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Controller Tuning by means of Multi-objective Optimization Algorithms: a Global Tuning Framework

Gilberto Reynoso-Meza, Sergio García-Nieto, Javier Sanchis, and Xavier Blasco

Abstract

A holistic multi-objective optimization design technique for controller tuning is presented. This approach gives control engineers greater flexibility to select a controller that matches their specifications. Furthermore, for a given controller it is simple to analyse the trade-off achieved between conflicting objectives. By using the multi-objective design technique it is also possible to perform a global comparison between different control strategies in a simple and robust way. This approach thereby enables an analysis to be made of whether a preference for a certain control technique is justified. This proposal is evaluated and validated in a non-linear MIMO system using two control strategies: a classical PID control scheme and a feedback state controller.

Index Terms

multiobjective optimization, controller tuning, pid tuning, evolutionary algorithm, decision making.

ACRONYMS

DM	Decision maker
EA	Evolutionary algorithm
GPP	Global physical programming
IADU	Integral of the absolute value of the derivative control signal

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22	IAE	Integral of the absolute value of the error
23	ISA	Instrumentation, systems and automation society
24	LD	Level diagram
25	MIMO	Multiple-input multiple-output
26	MOEA	Multi-objective evolutionary algorithm
27	mood4ct	Multi-objective optimisation design for controller tuning
28	MOO	Multi-objective optimisation
29	PI	Proportional-integral
30	PID	Proportional-integral-derivative
31	SISO	Single-input single-output
32	SS	State space
33	TITO	Two-input two-output
34	TRMS	Twin rotor MIMO system

35

36

I. INTRODUCTION

37 Satisfying a set of specifications and constraints required by real-control engineering prob-
38 lems is often difficult with traditional optimization approaches. From the control point of view
39 it is common to face a variety of requirements and specifications. These range from time-
40 domain specifications (such as maximum overshoot, settling time, steady state error, raise time)
41 to frequency-domain requirements (noise rejection or multiplicative uncertainty, for example).
42 Furthermore, constraints such as saturations, or the maximum changes enabled for a control
43 signal may be considered. Such problems, when multiple objectives must be fulfilled, are known
44 as multi-objective problems.

45 A traditional approach for solving a multi-objective problem is to translate it into a single-
46 objective problem using weighting factors to indicate the relative importance among objectives
47 (see for example [1]). The solution obtained strongly depends on which factors are used, and
48 it is not usually a trivial task to select the right weighting vector to assure a quality solution
49 with a reasonable trade-off among objectives [2]. This situation may be more complicated when
50 constraints are considered. More complex methods to tackle these issues have been developed
51 [3], such as lexicographic methods, goal programming methods or physical programming [4].

52 Multi-objective optimization (MOO) can handle these issues in a simple manner because of its
 53 simultaneous optimization approach. In MOO, all the objectives and constraints are significant
 54 from the designer point of view, and as a consequence, each is optimized to obtain a set of
 55 optimal non-dominated solutions. The MOO approach offers to the designer a set of solutions,
 56 a Pareto set approximation, where all the solutions are Pareto-optimal [3]. This set of solutions
 57 offers the decision maker (DM) greater flexibility. The role of the designer is to select the most
 58 preferable solution according to her/his needs and preferences for a particular situation.

59 There are several widely used algorithms for calculating this Pareto set approximation (normal
 60 boundary intersection method [5], normal constraint method [6], and successive Pareto front
 61 optimization [7]). Recently, multi-objective evolutionary algorithms (MOEAs) have started to be
 62 used because of their flexibility in dealing with non-convex and highly constrained functions
 63 [8], [9]. Some examples include NSGA-II [10], MOGA [11], ev-MOGA [12], pa ϵ -MyDE [13],
 64 and sp-MODE [14]. General methodologies for MOO have been developed [15]; nevertheless
 65 new approaches and methodologies using MOO are still required focusing on controller tuning.

66 In this work, a holistic MOO design technique using MOEA's is presented for controller tuning
 67 purposes. In Section II a review on MOO is given and in Section III the MOO approach for
 68 controller tuning (*moodAct*) is presented. In Section IV an engineering application example is
 69 developed and experimentally evaluated and discussed. Finally, some concluding remarks and
 70 future work are given.

71 II. MULTI-OBJECTIVE OPTIMIZATION REVIEW

72 A MOO problem, without loss of generality,¹ can be stated as follows:

$$\min_{\boldsymbol{\theta} \in \mathfrak{R}^n} \boldsymbol{J}(\boldsymbol{\theta}) = [J_1(\boldsymbol{\theta}), \dots, J_m(\boldsymbol{\theta})] \in \mathfrak{R}^m \quad (1)$$

73 where $\boldsymbol{\theta} \in \mathfrak{R}^n$ is defined as the decision vector, and \boldsymbol{J} as the objective vector. In general,
 74 there is no single solution because there is no solution that is better than the others for all the
 75 objectives. Therefore, a set of solutions, the Pareto set Θ_P , is defined and its projection into the

¹A maximization problem can be converted to a minimization problem. For each of the objectives that have to be maximized, the transformation $\arg \max_{\boldsymbol{\theta}} J_i(\boldsymbol{\theta}) = \arg \min_{\boldsymbol{\theta}} (-J_i(\boldsymbol{\theta}))$ can be applied.

76 objective space is known as the Pareto front \mathbf{J}_P (see Figure 1). Each point in the Pareto front
 77 is said to be a non-dominated solution (see Figure 2). A given solution θ^1 dominates a second
 78 solution θ^2 only if θ^1 has a better or equal cost value for all objectives (with, at least, one cost
 79 value being better).

80 **Definition** (Dominance relation): given a solution θ^1 with cost function value $\mathbf{J}(\theta^1)$, it domi-
 81 nates a second solution θ^2 with cost value $\mathbf{J}(\theta^2)$ if and only if:

$$\begin{aligned} & \{\forall i \in [1, 2, \dots, m], J_i(\theta^1) \leq J_i(\theta^2)\} \\ & \quad \wedge \\ & \{\exists q \in [1, 2, \dots, m] : J_q(\theta^1) < J_q(\theta^2)\} \end{aligned}$$

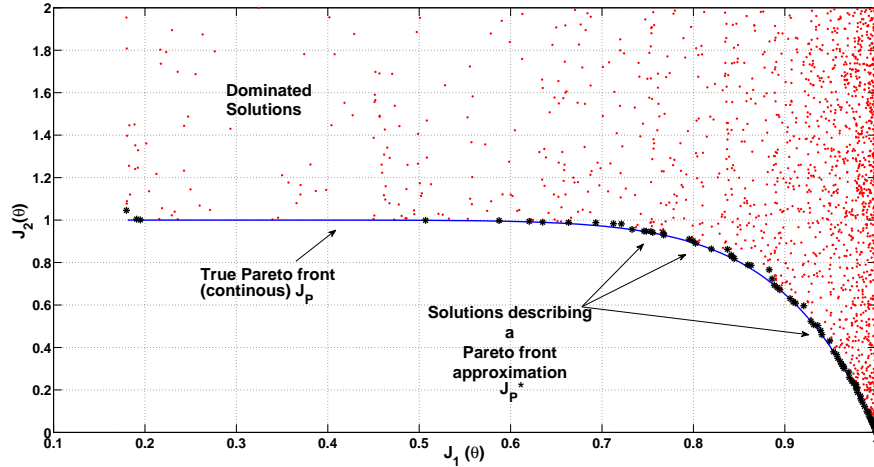


Fig. 1: Pareto front concept (example of a two objective optimization problem).

85 MOO techniques search for a discrete approximation Θ_P^* of the Pareto set Θ_P with a good
 86 description \mathbf{J}_P^* of the Pareto front. In this way, the DM has a set of solutions for a given problem
 87 and more flexibility for choosing a particular or desired solution.

88 III. MULTI-OBJECTIVE OPTIMIZATION DESIGN APPROACH FOR CONTROLLER TUNING

89 As a global framework, three main objectives need to be considered in a controller's tuning pro-
 90 cedure: performance, robustness and implementation issues. Usually, classical controller tuning
 91 techniques have been developed for only one of those objectives. Other tuning techniques are able

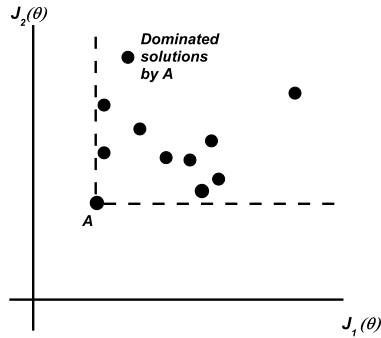


Fig. 2: Dominance concept. Solution A has a better cost value for all objectives.

92 to deal with these objectives. For example, H_2/H_∞ designs (or mixed-sensitivity techniques) have
 93 been shown to be powerful tools to address the trade-off between performance and robustness.
 94 However it is not easy to include constraints in the control and/or process variables and the
 95 performance objective interpretability could be lost. Strategies as Model Predictive Control [16]
 96 deal with this problem solving an optimization statement in each sampling time. A quadratic
 97 measure is usually used, whereas an absolute error measurement could be helpful to the designer
 98 for interpreting the performance of a proposed controller. However, useful or interpretable
 99 objectives considered by the DM could lead to complex non-convex and highly constrained
 100 cost functions.

101 Evolutionary algorithms (EAs) are a flexible tool for handling non-convex cost functions that
 102 are highly constrained in decision and objective spaces. They have been successfully applied in
 103 several control engineering areas [17] such as controller tuning [18], PI-PID tuning [19]–[21],
 104 multivariable control [22]–[26], and fuzzy control [27]–[30]. These algorithms have also been
 105 merged together with predictive control [31], H_∞ techniques [32], [33], linear matrix inequalities
 106 [34], and loop shaping [35]. The use of such a class of algorithms leads to a higher degree of
 107 flexibility, since more interpretable objectives can also be used to tune any kind of controller.
 108 Therefore, a multi-objective optimization design for controller tuning (*mood4ct*) by means of
 109 evolutionary algorithms will be proposed. Any multi-objective optimisation design approach
 110 must follow three main steps: problem definition, multi-objective optimisation process and
 111 decision making stage (see figure 3). The main contribution of this work consists in define
 112 a global optimisation problem statement for multivariable processes and its integration into the

113 optimisation procedure and the decision making stage (which is not a trivial task when the
 114 number of objectives is three or more). Any kind of MOEA can be used (NSGA-II [10]²,
 115 MOGA [15], [36]³, ev-MOGA [12]⁴, paε-MyDE [13], and sp-MODE [14], among others). Such
 116 algorithm must be capable of converging towards the Pareto front; it must have a good constraint
 117 handling mechanism and it must compute a useful well-spread approximation along the Pareto
 118 front.

