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Additional Information

Application of UV-Visible Light Absorption and Scattering technique to low absorption fuels under diesel-like conditions

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Abstract

Light Absorption and Scattering technique (LAS) has been applied for the measurement of fuel vapor distribution in diesel-type sprays. This technique is usually limited to fuels with relatively high absorptivity, which are sometimes not commonly used as surrogate fuels. In the present paper, a comparison of fuels with very different absorptive properties has been made to determine the range of application of the methodology. A calibration procedure has been applied to n-decane (DEC), a binary blend of n-decane and n-hexadecane (50DEC) and three blends of n-heptane with a highly-absorbing fuel (HEPB1, HEPB2 and HEPB3). This methodology enables the in-situ quantification of absorption coefficients at high pressure and temperature by creating a uniform mixture inside the cylinder. Results have been later applied for the quantification of fuel vapor distribution in sprays for DEC, 50DEC and HEPB3. Results obtained with these range of fuels have enabled to establish the limit in terms of absorption coefficient needed to get consistent results with the technique.

Keywords: UV-VIS Light Absorption and Scattering, direct injection, fuel, mixture formation, n-alkanes

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1. Nomenclature

LAS	Light Absorption and Scattering
UV	Ultraviolet light
VIS	Visible light
TDC	Top Dead Centre
DEC	n-Decane
50DEC	50%n-Decane/50%n-Hexadecane
HEP	n-Heptane
HAF	multi-component high absorption fuel
HEPB#	mixture of HEP and HAF
λ	Wavelength
MW	Molecular weight
L	Optical path
I_0	Reference light intensity
I	Attenuated light intensity
ε	Absorption coefficient
ρ_{vf}	vapor fuel partial density
Y_f	vapor fuel mass fraction
LoS	Line-of-sight
R	Ratio of droplet optical thickness at 280 and 560 nm
ϕ_{eq}	Equivalent diameter
d_0	Nozzle exit diameter
ρ_f	Fuel density at injection conditions
ρ_a	Ambient gas density
NO	Nominal thermodynamic conditions
LD	Low density thermodynamic conditions
HT	Hight temperature thermodynamic conditions

2. Introduction

Many efforts on internal combustion engine research are focused on reducing pollutant formation. The more and more restrictive regulations force the devel-

6 opment of new techniques and technologies, while improving the current ones.
7 One of the main research topics in this regard is the study of the evaporation
8 of fuel and subsequent mixing with air. Especially the latter process has been
9 proved to have a strong impact on combustion and pollutant formation in the
10 spray [1]. Over the past decades, many experimental diagnostic methods have
11 been developed in order to characterize quantitatively the fuel distribution. Ra-
12 man Spectroscopy allows the measurement of local fuel/air ratio [2]. However,
13 the low signal strength limits measurement to a reduced area and requires care-
14 ful signal-to-noise ratio considerations. In contrast, Planar Rayleigh Scattering
15 (PRS) and Planar Laser Induced Fluorescence (PLIF) present more intense sig-
16 nals and allow spatially resolved measurements. On the one hand, PRS can be
17 only applied in total absence of liquid droplets, which in practice means starting
18 measurements further downstream of the stabilized liquid length[2, 3, 4]. On the
19 other hand, PLIF has been widely employed to determine both vapor and liquid
20 phase concentrations simultaneously (Exciplex PLIF) [5, 6, 7, 8]. Nevertheless,
21 difficulties are usually found due to quenching with other molecules or cross-talk
22 between the monomer (vapor) and the exciplex (liquid) fluorescences. Besides,
23 quantitative measurements under high temperature become difficult due to a
24 strong dependence of fluorescence on this parameter [6].

25 Light Absorption and Scattering (LAS) technique is based on the fact that
26 the phenomena governing light interaction with fuel can be either absorption
27 or scattering, depending on the light spectrum and the size range of the fuel
28 particles (i.e. droplets or molecules) relative to wavelength. Mancaruso and
29 Vaglieco [9] showed extinction spectra of diesel fuel within an optical engine.
30 Their results evidence a strong absorption in the UV, mainly due to the pres-
31 ence of aromatic molecules, while the spectra in the visible range is flat, which
32 is due to liquid scattering. If absorption signal is isolated, fuel concentration
33 can be obtained by means of Lambert-Beer's law. The first applications of LAS
34 were based on the combination of infrared and visible wavelengths. However,
35 infrared extinction usually presents strong temperature dependence and it can
36 be interfered by the absorption of water vapor or heat radiation from hot sur-

37 faces. Based on the same principle, the Ultraviolet-Visible Light Absorption
38 and Scattering (UV-VIS LAS) was developed by Suzuki [10], and improved by
39 Zhang [11] for application under high pressure and high temperature condi-
40 tions. UV-VIS LAS is not influenced by water vapor or heat radiation, and
41 temperature dependence is weaker than in other techniques. Besides, as both
42 wavelengths are relatively close, several simplifications can be applied without
43 affecting measurement accuracy. Therefore, UV-VIS LAS technique is regarded
44 as a promising tool for quantitative measurement of concentration distribution
45 for fuel sprays, in presence of both vapor and liquid.

46 One of the main requirements for the application of UV-VIS LAS is that the
47 fuel under study has to be absorbent in the near UV range (between 250 and
48 300 nm) while being transparent for the visible wavelengths. The absorption
49 spectrum strongly depends on the fuel molecule itself. Most of the implemen-
50 tations available in the literature use complex fuels with high UV absorptivity
51 [10, 11, 12, 13, 14, 15]. With the aim of expanding the range of application of
52 UV-VIS LAS, this work addresses the application of this technique to measure
53 fuel vapor distribution of two n-alkanes under diesel-like conditions. These type
54 of fuels have been commonly used as surrogates of more complex ones. However,
55 they present low absorption in the near-UV range. In current work, n-Decane
56 and a 50% mass blend of n-Decane and n-Hexadecane have been investigated.
57 In parallel, some more absorptive fuel blends have also been evaluated and com-
58 pared with the other two to analyse the validity of results obtained. In addition,
59 a calibration methodology for in-situ measurements of the absorption coefficient
60 of each fuel is presented and validated.

61 **3. Experimental methodology**

62 *3.1. Experimental facility*

63 Tests have been performed in an optically accessible single cylinder engine.
64 A detailed description can be found at [16]. The facility is based on a 2-stroke
65 single cylinder engine (Jenbach JW 50), with 3 liter displacement. The engine is

66 motored at low speed (500 rpm) and the intake and exhaust processes are han-
67 dled by transfers on the liner. A schematic of the engine is depicted in figure 1
68 (left). The facility has been operated under non-reactive conditions in a closed
69 loop mode, where in-cylinder air is fully replaced by nitrogen. When the exhaust
70 gases leave the cylinder, they flow through an intercooler and a cyclonic filter
71 to remove the rests of fuel and oil. This ensures proper operating conditions
72 for a roots compressor, which is used to assist the engine charge management.
73 In-cylinder thermodynamic conditions are controlled by the intake air temper-
74 ature and pressure. The first one is regulated by two sets of electrical resistors.
75 Between them, the circuit is refilled with nitrogen through an electronic valve to
76 achieve the desired intake pressure, compensating blow-by and leak losses. The
77 engine is operated under skip-fired mode, so that in-cylinder conditions are not
78 influenced by the remaining residual gases from previous combustion/injection
79 cycles and temperature transients are avoided. Hence, an injection takes place
80 every 30 cycles.

