## Worked example: checking a steel beam in simple bending

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## 1 Summary

This document shows how to check a beam with an overhang, taking into account the Ultimate Limit States conditions considered in the Spanish Code DB SE A.

The geometry of the steel frame and the bracing system shown in the design must be taken into account in the analysis and design.

## 2 Introduction

Given the Steel frame shown in figure 1, it is requested to check BCD beam designed with a S 275 hot rolled IPE 240 considering only Ultimate Limit States.
To understand the steel structure performance, a 3D model showing the bracing systems can be seen in figure 2.

$$
\mathrm{q}_{\mathrm{d}}=12 \mathrm{kN} / \mathrm{m}
$$

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figure 1. Front view

figure 2. 3D model showing the bracing Systems

## 3 Aims

At the end of this document, the student will be able to check a simple supported beam with a cantilever designed with an S 275 IPE 240 cross section.
For that purpose, the steps to be followed are:

- Analyse a simple supported beam with a cantilever
- Check the designed beam in bending
- Check the designed beam in shear
- Check if the interaction Bending moments and shear forces has to be considered
- Check the lateral buckling condition
- Check the shear buckling of the web
- Check the local effects of the concentrated loads

The design will be accepted only when all the conditions have been fulfilled.

## 4 Worked example

### 4.1 Geometric properties of the cross section

Geometric properties and resistances of the cross sections are shown in tables land 2 respectively.


Table 1. Geometric properties of the cross section


Table 2. Resistances of the cross section

### 4.2 Obtaining the internal forces

In order to obtain the internal forces in the beam, and taking into account the principle of superposition, the load is divided into horizontal and vertical loads:

1. Considering vertical loads

Vertical reactions are obtained isolating the beam (see figure 3) and taking into account the vertical forces and bending moments equilibrium equations (Equation 3 and 4 respectively):

$\mathrm{q}_{\mathrm{d}}=12 \mathrm{kN} / \mathrm{m}$


Figure 3. Vertical loads
$\sum M_{B}=0 \rightarrow R_{C} \cdot 8-12 \cdot(8+2) \cdot 5=0$
Equation 2. Bending moments equilibrium equation
Calculating:
$R_{C}=\frac{12 \cdot(8+2) \cdot 5}{8}=75 \mathrm{kN} ; \quad R_{B}=120-75=45 \mathrm{kN}$
2. Considering horizontal loads (see figure 4)

Horizontal reactions are obtained taking into account an additional compatibility condition in equation 3 (beams compressive strain has been neglected)
$\delta_{h 1}=\delta_{h 2} \quad$ Equation 3. Compatibility condition

Being $\quad \delta_{h 1}=\frac{W_{d} \cdot h_{1}{ }^{4}}{8 \cdot E \cdot l_{1}}+\frac{(-P) \cdot h_{1}{ }^{3}}{3 \cdot E \cdot l_{1}}$

$$
\delta_{h 2}=\frac{P \cdot h_{2}^{3}}{3 \cdot E \cdot I_{2}}
$$



Figure 4. Horizontal loads

As in this case $h_{l}=h_{2}$ and $l_{l}=l_{2}: \frac{W_{d} \cdot h^{4}}{8 \cdot E \cdot l}=\frac{2 \cdot P \cdot h^{3}}{3 \cdot E \cdot l} \rightarrow P=\frac{3 \cdot W_{d} \cdot h}{2 \cdot 8}$
substituting and calculating: $P=\frac{3 \cdot 3 \cdot 5}{2 \cdot 8}=2,81 \mathrm{kN}$
Taking into account former values for reactions, the internal forces functions and diagrams are obtained in the following epigraph.

### 4.2.1 Axial forces

Axial function and diagram is (figure 5)
From B to C: $N_{(x)}=2,81 \mathrm{kN}$
From C to D $N_{(x)}=0 \mathrm{kN}$


Figure 5. Axial forces diagram

### 4.2.2 Shear forces

Shear forces functions and diagram is (see figure 6)

From B to C: $V_{(x)}=45-12 \cdot x$
Shear force on the left of C: $V_{E d}=51 \mathrm{kN}$
Shear force on the right of $C$ : $V_{E d}=24 \mathrm{kN}$


Figure 6. Shear forces diagram

Being the point where the shear force is zero: $45-12 \cdot x=0 \rightarrow x=3,75 m$ from $B$
From $C$ to $D: V_{(x)}=24-12 \cdot x$