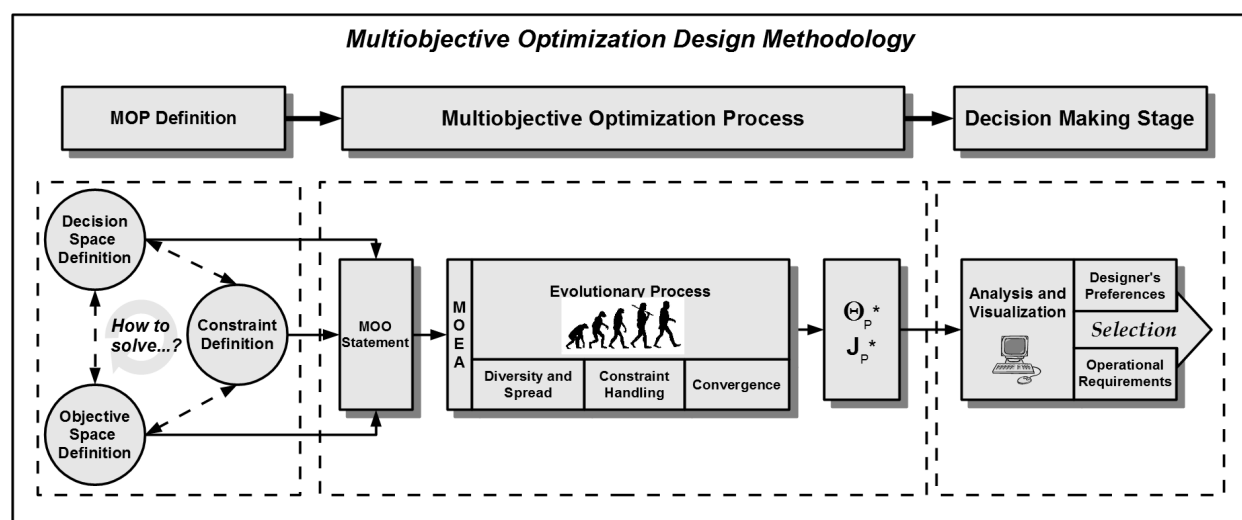


Fig. 3: Multi-objective optimisation design methodology.

119 The *mood4ct* approach, roughly speaking, is based on:

- 120 • A highly reliable process model to obtain a measurement of the performance for a given
 121 controller.
- 122 • Meaningful process objectives to facilitate the decision making stage.
- 123 • A MOEA with a constraint handling mechanism which can assure convergence, spread and
 124 diversity into the Pareto front.
- 125 • An intuitive and easy-to-use tool to analyze m -dimensional Pareto fronts.

²Source code available at: <http://www.iitk.ac.in/kangal/codes.shtml>; also, a variant of this algorithm is available in the global optimization toolbox of Matlab.

³Genetic Algorithm toolbox for Matlab available at <http://www.sheffield.ac.uk/acse/research/ecrg/gat>

⁴Available for Matlab at: <http://www.mathworks.com/matlabcentral/fileexchange/31080>

126 *A. Process objectives*

127 The use of a process model will lead to a higher degree of reliability for the controller's
 128 performance under practical considerations such as saturation, complex tracking references,
 129 and/or any kind of constraint. In this work, the integral of the absolute magnitude of the error
 130 (IAE) and the integral of the absolute value of the derivative control signal (IADU) are used
 131 due to their interpretability. Given a model, which will be controlled with a sampling time of
 132 T_s with $t \in [t_0, t_f]$ and with controller tuning parameters θ , the IAE and IADU are defined as:

$$IAE(\theta) = T_s \sum_{k=1}^N |r_k - y_k| \quad (2)$$

$$IADU(\theta) = \sum_{k=1}^N |u_k - u_{k-1}| \quad (3)$$

133 Where r_k , y_k and u_k are respectively the setpoint signal, the controlled and manipulated
 134 variables at sample k ; while N is the number of samples in $[t_0, t_f]$. The above mentioned
 135 objectives are defined for a SISO system. If a MIMO system with ρ inputs and ν outputs is
 136 under consideration, it is possible to have as many objectives IAE , $IADU$ as inputs and outputs.
 137 Nevertheless, this could lead to an exponential increase in the number of solutions in the Pareto
 138 front \mathbf{J}_P^* , and the analysis on the results could be more difficult. Moreover, a large subset of
 139 solutions will probably be undesirable for the DM (for example, controllers with an outstanding
 140 performance in one controlled variable at the expense of another). So, it is worthwhile trying
 141 to reduce the objective space to facilitate the analysis for the DM without losing any of the
 142 advantages of the MOO approach [37]. Let it be:

$$\mathbf{J}_E(\theta) = \left[\frac{IAE^{1,1}(\theta)}{\Delta R^1}, \frac{IAE^{2,2}(\theta)}{\Delta R^2}, \dots, \frac{IAE^{\nu,\nu}(\theta)}{\Delta R^\nu} \right] \quad (4)$$

$$\mathbf{J}_U(\theta) = \left[\sum_{j=1}^{\nu} \frac{IADU^{1,j}(\theta)}{\Delta U_{max}^1}, \sum_{j=1}^{\nu} \frac{IADU^{2,j}(\theta)}{\Delta U_{max}^2}, \dots, \sum_{j=1}^{\nu} \frac{IADU^{\rho,j}(\theta)}{\Delta U_{max}^\rho} \right] \quad (5)$$

143 Where $IAE^{i,j}(\theta)$ is the $IAE(\theta)$ for controlled variable i when there is a setpoint change
 144 ΔR^j for controlled variable j ; $IADU^{i,j}(\theta)$ is the $IADU(\theta)$ for control signal i when there is
 145 a change in setpoint signal j , and ΔU_{max}^i is the maximum change allowed for control signal i .
 146 Vectors 4 and 5 contain the IAE and IADU values for each variable normalized over a work

147 range. Because of this, it is possible to perform a comparison between controlled variables and
 148 between control signals.

149 Define a sorting function $\mathcal{Z} : \mathfrak{R}^{1 \times n} \rightarrow \mathfrak{R}^{1 \times n}$, $\mathcal{Z}(\mathbf{f}) = \mathbf{g}$ so that: $\mathbf{g} = [a_1, a_2, a_3, \dots, a_n]$, where
 150 $a_1 \geq a_2 \geq a_3 \geq \dots a_n$, where each a_i is an element of \mathbf{f} . The global index for IAE and IADU
 151 performance measurements are defined as $J_{\mathcal{E}}(\boldsymbol{\theta})$ and $J_{\mathcal{U}}(\boldsymbol{\theta})$ respectively:

$$J_{\mathcal{E}}(\boldsymbol{\theta}) = \mathcal{Z}(\mathbf{J}_{\mathcal{E}}(\boldsymbol{\theta})) \times \mathbf{w} \quad (6)$$

$$J_{\mathcal{U}}(\boldsymbol{\theta}) = \mathcal{Z}(\mathbf{J}_{\mathcal{U}}(\boldsymbol{\theta})) \times \mathbf{w} \quad (7)$$

152 Vector \mathbf{w} indicates it is most important to optimize the maximum value, thereby assuring a
 153 minimum worst performance for all objectives. As inputs and outputs are usually normalized in
 154 the range $[0, 1]$ an intuitive value ⁵ for \mathbf{w} is $\mathbf{w} = [10^0, 10^{-2}, \dots, 10^{-n}]^T$.

155 Please note that this objective reduction is important to facilitate the decision making step. In
 156 one hand, the multi-objective approach gives to the DM a better insight concerning the objective
 157 trade-offs; in the other hand, too much information (too many objectives) can hinder the DM
 158 task to select a desired solution. This topic, known as many-objectives optimization (usually
 159 more than 4 objectives) is not trivial, and some algorithms could face several problems due to
 160 their diversity improvement mechanisms [38], [39]. The objective reduction is an alternative to
 161 face the many-objectives optimization issue [40], and with this proposal the relevant information
 162 about the conflict between control actions and performance is retained.

163 Additionally, a measurement for coupling effects is required:

$$\mathbf{J}_C(\boldsymbol{\theta}) = \left[\max_{i \neq 1} \frac{IAE^{1,i}(\boldsymbol{\theta})}{\Delta R_{max}^i}, \max_{i \neq 2} \frac{IAE^{2,i}(\boldsymbol{\theta})}{\Delta R_{max}^i}, \dots, \max_{i \neq \nu} \frac{IAE^{\nu,i}(\boldsymbol{\theta})}{\Delta R_{max}^i} \right], i \in [1, 2, \dots, \nu] \quad (8)$$

$$J_C(\boldsymbol{\theta}) = \mathcal{Z}(\mathbf{J}_C(\boldsymbol{\theta})) \times \mathbf{w} \quad (9)$$

164 Where ΔR_{max}^i is the maximum allowable setpoint step change for controlled variable i .

⁵Notice that setting $\mathbf{w} = [1, 0, \dots, 0]$ is equivalent to set $J_{\mathcal{E}}(\boldsymbol{\theta}) = \|\mathbf{J}_{\mathcal{E}}(\boldsymbol{\theta})\|_{\infty}$. Nevertheless, any MOEA would not be able to differentiate, for example, between one solution $\mathbf{J}_{\mathcal{E}}(\boldsymbol{\theta}^1) = [0.9, 0.9, 0.9, 0.9, 0.9]$ with $\mathcal{Z}(\mathbf{J}_{\mathcal{E}}(\boldsymbol{\theta}^1)) \times \mathbf{w} = 0.9$ from another one $\mathbf{J}_{\mathcal{E}}(\boldsymbol{\theta}^2) = [0.9, 0.5, 0.01, 0.5, 0.7]$ with $\mathcal{Z}(\mathbf{J}_{\mathcal{E}}(\boldsymbol{\theta}^2)) \times \mathbf{w} = 0.9$. The latter should be preferred over the former.

165 Finally, it is not possible to rely only on the process model, due to un-modeled dynamics
 166 or parametric uncertainty. Therefore, a robustness objective is required to guarantee a robust
 167 stability. One possible choice is to use complementary sensitivity function $\mathcal{T}(s)$ with a linearized
 168 process model as follows:

$$J_{\mathcal{T}} = \sup_{\omega} \bar{\sigma}(\mathcal{T}(j\omega)W(j\omega)), \omega \in (\underline{\omega}, \bar{\omega}) \quad (10)$$

169 Usually $\mathcal{T}(s)$ together with weighting function $W(s)$ is stated as a hard constraint ($J_{\mathcal{T}} < 1$).
 170 Since $W(s)$ selection is not a trivial task [41], the *mood4ct* approach can manage this task as an
 171 optimization objective (*i.e.*, it will be minimized instead of being used as a hard constraint). The
 172 *mood4ct* can deal with constraints in the same way it deals with each objective and represents a
 173 feasible alternative to constraint-handling [42], [43]. This approach, combined with an adequate
 174 tool to analyze m -dimensional Pareto fronts, is useful to analyze the impact of relaxing, if
 175 possible, one or more constraints.