81 The cylinder head is specially designed to provide optical access to the
82 combustion chamber, which was designed with a cylindrical shape in order to
83 avoid wall impingement. The effective compression ratio is 15.7. The chamber
84 presents an upper port, where the injector is located, and four lateral accesses.
85 A pressure transducer is installed in one of them, whereas the other three are
86 equipped with oval-shaped quartz windows (88 mm long, 37 mm large and 28
87 mm thick). The cylinder head and engine temperature are controlled by a
88 coolant recirculation system. Temperature was set to 353 K, to guarantee good
89 lubricity.

90 A common-rail injection system was used, with a single-hole piezoelectric
91 injector. The orifice had conical shape (Ks factor equal to 1.5), with an outlet
92 diameter of 140 μm and 1 mm length. The injected mass is so low that thermo-
93 dynamic conditions inside the combustion chamber are barely affected by the
94 fuel evaporation. Temperature of the injector holder cooling was the same as
95 for the cylinder head. Hence, due to the low injection frequency, the injected
96 fuel temperature can be considered the same.

97 *3.2. Operating conditions*

98 An experimental matrix has been designed, which includes variations of both
99 in-cylinder pressure and temperature. A nominal point has been defined (NO),
100 together with lower density (LD) and higher temperature (HT) points. Com-
101 pared to NO, LD is obtained by lowering intake pressure at constant temper-
102 ature, while the HT is obtained by increasing intake temperature at constant
103 pressure. Conditions inside the cylinder have been calculated from measured
104 in-cylinder pressure, using a first-law thermodynamic analysis. A similar pro-
105 cedure has been followed in [16, 17], where a detailed explanation can be found.
106 It takes into account blow-by, heat losses and mechanical deformations. The
107 trapped mass is estimated using the intake temperature and volume at the
108 exhaust vent closure. Therefore, temperature along the engine cycle can be
109 calculated using the equation of state and correcting the air trapped mass with
110 blow-by estimations. Air mass and density are also required for the absorption
111 calibration methodology, as described in the upcoming sections. In-cylinder
112 pressure trace and the derived gas density for the three operating points is pre-
113 sented in figure 1(right). The injection pressure was set at 100 MPa for all the
114 cases.

115 The vapor fuel concentration has been measured for n-Decane (DEC) and a
116 50% blend in mass of n-Decane and n-Hexadecane (50DEC). A more absorptive
117 fuel has been also employed, which was obtained by diluting a highly absorption
118 blend of diferent single-component fuels (HAF), and pure n-Heptane (HEP),
119 which has a negligible absorption coefficient. All fuels were purchased with a
120 95% purity. Different blending dilutions have been considered to span a range
121 of absorption coefficient values of the blend. A summary of the composition of
122 the different fuels is summarized in table 1.

123 **4. UV-VIS LAS Methodology**

124 When light is transmitted through a mixture of vapor and droplets, it is
125 attenuated according to the Bouguer-Lambert-Beer law as follows:

$$\ln\left(\frac{I_0}{I}\right) = \int_0^L \frac{1}{MW} \varepsilon(\lambda) \rho_{vf} 100 dx + \int_0^L Q_{ext}(\lambda) dx \quad (1)$$

126 where λ is wavelength, ε is the absorption coefficient of fuel vapor ($l \cdot mol^{-1} \cdot$
 127 cm^{-1}), ρ_{vf} is the vapor fuel partial density (kg/m^3), MW is the molecular
 128 weight of fuel (g/mol), L is the optical path length (m), and Q_{ext} is the extinc-
 129 tion coefficient of a cloud of droplets. The first term on the right side of equa-
 130 tion 1 corresponds to light attenuation due to absorption by vapor molecules,
 131 while the second term is the extinction due to droplets, which includes scattering
 132 and absorption losses.

133 UV-VIS LAS is based on the combination of attenuation measurements at
 134 two wavelengths, the first one in the ultraviolet (UV) range and the other in
 135 the visible (VIS) range. In this work, 280 and 560 nm were chosen. Two main
 136 hypotheses are assumed:

- 137 • Fuel molecules will not absorb light in the visible range neither in the form
 138 of droplets nor in vapor phase.
- 139 • UV absorption by fuel droplets is negligible compared to scattering.

140 Suzuki et al. [10] evaluated the drop optical thickness at 280 nm and 560 nm
 141 for α -dimethylnaphthalene and concluded that the hypothesis of non-absorbance
 142 from liquid droplets can be applied. However, close to the nozzle region a
 143 certain error can affect the measurement since the droplet number density is
 144 too high. This error is minimized if vapor optical thickness dominates the total
 145 extinction. If both 280 and 560 signals are combined, the following expression
 146 can be derived from equation 1:

$$\overline{\rho_{vf}} = \frac{MW}{100 \cdot \varepsilon(\lambda_{UV})} \left[\ln\left(\frac{I_0}{I}\right)_{UV} - R \ln\left(\frac{I_0}{I}\right)_{VIS} \right] \quad (2)$$

147 where $\overline{\rho_{vf}}$ is the average vapor partial density along the optical path, as LAS
 148 technique is based on Line-of-Sight (LoS) measurements. The R term is the ratio
 149 of the drop optical thickness at the two wavelengths. From now on, the term
 150 within brackets will be referred to as *absorption*. Billings et al. [18] examined

151 the variation in drop optical thickness for their application at 3390 nm and 632
152 nm. Calculations conducted for the present work show similar results in the
153 UV-VIS range [19, 20]. It was observed that R varies mainly with the droplet
154 diameter (for two fixed wavelengths). Below 25 μm , R varies between 0.9 and
155 1 while for droplets larger than this size R is almost 1. For the present work a
156 range between 20 and 60 μm was considered and an average value of $R=0.976$
157 was used.

158 The optical set-up is presented in figure 2. A continuous broadband 1000 W
159 Hg(Xenon) Arc lamp was utilized, in combination with a diaphragm and a
160 diffuser to create a uniform point light source. This lamp is characterized by
161 a continuous emission spectra from 250 to 2400 nm, with enough intensity to
162 replace the commonly used Nd:YAG pulsed laser [11, 12, 13, 14, 15]. A parabolic
163 mirror of 150 mm diameter was employed to create a collimated light beam
164 through the combustion chamber. The light beam is collected at the other
165 side of the engine by a 75-mm diameter quartz lens, which focalizes light on
166 both detectors. A square quartz beam splitter (50 mm side) was positioned
167 just after the lens, to divide light in two different beams (50% transmitted
168 and 50% reflected intensity in the whole working range). For both UV and
169 VIS wavelengths, a digital ICCD image system Andor iStar was utilized, with
170 a 50 μs exposure time. Light was filtered just prior to the detectors by two
171 interference filters centred at 280 and 560 nm respectively (10 nm FWHM).

