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Additional Information

Rotor Bar Breakage Mechanism and Prognosis in an Induction Motor

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Abstract— This paper proposes a Condition-Based Maintenance and Prognostics and Health Management (CBM/PHM) methodology for a rotor bar in an induction motor, based on the results of a fatigue test intended to reproduce in the most natural way a bar breakage in order to carry out a comparison between transient and stationary diagnosis methods for incipient fault detection. Newly developed techniques in stator current transient analysis have allowed tracking the developing fault during the last part of the test, identifying the failure mechanism and establishing a physical model of the process. This non-linear failure model is applied by a Particle Filtering algorithm to diagnose the defect at an early stage and predict the Remaining Useful Life (RUL) of the bar. An initial generalization of the results to conditions differing from the ones under which the fatigue test was developed is studied.

Index Terms—Induction motors, rotors, fault diagnosis, failure analysis, fatigue, frequency-domain analysis, spectrogram, particle filters, prognostics and health management, remaining life assessment

NOMENCLATURE

a	Crack length.
C	Paris' equation constant.
e	Normalized energy.
f	Supply frequency.
I_1	Amplitude of the main current component.
I_{LSH}	Amplitude of the current's low sideband harmonic.
I_{USH}	Amplitude of the current's upper sideband harmonic.
I_0	Amplitude of the DC component.
I_{2fs}	Amplitude of the harmonic component at $2fs$ Hz.
I_{4fs}	Amplitude of the harmonic component at $4fs$ Hz.
i	Current waveform.
K_{lc}	Low cycle fatigue's equation constant.
K_p	Global Paris' equation constant.
k	Particle number.
m	Paris' equation exponent.
n	Predicted cycle number ($n=1,2,\dots,N$)
N	Predicted failure cycle
n_d	Process noise for the state variable

n_{lc}	Process noise for the asymmetry indicator, low cycle fatigue
n_p	Process noise for the asymmetry indicator, crack propagation
s	slip
t	Cycle number.
t_c	Time interval for integration.
$w_i(k)$	k -th particle weight at cycle t .
x_t^*	Predicted State of the system at startup t .
γ_{F1}	FFT's rotor asymmetry indicator (LSH).
γ_{F2}	FFT's rotor asymmetry indicator (LSH).
$\gamma_{Hilbert}$	Hilbert's rotor asymmetry indicator.
γ_χ	Wavelet's rotor asymmetry indicator of band χ
γ_W	Wigner-Ville's rotor asymmetry indicator.
γ_W^*	Predicted Wigner-Ville's rotor asymmetry indicator.
ΔK	Stress intensity factor.

I. INTRODUCTION

The new maintenance paradigm for critical equipment advocates for the determination of its current and probable future state of health with the aim of maximizing the asset's operational availability by optimally scheduled repairing activities. In order to achieve this objective, a complete Condition-Based Maintenance and Prognostics and Health Management (CBM/PHM) system must be developed, requiring a better comprehension of the physics behind the failure and their evolution. Incipient fault detection and failure models are critical stages in this methodology since, for the first case, prognostics rely on early detection to carry out a timely and trustworthy prediction, therefore the fault's detection threshold should be reduced as far as it is practical; and secondly, the development of physics-based modeling of components allows a better understanding of the influence of external factors, the failure mechanisms and its propagation times [1].

In addition, the non-linear nature of such physical models, and in general, the inherent complexity of the tasks involved, recommends the use of newly developed fault detection and prognosis algorithms, over existing ones such as the Kalman filter [2]. Particle Filtering has been developed as a powerful method with low computational cost [3], combining a model and data driven approaches [4-6], where model deviations are corrected by new measurements using Bayesian theory, having already been applied to predict the Remaining Useful Life (RUL) of equipment [1, 6].

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The aim of this paper consist of, following the framework outlined in [1], introducing a full CBM/PHM procedure for diagnosing and estimating the RUL of a rotor bar in an induction motor, based on the results of a fatigue test intended to develop in the most natural way a bar breakage. The architecture proposed here (Fig. 1) could be extended to the failure of more than one bar or other elements in the secondary circuit of an induction motor to predict the RUL of the whole machine.

With that objective, this work develops a natural bar breakage, validates a method for incipient fault detection of rotor asymmetries, identifies low cycle thermal fatigue as the bar's failure mechanism, links stator current derived indicators with crack progression in the cage, fits a crack progression expression to one of such indicators, uses that expression in a Condition-Based Maintenance and Prognosis and Health Management methodology based on Particle Filtering and finally studies an initial generalization of the results for startups whose severity differs to the one carried out during the fatigue test.

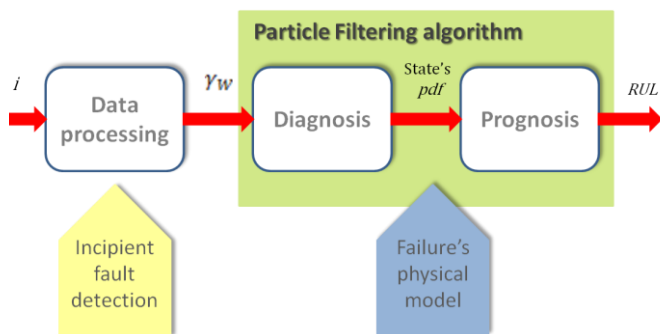


Fig. 1. Proposed CBM/PHM procedure. The results of the fatigue test have allowed establishing an incipient fault detection procedure and a physical model of the breakage needed for a correct prognosis of the fault.

Rotor asymmetries, and among them, bar and end ring breakages, are responsible for up to 10% of the failures of induction motors. Generally, this fault affects large units featuring a fabricated cage [7] and started at grid's frequency under high inertia, with high downtime and repair costs and often difficult replacement.

The diagnosis of rotor asymmetries has been a subject of detailed study in the last years, applying, in order to detect it, a wide battery of signal processing methods to magnitudes such as vibrations [8], voltages [9], flux and currents [7]; prevailing the techniques belonging to this later group due to their effectiveness and easy application, especially the ones that use the decomposition in frequency of the current waveform, known as Motor Current Signature Analysis (MCSA), which are of widespread use in the industry.