176 With the above mentioned objectives, it is possible to build a MOO statement to adjust any
 177 kind of parametric controller (see eq. 11). That is, given a control structure with numerical
 178 parameters to adjust, the latter MOO problem can be stated, using as performance measurement
 179 information from the simulation process. The objectives cover the most important requirements
 180 for a controller: performance, control effort, coupling effects and robustness. Although these
 181 performance measurements have been proposed as first approximation, some other measures can
 182 be used (or added) by the DM.

$$\min_{\theta \in \mathbb{R}^n} \mathbf{J}(\theta) = \left[J_{\mathcal{E}}(\theta), J_{\mathcal{U}}(\theta), J_{\mathcal{C}}(\theta), J_{\mathcal{I}}(\theta), J_{\mathcal{T}}(\theta) \right] \in \mathbb{R}^5 \quad (11)$$

183 Since the implementation objectives $J_{\mathcal{I}}$ are related with a particular controller, they will be
 184 considered according to each specific case. Constraint handling depends on the selected algorithm
 185 and its own mechanisms. In general, the guidelines stated in [44] can be used to incorporate
 186 them into the cost function evaluation or into the MOO statement as an additional objective
 187 [42], [43].

188 *B. Multiobjective evolutionary algorithm*

189 As it was noticed earlier, any kind of MOO algorithm can be used in the multi-objective
 190 optimisation design methodology. A MOEA is selected due to its flexibility to handle complex
 191 functions. The MOEA will adjust the parameters of a given controller to be used in the closed
 192 loop process simulation. Then it will use the performance calculated from the simulation process
 193 to evolve the population to the Pareto front. In particular, the sp-MODE algorithm is selected
 194 [14], due to its performance in academic benchmarks for MOO algorithms and its flexibility
 195 for control purposes. This algorithm is based on Differential Evolution technique, which is a
 196 real-coded evolutionary algorithm.

197 *C. Pareto front visualization*

198 It is widely accepted that visualization tools are valuable and provide decision makers with
 199 a meaningful method to analyze the Pareto front and take decisions [45]. For two-dimensional
 200 problems (and sometimes for three-dimensional) it is usually straightforward to make an accurate
 201 graphical analysis of the Pareto front, but the difficulty increases with the dimension of the
 202 problem. Tools as VIDEO [46] can plot a fourth dimension by using a color-coding in the a 3-
 203 dimensional plot. Nevertheless, it is usual to state more than four objectives in an MOO process.
 204 Common alternatives to tackle an analysis in higher dimension are: Scatter diagrams, Parallel
 205 coordinates [47] and Level Diagrams [48]. Scatter diagrams use a 2-dimensional graph for each
 206 pair of objectives whilst Parallel coordinates plot a m -dimensional objective vector in a two
 207 dimensional graphs. The former becomes difficult to analyze when visualizing several objectives
 208 (since at least $\frac{m(m-1)}{2}$ plots are required); the latter, is a very compact way, but it loses clarity
 209 with large sets of data.

210 Level diagram (LD) visualization [48] helps us to perform an analysis of the obtained Pareto
 211 front \mathbf{J}_P^* , which is not a trivial task when the number of objectives is larger than three. It has been
 212 used with success in control systems up to 15 objectives [49], safety systems analysis [50] and
 213 engineering design [51]. As pointed in [52], LD visualization is one of the most useful methods
 214 to visualize m -dimensional Pareto fronts. LD visualization is based on the classification of the
 215 approximation \mathbf{J}_P^* obtained. Each objective $J_q(\boldsymbol{\theta})$ is normalized with respect to its minimum
 216 and maximum values. That is:

$$\hat{\mathbf{J}}_q(\boldsymbol{\theta}) = \left[\hat{J}_1(\boldsymbol{\theta}), \hat{J}_2(\boldsymbol{\theta}), \dots, \hat{J}_q(\boldsymbol{\theta}) \right], q \in [1, \dots, m]. \quad (12)$$

217 where

$$\hat{J}_q(\boldsymbol{\theta}) = \frac{J_q(\boldsymbol{\theta}) - J_q^{min}}{J_q^{max} - J_q^{min}}, q \in [1, \dots, m]. \quad (13)$$

218 and

$$\mathbf{J}^{min} = \left[\min_{J(\boldsymbol{\theta}) \in J_P^*} J_1(\boldsymbol{\theta}), \dots, \min_{J(\boldsymbol{\theta}) \in J_P^*} J_m(\boldsymbol{\theta}) \right] \quad (14)$$

$$\mathbf{J}^{max} = \left[\max_{J(\boldsymbol{\theta}) \in J_P^*} J_1(\boldsymbol{\theta}), \dots, \max_{J(\boldsymbol{\theta}) \in J_P^*} J_m(\boldsymbol{\theta}) \right] \quad (15)$$

219 To each normalized objective vector $\hat{\mathbf{J}}(\boldsymbol{\theta})$ a p-norm $\|\mathbf{x}\|_p := \left(\sum_{q=1}^m |x_q|^p \right)^{1/p}$ is applied to
 220 evaluate the distance to an ideal solution $\mathbf{J}^{ideal} = \mathbf{J}^{min}$. Common norms are:

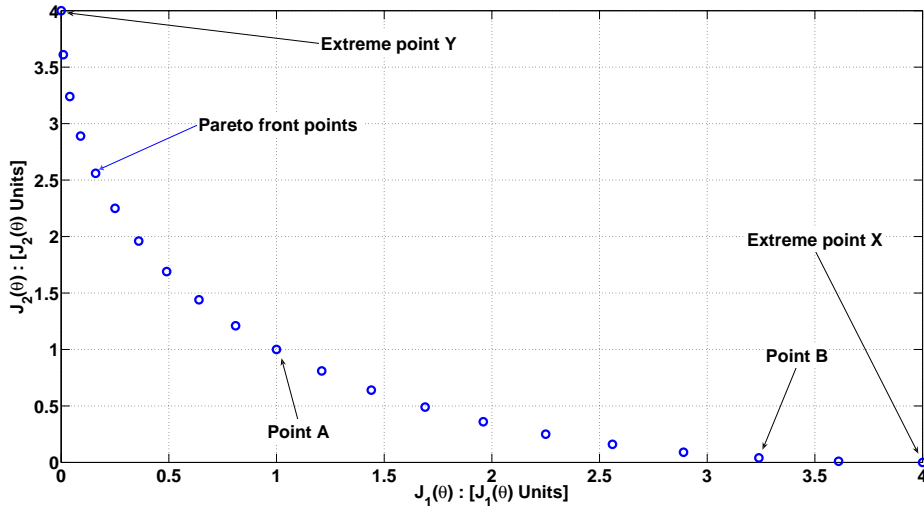
$$\|\hat{\mathbf{J}}(\boldsymbol{\theta})\|_1 = \sum_{q=1}^m \hat{J}_q(\boldsymbol{\theta}) \quad (16)$$

$$\|\hat{\mathbf{J}}(\boldsymbol{\theta})\|_2 = \sum_{q=1}^m \hat{J}_q(\boldsymbol{\theta})^2 \quad (17)$$

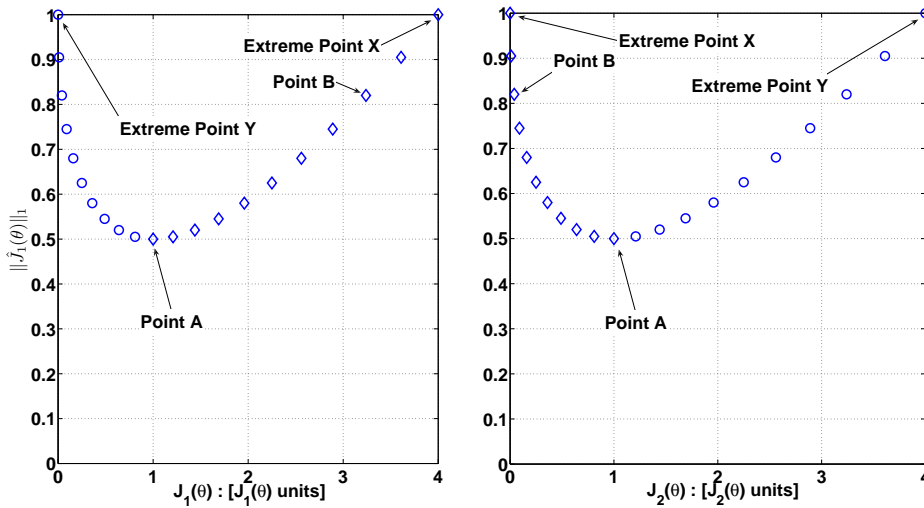
$$\|\hat{\mathbf{J}}(\boldsymbol{\theta})\|_\infty = \max \hat{\mathbf{J}}(\boldsymbol{\theta}) \quad (18)$$

221 The LD visualization uses a two dimensional graph for every objective and every decision
 222 variable. The ordered pairs $(J_q(\boldsymbol{\theta}), \|\hat{\mathbf{J}}(\boldsymbol{\theta})\|_p)$ in each objective sub-graph and $(\theta_l, \|\hat{\mathbf{J}}(\boldsymbol{\theta})\|_p)$
 223 in each decision variable sub-graph are plotted. Therefore, a given solution will have the same
 224 y -value in all graphs (see Figure 4). This correspondence will help to evaluate general tendencies
 225 along the Pareto front and compare solutions according to the selected norm. For example, an
 226 euclidian norm is helpful to evaluate the distance of a given solution with respect to the ideal
 227 solution, meanwhile a maximum norm will give information about the trade-off achieved by this
 228 solution. Using a norm to visualize tendencies in the Pareto front does not deform the MOP
 229 essence, since this visualization process take place after the optimization stage.

230 In all cases, the lower the norm, the closer to the ideal solution \mathbf{J}^{min} . For example, in figure
 231 4, point A is the closest solution to \mathbf{J}^{min} with the $\|\cdot\|_1$ norm. This does not mean that point A



(a) Typical visualization of the Pareto front for bi-objective problems.



(b) Representation using LD visualization.

Fig. 4: LD visualization. Points at the same level in LD correspond on each graphic.

232 must be selected by the DM. Selection will be performed according with the visual information
 233 from the LD visualization and the DM preferences. In the same figure, it is possible to visualize
 234 how the tradeoff rate changes in solution A. That is, it is possible to appreciate two different
 235 tendencies around solution A: in one hand, the better $J_2(\theta)$ value, the worst $J_1(\theta)$ value (circles).
 236 In the other hand, the worst $J_2(\theta)$ value, the better $J_1(\theta)$ value (diamonds). It is difficult to
 237 appreciate such tendencies with classical visualizations with more than three objectives. For the

238 remainder of this paper, the $\|\cdot\|_2$ norm will be used.

239 The LD visualization also enables the comparison of Pareto fronts obtained for different
240 design concepts [53] (in this case, controller schemes). In such visualization, it will be possible
241 to analyze the different trade-offs achieved by different control solutions, and determine under
242 which circumstances it is justified to use one over another. For example, in figure 5, it is possible
243 to see how a PID can achieve a better trade-off than a PI controller between load rejection and
244 step setpoint change (Zone Y). In the same way, it is possible to determine under which conditions
245 performance will be the same (Zone W).