172 Simultaneously to LAS measurements, MIE scattering images from the liq-
173 uid droplets were registered to identify the maximum liquid length. For that
174 purpose, a third camera (ICCD LaVision Dynamight) was utilized, due to the
175 low intensity of the collimated light beam. The procedure followed to register
176 and process the signal is described in [21].

177 *4.1. Absorption coefficient calibration*

178 According to the literature, the absorption coefficient can be strongly af-
179 fected by thermodynamic conditions [12, 22]. Moreover, significant differences
180 have been reported among different fuels. For this reason, it becomes necessary

181 to characterize the absorptivity of each fuel under engine operating conditions.
182 A methodology is proposed in the current work, based on creating a homo-
183 geneous mixture inside the cylinder with known concentration, temperature
184 and pressure. Thus, if light absorption is measured under this conditions, it
185 is possible to apply equation 2 to obtain the absorption coefficient at known
186 thermodynamic conditions.

187 Trapped air mass and in-cylinder density were derived from the pressure
188 signal, while the amount of fuel injected was previously measured for all the
189 fuels, as described in [23, 24]. Then, the average fuel mass fraction (\overline{Y}_f) inside
190 the cylinder can be calculated and hence $\overline{\rho}_{vf} = \overline{Y}_f \cdot \rho_c$. In order to achieve
191 the homogeneous mixture, fuel was injected early in the cycle, just after the
192 transfer closing (at -80.5 CAD). Due to the large displacement of the engine,
193 long energizing times and high injection pressures were required to introduce
194 enough mass of fuel to obtain a measurable concentration.

195 Characterization was performed for the different blends at the operating con-
196 ditions summarized in table 2. For each test conditions, 50 images of the light
197 beam with fuel (I) and without fuel (I_0) were registered alternatively. Each
198 set of images was averaged and by comparing to the vapor concentration, the
199 absorption coefficient was calculated. For most of the cases, the procedure was
200 repeated at different crankangle positions after TDC, which made it possible
201 to calculate ε for different combinations of pressure and temperature caused
202 by piston motion. Moreover, measurements at different in-cylinder conditions
203 but at the same crankangle positions enabled the comparison of points with the
204 same pressure but different temperature or vice-versa.

205 Finally, the absorption spectrum and the absorption coefficient at 280 nm
206 were measured at standard temperature and pressure (STP), for all the fuels
207 included in the calibration process. The same light source as the one described
208 previously was used, while absorption was measured with a UV-VIS spectrom-
209 eter AvaSpec 2048 L and quartz sample cell of 5 mm.

210 *4.2. Spray measurements*

211 For measurements of fuel spatial distribution within sprays under engine
212 conditions, a long energizing time was set so the spray was stabilized before the
213 injection finished. The injector was triggered at 6 CAD before TDC, while the
214 actual injection started approximately at 5 CAD before TDC. The energizing
215 time of the injector was set to 2000 μs (6 CAD) resulting on a total injection
216 duration of around 5000 μs (15 CAD) due to the hydraulic delay. Images were
217 taken at -3 CAD (1000 μs after injector was triggered) before TDC.

218 The reduced size of the neutral density filter limited the field of view, so all
219 the receiving optics were spatially shifted to measure the whole spray, with a
220 precision translational stage. Light was registered at three positions along the
221 spray axis. The effective length of the field of view was 45 mm, while the optics
222 were displaced 25 mm between two consecutive positions. Thus, an overlap of
223 20 mm was ensured, which was the base to merge the three images into a single
224 one.

225 For each test condition and measuring position, 50 images were registered.
226 Each set of images was averaged, merged and finally the attenuation was cal-
227 culated at each wavelength. The VIS signal is spatially transformed to obtain
228 the best correspondence pixel by pixel with the UV signal. This transformation
229 comprises translation, rotation and scaling. Then, the vapor absorption signal
230 was calculated $(\ln(I_0/I)_{UV} - R \ln(I_0/I)_{VIS})$.

231 At this point, the result is line-of-sight integrated. Thus, a deconvolution
232 (inversion) algorithm is required to obtain the corresponding signal at the sym-
233 metry plane of the spray. This algorithm is applied to one half of the spray, thus
234 the original absorption signal is divided into two halves (along spray axis), which
235 are averaged before applying the deconvolution algorithm. The Onion-Peeling
236 method is the most commonly used algorithm for numerical deconvolution (in-
237 version) of a LoS attenuation signal [11, 12, 25, 26]. Nevertheless, in this work,
238 the Three-Point Abel Inversion was chosen, as it has some advantages in terms
239 of noise when comparing with the Onion-Pelling [27] method. Besides, it was
240 combined with the Tikhonov regularization methodology [26, 28] to minimized

241 the influence of noise over deconvoluted signal. A regularization parameter has
242 to be optimized for each radial profile of the spray along its axis, to improve
243 accuracy of the algorithm. In this regard, an automatic selection method was
244 employed, proposed by Åkesson et al. [28].

245 Eventually, equation 2 is utilized to calculate the vapor fuel partial density
246 (ρ_{vf}) from the deconvoluted attenuation signal. It has to be noted that the
247 form at which this equation has been presented corresponds to the calculation
248 of the LoS averaged partial density ($\overline{\rho_{vf}}$). When applying this equation to the
249 symmetry plane, the optical path (L) considered is the minimum spatial unit
250 (i.e. 1 pixel). To solve the possible dependence of the absorption coefficient with
251 local temperature, a mixing model (state relationship) was employed, which is
252 based on the assumption that the mixture state corresponds to the result of
253 an adiabatic mixing process. Therefore, it is possible to correlate the local
254 fuel partial density with its temperature. A detailed description is presented
255 in [29]. Pressure within the spray has been assumed to be the same as for the
256 surrounding gas. The state relationship was also utilized to obtain the fuel mass
257 fraction distribution from the fuel partial density.

258 **5. Results and discussion**

259 *5.1. Absorption coefficient*

260 As previously presented, absorption coefficient measurements were performed
261 according to the conditions in table 2. Figure 3 shows an assembled image of at-
262 tenuation at 280 nm, which was obtained by injecting 54.37 mg of DEC (actual
263 injection timing from -80.5 to -45 CAD), at in-cylinder conditions corresponding
264 to the LD case. The overall spatial distribution of attenuation is practically ho-
265 mogeneous along the whole combustion chamber, so that optics shifting was not
266 necessary for calibration. Therefore, attenuation at 280 and 560 nm was mea-
267 sured only at the centre of the optical access. Images also indicate the existence
268 of small scale inhomogeneities, which are most probably due to beam steer-
269 ing, as such a pattern can also be observed in the background part of schlieren

270 images [21]. For all cases, measured attenuation at 560 nm was one order of
271 magnitude lower than the standard deviation from the image sample, i.e. signal
272 is in the range of the background noise, which confirmed the initial hypothesis
273 of no absorption by vapor in the visible.