Recently, research has been focused on special aspects of rotor asymmetries, such as the load's influence in the traditional MCSA-based diagnosis [10], the effect of rotor ducts [11], partial breakage detection [12, 13, 14] and diagnosis in transient regime [15], under which the FFT analysis proves useless. In fact, one of the seminal works on transient current analysis, by Elder et Al., pointed out towards

combining both of these approaches [16] in the study of direct startup transient since under such condition the machine is under severe electrical and mechanical stress, and thus faults as broken rotor bars could be detected at an earlier stage. The fatigue test whose results are presented in this paper was conceived to prove this hypothesis.

Destructive studies of rotor bars have already being carried out by Cabanas et Al. in [17], in a combined FEM and experimental approach that computed forces acting on this component under its most stressful working conditions, that is, during a direct on-line startup, and later cyclically replicated them on test especimens. The conclusions of this work showed that the mechanical loads alone on the bars of a constructed cage are not responsible for their failure, even if manufacturing defects exist, thus recognizing in this study the importance of thermal aspects [18, 19] in the development of the fault.

Furthermore, the authors in [17] compare their results with high cycle fatigue models, failure theory that is focused in evolutions ranging between 10^4 to 10^6 cycles, whilst large motor manufacturers design the rotor cages for just several thousand startups [20]. Therefore, just drawing the right conclusions from already published data, it can be hypothesized that a bar breakage should be modeled using low-cycle thermal fatigue expressions [21] (conceived for 10^2 to 10^3 cycles). In this case, crack propagation models, just spanning from 10^2 to 10^3 cycles and therefore worthless under high fatigue, may explain a significant part of the damage's evolution (Fig. 1 of [17]).

Thus, the main contributions of this work are:

- Providing for the first time a procedure to accurately monitor the evolution of a rotor bar breakage in large induction motors by comparing different techniques during a fatigue test. Previous studies have only been focused on diagnosing partial breakages [12, 13, 14].
- Applying for the first time crack propagation expressions in order to properly model that observed evolution, as opposed to the high cycle fatigue ones used in [17, 18, 19].
- Confirming the effect of constructive rotor asymmetries in the incipient detection of this fault, not detected in previous partial breakage studies [12, 13, 14].
- Yielding a complete CBM/PHM system able to monitor and predict the state of one rotor bar.
- Studying a first generalization of the results to different severity startups.

For these purposes, this work is structured as follows: in Section II the experimental procedure used for development of the fault –already introduced in [22]– is completed, Section III validates a method to track the evolution of the fault and expands the results presented in [23], followed in Section IV by a discussion of them, linking transient current indicators to the evolution of the breakage. This enables in Section V to propose a physical model of the failure and introducing a novel state estimation and prognosis module using the Particle Filtering approach based on this fault model, and finally,

determining and comparing their fault detection thresholds. Three indicators based on the study of the stator current during the last 10 seconds of stationary operation of each test were calculated.

1) The first one compares the amplitude of the low sideband harmonic, LSH, to the amplitude of the fundamental component from the current spectrum (1). In order to increase its accuracy, a Blackman window was used to avoid leakage effects and the Teager-Kaiser algorithm, having a minimal computational cost, was utilized to enhance the automatic peak detection [24] in an interval determined by the encoder reading.

$$\gamma_{F1}(dB) = 20 \cdot \log \left(\frac{I_1}{I_{LSH}} \right) \quad (1)$$

2) The second indicator modifies (1) also taking into account the amplitude of the upper sideband harmonic (2), as it is recommended in the bibliography [7].

$$\gamma_{F2}(dB) = 20 \cdot \log \left(\frac{I_1}{I_{LSH} + I_{USH}} \right) \quad (2)$$

3) Finally, also for the stationary operation, a third indicator was computed. It is based on studying the spectrum of the envelope obtained from the application of the Hilbert transform to the stator current [15]. This enables the suppression of the main current component and thus improved sideband detection. Their energy is compared to the amplitude of the DC component (3).

$$\gamma_{Hilbert}(dB) = 20 \cdot \log \left(\frac{I_0}{I_{2fs} + I_{4fs}} \right) \quad (3)$$

C. Transient indicators computed

For transient operation, Wavelet-based indicators (4) and the integration of the energy of the LSH using the Wigner-Ville distribution (5), both having low computational requirements and not needing a speed profile, were computed:

4) The Wavelet indicator [22], takes into account the energy present in a whole band whose limits are determined by the decomposition, by comparing the energy present in one detail or approximation χ with the energy in the original current waveform i , between the limits Ns and Nb , to avoid edge effects (4).

$$\gamma_{\chi}(dB) = 10 \cdot \log \left[\frac{\sum_{j=Nb}^{Ns} i_j^2}{\sum_{j=Nb}^{Ns} [\chi(j)]^2} \right] \quad (4)$$

5) The Wigner-Ville indicator, proposed in [25], was developed during the fatigue test with the objective of reducing the influence of noise and other minor amplitude components evolving during the startup that caused a high value of the standard deviation (Table II) for the most critical Wavelet indicators (4), which could hinder adequate incipient fault detection. In this case, the Wigner-Ville transform is utilized to obtain a time-frequency distribution of the energy of the current in a range whose limits can be set freely by use of band pass and notch filters. This time-frequency boxes are selected to isolate prevailing fault harmonics, whose evolution

can be tracked by means of the computation of the instantaneous frequency and their energy $e_{if,bw}$ integrated on that track, therefore avoiding contributions of other harmonics or interferences (5):

$$\gamma_w(dB) = 10 \cdot \log \left[\frac{2 \sum_{t_c} i_j^2}{e_{if,bw}} \right] \quad (5)$$

The selected time-frequency box studied here corresponds to the last part of the evolution of the LSH harmonic, between 0 and 45 Hz during the second part of the startup, in which the effect of cage's resistance is predominant in the rotor currents. The indicator (5) has been normalized, yielding values comparable to the application of (1) and (2) to the analysis of a rotor asymmetry in stationary operation.

III. RESULTS OF THE FATIGUE TEST

The fatigue test comprises 82,265 cycles. As no evolution was seen in the indicators, the rotor was progressively weakened. This procedure was carried out removing material from the end rings, in their junction with the rotor bars. In [22] the results of a symmetric lathing process, carried out after test number 33,477, were presented. In spite of that, the prolongation of the test up to 79,007 cycles showed no trend in the bar breakage indicators (both for stationary and transient operation), and therefore it was decided to further debilitate one of the bars. No procedure was carried out to select a specific one.