246 To plot the LD, the LD visualization tool (LD-tool)⁶ will be used. This is a *posteriori*
247 visualization tool (*i.e.* is used after the optimization process) and enables the DM to identify
248 preferences zones along the Pareto front, as well as selecting and comparing solutions. With this
249 tool, it is possible to remove objectives or to add new performance measurements, not used in the
250 optimization stage. Furthermore, it is possible to integrate the DM preferences in a lexicographic
251 environment (as the one proposed by physical programming) to identify preferred solutions.

252 The aforementioned steps (problem definition, MOO process and the decision making stage)
253 are important to guarantee the overall design methodology. With a poor problem definition, not
254 matter how good our MOEA and decision making methodologies are, we will not have solutions
255 which guarantee a good performance on the real system. If the MOEA have a low performance,
256 the DM will not have a useful Pareto set to analyze and select a solution according with his/her
257 preferences. Finally, a lack of decision making tools and methodologies imply a lower degree
258 of embedment of the DM into the solution selection and tradeoff impacts. Furthermore it could
259 lead the DM to a lack of interest in the MOO approach.

260 IV. EXPERIMENTAL VALIDATION OF THE MOOD4CT PROCEDURE

261 To show the applicability of the method, two different approaches of controller tuning for a
262 non-linear twin rotor MIMO system (TRMS) are presented.

263 The TRMS is an academic workbench and a useful platform to evaluate control strategies
264 [54]–[56] due to its complexity, non-linearities, and inaccessibility of states. It is a TITO (two
265 inputs, two outputs) system, where two DC motors have control over the vertical angle (main

⁶Available at <http://www.mathworks.com/matlabcentral/fileexchange/24042>

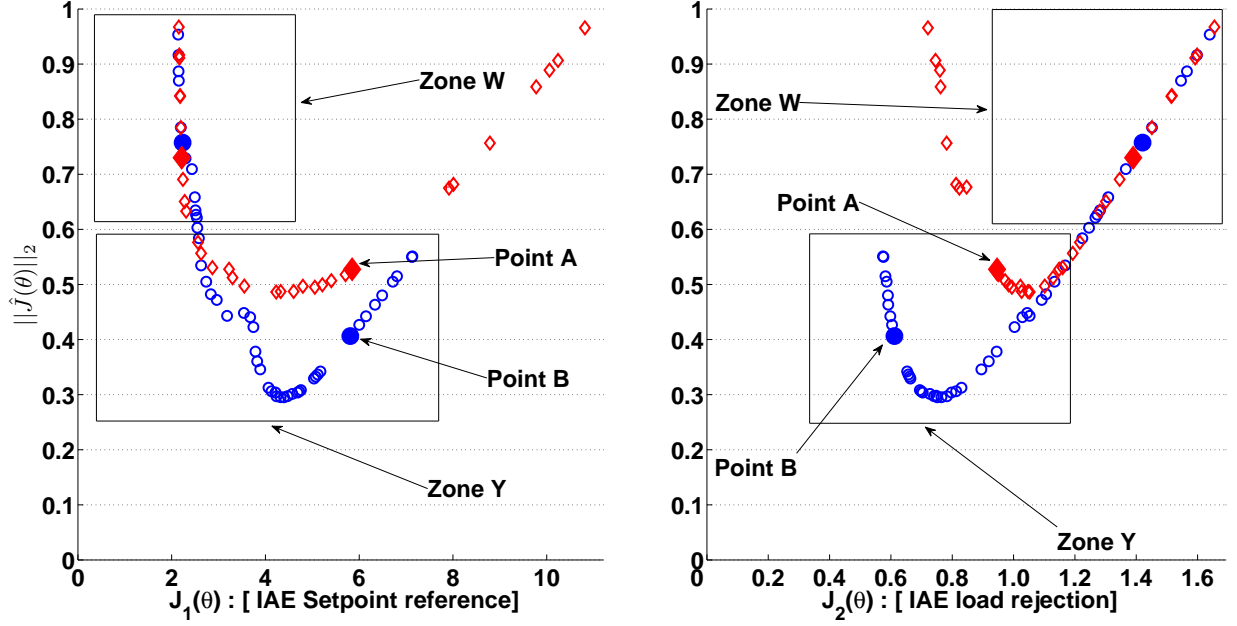


Fig. 5: Typical LD comparison for a SISO using a PI (\diamond) and a PID controller (\circ).

266 angle) and horizontal angle (tail angle) respectively. Both inputs are limited in the normalized
 267 range ± 1 , the main angle being in the range $[-0.5, 0.5]$ rad. And the tail angle in $[-3.0, 3.0]$
 268 rad.

269 The *mood4ct* procedure is validated in two steps:

- 270 1) An optimization stage using an identified process model to obtain Θ_P^*, J_P^* .
- 271 2) An experimental validation of the MOO results Θ_P^*, J_P^* on the real TRMS.

272 A. Optimization stage

273 A non-linear state-space model was identified as a part of the controller tuning-design pro-
 274 cedure. Details on the system modeling and the observer design can be consulted in [57] and
 275 Appendix A.

276 To evaluate the performance of a given controller a Simulink© model with the identified
 277 non-linear model was used. Two simulations were carried out with different patterns:

- 278 • Simulation pattern 1: Setpoint step change for main from 0 rad to 0.4 rad while tail setpoint
 279 is maintained at 0.

- 280 • Simulation pattern 2: Setpoint step change for tail from 0 rad to 2.4 rad while main setpoint
281 is maintained at 0.

282 The objectives defined in equations (6), (7), (9) and (10) are used according to a TITO system:

$$J_{\mathcal{E}}^{TITO}(\boldsymbol{\theta}) = T_s \left[\begin{array}{c} \max \left(\frac{IAE^{1,1}(\boldsymbol{\theta})}{\Delta R^1}, \frac{IAE^{2,2}(\boldsymbol{\theta})}{\Delta R^2} \right) \\ \min \left(\frac{IAE^{1,1}(\boldsymbol{\theta})}{\Delta R^1}, \frac{IAE^{2,2}(\boldsymbol{\theta})}{\Delta R^2} \right) \end{array} \right]^T \times \boldsymbol{w} \quad (19)$$

$$J_{\mathcal{U}}^{TITO}(\boldsymbol{\theta}) = \left[\begin{array}{c} \max \left(\sum_{j=1}^2 \frac{IADU^{1,j}(\boldsymbol{\theta})}{\Delta U_{max}^1}, \sum_{j=1}^2 \frac{IADU^{2,j}(\boldsymbol{\theta})}{\Delta U_{max}^2} \right) \\ \min \left(\sum_{j=1}^2 \frac{IADU^{1,j}(\boldsymbol{\theta})}{\Delta U_{max}^1}, \sum_{j=1}^2 \frac{IADU^{2,j}(\boldsymbol{\theta})}{\Delta U_{max}^2} \right) \end{array} \right]^T \times \boldsymbol{w} \quad (20)$$

$$J_{\mathcal{C}}^{TITO}(\boldsymbol{\theta}) = T_s \left[\begin{array}{c} \max \left(\frac{IAE^{1,2}(\boldsymbol{\theta})}{\Delta R_{max}^1}, \frac{IAE^{2,1}(\boldsymbol{\theta})}{\Delta R_{max}^2} \right) \\ \min \left(\frac{IAE^{1,2}(\boldsymbol{\theta})}{\Delta R_{max}^1}, \frac{IAE^{2,1}(\boldsymbol{\theta})}{\Delta R_{max}^2} \right) \end{array} \right]^T \times \boldsymbol{w} \quad (21)$$

283 Where \boldsymbol{w} is set to $\boldsymbol{w} = [10^0, 10^{-1}]$. To evaluate $J_{\mathcal{T}}(\boldsymbol{\theta})$ a linearized model is used. As a
284 weighting function for the robustness objective, the transfer function $W(s) = \frac{0.7s+2}{s+1.1}$ will be
285 used.

286 With the *mood4ct* approach, any kind of controller can be tuned. In this work, two schemes
287 are used: an ISA-PID controller [58] and a state-space controller (see figures 6 and 7). For
288 both cases, the controller is required to work with a sampling time of 20/1000 seconds with a
289 saturated control signal in the normalized range ± 1 .

290 1) *PID controller tuning*: PID controllers currently represent a reliable digital control solution
291 due to their simplicity. They are often used in industrial applications and so there is ongoing
292 research into new techniques for robust PID controller tuning [59]–[63]. For this reason, the PID
293 scheme will be the first to be evaluated.

294 A two degrees of freedom ISA-PID controller with a derivative filter and an anti-windup
295 scheme will be used:

$$\begin{aligned} U(s) &= K_c \left(b + \frac{1}{T_i s} + c \frac{T_d}{T_d/Ns + 1} \right) R(s) \\ &- K_c \left(1 + \frac{1}{T_i s} + \frac{T_d}{T_d/Ns + 1} \right) Y(s) \end{aligned} \quad (22)$$

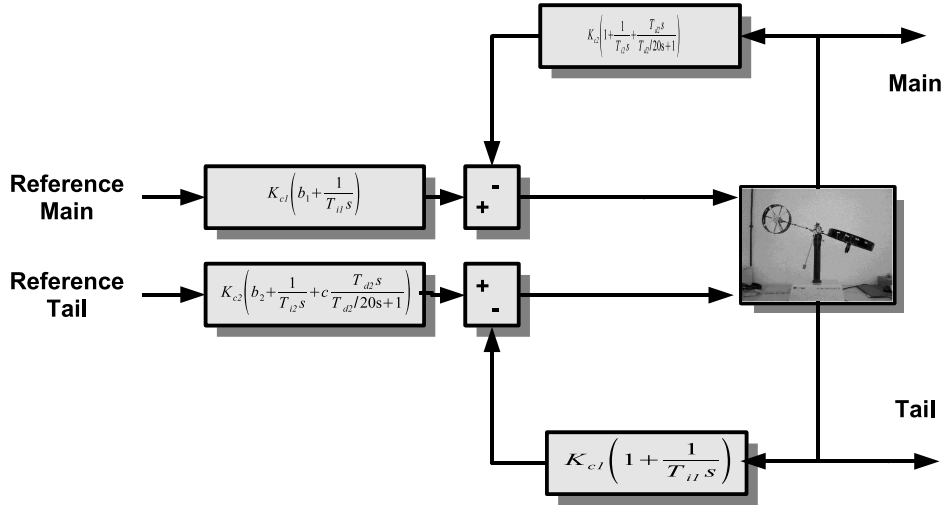


Fig. 6: PID controller scheme.