274 In figure 4, average ε values at 280 nm are presented for the investigated
275 fuels. The first point (lowest temperature) of each series correspond to the value
276 obtained at 0.1 MPa and 298 K (STP), measured with the spectrometer. The
277 rest of the points correspond to different combinations of mean temperature and
278 pressure inside the cylinder at the moment of image acquisition. The comparison
279 of two series with similar in-cylinder pressure at TDC enables the analysis of
280 the temperature influence, while the comparison of two series with similar in-
281 cylinder temperature at TDC makes it possible to study the effect of pressure.
282 For the sake of clarity, different engine conditions are only shown for HEPB3.
283 For this fuel, the absorption coefficient corresponding to 560 nm has been also
284 included. Results show that this value is negligible, confirming the hypothesis
285 that no absorption occurs at this wavelength.

286 The absorption coefficient can be observed to increase with the fraction of
287 aromatic fuels (HEPB1 to HEPB3), while DEC and 50DEC present significantly
288 lower ε values. For all the fuels, a large difference in ε is observed between STP
289 and engine conditions. Note that the STP is intended here to be used only as a
290 reference for the in-cylinder measured values. Furthermore, little sensitivity to
291 in-cylinder pressure and temperature can be observed for the different blends.
292 This is consistent with results presented by Zhang et al. [11], who reported
293 a large reduction of the absorption coefficient when pressure and temperature
294 increased. However, above a certain level (ambient pressure above 3 MPa and
295 ambient temperature above 650 K) this sensitivity tends to decrease. More-
296 over, the sensitivity is clearly dependent on the type of molecule, as they report
297 variations of the absorption coefficient around 10% for 1,3-Dimethylnaphtalene
298 and 60% for α -Methylnaphtalene, when temperature changed from 575 to 650
299 K at 3 MPa ambient pressure. Yamakawa et al. [12] also reported that the
300 absorption coefficient of p-xylene is almost not affected by thermodynamic con-

301 ditions above 1.5 MPa and 400 K. Summing up, literature results conclude that
302 the sensitivity of ε to ambient thermodynamic conditions tends to minimize or
303 even disappear at high pressures and temperatures, which is consistent with the
304 results presented in this work.

305 A similar behaviour is observed for DEC and 50DEC. Furthermore, for these
306 two low absorption fuels two different energizing times have been used (table 2),
307 and therefore two ε values can be observed at each ambient condition, which
308 fall onto each other. On the one hand, this indicates that the procedure is
309 independent of the injected mass. On the other hand, it also confirms that
310 the hypothesis of complete evaporation of the fuel is valid for the investigated
311 conditions, and discards any systematic error on $(\overline{\rho_{vf}})$ calculation due to spray
312 wall impingement or liquid formation.

313 5.2. Signal-to-noise considerations for spray measurements

314 LoS attenuation along spray axis is depicted for HEPB3 (upper plot) and
315 50DEC (lower plot) in figure 5 for 280 nm, 560 nm and the corresponding dif-
316 ference. Data correspond to NO conditions. Closer to the nozzle, visible and UV
317 signals are similar as scattering dominates due to the low amount of vaporized
318 fuel. At some point (depending of the amount of vaporized fuel and the absorp-
319 tion coefficient), the visible signal becomes lower and the single contribution
320 of the vapor absorption can be measured. In this figure, it is possible to see
321 that the net vapor absorption signal calculated for 50DEC is of the same order
322 of magnitude as the attenuation obtained for the visible wavelength. If this
323 last signal is considered as noise (mainly caused by beam steering), the corre-
324 sponding signal-to-noise ratio (calculated between 25 and 50 mm) is 1.80. In
325 contrast, the attenuation of HEPB3 at 280 nm is in general one order of magni-
326 tude higher. Even closest to the nozzle, where the dense liquid region is located,
327 some vapor absorption signal can be detected. In this case, the signal-to-noise
328 ratio is 26.46. Regarding DEC, a similar calculation was performed resulting
329 on a signal-to-noise ratio of 4.89, which is closer to 50DEC than to HEPB3.
330 These results evidence the advantage of using highly absorbing fuels to obtaine

331 reliable measurements under the investigated conditions.

332 5.3. Spray measurements

333 The ε calibration procedure has to be validated to guarantee the reliability
334 of results. For this purpose, the vapor fuel distribution was measured and
335 compared for the three HEPB blends at LD conditions. In figure 6 (upper plot),
336 the partial density of the three fuels are compared. A peak can be observed in
337 the fuel concentration evolution, which is a good estimation of the location of the
338 stabilized liquid length. Similar fuel concentrations were obtained for the three
339 fuels downstream of the peak, where the spray is fully vaporized. This result
340 is consistent with the fact that mass flow rate and spray momentum flux show
341 almost no change among blends, which should result in a very similar mixing
342 process for all three cases[21, 29]. For distances shorter than the maximum liquid
343 length, the fraction of each component in the vapor phase is unknown, and thus
344 the absorption coefficient cannot be strictly applied, as it was obtained only for
345 a fully vaporized mixture. For this reason, differences larger than expected are
346 observable upstream of the peak values of each case.

347 The second aspect that needs to be validated is the sensitivity of ε to in-
348 cylinder pressure and temperature. According to the results presented in figure
349 4, a constant value of ε has been used to obtain Y_f for each fuel, under different
350 thermodynamic conditions. The vapor fuel mass fraction of HEPB3 is shown
351 in figure 6 (lower plot), for the three operating conditions described in figure 1.
352 Data corresponds to the value along the spray axis. The X-coordinate of each
353 curve has been normalized with the equivalent diameter [30], which is defined
354 as $\phi_{eq} = d_0 \sqrt{\rho_f / \rho_a}$, where d_0 is the nozzle exit diameter, ρ_f is the fuel density
355 and ρ_a is the ambient gas density. The normalization process should enable the
356 comparison of all three cases at the same entrainment coordinate. The three
357 distributions are observed to collapse after the normalization, which confirms
358 that results are consistent. Therefore, it can be stated that ε is independent of
359 thermodynamic conditions for the fuels and operating conditions considered in
360 this study, as expected.

