the pole pairs, ν is the order of the stator time harmonics and f_1 is the mains frequency.

Finally, the mixed eccentricity (Figure 1 d) is a combination of both static and dynamic eccentricities. In this case the low rank fault frequencies that appear are defined by:

$$f_{m_{ecc}} = f_1 \pm k f_r \quad k = 1, 2, 3 \dots \quad (2)$$

where f_r is the rotor's mechanical angular speed.

A detection of these harmonic components is necessary but not sufficient to perform the diagnosis. IMs are not ideal and always have inherent levels of asymmetries, eccentricities,

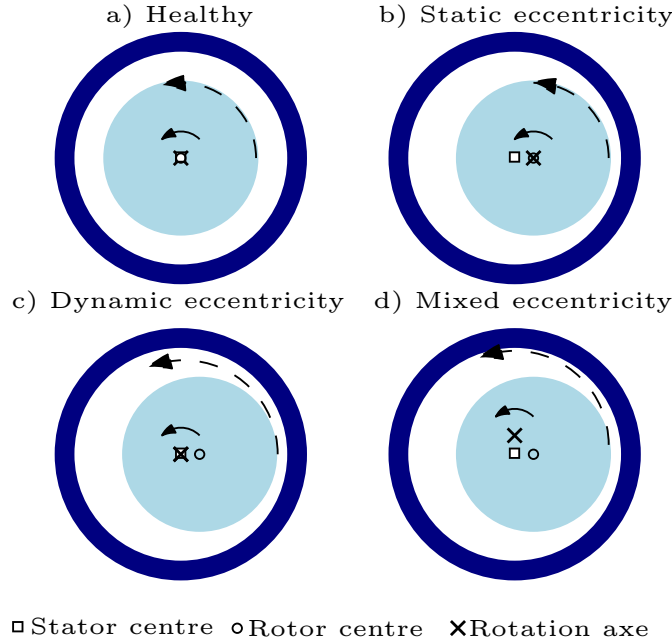


Figure 1: Eccentricity types. It can be seen that for static eccentricity the positions for the minimum and maximum airgap widths are fixed regarding the stator for any rotor orientation while for dynamic and mixed eccentricity those positions change as the rotor rotates.

38 etc. For example, in the case of static eccentricity, these inherent levels should be lower
 39 than 10% [10], but they may lead to wrong diagnoses. Therefore, the study of the amplitude
 40 evolution of these fault harmonic components regarding the fault severity degree will not
 41 only allow establishing theoretical thresholds from which detect incipient failures but also
 42 reducing misdiagnoses. Besides, the type of machine, the working conditions or the load
 43 condition are additional factors influencing the behaviour of the fault harmonic components
 44 which should be considered when developing a CM system.

45 On-line CM systems and artificial intelligence (AI) based fault diagnosis system would
 46 have a major impact in the detection of these faults at early stage. The on-line CM system
 47 continuously monitor the machine status obtaining a trend of the eccentricity fault harmonic
 48 components to monitor the severity of the upcoming fault. AI based CM systems need to
 49 trained with a large number of signals to learn to classify the machine status (healthy or
 50 faulty) and even to determine the severity degree of a given fault. Thus, not only the
 51 behaviour but also the reliability of the on-line CM system and the AI based fault diagnosis
 52 systems need to be checked with a large number of tests covering a wide variety of scenarios:
 53 types of machines, levels of severity of the fault, load conditions, working conditions (steady
 54 state, transient), etc.

55 The ideal is to fulfil these requirements (different IM with different levels of severity of
 56 the fault, wide variety of working conditions and load conditions) with IM working in the
 57 industry. However, that would require close and effective collaboration with the industry
 58 and, besides, there is a very limited number of IMs that could be running in the industry

59 under faulty conditions. Although laboratory test benches are a good alternative choice, the
60 tests are limited to the IMs and drives available in the laboratory. Besides, it is very costly
61 because it needs several destructive tests and, from a practical point of view, obtaining
62 several degrees of a given fault or even simultaneous faults is a very challenging task. That
63 is, although the experimental validation is inevitable at the last stage of the fully developed
64 CM, during the development stage other less costly alternatives must be considered.

65 Accurate models could help at reducing these drawbacks: it is possible to obtain different
66 models of several types of IMs, with different levels of severity of a given fault or even with
67 simultaneous faults at much lower cost than using laboratory test benches; moreover, these
68 models enable simulation of IMs under a wide variety of working conditions. To achieve these
69 benefits these models must consider the detailed IM structure to obtain simulation results
70 that truly reflect the real-world situations. Besides, these models have to run in real time to
71 properly test on-line condition monitoring systems. However, to achieve these requirements
72 (accuracy and the possibility of running in real-time) in a unique model is very challenging.
73 In the following subsection the main advances in the development of faulty models of IM
74 are reviewed.

75 *1.1. Faulty IMs models*

76 Several IMs models have been proposed in the technical literature. The well-known d-q
77 model [11, 12] is widely used in order to understand and design vector controlled drives. It is
78 simple to be implemented in a HIL but it does not consider the geometrical complexities, the
79 spatial distribution of the windings (i.e. the space harmonics) [13], the non-linearity of the
80 core materials and it cannot include the effects that a fault introduces in a machine and, thus,
81 it cannot be used for fault diagnosis purposes. To include these features in the model other
82 analytical approaches have been proposed in the technical literature such as the multiple
83 coupled circuit model (MMC) [14], the winding function approach (WFA) [15], the Con-
84 cordia transformations [16], the use of natural variables [17], the voltage-behind-reactance
85 formulation [18], the magnetic equivalent circuit (MEC) [19] or the sparse identification [20].
86 Nevertheless, these approaches cannot consider non-ideal conditions and cannot include the
87 effect of the rotor and stator slots in the air-gap magnetic force distribution, specially, when
88 the eccentricity is being modelled [21], as required in faulty IM models. Particularly, the
89 the sparse identification is proposed to improve the efficiency of motor control [20] and uses
90 the stator voltages and currents of an induction motor to compute the parameters of the
91 equivalent circuit. However, it assumes some simplifications such as uniform air-gap width
92 which is enough for motor control but not suitable for fault diagnosis purposes.

93 On the other hand, FEM models and their accuracy are widely accepted as they usually
94 take into account the geometrical complexities, the spatial distribution of the windings, the
95 non-linearity of the core materials, etc [22]. Unfortunately, time-stepping FEM simulations
96 require high computing power and memory resources. Besides, they take long simulation
97 times (from minutes to days) for short simulation periods. These constraints are even worse
98 with faulty IM models where simplifications to boost the time-stepping FEM simulation,
99 such as the symmetry boundary conditions, can no longer be applied. To sum up, FEM
100 highly increases the accuracy in machine simulation [22], but at significant computational

101 cost even with modern processing power computers [23]. Consequently, the savings in com-
102 putational effort are crucial in fields where a large number of results are required such as
103 fault diagnosis, either for on-line CM systems or for AI based fault diagnosis systems, motor
104 control optimization, etc.

105 Hybrid FEM-analytical models have been recently proposed as they are able to run in
106 real time in a HIL simulator and keep good accuracy [24, 25, 26, 27]. They are based on the
107 equivalent circuit parameters computation through magneto-static FEM simulations and on
108 using these parameters in the analytical model [27]. However, the hybrid FEM-analytical
109 model still has several limitations as the evaluation of each new scenario (fault conditions)
110 requires the full FEM analysis to compute the new coupling parameters with its long running
111 times and computational effort.

