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Additional Information

1 **Dynamic acousto-elastic test using continuous probe wave and transient**
2 **vibration to investigate material nonlinearity**

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7 **Abstract**

8 This study demonstrates the feasibility of the dynamic acousto-elastic effect of a
9 continuous high frequency wave for investigating the material nonlinearity upon transient
10 vibration. The approach is demonstrated on a concrete sample measuring 15x15x60 cm³.
11 Two ultrasonic transducers (emitter and receiver) are placed at its middle span. A
12 continuous high frequency wave of 500 kHz propagates through the material and is
13 modulated with a hammer blow. The position of the hammer blow on the sample is
14 configured to promote the first bending mode of vibration. The use of a continuous wave
15 allows discrete time extraction of the nonlinear behavior by a short-time Fourier
16 transform approach, through the simultaneous comparison of a reference non-modulated
17 signal and an impact-modulated signal. The hammer blow results in phase shifts and
18 variations of signal amplitude between reference and perturbed signals, which are driven
19 by the resonant frequency of the sample. Finally, a comprehensive analysis of the
20 relaxation mechanisms (modulus and attenuation recovery) is conducted to untangle the
21 coupled fast and slow hysteretic effects.

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1 **1. Introduction**

2 Investigation of the nonlinear dynamic properties of materials and structures is attracting
3 keen interest among several scientific communities such as geosciences¹, medicine², and
4 materials science³, thanks to their improved detection of microstructural features within
5 materials⁴. The link between the concerned materials among the different research areas is the
6 presence of defects in a wide range of scales (from nano- to macro-scale), which enhances the
7 nonlinear acoustic phenomena⁵⁻⁷. The internal friction between rough interfaces in their
8 imperfect microstructures causes a hysteretic behavior in terms of their quasi-static stress-strain
9 relationships. The resulting nonlinear modulus (M) including mechanical hysteresis can be
10 written as⁵

$$11 \quad M = M_o \left[1 + \beta \varepsilon + \delta \varepsilon^2 + \dots + U(\Delta \varepsilon, \dot{\varepsilon}) \right], \quad (1)$$

12 where the linear elastic modulus (M_o) is extended to include classical elastic higher order terms
13 of strain, β and δ , and a function that takes account of mechanical hysteresis; usually written as⁸

$$14 \quad U(\Delta \varepsilon, \dot{\varepsilon}) = \alpha (\Delta \varepsilon + \varepsilon \operatorname{sign}(\dot{\varepsilon})), \quad (2)$$

15 where α is a parameter that controls the magnitude of the hysteretic behavior, $\Delta \varepsilon$ is the strain
16 amplitude, ε is strain, and $\dot{\varepsilon}$ is strain rate. Under moderate dynamic strain amplitudes, $\sim 10^{-7}$ and
17 above⁹, the hysteretic behavior is manifested as an apparent softening of the material, so-called
18 non-classical behavior. The velocity of propagation and attenuation of the material depend on the
19 strain amplitude (fast dynamic effect), which in turn is accompanied by a long period of
20 relaxation after dynamic excitation (slow dynamic effect). The two mechanisms are thought to
21 coexist during dynamic excitation (material conditioning)¹⁰, and dominate the nonlinear behavior
22 of highly heterogeneous media such as those in concrete-like materials¹¹. The equation of state as

1 presented in Eq. 1 falls short of describing the slow dynamic effect, and only the fast dynamic
2 effect can be considered therein¹¹. The fast dynamic effect has been broadly investigated through
3 nonlinear resonant techniques. They consist in the investigation of the downward resonant
4 frequency shift observed in consecutive acquisitions by increasing the excitation amplitude⁸. The
5 normalized downward resonant frequency shift is in most materials proportional to the strain
6 amplitude¹⁰ and is related to the parameter α because of the fast dynamic effect as⁸

$$7 \quad \frac{\Delta f}{f_o} = \alpha \cdot \Delta \varepsilon, \quad (3)$$

8 where f_o is the resonant frequency obtained in the linear strain regime. Therefore, the
9 investigation of the material nonlinearity through nonlinear resonant spectroscopy-based
10 techniques relies on the fast dynamic effect (Eq.3), and the unavoidable effect of slow dynamics
11 contributes to the error in the estimation of the parameter α ^{10,11}. Such an effect is normally
12 minimized by increasing the time lapse between successive resonant frequency acquisitions.

13 Recent research on the application of nonlinear acoustics for the characterization of
14 biological tissues allowed investigation of the whole range of nonlinear phenomena with the
15 technique termed dynamic acousto-elastic test (DAET)¹². The DAET consists in monitoring the
16 variations of the speed of sound through the material, normally through variations in the time of
17 flight of ultrasonic pulses (probe wave), while a low frequency burst that matches a fundamental
18 resonant mode perturbs the media (pump wave). Assuming that Poisson's ratio, and density
19 variations are negligible during pump wave excitation, the resulted variations of modulus are
20 related to the speed of sound (c) as

$$\frac{M - M_o}{M_o} \approx \frac{c^2 - c_o^2}{c_o^2}. \quad (4)$$

Considering that the variations of the speed of sound (Δc) with respect to the speed of sound in the linear elastic regime (c_o) are so small that $c \approx c_o$ —and hence, $c^2 - c_o^2 \approx 2 \cdot (c - c_o)$ —, the corresponding relative variations of time of flight ($\Delta t/t_o$) are approximately related to the material nonlinearity as¹²⁻¹⁴

$$\frac{M - M_o}{M_o} \approx -2 \cdot \frac{\Delta t}{t_o} \approx C_E(\Delta \varepsilon) + \beta \varepsilon + \delta \varepsilon^2 + \dots + U(\Delta \varepsilon, \dot{\varepsilon}); \quad (5)$$

where, additional to the quadratic and cubic nonlinear classical parameters (β and δ) and strain rate-dependent hysteresis $U(\Delta \varepsilon, \dot{\varepsilon})$, the function $C_E(\Delta \varepsilon)$ takes into account the material conditioning effect: the apparent mix of slow and fast dynamics¹⁰ which offsets the relation between strain and the relative variation of modulus. In addition, the variations of material attenuation produced during low frequency burst excitation, are also derived from the amplitude of the ultrasonic pulses¹²⁻¹⁴.