361 In figure 7, the vapor mass fraction on the spray axis is compared for the
362 three ambient densities presented in figure 1 and DEC, 50DEC and HEPB3.
363 In the highly dense liquid region (i.e. first 10 - 20 mm), results for 50DEC
364 are not plotted due to the extremely high noise observed. The low amount of
365 vaporized fuel, in combination with a low absorption coefficient, leads to large
366 uncertainties on the experimental data. Thus, despite offering promising results
367 with DEC, the methodology described in this work can be observed not to be
368 sensitive enough to characterize this region if low absorption fuels such as the
369 50DEC are considered. **Nevertheless, it is important to highlight that**
370 **near the maximum liquid length, the technique is able to measure**
371 **the vapor fuel distribution in presence of liquid, even for 50DEC.**
372 **Although uncertainties on the accuracy within this region exist due**
373 **to the presence of droplets, this does not rule out the qualitative**
374 **evaluation of the vaporized fuel mass fraction.**

375 Downstream of the maximum vapor mass fraction, liquid is completely evap-
376 orated, air entrainment continues and fuel mass fraction decreases until the tip
377 of the spray is reached. Along this region, mass fraction distribution for both
378 HEPB3 and DEC coincide. However, it is not the case for 50DEC. For the NO
379 and LD cases, higher values of Y_f were obtained for this fuel in comparison
380 with the other two. When 50DEC and HEPB3 are compared (from 25 to 35
381 mm), differences are around 20% for the NO point and 40% for LD. As from
382 the previous section, the calibration methodology was able to characterize low
383 ε values. However, it can be observed (figure 4) that all the fuels present a
384 similar standard deviation, despite the fact that the value of ε can be more
385 than one order of magnitude different. The main consequence is that, while for
386 HEPB3 the deviation accounts for a maximum of 5% of the mean value, in the
387 case of 50DEC the standard deviation reaches almost 50% of the mean value.
388 This leaves much uncertainty over the calculated average value of ε , which di-
389 rectly affects Y_f distributions estimations. Based on these arguments on the
390 calibration, as well as on the evolution of on-axis fuel mass fraction in figures 5
391 and 7, it can be stated that 50DEC represents a limitation of sensitivity for

392 the methodology described in this work. The minimum threshold in absorption
393 coefficient for the adequate application seems to be between that of 50DEC and
394 DEC, as the latter fuel seems to be enough to improve significantly accuracy
395 and quality of results to acceptable levels.

396 Gas jet theory (e.g. [31]) shows that, in the fully vaporized region of the
397 spray, the fuel mass fraction should have self-similar profiles. This means that
398 fuel mass fraction normalized by the corresponding on-axis value ($(Y_f/Y_{f,cl})$)
399 should only be dependent of the radial to axial coordinates (R/X). From a
400 similar point of view, the radii where 15%, 50% and 90% of $Y_{f,cl}$ is located
401 should be a constant, if divided by the axial coordinate. This actually the type
402 of result that is shown in Figure 8 for DEC, 50DEC and HEPB3 and the three
403 test conditions defined in figure 1. Data below 15% have not been considered
404 in this analysis due to the large uncertainties observed in the outer regions of
405 spray and the low signal-to-noise ratio, especially for the low absorption fuels.

406 The first thing to be noticed is that the data scatter is, in general, smaller for
407 HEPB3 than for the other two fuels. Nevertheless, for 90%, a certain variability
408 is observed for all of them. It has been previously reported in the literature
409 [26, 27] that the deconvolution algorithm introduces errors close to the axis.
410 Besides, the numerical procedure followed in this work tends to flatten them
411 around this region, hindering the accurate calculation of the radii. A second
412 aspect to note is that, in general, radii values for the three fuels are similar. This
413 suggests that the discrepancies for 50DEC, reported in figure 7, are related to the
414 value of the absorption coefficient. As ε has been shown to have no dependency
415 on pressure or temperature, in practice it acts like a proportionality constant
416 to convert attenuation into fuel concentration. Therefore, when profiles are
417 normalized, the effect of the absorption coefficient is removed and the three
418 fuels are similar. Finally, the flat trends observed for the normalized radii
419 versus axial distance confirm that the radial profiles are self-similar in the fully
420 vaporized region.

421 The normalization of radial profiles depends on the accuracy with which the
422 numerical procedure is able to reconstruct the symmetry plane of the spray. As

423 commented above, these algorithms tend to accumulate errors at the inner parts
424 of the radial profiles [27]. To determine the effect of this issue on the global shape
425 of the inverted profiles and its normalization, a fitted Gaussian curve has been
426 compared with the experimental data, which is a profile shape usually found in
427 the literature. The fitting algorithm is based on the maximum gradient descent
428 methodology (according to Palomares [32]) and the experimental data consid-
429 ered for this purpose is the one comprised between 15% and 90% of the $Y_{f,cl}$.
430 In figure 9 (upper plot), a comparison between experimental and fitted radial
431 profiles is shown. Data corresponds to HEPB3 and DEC, at NO thermody-
432 namic conditions. It can be seen that the agreement between experimental and
433 fitted distributions is high in the range considered for the calculation. However,
434 as expected, the fitted curve presents higher values near the spray axis. This
435 comparison also reveals another region (especially for the DEC profiles), where
436 the Gaussian trend is not followed, namely the edge of the spray. In figure 9
437 (lower plot), the ensemble averaged normalized profiles calculated between 25
438 and 35 mm for HEPB3 and DEC at NO conditions are shown. In this case, it is
439 possible to see that the fitted profiles of the two fuels are more similar than the
440 experimental ones, which highlights the effect of noise over the deconvolution
441 algorithm. In case of DEC, with relatively higher noise, the experimental profile
442 does not even present a Gaussian shape, which is a more accurate description
443 for HPB3 measurements.

444 6. Conclusions

445 The UV-VIS LAS technique has been proposed to characterize the air-fuel
446 mixing process of two low absorption fuels (i.e. DEC and 50DEC). Three
447 additional fuels with progressively higher absorptivity (HEPB1, HEPB2 and
448 HEPB3) have also been characterized in order to compare and evaluate the
449 accuracy and reliability of the technique and the results obtained for the first
450 ones.

451 A calibration procedure has been designed to obtain in-situ measurements

452 of the absorption coefficient, using the same optical set-up as the one proposed
453 for spray measurements. For the conditions and fuels used in the calibration
454 procedure, the following conclusions were obtained:

- 455 • Fuel-air mixture inside the chamber was found to be homogeneous and
456 the absorption coefficient calculation was found to be independent of fuel
457 concentration
- 458 • Experimental results show that the methodology is sensitive to fuel prop-
459 erties.
- 460 • Measured ε values suggested a negligible sensitivity of this parameter to
461 pressure or temperature. These results have been also validated exper-
462 imentally, thanks to the consistence observed between fuel distributions
463 measured at different engine operating conditions
- 464 • It has been possible to characterize ε for low absorption fuels like DEC and
465 50DEC. However, results present uncertainties, which could even achieve
466 the 50% of the average value.

467 The values of ε have been used to obtain the fuel vapor distribution for DEC,
468 50DEC and HEPB3, from which following conclusions have been drawn:

- 469 • **Measurements of the vapor fuel distribution near liquid length**
470 **have been obtained for all the fuels, although uncertainties exist**
471 **in regions where droplets are present.**
- 472 • Accuracy and quality of results decrease with the absorption coefficient.
473 Similar results have been obtained for HEPB3 and DEC, while for 50DEC
474 values higher than expected have been measured.