A. Modifications of the rotor

In order to proceed with this weakening, two holes were drilled in the junction of a bar to the short circuit ring, inside the groove left by the lathing process. The diameter of the holes drilled, initially 3 mm, was increased to 4 mm (bar section reduced to 14%), and then to 4.5 mm one of them (bar section reduced to 7%), until the bar breakage developed.

This machining modified the original structure of the casted squirrel cage, becoming it similar to a welded one, since the lathing process increased the cantilevered disposition of the end rings, and the holes near the bar produced a configuration alike to a bar protruding from the rotor iron to reach the ring.

B. Indicators behavior

Table I shows the average value of the stationary indicators (1), (2), (3) during the fatigue test. The first epoch corresponds to a healthy rotor. The second one spans the operation of the rotor cage after the lathing process, which removed 70% of the material in contact with the bars. Finally, the third epoch covers the final startups of the fatigue test, when one bar was weakened drilling two holes around its junction to the end ring. Each indicator is calculated for both phases A and C. Table II introduces the results of the study of the LSH during the startup using the transient indicators (4), (5).

The difference of value before and after the bar breakage is noticeable for all the cases. It is remarkable that a clear increase in all indicators appears when the remaining section of the bar was reduced to 14% of the original one (see columns designed as "compensation" and (a1) in Fig. 5, 6, 7).

TABLE I
EVOLUTION OF THE STATIONARY INDICATORS DURING THE FATIGUE TEST.

Indicator Phase Cycles→		First epoch		Second epoch		Third epoch			
		1 – 33,477		33,478 – 79,007		Compensation 80,405 – 80,857		After Breakage 81,216 – 81,653	
		$\bar{\gamma}$	std(γ)	$\bar{\gamma}$	std(γ)	$\bar{\gamma}$	std(γ)	$\bar{\gamma}$	std(γ)
$\gamma_{F1(1)}$	A	42.8	0.8	40.9	1.3	49.3	0.5	39.8	0.5
	C	42.5	0.6	40.8	0.9	46	0.5	40.1	0.5
$\gamma_{F2(2)}$	A	41.3	0.7	39.5	1.1	46.2	0.4	37.5	0.5
	C	41	0.5	39.3	0.7	43.2	0.4	37.6	0.4
$\gamma_{Hilbert, (3)}$	A	46.7	1.1	45.7	1.3	47.5	1.1	43.6	1.1
	C	46.7	1.1	45.4	1.1	46.3	1	44.2	1

TABLE II
EVOLUTION OF THE TRANSIENT INDICATORS DURING THE FATIGUE TEST.

Indicator Phase Cycles→		First epoch		Second epoch		Third epoch			
		1 – 33,477		33,478 – 79,007		Compensation 80,405 – 80,857		After Breakage 81,216 – 81,653	
		$\bar{\gamma}$	std(γ)	$\bar{\gamma}$	std(γ)	$\bar{\gamma}$	std(γ)	$\bar{\gamma}$	std(γ)
$\gamma_{Apr6, (4)}$	A	46.2	0.3	44	0.5	47.2	0.3	35.7	0.2
	C	46.2	0.3	44	0.3	47.4	0.3	35.9	0.2
$\gamma_{Det7, (4)}$	A	46.5	0.3	44.5	0.5	47.7	0.3	36.8	0.3
	C	46.5	0.3	44.5	0.3	47.7	0.3	37	0.3
$\gamma_{Apr7, (4)}$	A	58.1	1.9	53.9	1.4	57.2	1.3	42.2	0.8
	C	58.3	2	53.9	1.1	58.8	1.1	42.8	0.9
$\gamma_{Det8, (4)}$	A	61.1	1	54.7	1.4	59.6	0.9	42.7	0.9
	C	61.2	1	54.7	1.2	59.9	0.9	43.1	0.9
$\gamma_W, (5)$	A	50	0.8	43.5	0.3	55.7	0.9	33.5	0.1
	C	50.1	0.7	43.5	0.3	56.2	1	33.9	0.1

This behavior is due to the fact that the LSH produced by the forced incipient fault is combined with the small LSH due to the inherent asymmetry of the cage, performing as a double incipient fault; but as it is shown in [26] and depending on the relative position of both fault components (incipient and inherent LSH in this case), the resultant LSH can grow or can be compensated reducing its value, as it happens in the performed test, yielding higher indicators.

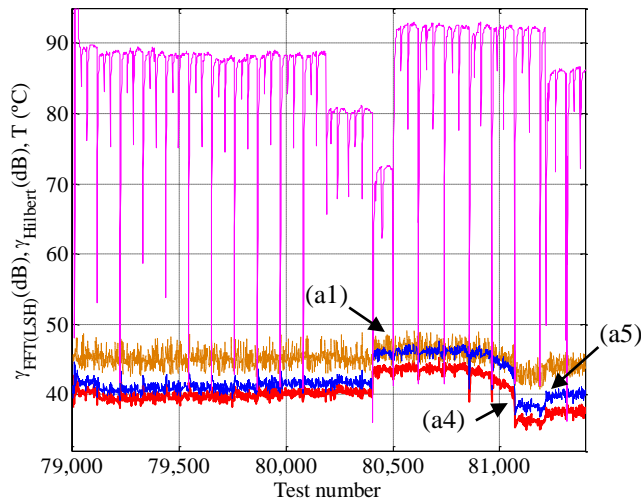


Fig. 5. Stationary rotor asymmetry indicators and stator temperature during the last stage of the fatigue test, phase C. Red, FFT indicator $\gamma_{F2(2)}$; Blue, FFT indicator $\gamma_{F1(1)}$; Brown, Hilbert indicator $\gamma_{Hilbert(3)}$. Rose, stator temperature.