Two conditions must be accomplished in DAET experiments. First, the time of flight (t_o) of the ultrasonic probe wave must be less than 1/10 times the value of the low frequency excitation¹³. In this way, the instantaneous variations of time of flight can be precisely related to the strain amplitude of the pump wave excitation. Second, the repetition rate of the ultrasonic pulse generator has to be set so that the ultrasonic wave is completely attenuated between pulses. Therefore, the succeeding ultrasonic pulses are not affected by the coda wave of the preceding ones¹²⁻¹⁴. In consequence of the last, the variations of time of flight and amplitude are obtained for few values of strain in several cycles of the low frequency burst (constant strain amplitude

1 excitation). Therefore, the test has to be repeated after changing the phase of the ultrasonic pulse
2 generator to obtain more experimental data, and describe properly the variations of time of flight
3 and amplitude over the whole strain range excitation¹²⁻¹⁴. These conditions make unpractical to
4 leverage the potential of DAET in on-site assessment of structures or in structural health
5 monitoring applications wherein the use of ambient vibrations and transient events (variable
6 strain amplitude excitation) are required to monitor passively the dynamic properties of
7 structures¹⁵⁻¹⁷.

8 In this study, a DAET is conducted by modulating a continuous monochromatic high
9 frequency probe that propagates through the material. The approach is demonstrated on a
10 prismatic concrete sample. The modulation of the continuous wave signal is achieved by a
11 hammer blow configured to promote its first bending mode of vibration. The signal analysis is
12 conducted with a short-time Fourier transform based approach. It is based on extracting the
13 phase and amplitude variations by the simultaneous comparison between a reference (non-
14 modulated) and an impact-modulated signal. Thus, the changes in velocity in the medium
15 produced upon transient vibration rely on the phase changes of the continuous probe. Unlike
16 previous DAET configurations^{13,14,18-20}, the approach presented herein has two main advantages.
17 It overcomes the inconvenience of changing the phase of the high frequency probe (in case of
18 ultrasonic pulses) in consecutive experiments for discretizing the variations of wave velocity
19 over the whole range of strain excitation, and it allows investigation of the discrete time variation
20 of material nonlinearity over only one cycle of low frequency excitation. Therefore, the
21 variations of modulus can be investigated during the ring down of the mechanical energy
22 introduced by a hammer blow. Otherwise, when a hammer blow is used as low frequency source
23 and ultrasonic pulses as probe wave^{21,22}, the only measurable parameter is the maximum offset of

1 the normalized time shift ($\Delta t/t_0$), since the strain amplitude is variable over time. Conversely, the
 2 parameters extracted herein show similarity with the nonlinear acoustic behavior derived from
 3 Eq.5, and previously observed in compressional mode DAET experiments^{13,14,18-20}. These are: 1)
 4 an offset of the material modulus upon low frequency excitation, 2) a phase delay between the
 5 onset of the strain amplitude and the relative variations of modulus, and 3) an incomplete
 6 recovery of the dynamic properties of the material when the strain energy of the low frequency
 7 signal is completely damped.

8 **2. Materials and methods**

9 **2.1. Materials**

10 The tests were conducted on a concrete sample measuring 15x15x60 cm³. The
 11 composition of concrete is listed in Table 1. At the moment of test, the concrete sample was fully
 12 matured. For reference, the compressional and shear wave velocities were determined using
 13 direct transmission of an ultrasonic pulse, along the straight-line path respect to the sample
 14 width. For compressional wave velocity measurement, two ultrasonic transducers GE-
 15 Measurement & Control (model G 0,25 G code 67422, central frequency of 250 kHz) were used.
 16 For shear wave velocity measurement, two transducers Panametrics-NDT (model V151, central
 17 frequency of 250 kHz) were used. The compressional and shear wave velocities were 4056 m/s
 18 and 2483 m/s. The density of the sample was $\rho = 2350 \text{ kg/m}^3$; thus $M_0 = 37.8 \text{ GPa}$, and
 19 Poisson's ratio $\nu = 0.20$.

20 Table 1. Mix design of concrete and properties.

Cement CEM I/52.5N (kg/m ³)	370
Water (l/ m ³)	212

Fine aggregates 0/4 (kg/m ³)	774
Coarse aggregates 4/14 (kg/m ³)	1069
Compressive strength at 28 days (MPa)	53

1

2 **2.2. Experimental configuration**

3 Figure 1 shows a schematic depiction of the experimental configuration. Two ultrasonic
4 transducers (Panametrics-NDT model V101, central frequency of 500 kHz) were placed on the
5 sample at its middle span with a distance (d) between them, center to center, of 10 cm. A
6 continuous sinusoidal high frequency probe was modulated by a hammer blow, while an
7 accelerometer (Bruel & Kjaer model 4525B, sensitivity of 1.046 mV/(m.s⁻²)) monitored the out-
8 of-plane acceleration at the center of the sample. The relative position between impact and
9 accelerometer was configured to promote the first bending mode of vibration. The strain
10 amplitude owed to the first bending mode of vibration was derived from the accelerometer
11 response (see section 2.4). An instrumented hammer (Bruel & Kjaer model 8207, sensitivity of
12 0.230 mV/N) with a polymeric tip was used to perform the measurements. Five different energy
13 impacts were conducted between 1kN to 5kN, which resulted in values of resonant frequency
14 between 1442 Hz and 1440 Hz. The time lapse between different acquisitions was set at one
15 minute.

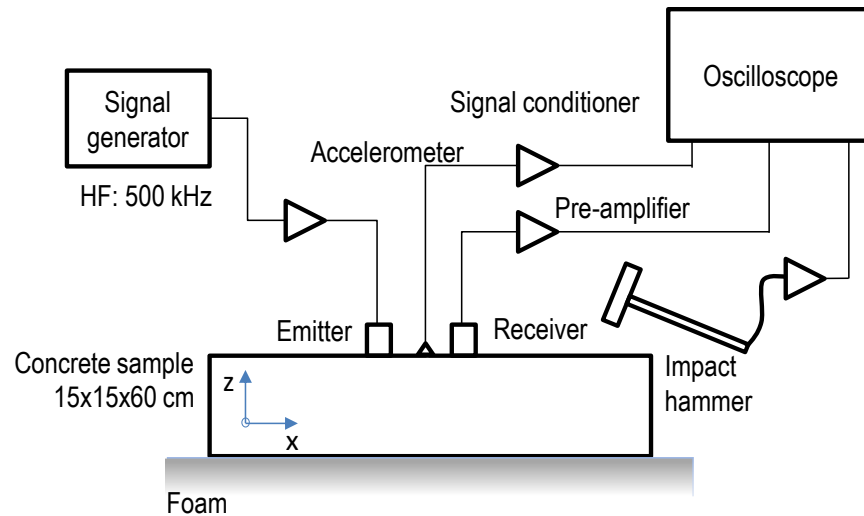


Figure 1. Schematic representation of the experimental setup.