475 Considering all the foregoing arguments, the methodology described in this
476 paper is of limited applicability when trying to characterize fuels with absorption
477 properties in the range of 50DEC. Fuels with $\varepsilon > 11 \text{ l mol}^{-1} \text{ cm}^{-1}$, such as
478 DEC, are suitable for this methodology. Furthermore, the larger the ε values,

479 the higher the validity of the results, as signal strength improves with this
480 parameter.

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Component	DEC	50DEC	HEPB1	HEPB2	HEPB3
n-heptane	0	0	95.3	93.5	91.7
n-decane	100	50	0	0	0
n-dodecane	0	0	0.5	0.6	0.8
n-hexadecane	0	50	1.4	1.9	2.4
n-octadecane	0	0	0.9	1.3	1.7
n-eicosane	0	0	0.6	0.8	1.1
1-methylnaphtalene	0	0	1.2	1.7	2.1
n-butylbenzene	0	0	0.1	0.2	0.2

Table 1: Fuel composition in percentage (mass) for the present study.

Fuel	ρ_f at 373 K [kg/m ³]	Test point	P_{inj} [MPa]	Energizing Time [μ s]	Total injected mass [mg/cc]	CAD of interest
DEC	669.1	LD, NO, HT	150	4500, 9000	37.53, 54.37	0, 6, 12, 18
50DEC	693.9	LD, NO, HT	150	4500, 9000	39.70, 64.03	0, 6, 12, 18
HEPB3	668.3	LD, NO, HT	150	9000	60.25	0, 6, 12, 18, 24, 30
HEPB2	666.3	LD	150	9000	60.07	0, 6, 12, 18, 24, 30
HEPB1	664.2	LD	150	9000	59.88	0, 6, 12, 18, 24, 30

Table 2: Test conditions and fuel properties for the absorption coefficient calibration.

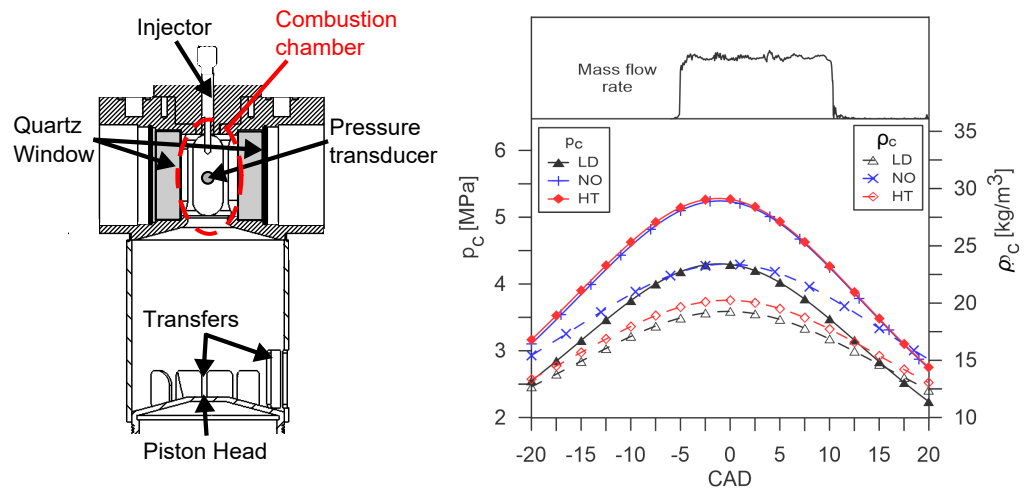


Figure 1: Scheme of the arrangement of the cylinder head and the liner (left). Evolution of in-cylinder pressure and density for the three operating points (right).

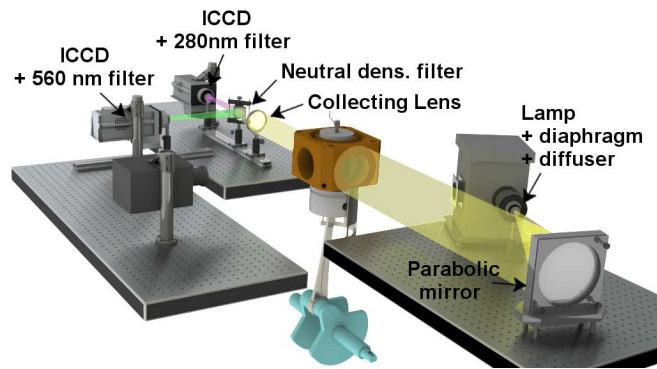


Figure 2: Scheme of the UV-VIS LAS optical set-up.

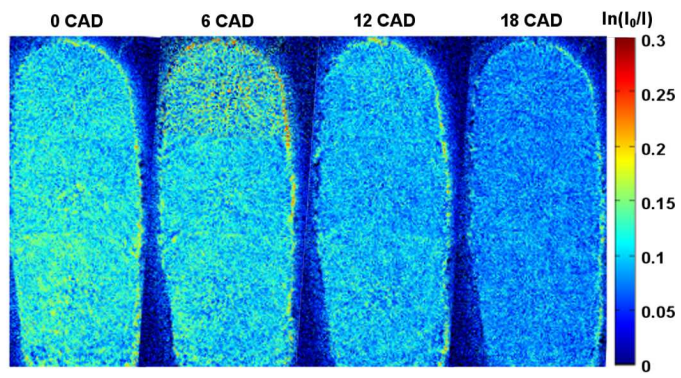


Figure 3: Example of in-cylinder homogeneous attenuation corresponding to 280 nm for DEC, at LD thermodynamic conditions.

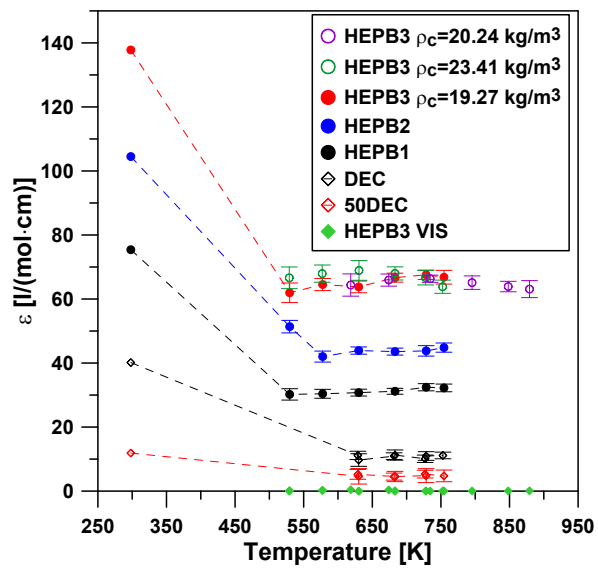


Figure 4: UV Absorption coefficient as a function of temperature for all the fuels and different engine conditions. VIS absorption coefficient is also included for HEPB3