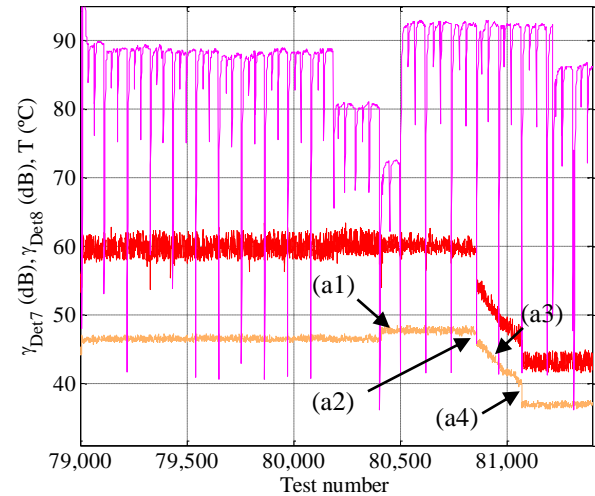


Fig. 6. Transient rotor asymmetry Wavelet indicators and stator temperature during the last stage of the fatigue test, phase C. Orange, Wavelet Detail 7 indicator $\gamma_{Det7, (4)}$; red, Wavelet Detail 8 indicator $\gamma_{Det8, (4)}$. Rose, stator

This result is interesting, since it suggests that an incipient fault, in its initial stage, should be detected through a slight variation of the fault indicators (increase or decrease) and not necessarily through a decrease of these parameters.

Moreover, if Fig. 5 showing the stationary indicators is compared to Fig. 6 and 7, which present the transient ones, a much clearer evolution is detected in the last ones once the section of the bar was reduced to 7% of its original value. First

of all, the increase of the diameter of one drill from 4 mm to 4.5 mm is clearly presented by a sudden reduction of the indicators (a2). Further, an unambiguous trend in the transient indicators is observed, as the bar weakens (a3). The full breakage occurred after the second startup following a long stop in which the machine reached room temperature (a4).

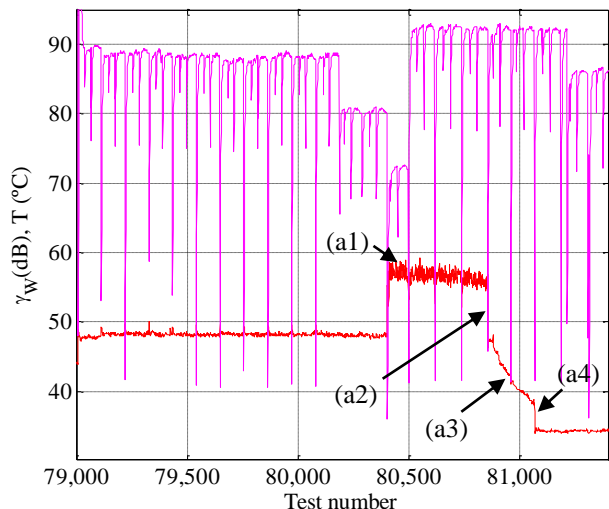


Fig. 7. Transient rotor asymmetry Wigner-Ville-based indicator γ_w (5) (red) for phase C, and stator temperature during the last stage of the fatigue test (rose).

Thus, the conclusions of the fatigue test support the hypothesis presented in [16]: the study of the direct startup of an induction motor yields a more accurate view of its rotor asymmetry [23].

C. Bar inspection

Once the rotor was inspected after the indicators stabilized, surprisingly no apparent damage was seen. The bridge between the holes was still in its position. However, a detailed comparison between the pictures taken before and after the breakage (Fig. 8 and Fig. 9) showed some damage, material lost at the junction between the ring and the bar and a crack down the 4 mm drill.

Fig. 10 presents a combination of pictures showing this crack (b2) using materials lenses, whose limited depth of field only allow to correctly focusing a few millimeters of the hole at a time. The crack clearly continues deep inside it. The cavities left by the material lost in the upper side of the bar are well seen (b1). Furthermore, an aluminum drop hanging from the hollow where it was formed is also identified (b3).

IV. DISCUSSION OF THE RESULTS

The bar breakage process achieved during the fatigue test showed a more complicated pattern than initially thought.

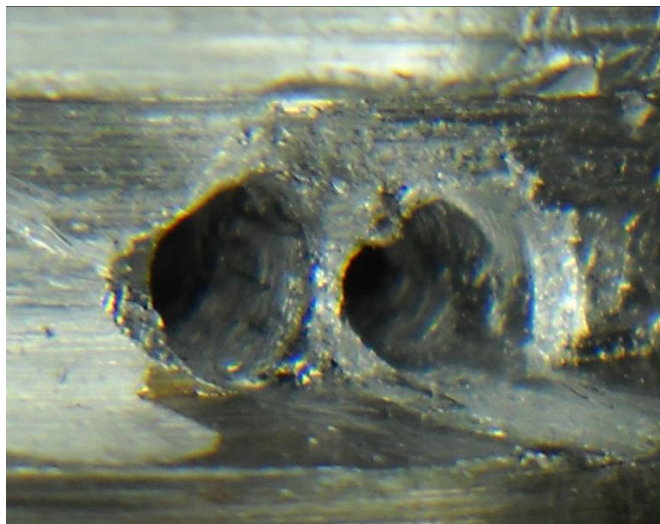


Fig. 8. Bridge between the bar and the end ring before the breakage.

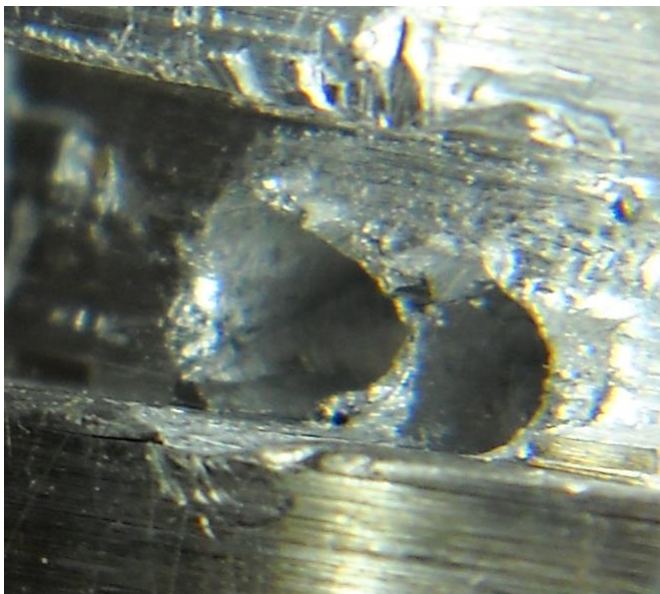


Fig. 9. Bridge between the bar and the ring after the breakage.