Recently, Fröjd and Ulrikssen²³ investigated the effect of a static load on the amplitude and phase of a continuous wave excitation at 47 kHz. In such study, the authors argued that the continuous wave source produced a steady-state diffuse field which consisted of direct propagation, scattered waves, and reflections from the boundaries of the concrete sample being investigated therein. In this study, the frequency of the continuous ultrasonic probe was set to 500 kHz. At such a frequency, multiple scattering effects occur in concrete and a strong attenuation of the energy in every direction is produced^{24,25}. Thus, the resulted field is mainly dominated by direct propagation, but reflection contributions from sample boundaries are dramatically minimized. For reference, the amplitude of the continuous wave excitation in a through-thickness continuous wave measurement was reduced more than 80% that obtained by using the original test configuration (as shown in Figure 1). Therefore, the reflected energy recorded by the receiver in the original position is not a significant contribution to the total received energy. The continuous wave is then perturbed by a hammer blow and as a result, the amplitude and phase of the continuous probe are modulated because of the material nonlinearity.

1 The distance between the transducers has to be set, so that the instantaneous variations of phase
2 and amplitude can be precisely related to the strain values derived from the acceleration
3 response. Despite, any specific bulk wave can be attributed to the continuous wave excitation,
4 the time of flight obtained for compressional wave was used to meet this requirement. To this
5 end, the distance between the transducers was set to $d= 10$ cm. Then, indirect propagation (as
6 shown in Figure 1) of five cycles at 500 kHz provided a time of flight of $t_o= 26.8 \mu s$ (p-wave
7 arrival). This value of t_o ensured that the instantaneous variations of amplitude and phase
8 correspond to less than 1/10 times the period of the low frequency excitation¹³. On other hand,
9 we noticed that the value of t_o was significantly lower than that obtained using direct
10 transmission (see section 2.1). This effect is most likely due to the different material properties in
11 the near-surface regions with respect to the inner core of the concrete samples^{26,27}.

12 **2.3. Signal processing**

13 The out-of-plane particle displacement because of the continuous monochromatic probe
14 wave before a hammer blow excitation ($y_r(t)$) can be written as

$$15 \quad y_r(t) = B \sin(\omega_r \cdot t) \cdot e^{-\gamma_r \cdot d} + \Psi_r(t), \quad (6)$$

16 where B is the amplitude of the signal, ω_r is the angular frequency of the probe, γ_r is the
17 ultrasonic attenuation, and $\Psi_r(t)$ is a function containing the non-coherent information of the
18 multiple scatters recorded on propagation from the emitter to the receiver. Subscript r stands for
19 “reference”, whereby the ultrasonic probe is assumed to propagate through the medium in the
20 linear elastic regime. Namely, in absence of hammer blow excitation. The high frequency probe
21 $y_r(t)$ is then perturbed by a hammer blow. As a result of the presence of contact-like defects
22 within the imperfect microstructure of concrete, the modulus of the material varies upon

1 compression and tension cycles induced by the pump wave excitation²⁸. Then, the phase and
 2 amplitude of the probe wave signal are modulated. The resulting impact-modulated signal $y_p(t)$
 3 can be written as

$$4 \quad y_p(t) = B \sin(\omega_r \cdot t + \Omega(t)) \cdot e^{-(\gamma_r + \Gamma(t))d} + \Psi_p(t). \quad (7)$$

5 The functions $\Omega(t)$ and $\Gamma(t)$ are the phase and amplitude modulation contributions. Both
 6 functions must consider the periodic variations of phase and attenuation, as well as the
 7 instantaneous offsets produced with the onset of the strain amplitude. Phase and amplitude
 8 modulation manifest as sideband peaks with respect to the frequency of the probe wave, when
 9 the whole modulated probe signal is transformed to the frequency domain. The relative energy of
 10 the sideband components, with respect to the energy of the probe wave, has been used as a
 11 measurement of material nonlinearity²⁸. However a more detailed description is proposed herein.
 12 The variation of material modulus because of the presence of contact-like defects can be
 13 quantified through the instantaneous phase difference of the impact-modulated probe (Eq.7) with
 14 respect the reference (Eq.6). The resulting instantaneous phase difference between both signals is
 15 the function $\Omega(t)$. To this end, a sliding window moves through the time domain signals of the
 16 probe waves (Eqs. 6 and 7) and transforms the time segments within the window to the
 17 frequency domain. At every windowed time segment (τ), the time shift (Δt) between perturbed
 18 and reference signals is obtained from their spectral phases ($\Phi_p(\omega)$ and $\Phi_r(\omega)$) at the probe
 19 frequency (ω_r) as

$$20 \quad \frac{\Delta t|_{\tau}}{t_0} = \frac{\phi_p(\omega)|_{\omega_r} - \phi_r(\omega)|_{\omega_r}}{t_0 \cdot \omega_r}. \quad (8)$$