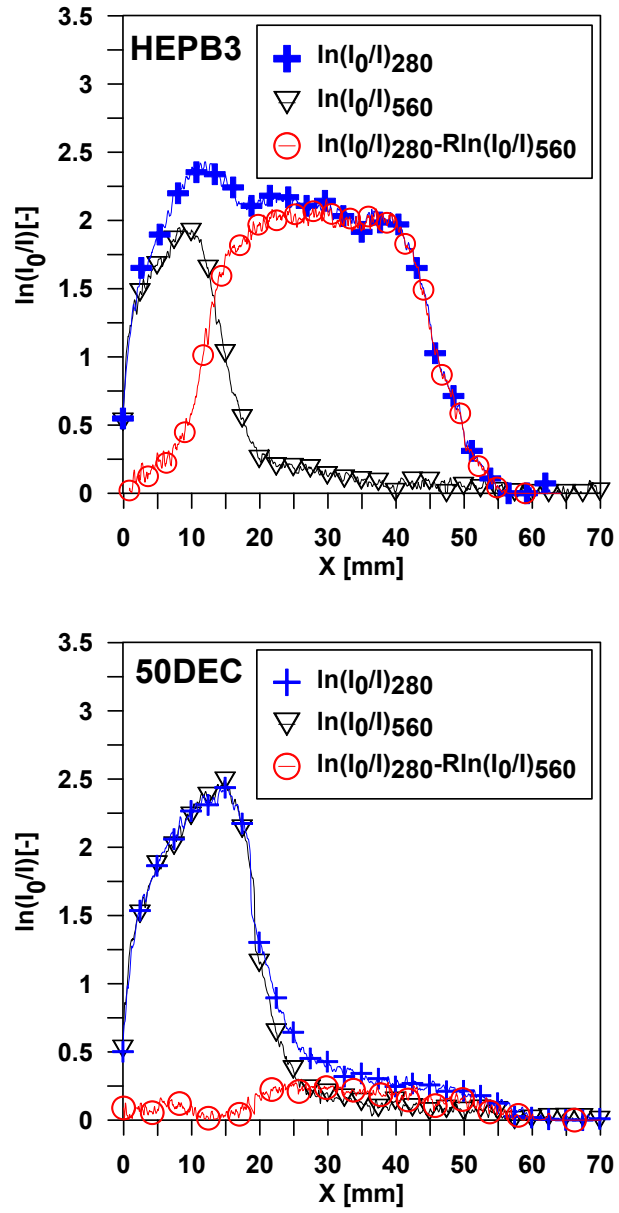


Figure 5: LoS attenuation for 560 and 280 nm on the spray axis for HEPB3 (upper plot) and 50DEC (lower plot). Data corresponds to NO conditions at -3 CAD before TDC.

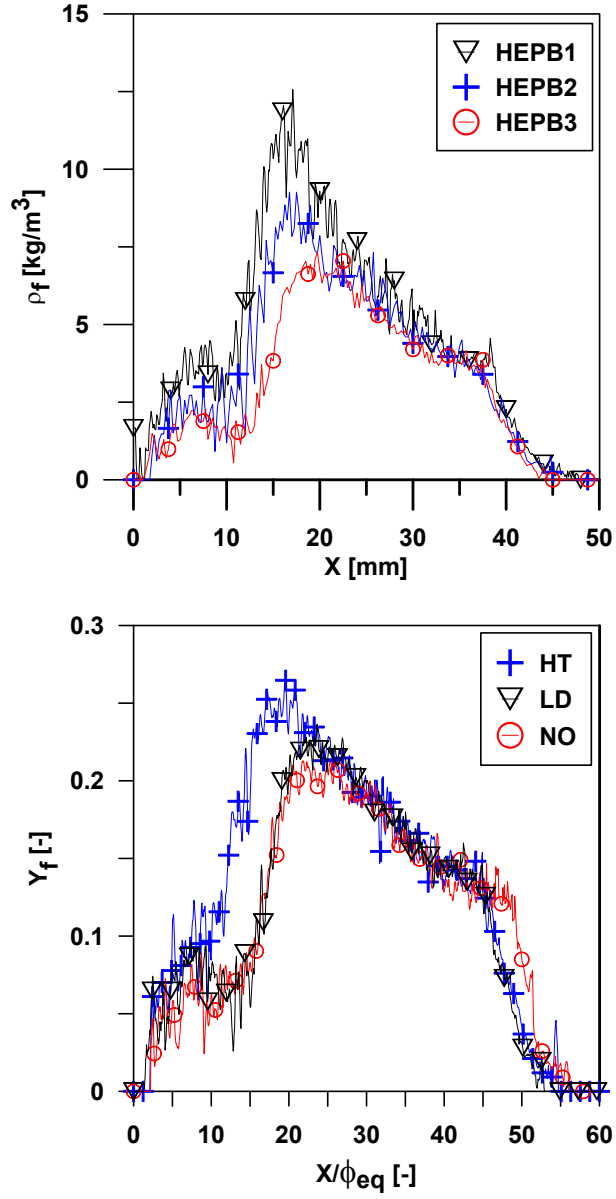


Figure 6: On-axis distribution of ρ_{vf} for the three mixtures of HEP and HAF, at LD conditions (upper plot). On-axis Y_f distribution of HEPB3, for the three conditions defined in figure 1 (lower plot). All the data were obtained at -3 CAD before TDC.