112 In an attempt to address these drawbacks [28] proposes the sparse subspace learning
113 (SSL) in combination with a hierarchical collocation strategy to compute a low-rank pre-
114 diction of the parametric solution of the FEM model. In fact, the SSL uses the outputs of
115 a deterministic solver to produce parametric solutions in a multi-level interpolation frame-
116 work. Thereafter, the deterministic solver uses these predicted solutions as input, as initial
117 guess, to obtain the solution in a new sampling point. In that case, the initial guess is so
118 close to the solution that the iteration time of the solver is drastically reduced or might
119 not even be required to run. From the point of view of the fault diagnosis purposes, this
120 approach does not mean any substantial improvement because it does not reduce the large
121 number of magneto-static FEM simulations required to obtain the coupling parameters [27]
122 for the analytical model. To alleviate this problem the sparse identification was proposed
123 in [29] to obtain an faulty induction machine model. However, to perform the parameter
124 identification of a new faulty IM model the method requires the input of not only a wide
125 range of fully FEM computed coupling parameters of the same machine with different sever-
126 ity degrees of the same fault under study. Therefore, it reduces the computational effort
127 compared with traditional methods, but it still requires a large number of FEM simulations.
128 Besides, it is very limited to be applied to other machines or even to other faults.

129 To ease the limitations of traditional SSL implementation, and as a novelty, this paper
130 proposes the use of the sparse identification technique aimed at reducing the number of
131 magneto-static FEM simulations. This will avoid the need of a FEM simulation for every
132 new sampling point, as in [28]. Therefore, this paper proposes the sparse identification
133 to compute the coupling parameters of the faulty IM model based on the results of a very
134 reduced number of magneto-static FEM simulations. This will not only reduce the compu-
135 tational effort but will also guarantee good accuracy of the obtained model. In this paper,
136 the proposed method is applied to obtain models with static eccentricity faults as it is a
137 fault that may lead to catastrophic failures and because it is very difficult to artificially force
138 different degrees of the fault in IM to be used in test benches. These models will provide a
139 better understanding of the physical phenomena while tracking the behaviour of the fault
140 harmonic components, to establish thresholds, etc. Besides, as these models are capable to
141 run in real time under different working conditions (power supply variations, load changes,
142 etc), they will be useful for developing on-line CM systems and to train AI based automatic
143 diagnostic systems.

144 The paper is structured as follows. In section 2 the hybrid FEM-analytical model is
 145 described and the methodology to compute the coupling parameters is introduced showing
 146 the main drawbacks of the approach. Section 3 introduces the proposed method to compute
 147 the coupling parameters and the main benefits are presented in terms on computing time
 148 and memory requirements. The main results are presented in section 3.3 In this section the
 149 coupling parameters computed with the proposed method are compared with those obtained
 150 with traditional methods. Moreover, the achieved accuracy is as good as the saving in terms
 151 on computing time and data storage memory. In section 4 the simulations results in terms
 152 of fault diagnosis purposes are presented while in section 5 the experimental validation is
 153 described. Finally, in 6 the main conclusions are presented.

154 2. Hybrid FEM-analytical model of an IM

155 The electromagnetic behavior of a general IM with M stator and N rotor phases can be
 156 modelled as [30, 31, 32]:

$$\begin{bmatrix} U_{s_1} \\ \vdots \\ U_{s_M} \\ U_{r_1} \\ \vdots \\ U_{r_N} \end{bmatrix} = \begin{bmatrix} R_{s_1} & & & & & \\ & \ddots & & & & \\ & & R_{s_M} & & & \\ & & & R_{r_1} & & \\ & & & & \ddots & \\ & & & & & R_{r_N} \end{bmatrix} \begin{bmatrix} I_{s_1} \\ \vdots \\ I_{s_M} \\ I_{r_1} \\ \vdots \\ I_{r_N} \end{bmatrix} + \frac{d[\Psi]}{dt} \quad (3)$$

157 where the subscripts s and r stand for stator and rotor respectively, $[U]$ is the phase voltage
 158 matrix, $[R]$ is the resistances matrix, $[I]$ is the phase current matrix and Ψ stands for the
 159 flux linkage that can be computed as:

$$\begin{bmatrix} \Psi_{s_1} \\ \vdots \\ \Psi_{s_M} \\ \Psi_{r_1} \\ \vdots \\ \Psi_{r_n} \end{bmatrix} = \begin{bmatrix} L_{s_1s_1} & \cdots & L_{s_1s_M} & | & L_{s_1r_1} & \cdots & L_{s_1r_N} \\ \vdots & \ddots & & & \vdots & & \\ L_{s_Ms_1} & \cdots & L_{s_Ms_M} & | & L_{s_Mr_1} & \cdots & L_{s_Mr_N} \\ \hline L_{r_1s_1} & \cdots & L_{r_1s_M} & | & L_{r_1r_1} & \cdots & L_{r_1r_N} \\ \vdots & & & & \ddots & & \vdots \\ L_{r_Ns_1} & \cdots & L_{r_Ns_M} & | & L_{r_Nr_1} & \cdots & L_{r_Nr_N} \end{bmatrix} \begin{bmatrix} I_{s_1} \\ \vdots \\ I_{s_M} \\ I_{r_1} \\ \vdots \\ I_{r_N} \end{bmatrix} \quad (4)$$

160 where $[L]$ is also known as the inductances matrix, that is to say, the coupling parameters
 161 between the different electromagnetic circuits inside of an IM. To simplify, (4) can also be
 162 expressed as:

$$\begin{bmatrix} \Psi_s \\ \Psi_r \end{bmatrix} = \begin{bmatrix} L_{ss} & | & L_{sr} \\ \hline L_{sr}^T & | & L_{rr} \end{bmatrix} \begin{bmatrix} I_s \\ I_r \end{bmatrix} \quad (5)$$

163 where $[L_{ss}]$, $[L_{sr}]$ and $[L_{rr}]$ are the coupling parameters between stator phases, between
 164 stator and rotor phases and between rotor phases respectively. The electromagnetic torque
 165 generated T_e by the IM is modelled as:

$$T_e = [I_s]^T \frac{d[L_{sr}]}{d\theta} [I_r] \quad (6)$$

166 and finally, the equation that models the mechanical behaviour is:

$$T_e - T_{Load} = J \frac{d^2\theta}{dt^2} + b \frac{d\theta}{dt} = J\alpha + B\omega \quad (7)$$

167 where T_{Load} is the load torque, J is the inertia constant, B is viscous friction constant, α is
 168 the angular acceleration and ω is the rotational speed. The analytical model of an IM using
 169 the system of equations (3) to (7) can be easily implemented in a Simulink model as shown
 170 in Figure 2. The main advantage of this type of model is that can be run in real time in a
 171 HIL system. It must be highlighted that all blocks used are standard Simulink blocks except
 172 for the “OpComm” block belonging to the HIL library and used to connect the input signals
 173 of the model. The model shown in Figure 2 uses the stator voltages and the load torque
 174 as inputs. For instance, to cover a wide variety of industrial scenarios the user can select
 175 to power the IM either using direct on-line (DOL) (balanced or unbalanced phase voltages)
 176 or through a variable speed drive (VSD) with its usual open/close loop controls. Moreover,
 177 during the real-time simulation the user could apply changes in the power supply such as
 178 the voltage and/or the reference speed of the VSD. On the other hand, the user can define
 179 load torque profiles to simulate industrial processes and/or modify the load torque during
 180 the real time simulation. Therefore, it allows to simulate the IM model under a wide variety
 181 of working conditions as required for the CM systems development.

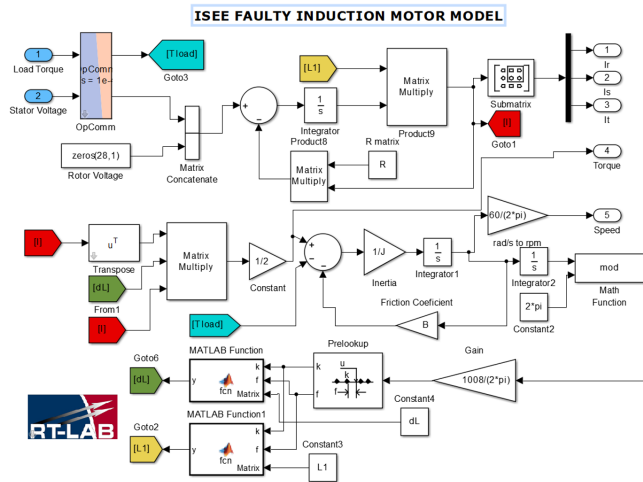


Figure 2: Analytical model using the system equations (3)-(7) using Simulink blocks ready to be run in a HIL system for real-time simulation.