A. Compensation of a previous rotor asymmetry

It is observed that the evolution of the rotor asymmetry indicators, contrary to what it is expected, increases its value as a bar is weakened (Fig. 5, 6, 7 and Tables I and II). This effect is attributed to the compensation of a slight previous asymmetry due to a constructive defect in a bar at half the pole pitch of the one that was weakened, as it is indicated in [26]. The study of the rotor asymmetry high order harmonics during the breakage confirms this hypothesis [27], which is also supported by the results of subjecting a faulty motor of the same characteristics, used in [25], to the same cycling. As shown in Table III, the initial inherent rotor asymmetry of this machine was lower and all the indicators suffer a wider variation.

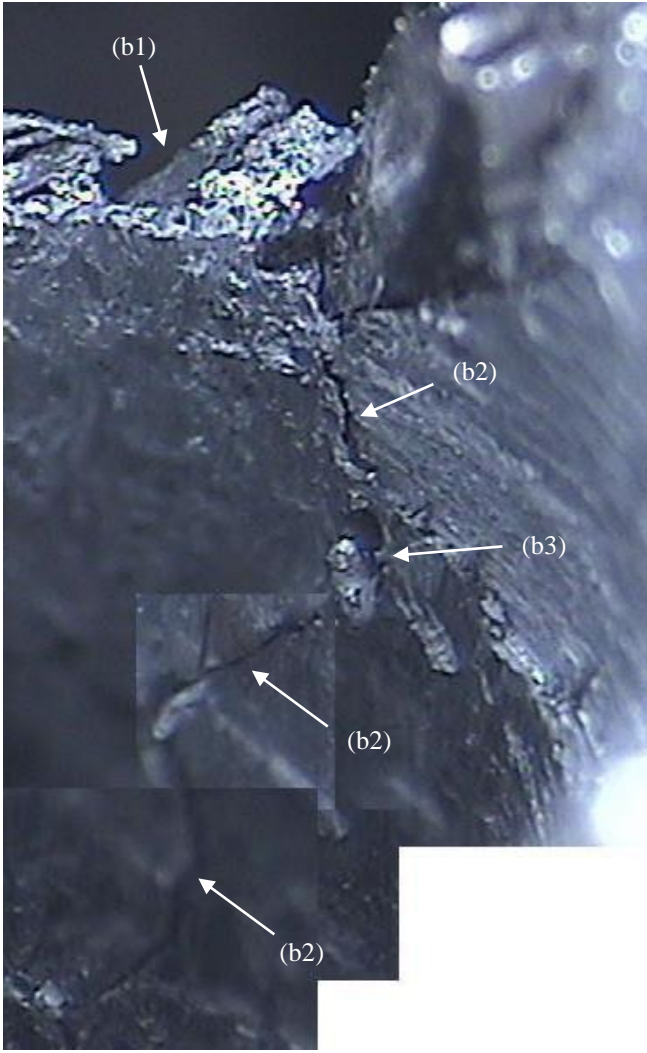


Fig. 10. 4 mm drill after the breakage. A crack and a molten aluminum drop are clearly visible.

TABLE III
ROTOR ASYMMETRY
INDICATORS FOR THE SECOND MOTOR.

Indicator	Phase	Healthy			Faulty		
		γ	$\bar{\gamma}$	std(γ)	γ	$\bar{\gamma}$	std(γ)
$\gamma_{F1(1)}$	A	45.4	32.6	0.4			
	C	41	32.4	0.4			
$\gamma_{F2(2)}$	A	43.5	31.8	0.4			
	C	41.9	31.8	0.4			
$\gamma_{Hilbert, (3)}$	A	49	39.8	1.1			
	C	47.2	39.7	1.1			
$\gamma_{Apr6, (4)}$	A	47.2	35.4	0.1			
	C	47.3	35.8	0.1			
$\gamma_{Det7, (4)}$	A	47.5	36.2	0.2			
	C	47.6	36.4	0.2			
$\gamma_{Apr7, (4)}$	A	59.6	43.3	0.8			
	C	60.6	44.3	1			
$\gamma_{Det8, (4)}$	A	66.9	43.8	0.9			
	C	67	44.6	1			
$\gamma_{W, (5)}$	A	54.5	33.9	0.1			
	C	54.3	34.4	0.1			

B. Influence of other factors on the traditional indicators

The disassembling and reassembling of the motor in order to check the breakage, has a clear effect in the stationary indicators that increase their value (Fig. 5, a5), whilst the transient based ones remain unaffected (Fig. 6, 7). The average variation is 1.7 dB for (1), 1.35 dB for (2) and 0.8 dB for (3). This effect can be a product of a modification of the static eccentricity as stated in [10]. Therefore, in experiments trying to detect a partial breakage using stationary indicators, some slight variations of the indicators may be product of unintentional modifications in the position of the rotor within the stator bore when it was extracted in order to artificially cause the damage, and not an effect of the incipient fault.

C. Trends observed during the bar breakage

As the section of the bar is reduced to an estimated 14% of its original area (test number 80,405) and the compensation is achieved, a slight trend in the rotor asymmetry wavelet indicators is appreciated (Fig. 6). In the latter developed method based in the Wigner-ville distribution, this evolution is wider (Fig. 7 and 11, (a1) and (c1) respectively).

For phase A, in 452 tests, the indicator is reduced by 1.7 dB (Fig. 11). In addition, the slope increases as this series of tests progresses.

Once the remaining section is reduced to the 7% of the original (test number 80,857), by increasing the diameter of a hole in 0.5 mm (Fig. 6, 7 (a2), Fig. 11 (c2)), the evolution is much faster (a3, c3) until the bar abruptly breaks (a4, c4) after 214 additional tests, that is, between cycles 81,070 and 81,071. There is no further evolution in the indicator (c5).

This behavior well fits into the evolution of a crack: a growing period (c1, c3) and once the section has been reduced enough, a sudden breakage (c4) occurs.