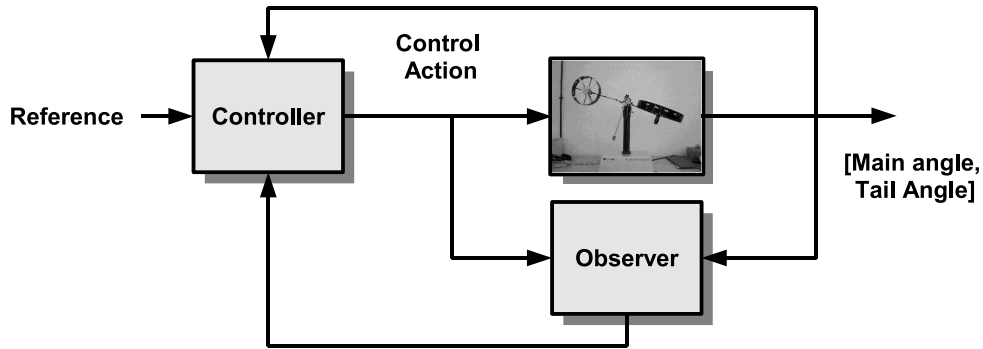


Fig. 7: State space controller proposal.

296 where

297 K_c is the proportional gain.

298 T_i represents the integral time (secs).

299 T_d is the derivative time (secs).

300 N represents the derivative filter. Common values for this filter lie in the range $N = [3, 20]$.

301 b is the setpoint weighting for the proportional action.

302 c is the setpoint weighting for the derivative action.

303 The antiwind-up is performed by conditional integration when the output signal is saturated
 304 [64]. The strategy to be implemented is a PI controller for the main angle and a PID controller
 305 for the tail angle. A setpoint weighting for the derivative action of $c = 0$ and a derivative filter of

TABLE I: MOO statement for the PID controller approach.

$\min_{\boldsymbol{\theta} \in \mathbb{R}^7} \mathbf{J}(\boldsymbol{\theta}) \in \mathbb{R}^5$	
$J_{\mathcal{E}}(\boldsymbol{\theta}) = T_s \left[\max \left(\frac{IAE_{step}^{Main}}{0.4}, \frac{IAE_{step}^{Tail}}{2.4} \right) + 10^{-2} \min \left(\frac{IAE_{step}^{Main}}{0.4}, \frac{IAE_{step}^{Tail}}{2.4} \right) \right]$	$K_{c1,c2} \in [0, 1]$
$J_{\mathcal{U}}(\boldsymbol{\theta}) = \max \left(\sum \Delta u_{step}^{Main} + \sum \Delta u_{pert}^{Main}, \sum \Delta u_{step}^{Tail} + \sum \Delta u_{pert}^{Tail} \right) + 10^{-2} \min \left(\sum \Delta u_{step}^{Main} + \sum \Delta u_{pert}^{Main}, \sum \Delta u_{step}^{Tail} + \sum \Delta u_{pert}^{Tail} \right)$	$T_{i1,i2} \in (0, 100]$
$J_{\mathcal{C}}(\boldsymbol{\theta}) = T_s \left[\max \left(\frac{IAE_{pert}^{Main}}{(2 \cdot 0.5)}, \frac{IAE_{pert}^{Tail}}{(2 \cdot 3)} \right) + 10^{-2} \min \left(\frac{IAE_{pert}^{Main}}{(2 \cdot 0.5)}, \frac{IAE_{pert}^{Tail}}{(2 \cdot 3)} \right) \right]$	$T_{d2} \in [0, 10]$
$J_{\mathcal{I}}(\boldsymbol{\theta}) = \sup_{\omega} \bar{\sigma}(\mathcal{S}(j\omega)), \omega \in (10^{-2}, 10^2)$	$b_{1,2} \in [0, 1]$
$J_{\mathcal{T}}(\boldsymbol{\theta}) = \sup_{\omega} \bar{\sigma}(\mathcal{T}(j\omega)W(j\omega)), \omega \in (10^{-2}, 10^2), s.t. J_5 > 0.8$	

306 $N = 20$ will also be used. Therefore, the *mood4ct* approach will be used to adjust the parameters
 307 K_{c1} , T_{i1} , b_1 for the PI controller and K_{c2} , T_{i2} , b_2 and T_d for the PID controller. Both will be
 308 tuned under SISO design considerations.

309 A total of five objectives are defined (see Table I). $J_{\mathcal{E}}(\boldsymbol{\theta})$, $J_{\mathcal{U}}(\boldsymbol{\theta})$, $J_{\mathcal{C}}(\boldsymbol{\theta})$, and $J_{\mathcal{T}}(\boldsymbol{\theta})$ are defined
 310 according to equations (19), (20), (21) and (10) respectively. Objective $J_{\mathcal{I}}(\boldsymbol{\theta})$ is included to prefer
 311 controllers with better disturbance rejection.

312 The Θ_P^* and \mathbf{J}_P^* from the *mood4ct* approach for PID tuning⁷ are shown in Figure 8. A total
 313 of 471 non-dominated controllers were found (a controllers subset Gk_{1i} is identified for further
 314 analysis). The following geometrical remarks (GR) on the level diagrams and their corresponding
 315 control remarks (CR) can be seen in Figure 8:

316 **GR 1:** It can be observed that two different subsets of solutions appear when solutions with

⁷A random search with the same number of function evaluations used by the MOEA was performed for comparison purposes. This approach calculates a Pareto front approximation with 161 solutions. The approximation calculated by the MOEA dominates 49 solutions of the random search approach; the random search approximation does not dominate any solution of the MOEA approximation.

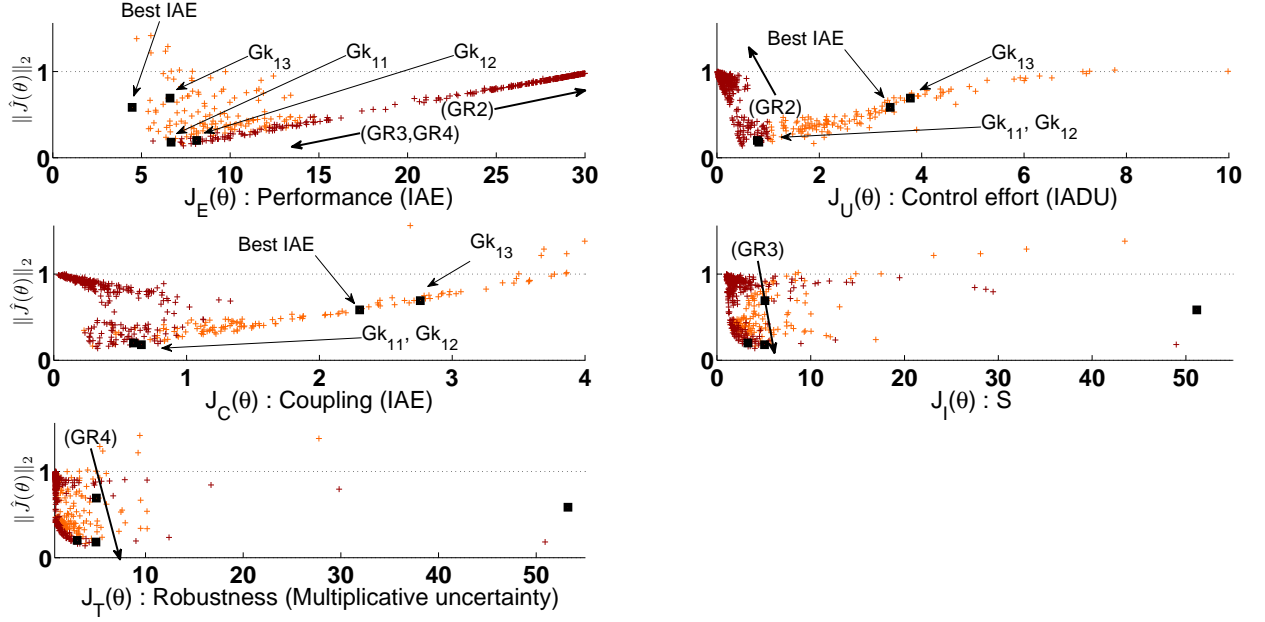


Fig. 8: J_P^* for PID controller. Dark solutions match the arbitrary requirement $J_U \leq 1$.

317

$J_U(\theta) \leq 1$ are separated.

318

CR 1: The IADU performance indicator for control action is a quality indicator to differentiate damping solutions along the Pareto front.

319

320

GR 2: For solutions with $J_U(\theta) \leq 1$, the lower $J_U(\theta)$, the higher $J_E(\theta)$.

321

CR 2: For overdamped solutions, the higher the control effort (IADU), the better the performance (IAE).

322

323

GR 3: For solutions with $J_U(\theta) \leq 1$, the lower $J_E(\theta)$, the higher $J_I(\theta)$.

324

CR 3: For overdamped solutions, the better the performance (IAE), the worse the disturbance rejection ($J_I(\theta)$).

325

326

GR 4: For solutions with $J_U(\theta) \leq 1$, the lower $J_E(\theta)$, the higher $J_T(\theta)$.

327

CR 4: For overdamped solutions, the better performance (IAE), the worse the robustness.

328

329

All of these points are well-known considerations in control theory. The Pareto front enables the visualization of this trade-off between objectives; and the DM can choose a solution that meets his own needs and preferences.

330

TABLE II: MOO statement for the state space controller approach.

$\min_{\theta \in \mathbb{R}^{16}} \mathbf{J}(\theta) \in \mathbb{R}^5$	
$J_{\mathcal{E}}(\theta) = T_s \left[\max \left(\frac{IAE_{step}^{Main}}{0.4}, \frac{IAE_{step}^{Tail}}{2.4} \right) + 10^{-2} \min \left(\frac{IAE_{step}^{Main}}{0.4}, \frac{IAE_{step}^{Tail}}{2.4} \right) \right]$	$\theta_i \in [-10, 10]$
$J_{\mathcal{U}}(\theta) = \max \left(\sum \Delta u_{step}^{Main} + \sum \Delta u_{pert}^{Main}, \sum \Delta u_{step}^{Tail} + \sum \Delta u_{pert}^{Tail} \right) + 10^{-2} \min \left(\sum \Delta u_{step}^{Main} + \sum \Delta u_{pert}^{Main}, \sum \Delta u_{step}^{Tail} + \sum \Delta u_{pert}^{Tail} \right)$	$i \in (1, 2, \dots, 16)$
$J_{\mathcal{C}}(\theta) = T_s \left[\max \left(\frac{IAE_{pert}^{Main}}{(2 \cdot 0.5)}, \frac{IAE_{pert}^{Tail}}{(2 \cdot 3)} \right) + 10^{-2} \min \left(\frac{IAE_{pert}^{Main}}{(2 \cdot 0.5)}, \frac{IAE_{pert}^{Tail}}{(2 \cdot 3)} \right) \right]$	
$J_{\mathcal{I}}(\theta) = \text{trace}(K * K')$	
$J_{\mathcal{T}}(\theta) = \sup_{\omega} \bar{\sigma}(\mathcal{T}(j\omega)W(j\omega)), \omega \in (10^{-2}, 10^2), s.t. J_5 > 0.8$	

331 2) *State space feedback controller tuning*: The above proposal used a PI-PID SISO strategy
 332 to address the control of a MIMO system. Such an approach is sometimes not enough to gain
 333 satisfactory control in a wide operational working zone, mainly because of the coupling dynamics.
 334 For this reason, a matrix gain for a state space (SS) control approach is selected as a second
 335 strategy (see Figure 7).