182 In this approach, the key issue is to compute the coupling parameters $[L]$ of the faulty
 183 IM model. Specifically, the presence of derivatives in (6) requires an accurate enough com-
 184 putation of the coupling parameters for the reliable identification of different fault severity
 185 degrees. As the coupling parameters $[L]$ vary depending on the rotor position, specially in
 186 the case of a faulty machine, it is necessary to compute the mutual and self inductance for
 187 each rotor position requiring, thus, a large number of magneto-static FEM simulations.

188 To illustrate the main drawbacks associated with the coupling parameters $[L]$ computa-
 189 tion based on FEM software and to introduce and show the main benefits of the proposed
 190 method the machine whose main characteristics are shown in Table 1 is used. Given the
 191 importance of the static eccentricity fault, this fault has been used to illustrate the method
 192 but the same procedure of sections 2.1 and 3 is also valid for other different types of faults.

Table 1: Data of the simulated machine.

Electrical		Mechanical	
Power	1.1kW	Pole pairs	2
Voltage	230/400 V	Speed	1415 rpm
Current	4.4/2.55 A	N° of rotor bars	28
Frequency	50Hz	N° of stator slots	36
		Airgap length	0.28mm

193 2.1. Methodology to compute the parameters of the IM model using FEM for a generic case

194 FEM software allows to create accurate IM models which consider the non-uniform air-
 195 gap due to stator and rotor and other asymmetries due to faults. Therefore, in opposition
 196 the simulation requirements for healthy machines, the whole geometry of a faulty machine
 197 has to be considered in the simulation which results in a much more time-consuming task.
 198 Besides, a large number of simulations are required to compute the coupling parameters
 199 matrix.

200 Figure 3 shows the general diagram to compute the coupling parameters $[L]$ for a IM.
 201 The process starts by creating a FEM model of the IM in which the geometry of the machine
 202 as well as the specific characteristics of the fault are considered and the rotor is placed in
 203 the first position ($q = 1$). Starting from the first stator phase ($m = 1$) each of the M stator
 204 phases are fed with 1 A DC and the magneto-static FEM simulation is performed. With the
 205 results it is possible to compute the coupling parameters between stator phases $[L_{ss}]$ and
 206 between stator and rotor phases $[L_{sr}]$. Usually, in case of cage IM the coupling parameters
 207 between stator phases and rotor bars are considered instead of between stator and rotor
 208 phases. Subsequently, each of the N rotor phases is fed by 1 A DC, the FEM magneto-
 209 static simulation is performed and the coupling parameters between rotor phases (rotor bars
 210 in case of cage IM) $[L_{rr}]$ are computed. The rotor is moved in increments of $rd = 2\pi/K$,
 211 where K is the number of positions desired, and the aforementioned process is repeated for
 212 each rotor position. Clearly, the larger the K considered the higher the accuracy in the
 213 description of the coupling parameters $[L]$ for different rotor positions and the higher the
 214 number of FEM simulations required with its corresponding running time, computing power
 215 and memory resources. Finally a three dimension $(M + N, M + N, K)$ coupling parameters
 216 matrix $[L]$ is obtained, where the third dimension is related to the rotor position q . Hence,
 217 the coupling parameters matrix $[L_{abq}]$ designated from now as $[L_q]$ is the coupling parameter
 218 matrix $[L]$ of dimensions $(M + N, M + N)$ corresponding to the rotor position q .

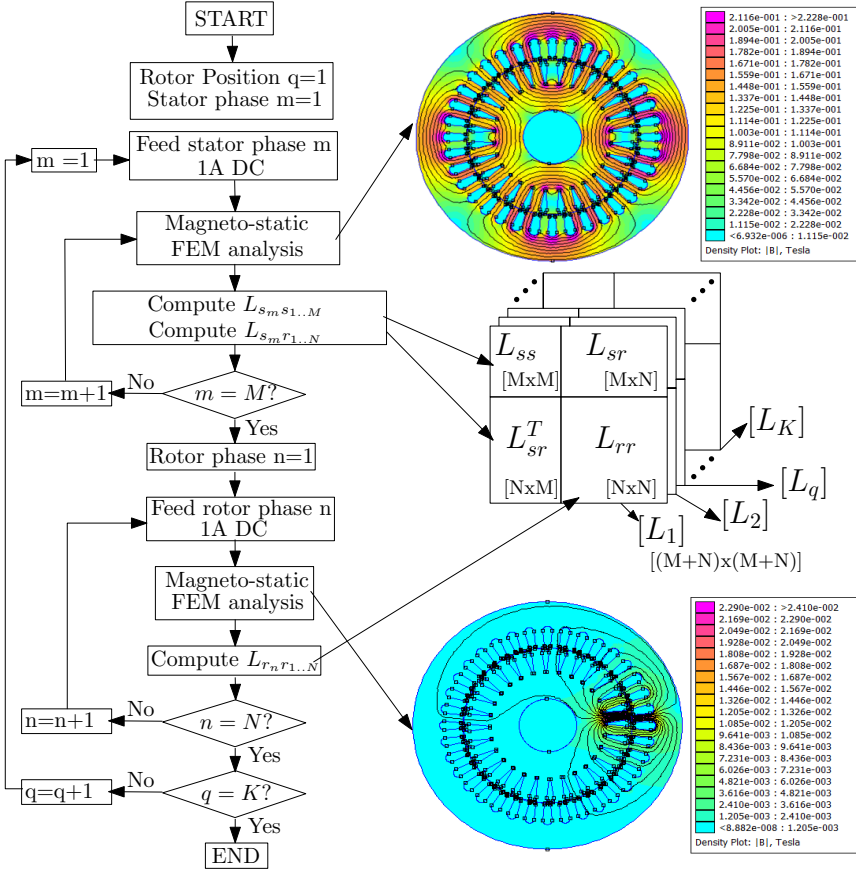


Figure 3: General diagram of the procedure to compute the coupling parameters $[L]$ of a IM model using FEM software.

219 The method proposed in this paper assumes linear conditions for the computation of the
 220 inductance matrix of the eccentric machine. From a diagnostic point of view, saturation
 221 generates fault harmonics which are different from those generated by the eccentricity fault.
 222 Indeed, the purpose of the presented approach is to develop an efficient method for computing
 223 an inductance matrix that accurately captures the effect of the eccentricity fault. The
 224 proposed paper aims at developing an analytical model able to run in real-time with high
 225 accuracy (based on FEM results) but a much lower cost. In this case, considering only the
 226 linear, incremental problem, balances the results with the computational costs obtaining a
 227 reasonably accurate solution for eccentricity fault diagnosis purposes. Nevertheless, if the
 228 saturation effects are also to be reproduced, there are several methods presented in the
 229 technical literature where the inductance matrix obtained with the proposed approach can
 230 be used. One of such methods is the the “incremental permeability” solution proposed in
 231 [33], which is used in several FEM software packages to save computation time, such as FEM.
 232 In this solution, first, the nonlinear problem is solved at a particular nonlinear operating
 233 point (particular instantaneous currents, rotor orientation), using a full FEM simulation.
 234 Then, a linear, incremental problem based on the incremental permeabilities is solved, by

235 feeding each phase with a current of 1A with all other excitations are turned off. This
 236 linear problem is solved in [33] using again FEM simulations, which could be replaced by
 237 the solutions provided by the proposed method, in a much faster way, while keeping their
 238 accuracy.