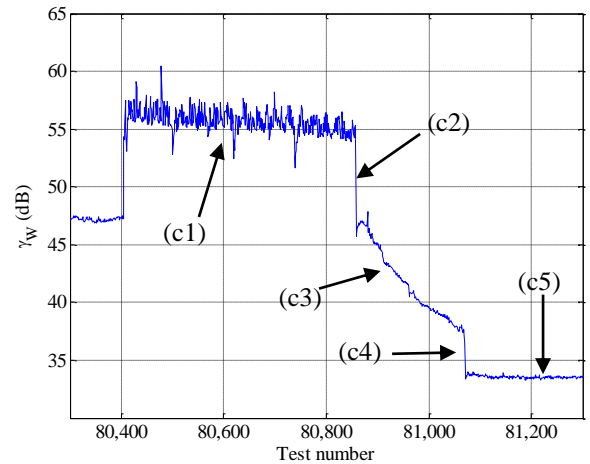


Fig. 11. Wigner-Ville indicator γ_w (5) for phase A assessing the energy of the LSH in the startup during the bar weakening and breakage.

It must be noted that although there are signs of creep at the upper part of the remaining bar, with material lost, and in the hole itself (the aluminum drop), similar as detected on large casted aluminum cage's faults [18], the breakage mechanism according to Fig. 11, corresponds to the behavior of a crack propagation, as observed during the visual analysis of broken

bars in fabricated cages [17].

Qualitatively, from the relatively rapid development of the failure, below ten thousand cycles, it is inferred that the breakage can be modeled as low cycle fatigue [21]. Furthermore, both the behavior of the transient indicators and the effects that appeared on the surface of the aluminum (Fig. 10, b2), point out to the predominance of stress-strain, caused by thermal expansion and contraction, over creep. Therefore and contrary to what has been tried in previous studies [17-19], the equations modeling low cycle (thermal) fatigue and crack propagation could be applied to estimate the remaining life of a bar, once an evolution in the rotor asymmetry indicators based on the study of the transient current has been detected.

D. Conclusions of the fatigue test

The objective of causing a natural bar breakage on a small induction motor featuring a casted aluminum cage proved difficult to achieve through heavy transients without artificially decreasing the section of a bar. No natural evolution could be appreciated in the rotor asymmetry indicators based on the analysis of a stator current until the remaining section was reduced to 14% of the original. Even in this case, the existence of small constructive defects in the secondary circuit of the motor may cause indicators based on the study of low frequency harmonics to initially react opposite to what is expected.

The results of the fatigue test validated the initial hypothesis that transient analysis of stator startup currents can earlier detect a bar fault and, in addition, yielded a method to accurately track the breakage process. This and the inspection of the bar after the failure have allowed characterizing this process as low cycle thermal fatigue, with a significant final stage of crack propagation. The fact that the growing of this crack is clearly depicted by the transient indicators links the failure mechanism to a magnitude that can easily be measured externally to the machine, thus permitting in the next section to propose a model easy to implement in real applications, useful not only in the diagnosis but, especially in the prognosis of the fault.

V. FAULT DETECTION AND PROGNOSIS MODULE

In this section an implementation of a state estimation and prognosis module for a rotor bar is proposed. This algorithm takes advantage of a physical model of the failure obtained from the data yielded by the fatigue test.

A. Crack propagation model

The rapid evolution (c3) observed in Fig. 11 once the section of the bar had been decreased to a 7% of its original value and its sudden collapse (c4) invites the application of fracture theory to model this behavior.

The Paris equation [5] relates the growth rate of a crack having a length a to the stress intensity factor ΔK , which is linked to the range of cyclic stress, by means of two constants of the material, C and m :

$$\frac{da}{dn} = C \cdot \Delta K^m \quad (6)$$

For aluminum ($m=3$, $C=10^{-12}$) the integration of this equation yields:

$$|a_i^{-1/2} - a_0^{-1/2}| = -\frac{1}{6} \cdot C \cdot \Delta K^3 \cdot N_f \quad (7)$$

being N_f the number of load cycles.

Taking into account the evolution of the Wigner-Ville parameter as the development of the crack and fitting the expression to the total number of cycles, it gives $\Delta K = 6.4 \cdot 10^2$.

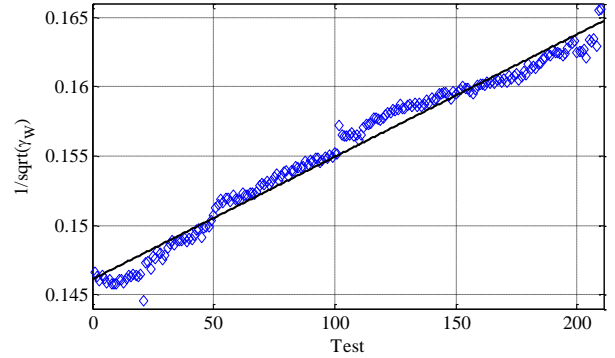


Fig. 12. Fitting of the Paris equation to the evolution of the Wigner-Ville indicator during the final stage of the breakage (c3).

In principle this result links the stress intensity factor to the transient indicators of the bar breakage obtained during a direct startup.

B. Diagnosis module

Following the system architecture depicted in [1, 6], a Particle Filtering approach is implemented as diagnosis module, which permits the combination of both Boolean and continuous variables, the use of non-linear models (7) and an easy linking to prognosis routines, among other advantages.

Given the interest of monitoring a bar breakage and estimating the Remaining Useful Life (RUL) in large machines, this implementation is based on their expected operational conditions, similar to the ones in which the fatigue test was conducted.

A scheme of the implemented module is shown in Fig. 13 (left). The diagnosis routine features the three stages of the Particle Filtering approach [4]. In the first step, *prediction*, the system transition equations are used to forecast the actual state of the system x_t^* at cycle t based on the previous known value x_{t-1} . The *update* stage confronts the prediction to the new measurement $\gamma_{W,t}$ when it is available, using the observation equation via the Bayes rule and associating a weight w to each particle k . Finally, in the *resampling*, step the particles are resampled according to their weight and sent back to the prediction stage. The output of the update stage, $\{x_t(k), w_t(k)\}$, constitutes a discrete approximation of the probability distribution function of the system's state given the

measurement, and therefore, statistical quantities can be derived from it.