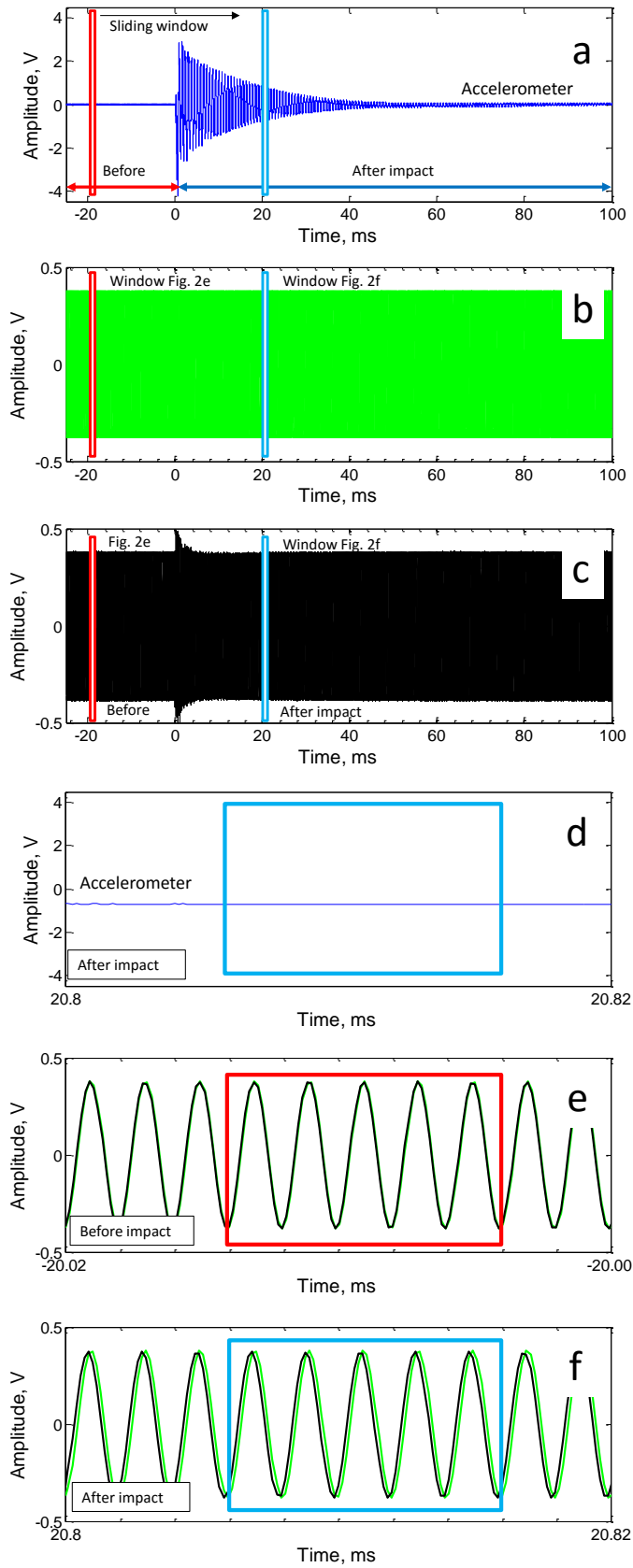
1 Given that a continuous probe lacks a clear wave path, and it does not allow extraction of phase
2 velocity, in this study we assume that the time shift variations between reference and modulated
3 signals are related to the variation of the material modulus. Therefore, the relative time shift
4 modulation obtained from Eq. 8 provides insight into the relative variation of the modulus as
5 deduced in Eq. 5. A normalization of the time shift (Δt) by t_0 is still needed, especially when
6 different materials are to be compared²². The time of flight obtained for indirect propagation of 5
7 cycles at 500 kHz was selected as the reference value ($t_0 = 26.8\mu\text{s}$). In addition, at every window
8 position, the variation of the attenuation properties between reference and impact-modulated
9 probe can be considered as¹⁴

$$10 \quad \gamma_p - \gamma_r \Big|_{\tau} = -\ln \left(\frac{|F_p(\omega)|_{\omega_r}}{|F_r(\omega)|_{\omega_r}} \right) / d, \quad (9)$$

11 where $|F_r(\omega)|$ and $|F_p(\omega)|$ stand for the spectral amplitude at the probe frequency ω_r ,
12 corresponding to the windowed time segment τ of the signals $y_r(t)$ and $y_p(t)$. The instantaneous
13 variations of attenuation are $\Gamma(t)$, since $\gamma_p(t) = \gamma_r(t) + \Gamma(t)$.

14 Figures 2a-2c show typical recorded signals and a schematic representation of the signal
15 processing. At overall, the signal processing consists in a sliding window that simultaneously
16 transforms the time segment of the reference (y_r) and perturbed (y_p) signals, and computes the
17 relative time shift (Eq. 8) and amplitude variations (Eq. 9) between them. Figures 2d-f show
18 magnified plots of the signals. Herein, the window length was set to five cycles of the high
19 frequency probe that corresponds to 1/69 times the resonant frequency corresponding to the first
20 bending mode of vibration. Such a window length provides an almost constant value of
21 acceleration (see Figure 2d). Then, time shift and amplitude variations can be precisely related to

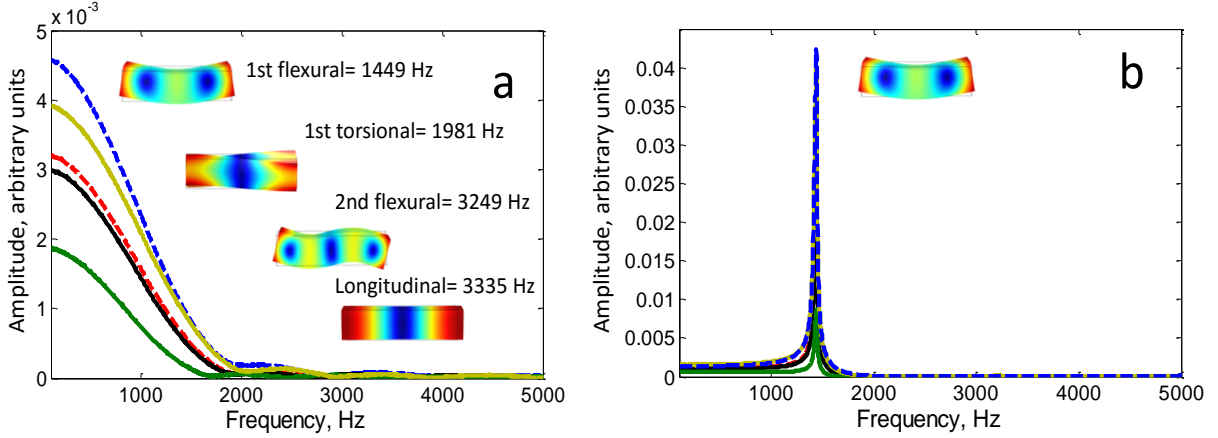
1 the in-plane strain which is derived from the accelerometer response (see section 2.4). Figures 2e
2 and 2f compare a time segment of y_r and y_p , before and after an impact event. Before the impact
3 event, the phase differences between the signals y_r and y_p are close to zero (Figure 2e), whereas
4 they become evident after the hammer blow (Figure 2f).