239 *2.1.1. Particularizing the method to compute the coupling parameters to the IM under study.*

240 To illustrate the cost in terms of computing power, memory resources and processing
 241 time, the method introduced in section 2 and illustrated in the Figure 3 has been applied to
 242 obtain the coupling parameters matrix $[L]$ of the IM whose main characteristics are shown
 243 in Table 1. The software used for the FEM simulations is the open source femm 4.2 and the
 244 computer used has an intel processor (R) Core (TM) i5-6400 CPU@2.70GHz 2.20GHz and
 245 16GB of RAM memory.

246 The first step, is to decide the rotor movement steps ($rd = 2\pi/K$) to be used. This
 247 term is related to the accuracy, in terms of rotor position, needed to compute the coupling
 248 parameters $[L]$. Usually, the result of multiplying the number of stator slots by the number
 249 of rotor bars for cage IM (or rotor slots in case of wound rotor IM) provides a good value of
 250 K :

$$K = n^{\text{er}} \text{ of rotor bars} \times n^{\text{er}} \text{ of stator slots} = 28 \times 36 = 1008 \quad (8)$$

251 and, therefore $rd = 2\pi/K = 2\pi/1008 = 0.00632\text{rads}$. Considering the process shown in
 252 Figure 3 where for the IM studied $K = 1008$, a rotor phase is the loop of two adjacent rotor
 253 bars $N = 28/2$ and the number of stator phases is $M = 3$, the computation of the coupling
 254 parameters matrix $[L]$ in this generic case requires $(M + N) \cdot K = (3 + 28/2) \cdot 1008 =$
 255 $17,136$ magneto-static FEM simulations. Taking into account that each magneto-static
 256 FEM simulation takes 1 minute and needs 22.5 MB for data storage, the computation of
 257 the coupling parameters $[L]$, for one machine and with only one severity degree for a given
 258 fault, would require 11 days 21 hours and 36 minutes and 376.52 GB for data storage if
 259 all the FEM results need to be saved; if only the coupling parameters $[L]$ are needed each
 260 $[L_q]$ (coupling parameters for a rotor position) require 1 kB, so that, the memory resources
 261 could be reduced to just 1008 kB. It must be highlighted that these resources are needed
 262 to compute the coupling parameters of just 1 machine and only for 1 severity degree of a
 263 given fault whereas to fulfil the requirements for developing on-line CM systems and expert
 264 systems a wide variety of models of different machines and with several severity degrees of
 265 a given fault are needed.

266 *2.1.2. Particularizing the method to compute the IM coupling parameters considering the
 267 static eccentricity fault.*

268 Sometimes, the particularities of a specific fault enable the use of some simplifications to
 269 reduce the computing effort for computing the coupling parameter matrix L . To reproduce
 270 the static eccentricity fault the rotor symmetry axis have to be displaced from the stator
 271 centre. It allows the definition of different degrees of static eccentricity fault, from 0% for
 272 healthy conditions to 100% for the maximum displacement of rotor symmetry axis. It results
 273 in a faulty IM model where the positions of the maximum and minimum air-gap width with
 274 respect to the stator do not depend on the rotor position. Therefore, each rotor phase (rotor

275 bar) will have the same flux linkage but with a certain geometric offset. As a consequence,
 276 two main simplifications can be applied; only one rotor phase (or rotor bar) must be fed to
 277 compute the coupling parameters between rotor phases (or rotor bars) $[L_{rr}]$; and this bar
 278 has to be displaced only along half of all the possible rotor positions, that is to say, between
 279 $[0, \pi]$ which, in the case of study, is 504 of the 1008 total positions.

280 On the other hand, to compute the coupling parameters between stator and stator phases
 281 $[L_{ss}]$ and between stator phases and rotor phases (bars) $[L_{sr}]$ just the positions of a rotor bar
 282 travelling through a stator slot are required. Therefore, the computation of $[L_{ss}]$ and $[L_{sr}]$
 283 requires just the simulation of the model placing the rotor in the first 36 of the 1008 positions
 284 while feeding, sequentially, each stator phase. It reduces the required FEM simulations to
 285 $3 \cdot 36 + 504 = 612$ FEM simulations with a computing time of 10 hours and 12 minutes and
 286 a data storage of 13.45 GB. It is a significant improvement but it does not go far enough to
 287 meet the needs of the development of CM systems as they require to test a wide variety not
 288 only of IMs but also a wide range of severity degrees of the faults.

289 **3. Proposed method: sparse identification to compute the parameters of the IM** 290 **model with similar accuracy to FEM**

291 *3.1. The Sparse Subspace Learning (SSL) and the Hierarchical Lagrange interpolation (HLI)*

292 The SSL in combination with a hierarchical Lagrange interpolation (HLI) as polynomial
 293 basis introduced in [28] reduces the computing time needed to solve parametric problems
 294 in FEM software. It proposes a collocation strategy to reduce the time computing require-
 295 ments of parametric models. The SSL strategy selects specific sampling points in which the
 296 simulation has to be performed; thereafter, the output of the deterministic solver is used to
 297 obtain a HLI polynomial basis which allows to compute an approximate low-rank parametric
 298 solution at new sampling points. This approximated solution is used to initialize the FEM
 299 solver which speeds up the convergence of the iteration process as the predicted solutions
 300 are very close to the FEM solution; in fact, in some cases, the iteration process is not
 301 even required to be run. In many cases with moderate dimensionality, the iteration process
 302 is not needed as the hierarchical predicted solution yields precise enough results for most
 303 engineering problems at a reasonable computational costs [34].

304 *3.2. Sparse identification to compute the coupling parameter matrix.*

305 With the sparse identification strategy proposed in [28] the polynomial basis obtains
 306 a prediction of the solutions in the nodes of the FEM model. Although it reduces the
 307 computing time for complex models, it keeps the memory requirements for data storage. If
 308 we contextualize to compute the coupling parameters of a faulty IM, where a magneto-static
 309 FEM simulation takes just 1 minute, the main benefits of the sparse identification shall not
 310 have a major impact in the overall computation time. Therefore, what is proposed in this
 311 paper, is to go one step further by performing the sparse identification to obtain the coupling
 312 parameters matrix $[L]$ of a faulty IM. This matrix will be computed with a polynomial basis
 313 built from results of a few FEM simulations. Hence, it will reduce not only the computing
 314 time but also the memory requirements for data storage.

315 In this case, for a faulty IM model with a specific degree of static eccentricity fault, the
 316 coupling parameters $[L]$ vary depending on the rotor position θ . This paper proposes the
 317 SSL strategy to select the rotor positions θ in the parametric space $[\theta_{min}, \theta_{max}]$ in which the
 318 FEM simulations have to be performed. Thereafter, the results, i.e. the coupling parameters
 319 for these specific rotor positions, will be used to compute the polynomial basis with which
 320 compute the coupling parameters $[L]$ for the remaining rotor positions.

321 When using polynomial approximation an optimal choice for the sampling is defined by
 322 the set of Gauss-Chebyshev-Lobatto (GCL) points:

$$\mathcal{P}^{(k)} \equiv \begin{cases} \{\theta_{min}, \theta_{max}\} & \text{if } k = 0 \\ \left\{ \theta_j = \left(\frac{\theta_{min} + \theta_{max}}{2} \right) \cdot \left(\cos \left(\frac{2j-1}{2^k} \pi \right) + 1 \right) \right. & \text{if } k > 0 \\ \left. \forall j = 1, \dots, 2^{k-1} \right\} \cup \mathcal{P}^{(k-1)} & \end{cases} \quad (9)$$

323 where $\mathcal{P}^{(k)}$ are the selected points for the hierarchical k level. This means that the corre-
 324 sponding set of points $\mathcal{P}^{(k)}$ for the hierarchical level k has $N^{(k)}$ elements. It implies that
 325 each level contains the $N^{(k-1)}$ elements of the previous levels plus the $N^{(k)} - N^{(k-1)}$ addi-
 326 tional points [28]. To build the polynomial basis the Lagrange interpolation is considered.
 327 Therefore, for a given hierarchical level k and $N^{(k-1)} < j < N^{(k)}$, the HLI polynomial basis
 328 is constructed as:

$$\mathcal{L}_j^k(\theta) = \prod_{\theta_i \in \mathcal{P}^{(k)}, i \neq j} \frac{\theta - \theta_i}{\theta_j - \theta_i} \quad (10)$$