336 The *mood4ct* approach will be used to adjust a feedback gain matrix $K_{2 \times 8}$ to control the
 337 system. A total of five objectives are defined (see Table II). Objectives $J_{\mathcal{E}}$, $J_{\mathcal{U}}$, $J_{\mathcal{C}}$, and $J_{\mathcal{T}}$
 338 are again defined according to equations 19, 20, 21 and 10. Objective $J_{\mathcal{I}}$ is included to have
 339 preference over controllers with lower numerical sensibility, *i.e.* well balanced controllers at the
 340 implementation stage.

341 The Pareto front approximation \mathbf{J}_P^* ⁸ is shown in Figure 9. As a result, 589 non-dominated
 342 solutions were found (a controllers subset Gk_{2i} is identified for further analysis). The following
 343 geometrical remarks (GR) and their corresponding control remarks (CR) can be seen in Figure

⁸A random search with the same number of function evaluations used by the MOEA was performed for comparison purposes. This approach calculates a Pareto front approximation with 86 solutions. The approximation calculated by the MOEA dominates 85 solutions of the random search approach; the random search approximation does not dominate any solution of the MOEA approximation.

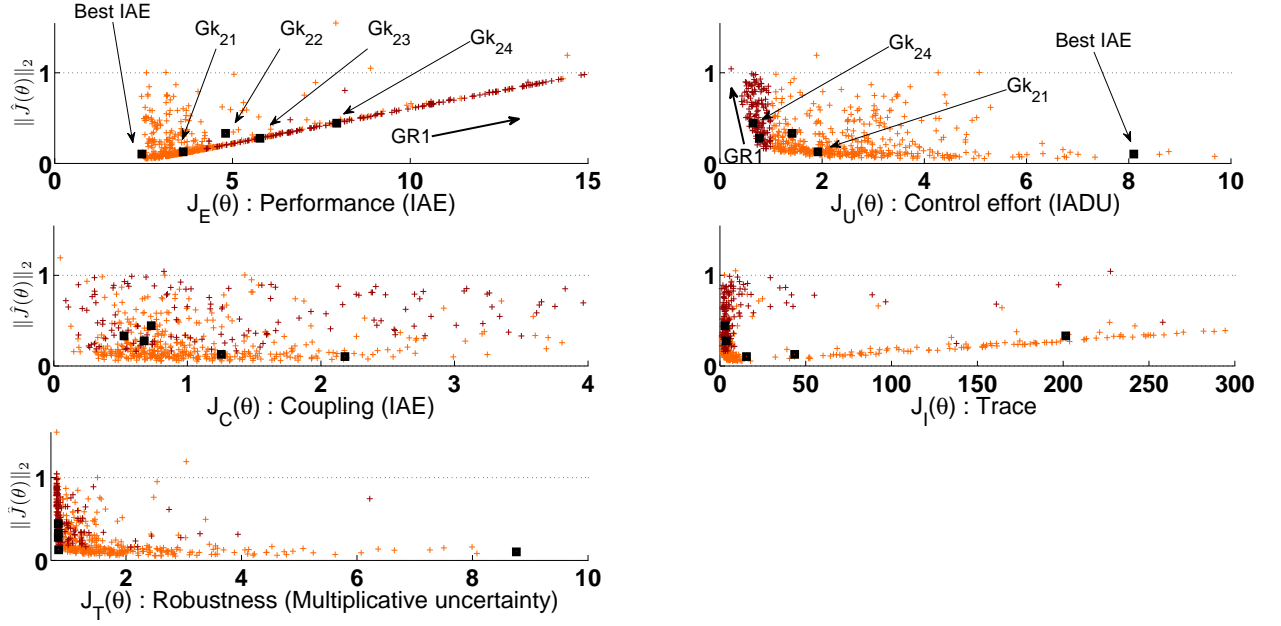


Fig. 9: J_P^* for the SS controller. Dark solutions match the arbitrary requirement $J_U(\theta) \leq 1$.

344 9:

345 GR 1: For solutions with $J_U \leq 1$, the lower $J_U(\theta)$, the higher $J_E(\theta)$.

346 CR 1: For overdamped solutions, the higher the control effort (IADU), the better the perfor-
347 mance (IAE).

348 GR 2: For objective $J_T(\theta)$, solutions matching the requirement $J_U(\theta) \leq 1$ have the lower
349 trace.

350 CR 2: Solutions with more balanced coefficients in the matrix gain are solutions that offer
351 less damping responses.

352 B. Experimental validation

353 To validate both approaches, the setpoint pattern on Figure 10 is used on the real TRMS ⁹.
354 It is important to note that such a pattern is different from the one used at the optimization
355 stage. In this way, it will be possible to evaluate and validate the *mood4ct* approach. The new

⁹Controllers from Tables III and VI were implemented in a National Instruments PXI-1002 System.

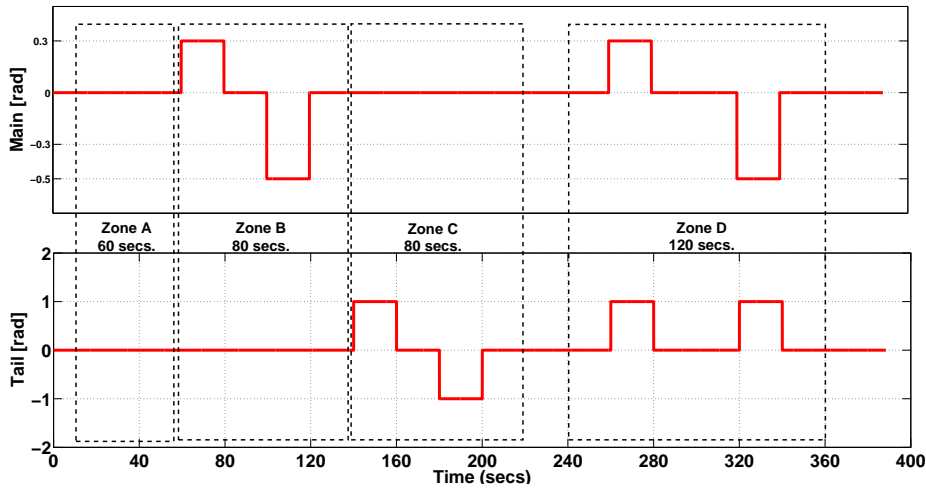


Fig. 10: Pattern for test on real TRMS. Idle state value for the main angle is around 0.3 rad.

356 pattern evaluates the performance of a given controller in maintaining zero-reference (zone A);
 357 a setpoint change in the main angle (zone B); a setpoint change in the tail position (zone C);
 358 and simultaneous changes in reference (zone D).

359 1) *PID controller - experimental results*: A subset of three controllers (see Table III) are
 360 selected from the Pareto set (Figure 8) for further analysis on the TRMS. Controller Gk_{13} is
 361 selected due to its performance on $J_{\mathcal{E}}(\theta)$; controller Gk_{11} due to its trade-off for objectives
 362 $J_U(\theta)$ and $J_C(\theta)$ (some performance is sacrificed in order to obtain a better control effort and
 363 less coupling between the main and tail closed loops). Finally, controller Gk_{12} is selected due to
 364 its robustness (this is a controller capable of working with a larger set of plants because it has
 365 a smaller $J_{\mathcal{T}}(\theta)$ value). In all cases, it is observed that the robustness requirement $J_{\mathcal{T}}(\theta) < 1$
 366 is not achieved. The reason for this could be: 1) it is not possible to use a PID scheme to
 367 control the system; or 2) the weighting function for robustness has not been chosen correctly
 368 (*i.e.* it is an excessive constraint) and the control engineer needs to evaluate if this constraint
 369 could be relaxed. After some analysis on the closed loop frequency response, it is determined
 370 that it is possible to use these controllers in a small operation range. The performances of these
 371 controllers with the reference pattern for the real test (see Figure 10) are shown in Tables IV,
 372 V and Figure 11.

373 As expected, controller Gk_{12} had the worst performance, but fewer coupling effects and the

TABLE III: PID controllers selected from Θ_P^* (Figure 8).

	$J_{\mathcal{E}}(\boldsymbol{\theta})$	$J_{\mathcal{U}}(\boldsymbol{\theta})$	$J_{\mathcal{C}}(\boldsymbol{\theta})$	$J_{\mathcal{I}}(\boldsymbol{\theta})$	$J_{\mathcal{T}}(\boldsymbol{\theta})$	$\boldsymbol{\theta} = (K_{c1}, T_{i1}, b_1, K_{c2}, T_{i2}, T_{d2}, b_2)$
Gk_{11}	6.83	0.82	0.65	4.76	4.58	$\boldsymbol{\theta} = (0.001, 0.006, 0.99, 0.269, 8.258, 1.420, 0.626)$
Gk_{12}	8.60	0.79	0.59	2.94	2.61	$\boldsymbol{\theta} = (0.001, 0.008, 0.68, 0.2533, 8.45, 1.14, 0.84)$
Gk_{13}	6.81	3.76	2.74	4.76	4.58	$\boldsymbol{\theta} = (0.001, 0.006, 0.70, 0.999, 7.396, 1.887, 0.6721)$

TABLE IV: Performance of PI-PID controllers on the real TRMS (Zones A and B)

		Zone A		
		IAE	IADU	Obj
Gk_{11}	Main	4.76E+000	2.85E-002	$J_1 = 1.31E - 001$
	Tail	1.07E+001	4.67E+000	$J_2 = \mathbf{4.67E + 000}$
		—	—	$J_3 = - - - - -$
Gk_{12}	Main	6.45E+000	3.05E-002	$J_1 = 2.43E - 001$
	Tail	3.42E+001	4.81E+000	$J_2 = 4.81E + 000$
		—	—	$J_3 = - - - - -$
Gk_{13}	Main	3.58E+000	2.03E-002	$J_1 = 9.89E - 002$
	Tail	8.17E+000	1.65E+001	$J_2 = \mathbf{1.65E + 001}$
		—	—	$J_3 = - - - - -$
		Zone B		
		IAE	IADU	Obj
Gk_{11}	Main	3.73E+002	2.23E+000	$J_1 = \mathbf{2.49E + 001}$
	Tail	1.14E+003	5.74E+001	$J_2 = \mathbf{5.74E + 001}$
		—	—	$J_3 = 3.81E + 000$
Gk_{12}	Main	4.44E+002	2.11E+000	$J_1 = 2.96E + 001$
	Tail	1.27E+003	5.91E+001	$J_2 = 5.91E + 001$
		—	—	$J_3 = 4.24E + 000$
Gk_{13}	Main	3.86E+002	2.20E+000	$J_1 = 2.57E + 001$
	Tail	3.12E+002	1.80E+002	$J_2 = 1.80E + 002$
		—	—	$J_3 = \mathbf{1.04E + 000}$