1 Figure 2. (Color online) Typical time domain signals and schematic representation of the signal
2 processing: a) signal recorded by the accelerometer, b) ultrasonic monochromatic reference
3 signal (y_r), c) impact-modulated ultrasonic signal (y_p), d) a time segment of the acceleration signal
4 after the hammer blow; it shows an almost constant value of acceleration (and hence strain) within the
5 length of the window ($10 \mu\text{s}$), e) a time segment of the ultrasonic probes y_r and y_p before the hammer blow
6 (the window is shown in red), and f) a time segment the ultrasonic probes y_r and y_p after impact (the
7 window is shown in blue).

8 ***2.4. Measurement of strain from acceleration response***

9 Different resonant modes are generated upon hammer blow excitation. For reference, the
10 frequency domain spectra for the five hammer blows are represented in Figure 3a. The
11 broadband excitation generated upon a hammer blow ranges from 0 to nearly 2 kHz. A numerical
12 eigen-frequency analysis of linear isotropic material, and for the specific sample the geometry
13 and material properties of the concrete sample — M_0 , ν and ρ , see section 2.1— leads to the first
14 bending mode occurring at 1449 Hz, the first torsional mode at 1981 Hz, the second bending
15 mode at 3249 Hz, and the longitudinal mode at 3335 Hz. For the given impact point and
16 frequency band generated, only the first bending mode (1442 Hz obtained experimentally)
17 contributes to the strain field, and other fundamental and higher order modes will not be excited
18 (Figures 3a and 3b).



1
2 Figure 3. (Color online) a) Frequency response for the five impulse signals as acquired by the
3 hammer load cell, and b) frequency spectra from acceleration signals. Inset figures show the
4 mode shapes obtained by the numerical eigen-frequency analysis.

5 From the acceleration response, the dynamic strain on the outermost fiber of the beam
6 owed to the first bending mode of vibration can be derived from the solution of the Euler-
7 Bernoulli beam equation²⁹. However, the assumptions of the Euler-Bernoulli beam theory do
8 not apply in the case of short beams³⁰. Alternatively, Payan and colleagues³¹ proposed estimating
9 the strain from a numerical finite element model (FEM), relating the out-of-plane acceleration
10 (a_{zz}) to the in-plane strain (ε_{xx}); in this case that owed to the first bending mode of vibration, as

$$11 \quad \varepsilon_{xx} = \frac{K \cdot \varepsilon_{num}}{K \cdot a_{zz,num}} \cdot a_{zz,exp} \quad ; \quad (10)$$

12 where K is an amplification factor used by the FEM software to display relative values of strain
13 and acceleration in the solution of free boundary conditions. The factor K is then divided out,
14 ε_{num} and $a_{zz,num}$ are the in-plane strain and out-of-plane acceleration obtained in the eigen-
15 frequency analysis; $a_{zz,exp}$ is the out-of-plane acceleration obtained experimentally. In the
16 following, we use the convention of negative strains for compression of the material.

1 3. Results and discussion

2 Figures 4a-d show representative results of the relative variations of time shift and the
3 variation of signal attenuation as the mechanical energy input naturally decreases. The results
4 reveal a sudden softening of the material produced after a hammer blow, which manifests as a
5 decrease of material modulus and an increase of attenuation (decrease of signal amplitude). The
6 instantaneous variations of both extracted signals —derived from Eqs. 8 and 9, and shown in
7 Figures 4a and 4c— are mainly driven by the first resonant bending mode of the sample, and
8 harmonics at twice the fundamental resonance frequency. Simultaneously, the signals are
9 instantaneously offset because of the non-classical contributions: the fast and slow hysteretic
10 motions. For better understanding of the classical and non-classical contributions, the signals
11 were decomposed, to untangle the effect of material conditioning from the relative variation of
12 time shift and the variation of attenuation properties: C_E and C_D respectively. Such
13 decomposition can be attained by applying a low pass filter to the resulting time shift and
14 amplitude variation signals¹³. Herein, the cutoff frequency was set to 1/3 of the low frequency
15 excitation. The instantaneous offsets ($C_E(t)$ and $C_D(t)$) were then subtracted from the whole time
16 shift and amplitude variation curves. Such decomposition allows one to analyze separately the
17 relaxation kernel (material conditioning) after a hammer blow, and the instantaneous
18 modification of the classical material nonlinearity (β and δ) with decreasing strain amplitude
19 ($\Delta\varepsilon$). Once the relaxation kernels (the instantaneous offsets C_E and C_D) of the signals have been
20 untangled, the resulting mean-centered variations of the time shift signal include the low
21 frequency excitation ($h=1$), and H harmonics (see Figure 4e). The resulting signal can be
22 described as

$$-2 \cdot \frac{\Delta t}{t_0}(t) - C_E(t) = \sum_{h=1}^{h=H} A_p e^{i h \omega_M t + \xi_h t}, \quad (11)$$

where A_p is the amplitude of the signal corresponding to the harmonic h ; the variable ξ_h takes into account the decaying energy for every harmonic h of the relative variation of modulus (see inset plot in Figure 4e). On other hand, the strain owed to the first bending mode resonant frequency (ω_M) can be described as

$$\varepsilon_{xx}(t) = S_1 \cdot e^{i \omega_M t + \xi_M t}, \quad (12)$$

where S_1 is the amplitude of the strain signal (see Figure 4d), and ξ_M is the modal damping ratio. For simplicity, Eqs. 11 and 12 do not consider time-dependent frequency and attenuation behavior, which are normally observed during the ring down of the signals in concrete^{32,33}. After substituting Eq. 11 and Eq. 12 into Eq. 5, a relative quantification of the classical higher order terms can be obtained as $\beta' = A_1/S_1$ and $\delta' = A_2/S_1^2$. Herein, the amplitudes A_1 , A_2 , and S_1 were estimated using a sliding window that computes the spectral amplitude over ten cycles of their corresponding signals (Figures 4e and 4f). Therefore, these amplitudes are weighted averaged values obtained through a Fourier transform. The obtained values are $\beta' = -300$ and $\delta' = -7.4 \cdot 10^7$, within the strain amplitudes of $S_1 = 3 \cdot 10^{-6}$ and $1 \cdot 10^{-6}$. Such analysis leads to similar values of β' and δ' and C_E obtained through a second order polynomial fit of the relation between the relative time shift (as $-2 \cdot \Delta t/t_0$) and strain (see polynomial fit in Figure 4b). The main drawback of the polynomial fit method^{14,18-20}, is the lack of fit of the second order polynomial to the observed behavior (Figure 4b), and hence, a Fourier analysis can be preferred. However, the amplitudes A_1 , A_2 and S_1 obtained from the Fourier analysis of transient response signals are weighted

1 averaged values of the actual amplitude, which can result in an underestimation of the actual
 2 amplitudes.