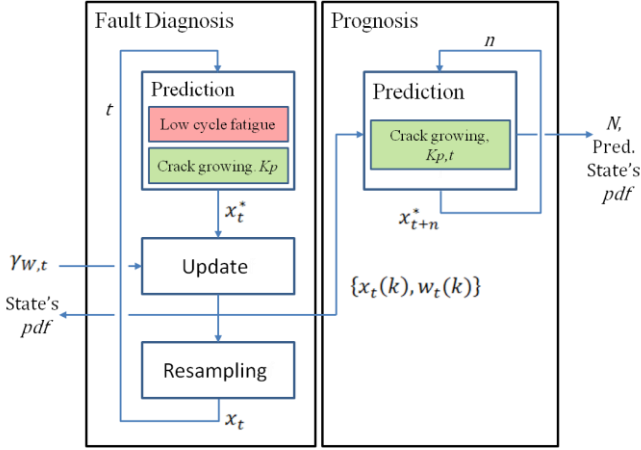


Fig. 13. Diagram showing the proposed Fault Diagnosis and Prognosis module. Given the input of the rotor asymmetry indicator γ_w at cycle t the Particle Filter-based diagnosis module is able to yield a pdf of the state of the bar. In addition, the particles depicting such estimation can be further propagated using the prediction stage in order to obtain an estimation of the future states at cycle $t+n$.

In this work, the state vector x_k is composed by two variables, one discrete signaling the state of the bar and the other continuous evaluating a defect's indicator value (8). For the first one, four states are defined, according to [28]: healthy ("1"), which in this case also includes the compensation stage; hot spots ("2"), breaking ("3") and broken bar ("4"). In the case of the second one, the Wigner-Ville indicator (5) for phase A (Fig. 11) which accurately depicts the evolution of the fault is selected.

$$x_t = \begin{bmatrix} x_{d,t} \\ \gamma_{W,t} \end{bmatrix} \quad (8)$$

Eq. (9-11) compute the estimated state vector x_t^* in the prediction stage. Although in this paper $K_{lc}=1$ has been used, Eq. (10) can take into account the natural wear off of the machine during its designed number of startups for $K_{lc} < 1$, or can also be replaced by a low-cycle fatigue expression such as the one proposed in [23]. However, if a breaking bar-state is detected, (10) is substituted by the Paris' equation (7, 11) in order to estimate the next value of the asymmetry indicator $\gamma_{W,t}^*$:

$$x_{d,t}^* = x_{d,t-1} + n_{d,t-1} \quad (9)$$

$$\gamma_{W,t}^* = K_{lc} \cdot \gamma_{W,t-1} + n_{lc,t-1} \quad (10)$$

$$\gamma_{W,t}^* = (\gamma_{W,t-1}^{-1/2} + K_p)^{-2} + n_{p,t-1} \quad (11)$$

where $K_p = \frac{1}{6} \cdot C \cdot \Delta K^3$, and n_{t-1} , $n_{lc,t-1}$, $n_{p,t-1}$ are zero mean Gaussian noise with standard deviation 0.5, 1 and 0.5.

In the update stage, bibliography thresholds presented in [28] for the FFT indicator used commonly in the industry to assess a bar breakage in large units are used, since the Wigner-

Ville indicator (5) yields similar values (see Table IV). The procedure is complemented by a trend detection code that recognizes long term evolutions in the data that reflect progressing damage. The probability of each particle according to the measurement, which defines its weight, is derived accordingly. The measurement noise for the Wigner-Ville indicators during the final stages of the fatigue test fits a normal distribution, with a standard deviation as presented in Table II.

TABLE IV
THRESHOLD VALUES FOR EACH STATE.

State	Value	Threshold (dB)
Healthy ("Compensation")	1	$\gamma_w \geq 49$ dB
Hot Spots	2	$44 \text{ dB} \geq \gamma_w > 49 \text{ dB}$
Breaking Bar	3	$39 \text{ dB} \geq \gamma_w > 44 \text{ dB}$
Broken Bar	4	$\gamma_w < 39 \text{ dB}$

The particles obtained following this procedure constitute a discrete approximation of the probability density function pdf of the state of the bar. Its mean yields the current state whilst the gap between it and the theoretical one's mean allow obtaining a confidence value.

Fig. 14 shows the results of applying the particle filtering approach for the state estimation of the bar during the breakage. In this case, 35 particles have been used ($k=1,2,\dots,35$). This procedure detects the damage progression at test number 80,885, when the asymmetry indicator (blue line in Fig. 14, acquires a downward trend. A broken bar is declared when the resulting state pdf 's mean acquires a value below 39 dB, according to Table IV, which happens at test number 81,040, that is, 31 cycles before the actual breakage takes place.

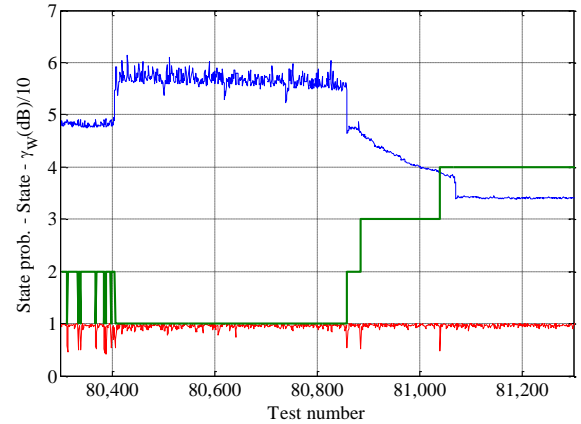


Fig. 14. Results of the state estimation by the particle filtering approach applied to the indicator 5A. Blue line, 5A indicator value divided by 10; green line, state (1, healthy ("compensation")); 2 hot spots, 3 breaking bar, 4 broken bar); red line, state probability.

C. Prognosis module and generalization

The prognosis module (Fig. 13, right side) has been implemented according to [1, 6] as well. Two assumptions are made for its implementation:

- 1) The machine is operated under constant conditions, as it would correspond to a big unit working in a fixed application.
- 2) Since the fracture mechanics have not been studied in

this work, the objective of the prognosis module is not to estimate the remaining operational cycles until the actual breakage occurs, but instead, until the threshold level of 39 dB is reached. Beyond this point the rupture would be considered imminent.

The prognosis module works, once a breaking bar condition is detected by the diagnosis module, propagating the actual *pdf* of system's state at time t (approximated by the particles $\{x_t(k), w_t(k)\}$,) by successively using the Paris' equation (11) in the prediction stage, until the forecasted level for the asymmetry indicator $\gamma_{W,t+n}^*$ for cycle n reaches a mean value of 39 dB. The number of cycles needed N represents its RUL computed at t . Since there is no update stage, the particle weights remain constant. The result is also a *pdf* of the future state, which can as well yield statistical quantities in order to reflect the accuracy of the prediction (Fig. 15).