### 4.2.3 Bending moments

The maximum positive bending moment B is obtained considering the equilibrium of bending moments equation at $3,75 \mathrm{~m}$ form B (equation 4)

$$
M_{\max +}-45 \cdot 3,75+12 \cdot \frac{3,75^{2}}{2}=0
$$

equation 4. Maximum bending moment
Calculating: $M_{\text {max+ }}=84,375 \mathrm{kN} \cdot \mathrm{m}$.
The bending moments diagram is sketched in figure 7, being the maximum negative bending moment at $C$ equal to:
$M_{\text {max }-}=-12 \cdot 2 \cdot 1=-24 \mathrm{kN} \cdot \mathrm{m}$


Figure 7. Bending moments diagram

### 4.3 Ultimate Limit States Conditions

### 4.3.1 Resistance

- Bending moment condition: In the absence of shear force, the design value of the bending moment MEd at each cross-section will satisfy equation 5

$$
M_{E d} \leq M_{c, R d}
$$

with $M_{E d}=84.375 .000 \mathrm{~N} \cdot \mathrm{~mm}$ (maximum value at $3,75 \mathrm{~m}$ from B)

And $\quad M_{c, R d} \equiv M_{p l, R d}=\frac{W_{p l} \cdot f_{y}}{\gamma_{M 0}}$ because S 275 IPE 240 in bending is Class 1 As $\quad M_{p l, y, R d}=95.857 .142 N \cdot m m \quad$ (see table 2)

And $M_{E d}=84.375 .000 \mathrm{~N} \cdot \mathrm{~mm}<M_{p l, y, R d}=95.857 .142 \mathrm{~N} \cdot \mathrm{~mm} \rightarrow$ the beam fulfils the bending condition.

- Shear forces condition: The design value of the shear force $V_{E d}$ at each cross-section will satisfy equation 6 .

$$
V_{E d} \leq V_{p l, R d}
$$

equation 6. Shear force condition
with $\quad V_{E d}=51.000 \mathrm{~N} \cdot \mathrm{~mm}$ (maximum value on the left of B )
And $\quad V_{p l, y, R d}=288.812 \mathrm{~N}$ (see table 1)
As $\quad V_{E d}=51.000<V_{p l, y, R d}=288.812 \mathrm{~N}$ (see table 1) $\rightarrow$ complies shear condition

## - Shear and bending moments interaction:

The theoretical plastic bending moment resistance of a cross-section is reduced by the presence of shear. For small values of the shear force this reduction is so small that may be neglected. Provided that the design value for the shear force $V_{\text {Ed }}$ does not exceed the $50 \%$ of the design plastic shear resistance $V_{\text {pl.Rd }}$ no reduction needs to be made in the value of the former plastic bending moment resistance.
The point to analyse if shear and bending interaction must be taken into account is $C$, where $V_{E d}=51.000 \mathrm{~N}$ and $M_{E d}=24 \mathrm{kN} \cdot \mathrm{m}$

Provided that $V_{E d}=51.000<50 \% \cdot V_{p l, y, R d}=144.406 \mathrm{~N}$ the shear and bending interaction is not considered.

## - Axial forces and bending moments interaction:

In I cross-sections, the reduction of the theoretical plastic moment resistance by the presence of small axial forces is balanced by strain hardening and may be neglected. When the axial force is smaller than the axial resistance of the web (equation 7) bending and axial force interaction can be neglected.

$$
\begin{equation*}
N_{E d}<0,5 \cdot A_{v} \cdot \frac{f_{y}}{\gamma_{M O}} \tag{equation 7}
\end{equation*}
$$

Provided that $\quad N_{E d}=2810 \mathrm{~N}<0,5 \cdot 1910 \cdot \frac{275}{1,05}=250.119 \mathrm{~N}$ the axial and bending interaction is not considered.

### 4.3.2 Lateral buckling condition

Lateral-torsional buckling condition is shown in equation 8.

$$
M_{E d} \leq M_{b, R d} \quad \text { equation 8. Lateral buckling condition }
$$

with $\quad M_{E d}=84.375 .000 \mathrm{~N} \cdot \mathrm{~mm}$ (maximum value at 3.75 m from B)
and $\quad M_{b, R d}=\frac{\chi_{L T} \cdot W_{y} \cdot f_{y}}{\gamma_{M 1}} \quad$ equation 9. Lateral-torsional buckling resistance for beams as the cross-section is Class 1 , equation 9 es equivalent to $M_{b, R d}=\chi_{L T} \cdot M_{p l, y, R d}$
being $\chi_{L T}$ the reduction factor for lateral-torsional buckling, which value can be obtained from the buckling curves in figure 8, considering the non-dimensional buckling slenderness $\bar{\lambda}_{L T}$, calculated in equation 9 , while $M_{c r}$ value can be obtained following equation 10 , once $M_{L T, v}$ and $M_{L T, w}$ (equations 12 and 12 respectively) are known. For that purpose, values of $b_{L T, v}, b_{L T, w}$ coefficients are shown in figure 8

$$
\begin{aligned}
& \bar{\lambda}_{L T}=\sqrt{\frac{W_{y} \cdot f_{y}}{M_{c r}}} \\
& M_{c r}=\sqrt{M_{L T, v}^{2}+M_{L T, w}^{2}} \\
& M_{L T, v}=b_{L T, v} \cdot \frac{C_{1}}{L_{c}} \\
& M_{L T, w}=b_{L T, w} \cdot \frac{C_{1}}{L_{c}^{2}}
\end{aligned}
$$

equation 10. non-dimensional buckling slenderness
equation 11. Elastic critical bending moment
equation 12 . Uniform torsional rigidity component
equation 13. Non-uniform torsional rigidity component