329 and the coupling parameter row a column b for the rotor position θ , in the coupling param-
 330 eter's matrix with the proposed method L_{ab}^{HLI} is computed as:

$$L_{ab}^{HLI^{(k)}}(\theta) = \sum_1^k \sum_{j \in \mathcal{P}^{(k)}} \left(L_{ab}^{FEM}(\theta_j) - \sum_{i \in \mathcal{P}^{(k-1)}} \mathcal{L}_i^{k-1}(\theta_j) \right) \cdot \mathcal{L}_j^k(\theta) + \sum_{j \in \mathcal{P}^{(0)}} L_{ab}^{FEM}(\theta_j) \mathcal{L}_j(\theta) \quad (11)$$

331 As stated in section 2.1.2, for the static eccentricity fault the computation of $[L_{ss}]$ and
 332 $[L_{sr}]$ only requires the results of the FEM simulations feeding each stator phase for the first
 333 36 rotor positions which means that the rotor has to be moved between 0 to $2\pi/1008 \cdot 36 =$
 334 $\pi/14$ rad. Hence the parametric space is defined as $[0, \pi/14]$ rad. In the same way, to
 335 compute the coupling parameters associated to rotor phases (or rotor bars) one must feed
 336 a rotor phase and performing the magneto-static FEM simulation for the first 504 positions
 337 which implies the parametric space $[0, \pi]$ rad. Tables 2 and 3 show the set of GCL points,
 338 i.e. the rotor positions θ , for the different hierarchical levels k in which the magneto-static
 339 FEM simulation should be performed. Thereafter, the polynomial basis according to (11) is
 340 computed with the results of FEM simulations with the rotor placed in the positions of
 341 Tables 2 and 3 and used to compute the coupling parameters between stator phases $[L_{ss}]$,
 342 between stator phases and rotor bars $[L_{sr}]$ and between rotor bars $[L_{rr}]$ respectively.

343 However, to compute the polynomial basis for the case of coupling parameters between
344 rotor bars a deeper analysis is required. Figure 4 shows the coupling parameters between
345 the 1st rotor bar with itself based on FEM simulations. It can be seen as these coupling
346 parameters can be computed as a sum of two functions. Due to the slot effect, there
347 is a first function with $\pi/14$ period associated to the movement of a rotor bar through
348 a stator slot. On the other hand, due to the static eccentricity fault, there is a second
349 function with a 2π period that behaves similarly to the air-gap width depending on the
350 rotor position. It can be seen that the higher fault severity the higher the amplitude of this
351 second function. Therefore, as the fault severity degree increases this 2π period function
352 becomes more prominent whereas for low severity degrees the slot effect is more prominent.
353 Thus, both effects have to be considered to accurately compute the coupling parameters
354 between rotor bars. This fact suggests that two parametric spaces should be considered to
355 obtain the polynomial basis to compute the rotor rotor coupling parameters: $[0, \pi/14]$ rad
356 to include the slot effect, $[0, \pi]$ rad to consider the effect of the static eccentricity fault.
357 To better illustrate this fact, Figure 5 compares the coupling parameters obtained with
358 FEM analysis between those obtained with the polynomial basis obtained in: the $[0, \pi]$ rad
359 parametric space, the $[0, \pi/14]$ rad parametric space or as proposed, i.e., as a combination
360 of both polynomial basis. It can be seen, that the proposed polynomial combination yields
361 better results and reflects more accurately the coupling parameters if FEM analysis was
362 performed but a lower cost.

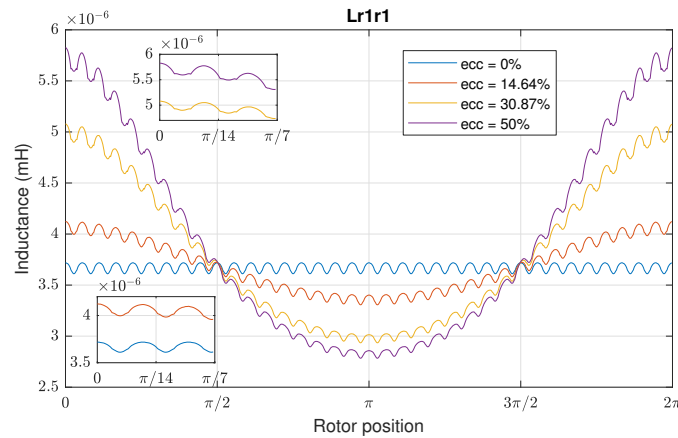


Figure 4: Coupling parameters between rotor bar 1 and itself for different rotor positions for four different degrees of static eccentricity fault. These coupling parameters have been computed through FEM simulations. Zoom of the coupling parameters have been included to show the effect of the slots.

363 3.3. Results

364 To check the effectiveness of the proposed method different polynomial basis have been
365 computed considering the different hierarchical levels 0 to 3 of Tables 2 and 3. Considering
366 the polynomial basis obtained for each hierarchical level k , the coupling parameter matrix,
367 $[L]$, has been computed and compared with the results obtained using the FEM simulations

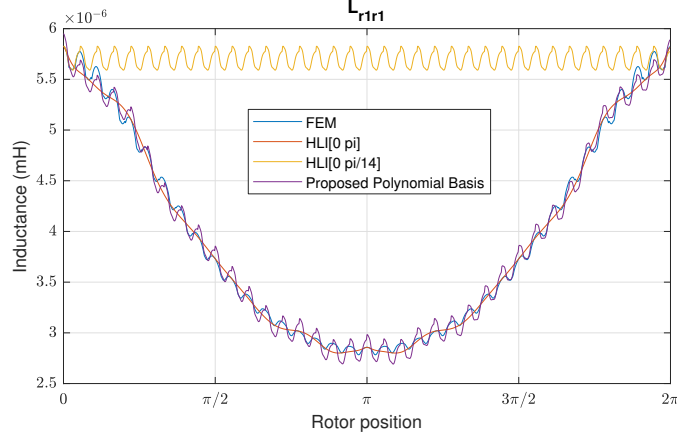


Figure 5: Comparison of coupling parameters between rotor bar 1 and itself for different rotor position and for a static eccentricity of 50% computed with FEM, HLI considering the parametric space $[0, \pi]$ rad, HLI considering the parametric space $[0, \pi/14]$ rad and with the polynomial basis proposed as a combination of the HLI obtained with both parametric spaces $[0, \pi/14]$ rad and $[0, \pi]$ rad.

Table 2: Set of the GCL points in the parametric space $[0, \pi/14]$ to compute $[L_{ss}]$, $[L_{sr}]$ and include the slot effect in $[L_{rr}]$.

Level (k)	Rotor position θ ($\mathbf{P}^{(k)}$) rad	Level (k)	Rotor position θ ($\mathbf{P}^{(k)}$) rad
0	0	3	0.0085
0	$\pi/14 = 0.2244$	3	0.0693
1	$\pi/28 = 0.1122$	3	0.1551
2	0.0329	3	0.2159
2	0.1915		

368 and the procedure depicted in Figure 3 particularized for the case of static eccentricity fault
369 described in section 2.1.2.

370 Figure 6 shows the evolution of three elements of the coupling parameters matrix $[L]$
371 depending on the rotor position: between the stator phase 1 with itself $[L_{s_1s_1}]$, between the
372 stator phase 1 and the rotor bar 1 $[L_{s_1r_1}]$ and between the rotor bar 1 with itself $[L_{r_1r_1}]$
373 for three different levels of static eccentricity (0% (Healthy), 14.64% and 50%). This figure
374 compares the results of computing the coupling parameters using the full FEM analysis with
375 the results obtained with the proposed method considering different hierarchical levels k to
376 compute the polynomial basis $L^{HLI^{(k)}}$ according to (11). It must be highlighted that the
377 proposed method uses, for each hierarchical level k , just the results of the FEM simulations
378 for the rotor positions in the set of the GCL points shown in Tables 2 and 3.