3 Normally, the relation between time shift variations and strain resembles a bow tie, and
 4 likewise for the attenuation variations, since both signals are delayed in phase with respect to the
 5 strain signal (Figures 4b and 4d). Such phase delay depends on the storage and loss modulus of
 6 the material, namely the relation between the real and imaginary parts of the frequency domain
 7 signals³⁴, as well as on the hysteretic contribution $U(\Delta\varepsilon, \dot{\varepsilon})$. Further, by integrating the relation
 8 between the relative variations of modulus with strain, the stress-strain relationship can be
 9 obtained as

$$10 \quad \sigma(\varepsilon) = \sigma_L(\varepsilon) + \sigma_{NL}(\varepsilon) = M_o \cdot \varepsilon + M_o \int \frac{\Delta M}{M_o} \cdot d\varepsilon, \quad (13)$$

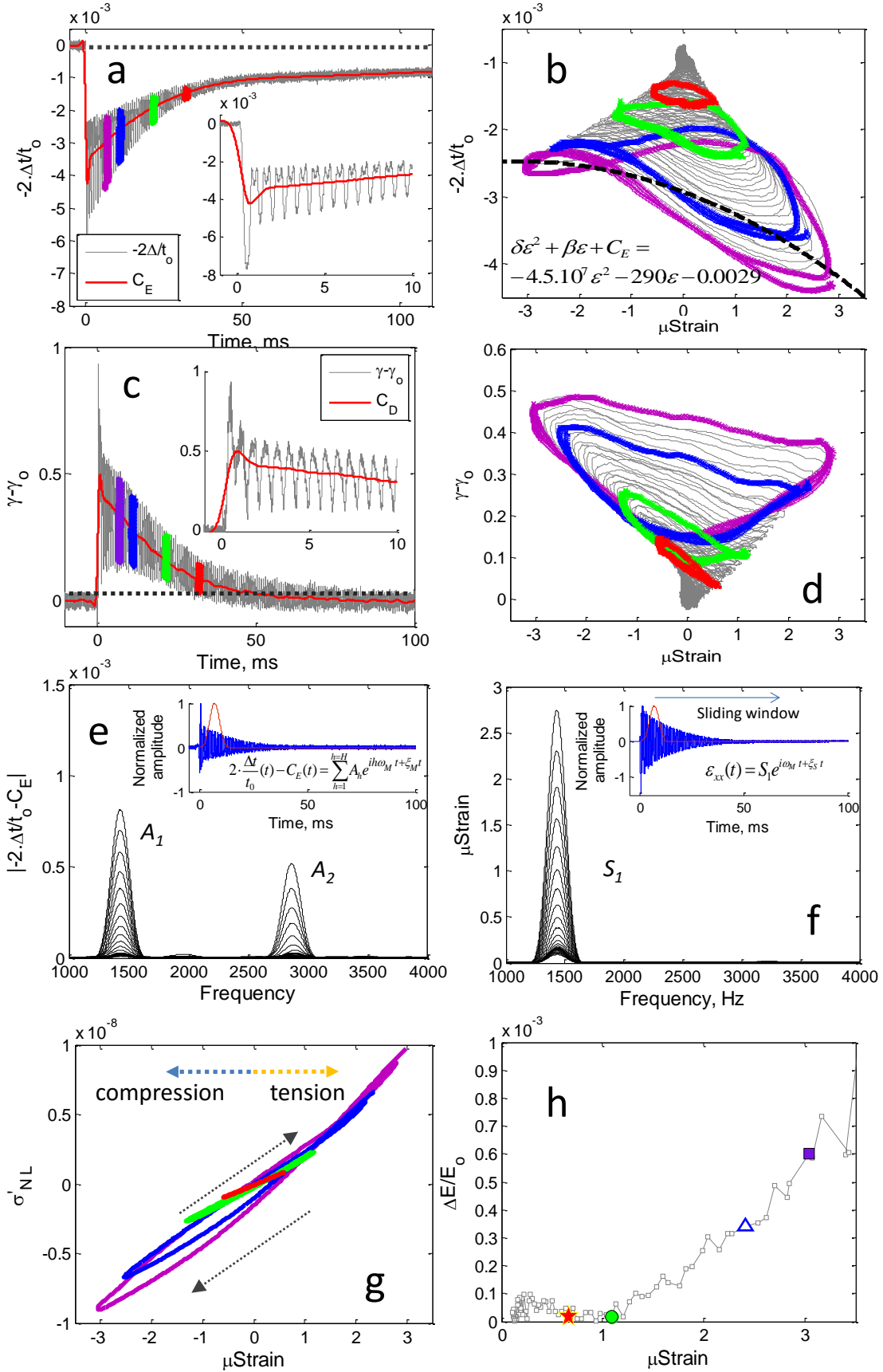
11 where σ_L and σ_{NL} are the linear and nonlinear contributions to the stress-strain relationship.
 12 However, because of the relative characteristic of the measure of material nonlinearity conducted
 13 herein, the nonlinear stress-strain counterpart can be quantified as

$$14 \quad \sigma'_{NL}(\varepsilon) = \int -2 \cdot \frac{\Delta t}{t_0} \cdot d\varepsilon, \quad (14)$$

15 where the nonlinear stress counterpart σ_{NL} is proportional to σ'_{NL} (results shown in Figure 4g). On
 16 other hand, the energy dissipated (ΔE) during a hysteretic stress-strain cycle, because of internal
 17 friction, can be quantified as³⁵

$$18 \quad \frac{\Delta E}{E_o} = \frac{\oint \sigma_{NL} \cdot d\varepsilon}{M_o \varepsilon^2}, \quad (15)$$

1 where E_o is the elastic energy stored in the specimen when the strain is a maximum. The results
2 obtained from the analysis conducted through Eqs. 14 and 15 show how the hysteretic behavior
3 progressively vanishes, as a measure that the strain amplitude decreases. As a result, the energy
4 dissipated in every cycle (Figure 4h) decreases with decreasing strain amplitude.