Contrary to the work developed in this paper, in which K_p , that is, the rate of propagation of the crack used in the Paris' equation (7, 11) has been experimentally determined by a fatigue test, in real applications, it would be unknown and linked to design and construction of the cage and the severity of the startup through the stress intensity factor ΔK . Given that Particle Filtering is a combined model and data driven approach and model deviations are corrected by new measurements, an approximation of K_p can be used in the diagnosis routine presented in the previous point, however, for prognosis the calculation of its value becomes critical in order to obtain an accurate prediction. Therefore, the first task once a breaking bar is detected would consist of determining K_p , provided that the long term evolution should fit (7). For this duty, once new particles are available from the update stage at time t , a regression is being used to compute a new $K_{p,t}$ coefficient. The prognosis at t is carried out using this new value.

Fig. 15 portrays the smoothed estimation of the bar's RUL once the breaking condition has been detected (test number 80,855), for seeded values of K_p ranging from $5 \cdot 10^{-5}$ to $9 \cdot 10^{-5}$, compared to the real cycles to failure represented by the black dotted line. The red (upper) and green (lower) dotted lines are established at three times the smoothed standard deviation of the predicted particle values x_{t+N}^* . As new data becomes available, the accuracy of the prediction increases even for initial values that differ considerably. These results provide an initial generalization of the proposed approach to different severity startups.

Nevertheless, since an induction machine can operate with several broken bars and there exist other kind of defects in the rotor cage, this approach by itself cannot predict the Remaining Useful Life of the machine. This work is focused on a particular failure mode under certain operational conditions. Besides, additional tests with different rotor cage configurations and multiphysics finite element analysis, should be carried out to establish the variation of the stress intensity factor ΔK in those situations, however, the versatility of the Particle Filtering algorithm proposed here may provide a framework in which new results can be easily integrated.

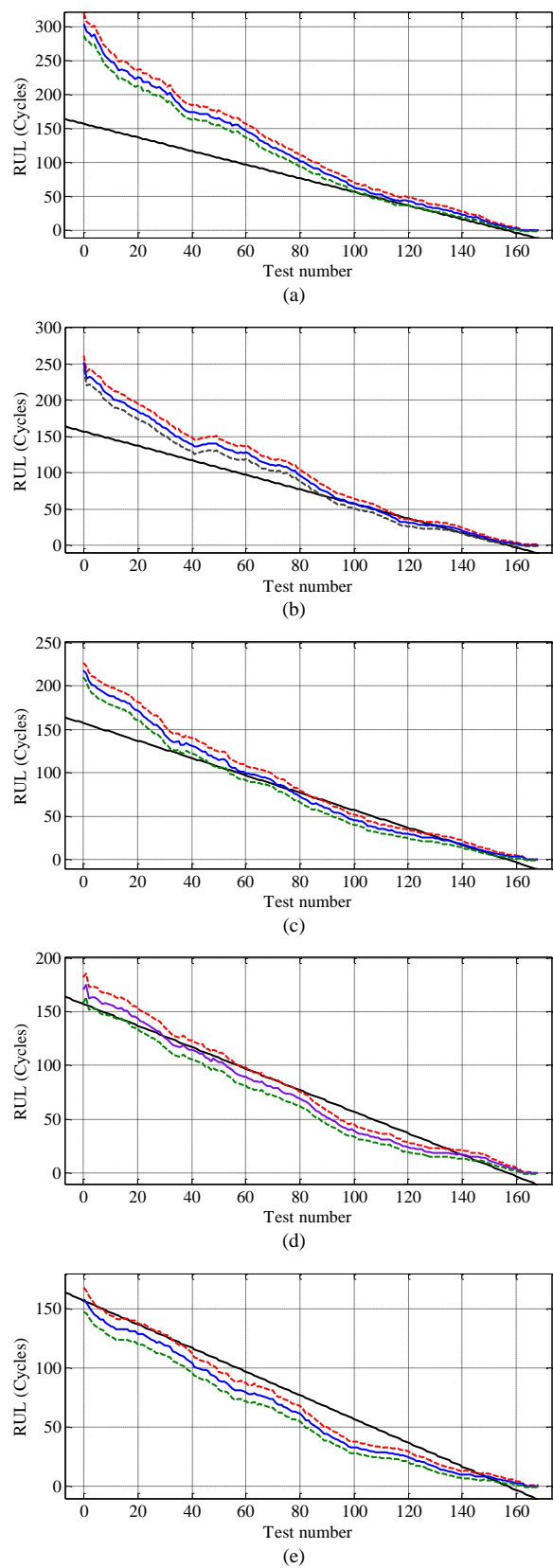


Fig. 15. Results of the RUL estimation N through the particle filtering approach since the breaking bar condition is detected. Black line, cycles to failure; blue line, estimation for seeded value of $K_p = 5 \cdot 10^{-5}$ (a), for $6 \cdot 10^{-5}$ (b), for $7 \cdot 10^{-5}$ (c), for $8 \cdot 10^{-5}$ (d) and for $9 \cdot 10^{-5}$ (e); red dotted line, upper 3σ limit; green dotted, lower 3σ limit.

VI. CONCLUSIONS

A Condition Based Maintenance and Prognosis and Health Management architecture is presented for determining the state and the remaining useful life (RUL) of a rotor bar in an induction machine, based on the results of a fatigue test designed to produce in the most natural way possible a breakage.

The test has shown that the study of a direct startup's stator current allows tracking the evolution of breaking bar accurately, detecting its failure in an incipient stage, and providing a pattern of its development that permits the fitting of physical failure models with the objective of carrying out the prognosis of the fault.

Those models and the entire Particle Filtering diagnosis and prognosis algorithm provided here, offer a framework to further extend the results to different fault modes and operational conditions, such a multiple bar breakages or end ring failure, with the aim of determining the RUL of the entire machine. Multiphysics Finite Element analysis and fatigue testing would be necessary to establish the stresses on the rotor cage under those configurations.

APPENDIX

The characteristics of the motor tested are: star connected, rated voltage (U_n): 400V, rated power (P_n): 1.5 kW, 1 pole pairs, stator rated current (I_{1n}): 3.25 A rated speed (n_n): 2860 rpm.

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