TABLE V: Performance of the PI-PID controllers on the real TRMS (Zones C and D)

Zone C				
		IAE	IADU	Obj
Gk_{11}	Main	5.68E+001	3.45E-001	$J_1 = 1.13E + 001$
	Tail	5.65E+002	4.26E+001	$J_2 = 4.26E + 001$
		—	—	$J_3 = 1.14E + 000$
Gk_{12}	Main	5.71E+001	2.74E-001	$J_1 = 1.28E + 001$
	Tail	6.42E+002	3.87E+001	$J_2 = 3.87E + 001$
		—	—	$J_3 = 1.14E + 000$
Gk_{13}	Main	6.36E+001	3.69E-001	$J_1 = 8.64E + 000$
	Tail	4.32E+002	1.21E+002	$J_2 = 1.21E + 002$
		—	—	$J_3 = 1.27E + 000$
Zone D				
		IAE	IADU	Obj
Gk_{11}	Main	3.97E+002	2.36E+000	$J_1 = 5.48E + 001$
	Tail	1.41E+003	7.45E+001	$J_2 = 7.45E + 001$
		—	—	$J_3 = - - - - -$
Gk_{12}	Main	6.03E+002	1.97E+000	$J_1 = 7.76E + 001$
	Tail	1.87E+003	6.34E+001	$J_2 = 6.34E + 001$
		—	—	$J_3 = - - - - -$
Gk_{13}	Main	3.88E+002	2.19E+000	$J_1 = 3.70E + 001$
	Tail	5.57E+002	2.24E+002	$J_2 = 2.24E + 002$
		—	—	$J_3 = - - - - -$

374 best control effort on zones C and D. Controller Gk_{13} , as indicated by the Pareto front, has the
375 highest control effort in all cases and the best performance on zones A and D. Finally, controller
376 Gk_{11} presents a good trade-off between performance and control effort.

377 2) *State space approach - experimental results:* A subset of six controllers (Table VI) was
378 selected from the Pareto set (Figure 9), according to the control requirements and the closed
379 loop frequency response on the linear model. Notice that it is possible to fulfill the requirement
380 $J_{\mathcal{T}}(\theta) < 1$, meaning that a larger set of plants can be controlled by the state space approach.
381 Controller Gk_{21} is selected because it is the controller with the lowest 2-norm on the level
382 diagram, while controller Gk_{22} is selected to analyze the impact of $J_{\mathcal{T}}(\theta)$ on performance.
383 Controllers Gk_{23} and Gk_{24} are selected to validate the trade-off achieved by decreasing the

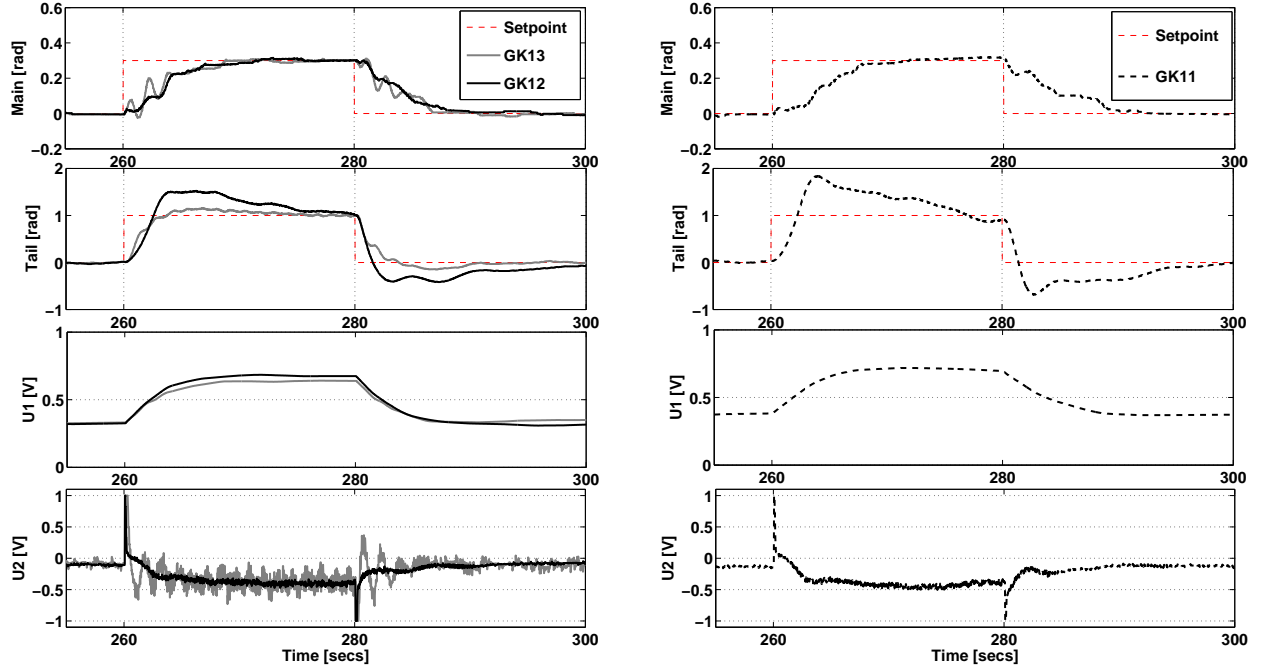


Fig. 11: Performance on the real TRMS of the *mood4ct*-PID approach for the setpoint pattern.

384 performance in order to gain a better control action and less coupling effects between the main
 385 and tail angles. The performance of these controllers with the reference step pattern for the real
 386 test (see Figure 10) is shown in Tables VII, VIII and in Figure 12.

TABLE VI: State space controller and their performances at the optimization stage.

	$J_{\mathcal{E}}(\boldsymbol{\theta})$	$J_{\mathcal{U}}(\boldsymbol{\theta})$	$J_{\mathcal{C}}(\boldsymbol{\theta})$	$J_{\mathcal{I}}(\boldsymbol{\theta})$	$J_{\mathcal{T}}(\boldsymbol{\theta})$
Gk_{21}	3.61	1.91	1.25	43.58	0.83
Gk_{22}	4.82	1.41	0.53	201.52	0.83
Gk_{23}	5.77	0.77	0.68	3.67	0.83
Gk_{24}	7.93	0.65	0.71	2.96	0.83

387 Gk_{21} and Gk_{22} are controllers with outstanding performance at the expense of high control
 388 efforts ($J_{\mathcal{U}}(\boldsymbol{\theta})$) and larger trace values ($J_{\mathcal{I}}(\boldsymbol{\theta})$). Controller Gk_{21} exhibits more coupling effects as
 389 was pointed by $J_{\mathcal{C}}(\boldsymbol{\theta})$, and noise sensitivity ($J_{\mathcal{I}}(\boldsymbol{\theta})$). Controller Gk_{22} exhibits a better performance
 390 than Gk_{21} due to coupling effects ($J_{\mathcal{C}}(\boldsymbol{\theta})$), but also shows a higher noise control effort ($J_{\mathcal{I}}(\boldsymbol{\theta})$).

391 Controller Gk_{23} and Gk_{24} has almost the same performance for objectives $J_{\mathcal{U}}(\boldsymbol{\theta})$, $J_{\mathcal{C}}(\boldsymbol{\theta})$,

392 $J_{\mathcal{I}}(\boldsymbol{\theta})$, $J_{\mathcal{T}}(\boldsymbol{\theta})$ and it is possible to see the tradeoff predicted by the Pareto front approximation.
 393 Controller Gk_{24} shows worse performance than controller Gk_{23} , but with less control effort.

TABLE VII: Performance of the state space controller on the real TRMS (Zones A and B).

Zone A				
		IAE	IADU	Obj
Gk_{21}	Main	8.64E+000	3.07E+001	$J_1 = 2.18E - 001$
	Tail	1.36E+001	2.17E+001	$J_2 = 3.07E + 001$
		—	—	$J_3 = - - - - -$
Gk_{22}	Main	6.47E+000	7.71E+001	$J_1 = 1.88E - 001$
	Tail	1.74E+001	2.90E+001	$J_2 = 7.71E + 001$
		—	—	$J_3 = - - - - -$
Gk_{23}	Main	9.96E+000	7.94E+000	$J_1 = 2.79E - 001$
	Tail	2.39E+001	8.61E+000	$J_2 = 8.61E + 000$
		—	—	$J_3 = - - - - -$
Gk_{24}	Main	9.67E+000	6.71E+000	$J_1 = 2.66E - 001$
	Tail	2.19E+001	5.11E+000	$J_2 = 6.71E + 000$
		—	—	$J_3 = - - - - -$
Zone B				
		IAE	IADU	Obj
Gk_{21}	Main	2.53E+002	1.61E+002	$J_1 = 1.69E + 001$
	Tail	1.63E+002	1.24E+002	$J_2 = 1.61E + 002$
		—	—	$J_3 = 5.42E - 001$
Gk_{22}	Main	2.11E+002	4.18E+002	$J_1 = 1.40E + 001$
	Tail	3.46E+002	1.59E+002	$J_2 = 4.18E + 002$
		—	—	$J_3 = 1.15E + 000$
Gk_{23}	Main	3.17E+002	4.85E+001	$J_1 = 2.11E + 001$
	Tail	3.28E+002	5.72E+001	$J_2 = 5.72E + 001$
		—	—	$J_3 = 1.09E + 000$
Gk_{24}	Main	5.79E+002	4.33E+001	$J_1 = 3.86E + 001$
	Tail	3.28E+002	3.56E+001	$J_2 = 4.33E + 001$
		—	—	$J_3 = 1.09E + 000$

TABLE VIII: Performance of the state space controller on the real TRMS (Zones C and D).

		Zone C		
		IAE	IADU	Obj
Gk_{21}	Main	1.34E+002	1.57E+002	$J_1 = \mathbf{1.01E + 001}$
	Tail	5.07E+002	1.10E+002	$J_2 = 1.57E + 002$
		—	—	$J_3 = 2.67E + 000$
Gk_{22}	Main	4.86E+001	4.02E+002	$J_1 = 1.25E + 001$
	Tail	6.26E+002	1.58E+002	$J_2 = 4.02E + 002$
		—	—	$J_3 = \mathbf{9.73E - 001}$
Gk_{23}	Main	6.77E+001	3.70E+001	$J_1 = 1.04E + 001$
	Tail	5.20E+002	4.23E+001	$J_2 = 4.23E + 001$
		—	—	$J_3 = 1.35E + 000$
Gk_{24}	Main	1.06E+002	3.09E+001	$J_1 = 1.46E + 001$
	Tail	7.28E+002	2.52E+001	$J_2 = \mathbf{3.09E + 001}$
		—	—	$J_3 = 2.12E + 000$
		Zone D		
		IAE	IADU	Obj
Gk_{21}	Main	2.90E+002	2.25E+002	$J_1 = 3.01E + 001$
	Tail	5.34E+002	1.64E+002	$J_2 = 2.25E + 002$
		—	—	$J_3 = - - - - -$
Gk_{22}	Main	2.18E+002	6.37E+002	$J_1 = \mathbf{2.96E + 001}$
	Tail	7.54E+002	2.48E+002	$J_2 = 6.37E + 002$
		—	—	$J_3 = - - - - -$
Gk_{23}	Main	3.42E+002	4.99E+001	$J_1 = 3.61E + 001$
	Tail	6.64E+002	5.51E+001	$J_2 = 5.51E + 001$
		—	—	$J_3 = - - - - -$
Gk_{24}	Main	6.20E+002	5.15E+001	$J_1 = 6.26E + 001$
	Tail	1.06E+003	4.23E+001	$J_2 = \mathbf{5.15E + 001}$
		—	—	$J_3 = - - - - -$

394 C. Comparison between control approaches

395 With the multiobjective approach and the LD tool it is possible to perform an overall com-
396 parison between both control approaches. The comparison will be not limited by using just a
397 pair of solutions (controllers), and the whole set of controllers will be used in accordance with
398 the quality of their performances along the Pareto front approximation.