1 Figure 4. (Color online) Dynamic properties extracted for a single impact (force = 4.7 kN): a)
2 relative variation of time shift as a function of time; inset plot shows the cyclic response whose
3 frequency matches the first bending mode of vibration, b) relative variations of time shift as a
4 function of strain; dashed line shows a second order polynomial fit as $-2 \cdot \Delta t / t_0 = \delta' \cdot \varepsilon^2 + \beta' \cdot \varepsilon + C_E$,
5 c) variation of ultrasonic attenuation with time, d) variation of ultrasonic attenuation as a
6 function of strain, e) untangled mean-centered variations of the relative variations of time shift
7 (Eq. 11) in frequency domain, and analyzed in short windows of 10 cycles; inset plot shows the
8 corresponding time domain signal and the sliding window, f) strain signal in frequency domain,
9 g) relative nonlinear stress (σ'_{NL})-strain relationship derived from Eq. 14, and h) specific loss
10 (Eq. 15) as a consequence of the hysteretic behavior. Dashed lines in a) and c) show that the
11 recovery of the modulus takes a longer time than that found for the variation of the ultrasonic
12 attenuation.

13 When the material is investigated at different impact energies, the same behavior as
14 presented in Figure 4 is consistently reproduced (results not shown here). However, the more
15 energy is introduced by the hammer blow, the more the material modulus and attenuation are
16 offset. Figures 5a and 5b show the maximum offset of the relaxation kernels ($C_{E,min}$ and $C_{D,max}$,
17 shown as square symbols), and the behavior of C_E and C_D as a function of the strain amplitude
18 during the ring down. As regards the elastic softening of the material, it is observed that the
19 minimum value of C_E at every impact exhibits an inverse proportionality to the attained strain
20 amplitude as

$$21 \quad \min\{C_E(\Delta\varepsilon)\} = \alpha_{CE} \cdot \Delta\varepsilon, \quad (16)$$

1 which leads to a value of $\alpha_{CE} = -800$. This approach is in fact equivalent to the nonlinear
2 resonance spectroscopy-based techniques (Eq. 3), since the variable C_E represents the time-
3 averaged relative variations of modulus ($\Delta M/M_o$), as shown in previous studies¹⁸. However, the
4 behavior of $C_E(\Delta\varepsilon)$ and $C_D(\Delta\varepsilon)$ observed during ring down differs from the simple observation of
5 their maximum absolute offsets (Eq. 16 and Figures 5a and 5b), most likely because of the
6 contribution of the slow dynamic effect. Similar results were anticipated by Van Damme and
7 Van Den Abeele³⁶ in the investigation of fatigue damage in carbon fiber-reinforced polymer
8 plates. In that study, the dependence between resonant frequency and strain amplitude followed
9 different trends when the material nonlinearity was investigated at increasing excitation
10 amplitudes, and when it was investigated through the analysis of the ring down of the sample
11 (see Figure 3 in Van Damme and Van Den Abeele³⁶). These findings suggest that the relaxation
12 of the elastic properties after a nonlinear dynamic excitation is the result of the coupled
13 mechanisms of fast and slow hysteretic motions. Therefore, the instantaneous variations of the
14 resonant frequency during the ring down of the sample are a consequence of the attained strain
15 amplitude at every impact, as well as of the past history load³⁷.

16 Fast motion hysteresis is thought to be driven by the recovery of asperities that are
17 eventually broken during the strain field generated on resonance of the sample³⁸. Conversely, a
18 different volume of asperities remains in metastable equilibrium once the dynamic excitation
19 vanishes, and will return to the initial state because of thermal fluctuations³⁸ and other
20 mechanisms not yet established¹¹ which takes place on a longer timescale (slow dynamics).
21 Since the two mechanisms seem to coexist during a dynamic excitation¹⁰, the elastic softening C_E
22 can be written as the superposition of fast and slow contributions as

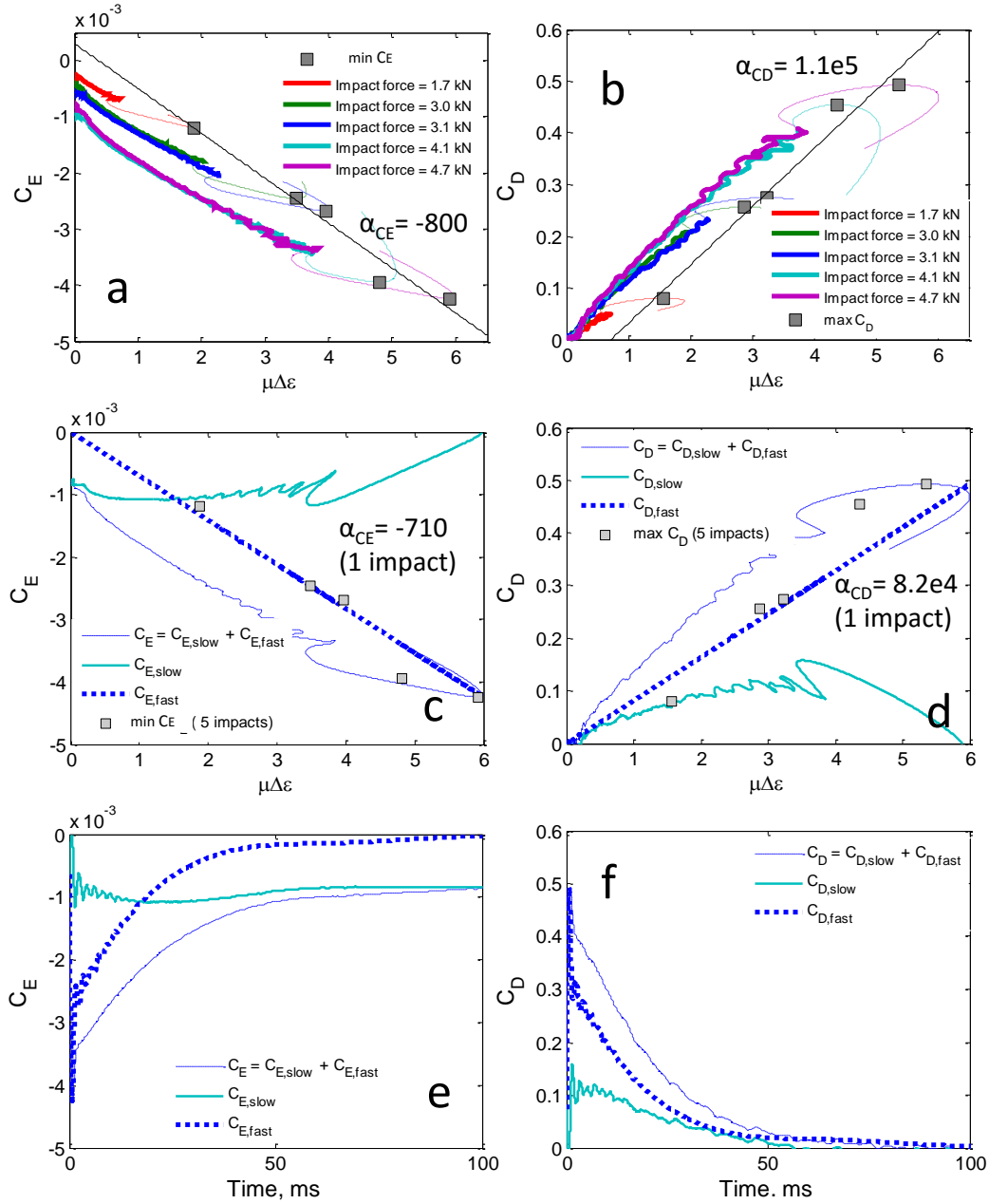
1 $C_E = C_{E,fast} + C_{E,slow};$ (17)