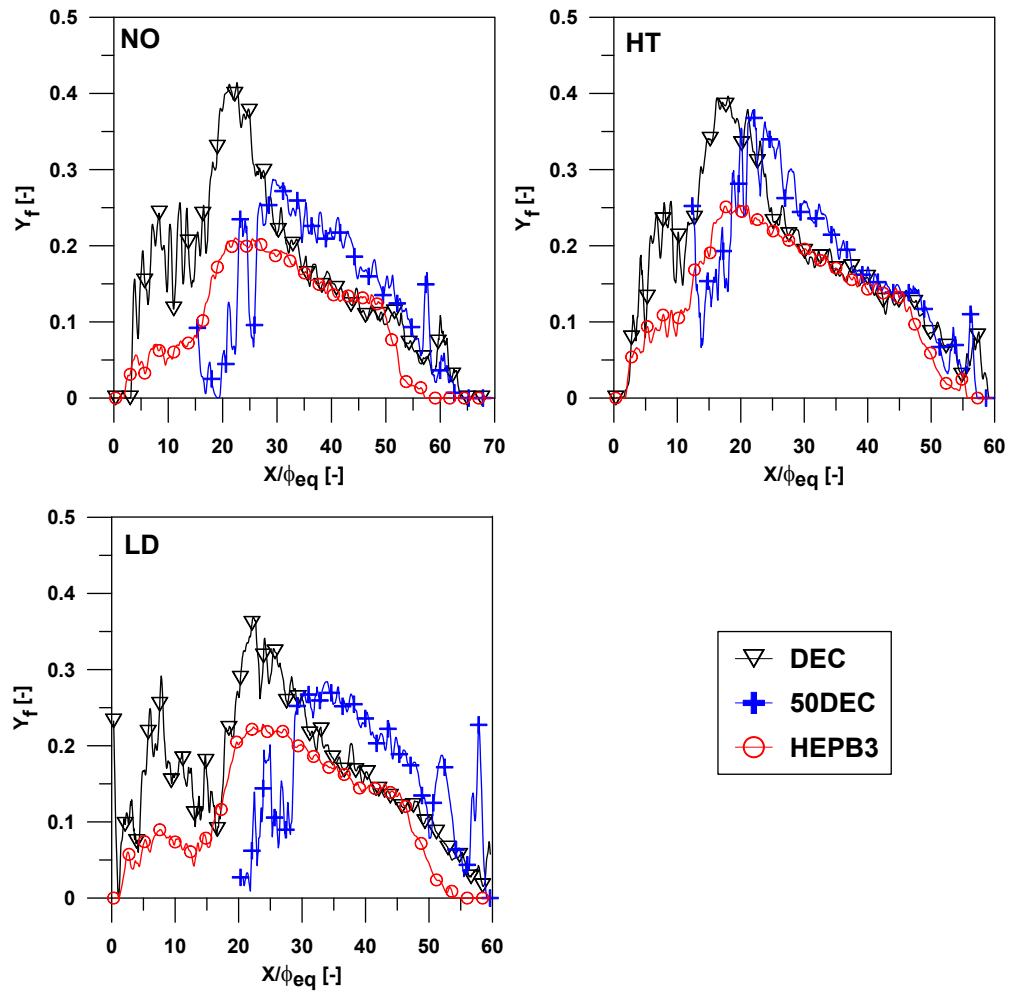


Figure 7: On axis distribution of Y_f of DEC, 50DEC and HEPB3. Data corresponds to the thermodynamic conditions defined in figure 1 at -3 CAD before TDC.

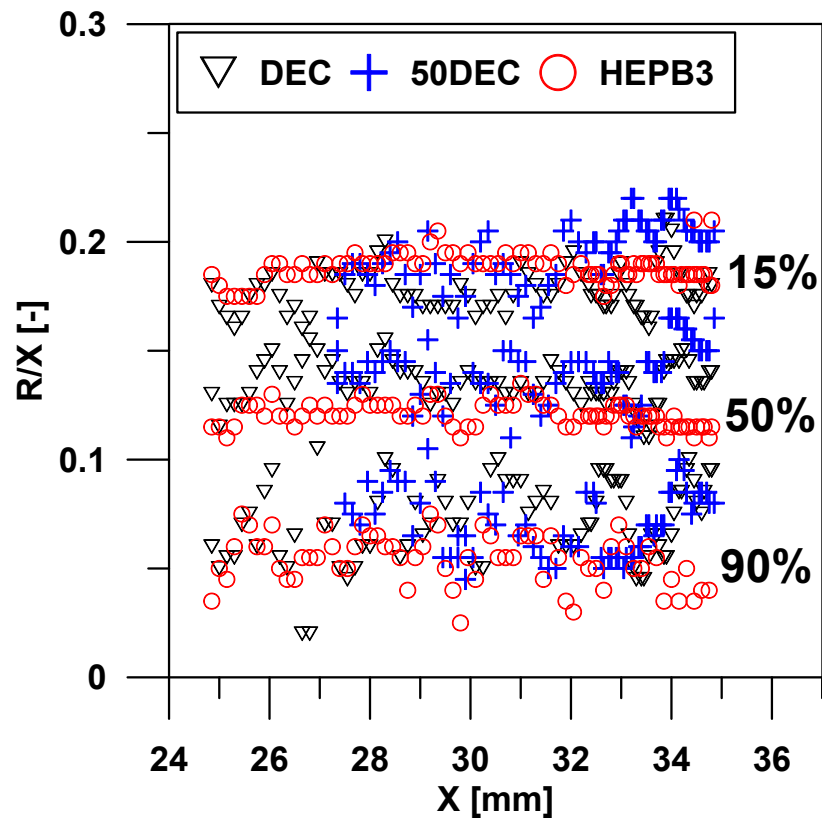


Figure 8: Radii for 15%, 50% and 90% of $Y_{f,cl}$ for DEC, 50DEC and HEPB3 along the spray axis. Data corresponds to the three operating conditions defined in figure 1 at -3 CAD before TDC.

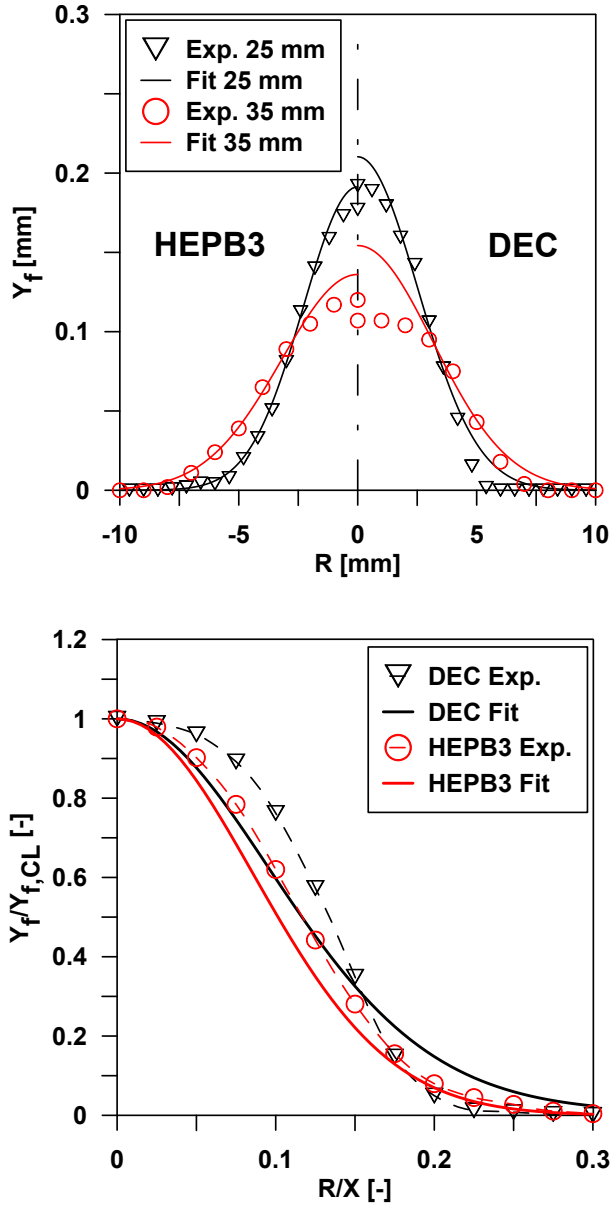


Figure 9: Comparison between experimental and fitted exponential curve for DEC and HEPB3, at 25 and 35 mm (upper plot). Comparison between experimental and fitted normalized profiles, averaged between 25 and 35 mm (lower plot). Data corresponds to NO thermodynamic conditions at -3 CAD before TDC.