379 To check the accuracy of the proposed method the error between the coupling parameters
380 computed with FEM L^{FEM} and with the proposed method $L^{HLI^{(k)}}$ for each hierarchical level
381 k is computed as:

Table 3: Set of the GCL points in the parametric space $[0, \pi]$ to include the static eccentricity effect in the $[L_{rr}]$ computation.

Level (k)	Rotor position θ ($\mathbf{P}^{(k)}$) rad	Level (k)	Rotor position θ ($\mathbf{P}^{(k)}$) rad
0	0	3	0.1196
0	π	3	0.9697
1	$\pi/2$	3	2.1719
2	0.4601	3	3.0220
2	2.6815		

$$\text{error}(\%) = \text{mean} \left| \frac{L^{FEM}(\theta) - L^{HLI^{(k)}}(\theta)}{L^{FEM}(\theta)} \right| \cdot 100 \quad (12)$$

382 Figure 6 bottom shows the mean error obtained in the computation of the three coupling
383 parameters ($L_{s_1 s_1}$, $L_{s_1 r_1}$, $L_{r_1 r_1}$) for the three different levels of static eccentricity depending
384 on the hierarchical level k considered. It can be seen that, as the hierarchical level k increases
385 the accuracy improves due to the more FEM simulations used to obtain the polynomial basis.
386 However, it can be also seen that the level $k = 2$ could be enough to compute the coupling
387 parameters. For this hierarchical level the error is less than the 2% and the use of more
388 points (higher hierarchical level) does not significantly improves the accuracy but adds more
389 computational effort.

390 This, in turn, implies that just with results of the FEM simulations for the rotor placed
391 in the five positions for the hierarchical levels $k = 0$ to $k = 2$ of Table 2 is enough to compute
392 the coupling parameters matrix. Additionally, for the rotor-rotor coupling parameters the
393 positions corresponding to levels $k = 0$ to $k = 2$ of Table 3 are also required to build the
394 two polynomial basis aforementioned.

395 Table 4 shows the computational costs in terms of FEM simulations, computation time
396 and memory resources for data storage for the generic case (according to the procedure shown
397 in subsection 2.1), the generic case particularized for the static eccentricity fault (according
398 the procedure shown in subsection 2.1.2) and for the proposed method (depicted in section
399 3). In view of the results, it must be highlighted that the proposed method computes the
400 coupling parameters 24.48 times faster and requires just one 4.09% of the memory resources
401 for the generic case particularized for the static eccentricity fault (Table 4: 3rd column) and
402 keeping the accuracy better than 2%.

403 4. Simulation

404 Once the coupling parameters are computed, they are used in the model shown in Fig-
405 ure 2. The model has been implemented in the HIL OP4500 whose main characteristics
406 are detailed in the Appendix Appendix A. The HIL runs the model in real time and the
407 stator currents as well as other other signals such as the speed can be sampled in real time

Table 4: Computational effort and memory resources needed to obtain the coupling parameters of a faulty IM model in a generic case (1st column), particularizing for the static eccentricity fault (2nd column) and with the proposed method (3rd column).

	Generic case	Static eccentricity	Proposed method
n^{er} FEM simulations	17, 136	612	25
Computation time	11 days 21 hours 36 min	10 hours 12 min	25 min
Data storage	376.52 GB	13.45 GB	0.55 GB

408 through the analog outputs. Hence, as shown in Figure 7 these signals can be used to test
 409 fault diagnosis techniques implemented in embedded devices, to train and test CM systems
 410 based on artificial neural network (ANN) or to be acquired through a digital oscilloscope
 411 and processed offline in a pc system to develop other fault diagnosis techniques.

4.1. Fault diagnosis using the proposed IM running in a HIL simulator

412 Transient based fault diagnosis methods have attracted a rising interest due to their
 413 reliability. They can be used to detect faults in wide variety of working conditions such
 414 as oscillating loads, inverter-fed motors with changes of speed, start-up transients, supply
 415 variations, etc. Hence, they allow to reduce misdiagnoses generally associated with steady
 416 state fault diagnosis techniques due to situations that could be confused with faults [1, 35,
 417 36, 37, 38, 39]. Indeed, the current analysis during the IM start-up transient is widely used
 418 because the evolution of the slip is well-known (from 1 to $\simeq 0$) and it is possible to identify
 419 the different patterns followed by the fault harmonic components. Moreover, it does not
 420 require the speed measurement.
 421

4.2. Evolution of the static eccentricity fault harmonic components

422 For the static eccentricity fault the frequencies of the harmonic components are defined
 423 by (1). Considering the specific parameters of the motor defined in Table 1 the evolution
 424 of the fault harmonic components during the start-up transient can be easily computed as
 425 the slip evolves from 1 to $\simeq 0$. Consequently, for the lower side harmonic (LSH), $\nu = -1$,
 426 its frequency evolves from 50 Hz to $\simeq 650$ Hz. Similarly, the upper side harmonic (USH),
 427 $\nu = 1$, evolves from 50 Hz to $\simeq 750$ Hz. Besides, the amplitude of the LSH remains almost
 428 constant regardless the fault severity degree while in case of the USH its amplitude increases
 429 as the severity degree increases [40].
 430

431 Therefore, to check the usability of the parameters computed with the proposed method,
 432 in terms of fault diagnosis purposes, three different simulations, with three degrees of static
 433 eccentricity (Healthy, 30% and 50%) fault have been performed. The simulations have been
 434 performed during the start-up transient and the currents have been sampled during 2 seconds
 435 from the analogue outputs of the HIL, OPAL OP4500, at a sampling frequency of 10kHz.

436 The main objective is to compare the results obtained from the model using the coupling
437 parameters computed with the proposed method and with the coupling parameters fully
438 computed with the traditional FEM analysis following the procedure shown in Figure 3.
439 For fault diagnosis purposes, these results have to be compared in terms of the evolution of
440 the static eccentricity fault harmonic components during the start-up transient. Figure 8
441 shows the spectrogram of the stator current for the simulated machine with the coupling
442 parameters computed with the proposed method and compared with those obtained using the
443 coupling parameters computed with FEM. As can be seen, the LSH appears regardless the
444 machine status and it is a reason why this component is used for speed prediction in some
445 control drives. On the other hand, the USH component appears only when the machine is
446 under faulty conditions and it becomes more clear as the severity degree increases. As can
447 be seen, these results reinforce the validity of the proposed method as it obtains the same
448 results, with minor errors, as those obtained with the traditional FEM analysis.

449 5. Experimental Validation

450 The proposed method has been validated with a commercial 1.1kW IM (whose main
451 characteristics are the same as the simulated model and shown in the Table 1) to observe
452 the presence of the fault harmonic components. The experimental set up is shown in Figure 9.
453 To achieve longer startup transients the IM has been fed to reduced voltage through an
454 auto-transformer and no external load has been used. To introduce the static eccentricity
455 fault the hood fastening holes have been slightly enlarged to achieve a small tolerance in the
456 positioning of the rotor axis as shown in Figure 10. The stator currents have been sampled
457 using a digital oscilloscope during 10 seconds at a sampling rate of 10 kHz.

458 The stator current spectrogram for the IM in the same conditions as bought (considered
459 as healthy) and for the IM faulty machine are shown in Figure 11 where the LSH and USH
460 harmonic components are highlighted. As can be seen, for the healthy machine the USH is
461 also visible as IMs are not ideal and each IM has an inherent eccentricity that should be
462 lower than 10% as stated in [10]. However, it should be highlighted that, as the fault severity
463 degree increases (faulty machine) the amplitude of the USH also increases as shown in the
464 simulation results and confirmed by this experimental validation. Therefore, tracking the
465 evolution of this fault harmonic component could be a useful tool for condition monitoring
466 of IM.