2 and analogously for C_D . If it is assumed that the fast hysteretic motion is mainly driven by the
 3 onset of the strain amplitude signal, the relaxation kernel owed to slow hysteretic motion ($C_{E,slow}$)
 4 can be found by scaling the envelope of the strain amplitude—in the case that only one resonant
 5 mode dominates the dynamic response—and subtracting it from the total contribution as

6 $C_{E,slow} = C_E - \min\{C_E\} \frac{\Delta\epsilon(t)}{\Delta\epsilon_{\max}}.$ (18)

7 We can therefore assume that the fast dynamic effect ($C_{E,fast}$) is proportional to the strain
 8 amplitude—as shown in Eq. 16—while the deviation of the proportionality behavior of C_E is
 9 gathered in the variable $C_{E,slow}$. The value of α_{CE} after untangling the slow dynamic contribution
 10 for a single impact is $\alpha_{CE} = -710$ (Figure 5c), and approximately the same for every different
 11 hammer blow. Interestingly, these values are close to those obtained through the quantification
 12 of α_{CE} from the minimum offset obtained for five impacts (Eq.16). Likewise for the analysis
 13 conducted on C_D (Figures 5b and 5d). Therefore, after the slow dynamics of a single impact are
 14 untangled, the estimation of the parameter α approximates to the value obtained with a nonlinear
 15 resonant spectroscopy approach, wherein the effect of the slow dynamics is minimized¹⁰. The
 16 contribution of the slow dynamics becomes more evident when the signals are shown as a
 17 function of time. Figures 5e and 5f show the untangled fast and slow hysteretic effects on the
 18 relaxation kernels as a function of time. The fast counterpart is the scaled envelope of the strain
 19 signal (Eq. 18), while the slow contribution directs the global behavior of the relaxation kernels,
 20 C_E and C_D , away from the scaled strain signal envelope. Remarkably, the slow dynamic
 21 contribution seems to be enhanced on C_E compared with C_D . This observation is not unique.

1 Similar behavior was also observed in compressional mode DAET experiments conducted on
2 rock¹⁷. Although more research is needed in this respect, the fast hysteretic motion is most likely
3 captured by the attenuation variation (C_D) which recovers completely to the initial properties, γ_o
4 (Figure 5f), at the time that the strain signal is completely damped. This observation suggests
5 that the underlying mechanisms of slow dynamics differ in the recovery processes of the elastic
6 and attenuation properties.



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Figure 5. Relaxation kernel observed for five different impact energies, for a) the elastic properties (C_E) for five different impact energies, b) the attenuation properties (C_D). Untangled fast and slow dynamic contributions from a single impact as a function of strain amplitude, for c) C_E and d) C_D , and untangled fast and slow dynamic contributions from a single impact as a function of time, for e) C_E , and f) C_D .

1 Despite only one concrete sample was tested herein, the extent of material nonlinearity is
2 reasonable for undamaged concrete when compared with previous studies³¹⁻³³. We expect
3 however, that the full range of nonlinear characteristics extracted herein be enhanced as a
4 measure that cracking-like damage progresses. To this end, prospective work will focus in
5 assessing distributed microcracking damage and surface-breaking cracks in concrete samples. On
6 other hand, the methodology proposed in this study can be transposed to various applications
7 which involve non-steady-state vibration, such as the analysis of earth response during ground
8 motion, in structure health monitoring applications by using ambient noise, as well as on-site
9 assessment of infrastructure materials, where usually only one face is available for
10 nondestructive inspection. Some of these applications were previously devised in other
11 studies^{21,22}. However, the approach and analysis conducted here allows a more detailed
12 description of the material nonlinearity.

13 **4. Conclusions**

14 This study shows the feasibility of extracting the full nonlinear behavior of materials by
15 using a hammer blow to modulate a continuous high frequency probe. The approach was
16 demonstrated in one undamaged concrete sample. The analysis focused on investigating the
17 amplitude and phase changes produced in the continuous probe wave upon transient vibration.
18 The extracted dynamic features in this study are those expected for materials that present
19 distributed contact-like defects such as concrete or rocks: 1) the relative variations of modulus
20 with strain resembled a bow tie, 2) the modulus decreased and progressively was recovered with
21 the onset of the strain amplitude, and 3) the modulus was not completely recovered at the time
22 that the strain amplitude was completely damped. Consequently, these phenomena were also
23 reproduced on the attenuation properties. In addition, the recoveries of modulus and attenuation

1 properties (C_E and C_D) during the ring down of the sample were analyzed. The analysis consisted
2 in decomposing the modulus and attenuation recoveries in those contributions strictly
3 proportional to the strain amplitude (fast dynamic effect), and those that directed away from such
4 proportional relationship the recoveries (slow dynamic effect). After untangling the slow
5 dynamic contribution, the hysteretic parameters (α_{CE} and α_{CD}) obtained from a single impact
6 event were similar to those obtained from a nonlinear resonant spectroscopy based approach; that
7 is obtaining the hysteretic parameters from multiple impact events.

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