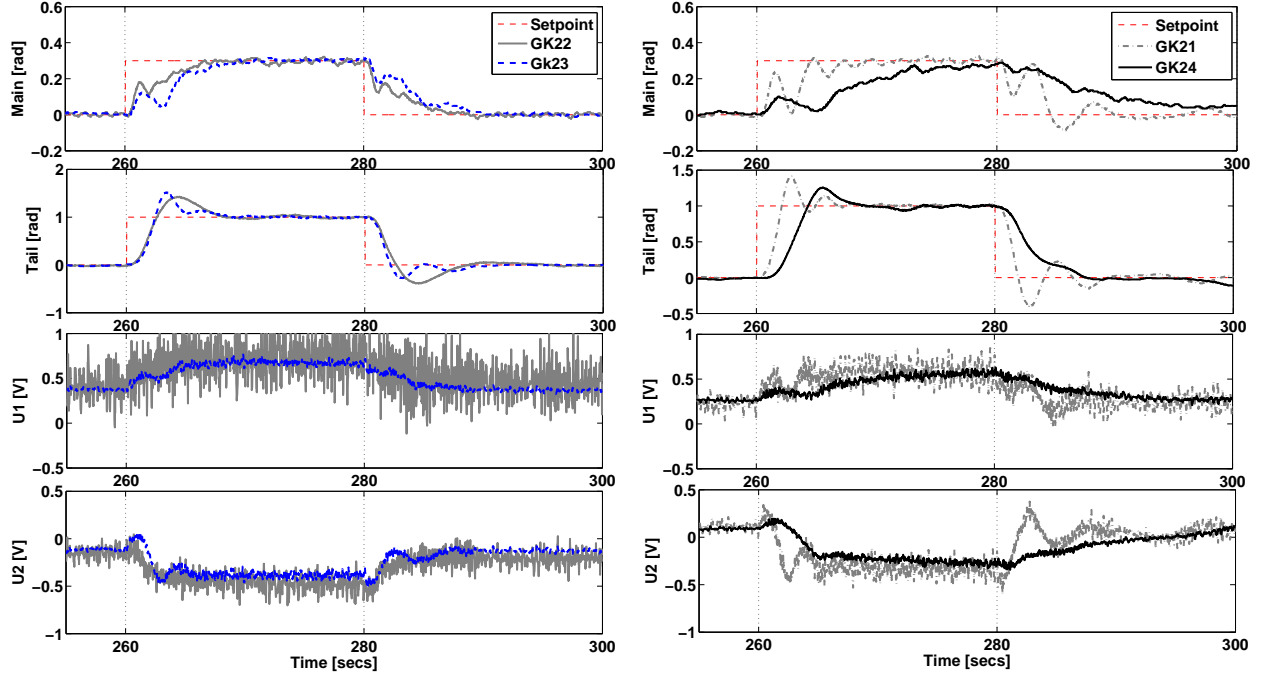


Fig. 12: Performance on the real TRMS of the *mood4ct*-SS approach on setpoint pattern.

399 As objective $J_{\mathcal{I}}(\boldsymbol{\theta})$ corresponds to the particular implementation of each controller, a com-
 400 parison can be performed in the objective subset $\mathbf{J}_s(\boldsymbol{\theta}) = [J_{\mathcal{E}}(\boldsymbol{\theta}), J_u(\boldsymbol{\theta}), J_c(\boldsymbol{\theta}), J_{\mathcal{T}}(\boldsymbol{\theta})]$. A new
 401 level diagram, using both set of solutions (with the ideal solution being the minimal offered by
 402 two approaches) is built (see Figure 13). Again, it is possible to make some geometrical remarks
 403 (GR) and their corresponding control remarks (CR):

404 GR 1: In objective $J_{\mathcal{E}}$ there is a range of solutions where both approaches coincide in the LD
 405 (Zone A).

406 CR 1: There are configurations for each controller capable of reaching the same level of
 407 performance in the range $IAE \approx [6, 15]$.

408 GR 2: For the above mentioned range, solutions of the frontal state space tend to have better
 409 values in $J_c(\boldsymbol{\theta})$ and $J_{\mathcal{T}}(\boldsymbol{\theta})$.

410 CR 2: For the performance range $IAE \approx [6, 15]$ the state space controller gives a better
 411 trade-off for control effort and robustness than a PID controller.

412 GR 3: Solutions below $\|\hat{\mathbf{J}}(\boldsymbol{\theta})\|_2$ (Zone B) correspond to second front solutions. These solu-
 413 tions tend to disperse with larger values in objectives $J_u(\boldsymbol{\theta})$, $J_c(\boldsymbol{\theta})$, and $J_{\mathcal{T}}(\boldsymbol{\theta})$.

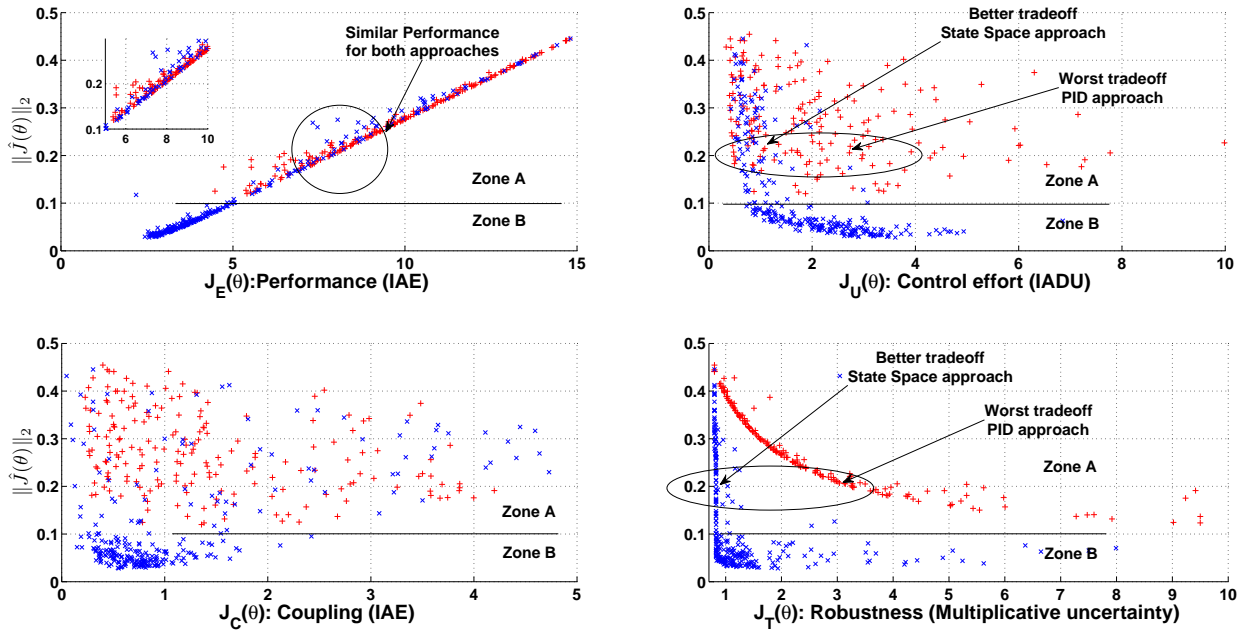


Fig. 13: Design concept comparison between: PID controllers (+) and state space controllers (x).

414 CR 3: The state space approach can reach closer values to the ideal solution. Nevertheless,
 415 these solutions may include the worst values for control effort, coupling effect, and
 416 robustness.

417 With such graphical analysis, it is possible to see the trade-off gained by using a modern
 418 control strategy such as a state space controller over a PID controller. In some instances, it will
 419 be worthwhile seeing if a complex control technique is justified over a classical technique (such
 420 as a PID controller) according with the DM preferences.

421

V. CONCLUSIONS

422 In this work, a holistic multi-objective optimisation design for controller tuning (*mood4ct*)
 423 has been presented. With *mood4ct*, it is possible to achieve a higher degree of flexibility for
 424 choosing a solution that matches the desired level of trade-off between conflicting objectives,
 425 such as performance, control effort, and robustness. The approach includes the use of mean-
 426 ingful performance objectives through simulation, and the use of a flexible tool to visualize
 427 m -dimensional Pareto fronts.

428 *Mood4ct* has been used to control a non-linear MIMO system. The controller tuning approach
 429 has been shown to be flexible for classical PID controllers and state space controllers tuning. It
 430 has also been shown to be reliable and robust enough to control the system with different
 431 reference patterns. This approach makes it possible to achieve a desired trade-off between
 432 performance and robustness, which leads to better implementation results on a real system than
 433 the results achievable by optimizing just a performance measurement. As the tendencies are those
 434 predicted by J_P^* from the optimization stage with the process model, the *mood4ct* procedure is
 435 validated as a tool for designing different control architectures.

436 Finally, using the level diagram tool a global comparison has been made between different
 437 control approaches, and this is useful to determine if a complex control technique is justified in
 438 preference to a classical technique that matches the DM preferences. Further research will focus
 439 on more interpretable objectives for robust control and stability.

440

APPENDIX

441 All models and controllers in this work are available to download (Simulink© format) from:

- 442 • <http://personales.upv.es/gilreyme/mood4ct/mood4ct.html>

443 A. State space linear model

$$\begin{pmatrix} \dot{x} = Ax + Bu \\ y = Cx \end{pmatrix}, C = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \end{bmatrix} \quad (23)$$

$$A = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ -4.74 & -0.03 & 5.66 & 0 & 0 & 0 \\ 0 & 0 & -0.75 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & -0.12 & -0.19 & 1 \\ 0 & 0 & 0 & 0 & 0 & -2.33 \end{bmatrix}, B = \begin{bmatrix} 0 & 0 \\ 0 & 0.239 \\ 0.752 & 0 \\ 0 & 0 \\ 0 & 0 \\ 0 & 2.326 \end{bmatrix} \quad (24)$$

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