467 6. Conclusions

468 This paper proposes the sparse identification to reduce the computational effort required
469 to compute the coupling parameters of a FEM-analytical model of a faulty IMs. The
470 proposed method achieves a significant improvement in the computing time and in the
471 memory resources while keeping a good accuracy. In fact, it is 24.48 times faster computing
472 the coupling parameters and requires just the 4.09% of the memory resources than a full
473 FEM analysis. Besides, the actual error between the coupling parameters computed with
474 FEM and with the proposed method is less than 2%. The proposed method takes into

475 account the special effects of the geometry of the induction motor such as the slot effect or
476 the winding effect. Moreover it can consider the variations the air-gap width due to a fault.

477 In this paper, the proposed method has been illustrated for the static eccentricity fault
478 with different fault degrees from incipient levels or even inherent levels of the machine
479 to more severe scenarios which would help in the aim of correlating the amplitude of the
480 fault harmonic components with the fault severity degree. After that, the models have
481 been tested with the coupling parameters computed and the fault diagnosis results (fault
482 harmonic components) have been compared with those obtained with the models where the
483 coupling parameters have been fully computed with FEM software. Besides, these results
484 have been confirmed by experimental validation. Following the same reasoning the method
485 could be extended to other types of faults or even to simultaneous faults.

486 Finally, it must be highlighted that the proposed method would have a major impact
487 in the fast development of IM models with different types of fault and/or with different
488 degrees of severity of a given fault. Therefore the method will favour the development of
489 fault diagnosis systems, especially on-line fault diagnosis system and AI based systems as
490 it will help not only to cover a wide variety of scenarios (machines, degrees of severity and
491 types of fault, working conditions) but also in establishing thresholds for the early detection
492 of a given fault.

493 **Appendix A. OPAL 4500 main features:**

494 Real-time target: 4 INTEL processor cores 3.3 GHz (only 1 core activated). Solid state
495 disk: 125 Gb. Memory RAM: 4 Gb. Real-time operating system: Linux RedHat. Xilinx
496 Kintex 7 FPGA (326.000 Logic cells and 840 DSP slice). Sampling Rate: 200MHz. 96 User
497 Inputs/Outputs (I/O): 16 analog inputs and 16 analog outputs, 24 digital inputs and 24
498 digital outputs, 8 RS422 digital inputs and 8 RS422 digital outputs.

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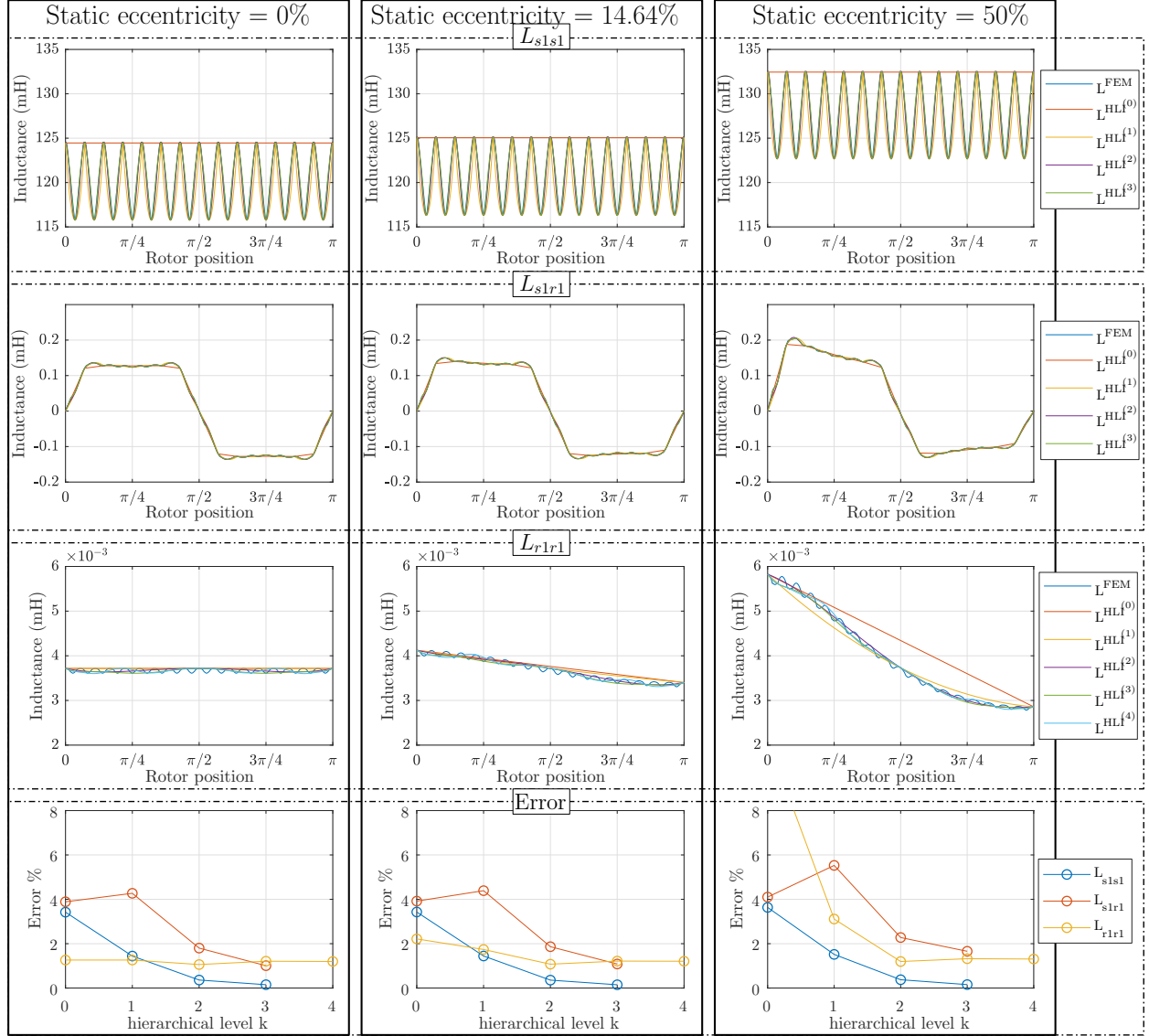


Figure 6: The figure shows the coupling parameters computed with FEM analysis and with the proposed method, L^{HLI^k} where k is the hierarchical level of the polynomial basis defined in (11) and computed with the GCL points of tables 2 and 3. From top to bottom: coupling parameters between stator phase 1 with itself, $L_{s_1s_1}$, between stator phase 1 and rotor bar 1, $L_{s_1r_1}$ and between rotor bar 1 with itself $L_{r_1r_1}$ for three different degrees of severity of static eccentricity fault (0% - Healthy, 14.64% and 50%). Finally, the bottom figure shows the error, computed according (12), committed in the computation of each coupling parameters with the proposed method depending on the hierarchical level k of the polynomial basis used.

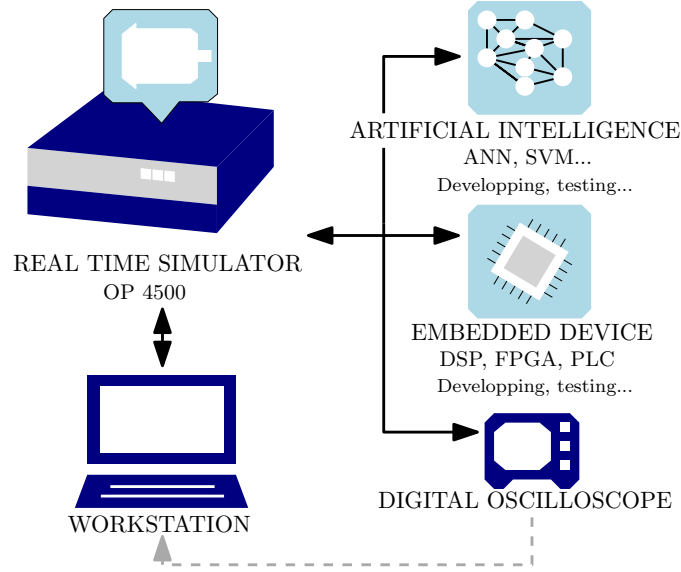


Figure 7: The hybrid FEM- analytical model developed is transferred to the real time simulator. The real-time signals needed (stator currents, speed) are connected to the analogue outputs of the HIL and used for different fault diagnosis purposes: to develop, train and test either AI based or continuous on-line (embedded devices) fault diagnosis systems. On the other hand, these signals can also be acquired through digital oscilloscope or directly transferred to the pc-station for further processing.

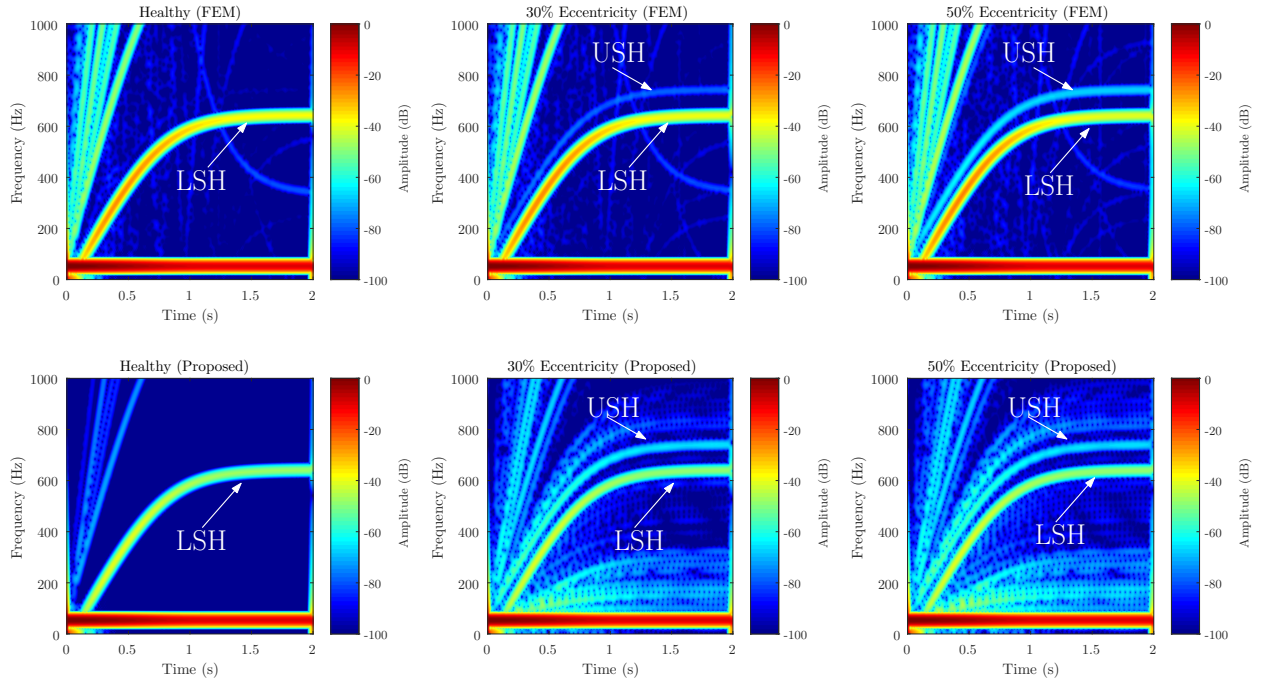


Figure 8: Stator current spectrogram for the three different levels of static eccentricity using the proposed method ($HLI^{(2)}$) and FEM software to compute the coupling parameters of the hybrid FEM-analytical model. The fault harmonic components LSH and USH due to static eccentricity fault have been highlighted for both methods.

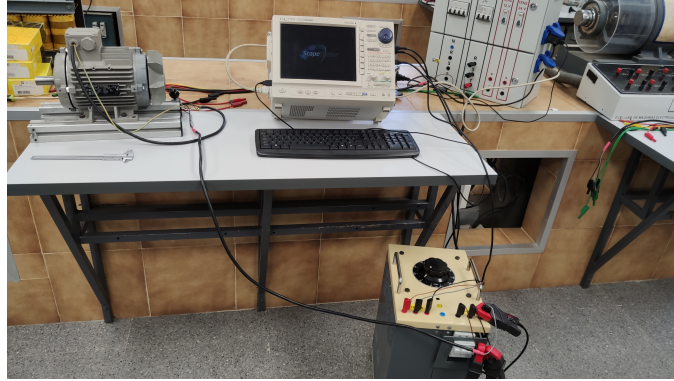


Figure 9: Test bed used for the experimental validation. The motor is fed through an autotransformer to a low voltage to achieve a longer start-up transient. The stator currents have been sampled using digital oscilloscope with the aid of current clamps.

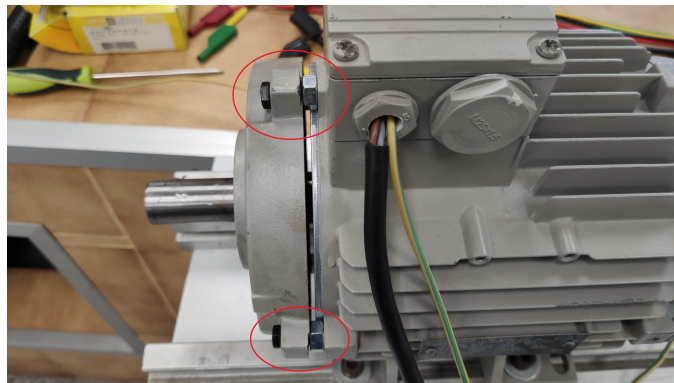


Figure 10: Detail of the hood fasten holes drilled to introduce the static eccentricity fault.

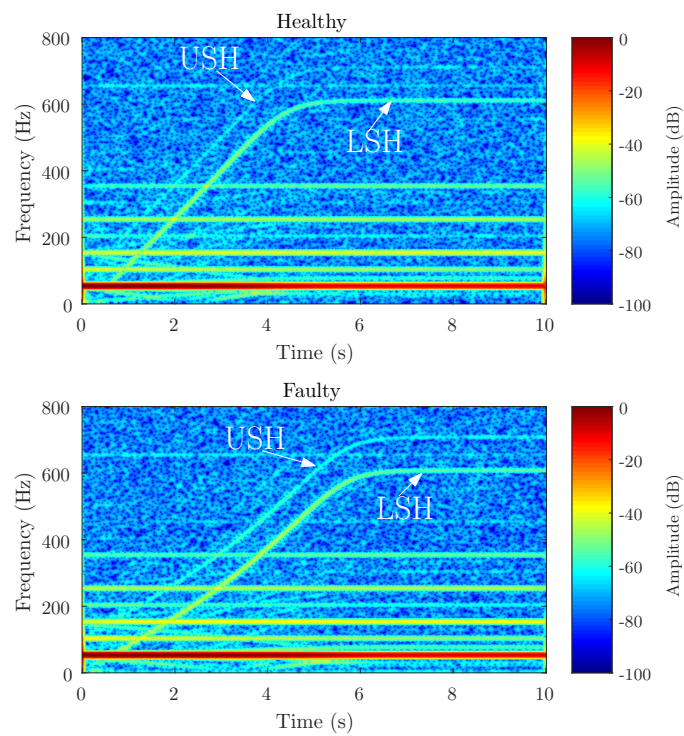


Figure 11: Stator current spectrogram for the IM in the same conditions as it was bought, i.e. in healthy conditions (top) and with the static eccentricity fault (bottom). It can be seen that the start up transient lasts 5 seconds and can be seen both USH and LSH fault harmonic components. It must be highlighted as the USH increases its amplitude as the fault severity degree increases as shown in the simulation results.