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Non-dimensional aleeral bucking slendemess $\overline{\bar{L}}_{L T}$

$\bar{\lambda}_{L T}=\sqrt{\frac{W_{y} \cdot f_{y}}{M_{c r}}}$ and $M_{c r}=\sqrt{M_{L T, v}^{2}+M_{L T, w}^{2}}$
$b_{L T, V}=\pi \cdot \sqrt{G \cdot I_{t} \cdot E \cdot I_{z}}$ to obtain $M_{L T, v}=b_{L T, v} \cdot \frac{C_{1}}{L_{c}}$
$b_{L T, w}=W_{e l, y} \cdot \pi^{2} \cdot E \cdot i_{f, z}{ }^{2}$ to obtain $M_{L T, w}=b_{L T, w} \cdot \frac{C_{1}}{L_{c}{ }^{2}}$
$i_{f, Z}$ radius of gyration of the compressed flange $+1 / 3$ web, along the $z$ axis

| Cross- <br> section | $i_{f, z}$ <br> $m m$ | $b_{L T, V}$ <br> $\times 10^{6} \mathrm{~N}$ <br> $m m^{2}$ | $b_{L T, w}$ <br> $\times 10^{9} \mathrm{~N} \cdot \mathrm{~mm}^{2}$ | $W_{\text {el, }} \cdot f_{y}$ <br> $\mathrm{~N} \cdot \mathrm{~mm}$ | $W_{\text {Dl, }} \cdot f_{Y}$ <br> $\mathrm{~N} \cdot \mathrm{~mm}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| IPE 240 | 29,4 | 247810 | 580441 | 89100000 | 100650000 |


| $\lambda$ | Buckling curve |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\mathrm{a}_{0}$ | a | b | c | d |
| $\leq 0,2$ | 1,00 | 1,00 | 1,00 | 1,00 | 1,00 |
| 0,3 | 0,99 | 0,98 | 0,96 | 0,95 | 0,92 |
| 0,4 | 0,97 | 0,95 | 0,93 | 0,90 | 0,85 |
| 0,5 | 0,95 | 0,92 | 0,88 | 0,84 | 0,78 |
| 0,6 | 0,93 | 0,89 | 0,84 | 0,79 | 0,71 |
| 0,7 | 0,90 | 0,85 | 0,78 | 0,72 | 0,64 |
| 0,8 | 0,85 | 0,80 | 0,72 | 0,66 | 0,58 |
| 0,9 | 0,80 | 0,73 | 0,66 | 0,60 | 0,52 |
| 1,0 | 0,73 | 0,67 | 0,60 | 0,54 | 0,47 |
| 1,1 | 0,65 | 0,60 | 0,54 | 0,48 | 0,42 |
| 1,2 | 0,57 | 0,53 | 0,48 | 0,43 | 0,38 |
| 1,3 | 0,51 | 0,47 | 0,43 | 0,39 | 0,34 |
| 1,4 | 0,45 | 0,42 | 0,38 | 0,35 | 0,31 |
| 1,5 | 0,40 | 0,37 | 0,34 | 0,31 | 0,28 |
| 1,6 | 0,35 | 0,32 | 0,31 | 0,28 | 0,25 |
| 1,7 | 0,31 | 0,30 | 0,28 | 0,26 | 0,23 |
| 1,8 | 0,28 | 0,27 | 0,25 | 0,23 | 0,21 |
| 1,9 | 0,25 | 0,24 | 0,23 | 0,21 | 0,19 |
| 2,011 | 0,23 | 0,22 | 0,21 | 0,20 | 0,18 |
| 2,111) | 0,21 | 0,20 | 0,19 | 0,18 | 0,16 |
| 2,211 | 0,19 | 0,19 | 0,18 | 0,17 | 0,15 |
| 2,311) | 0,18 | 0,17 | 0,16 | 0,15 | 0,14 |
| 2,4(1) | 0,16 | 0,16 | 0,15 | 0,14 | 0,13 |
| 2,52] | 0,15 | 0,15 | 0,14 | 0,13 | 0,12 |
| 2,62) | 0,14 | 0,14 | 0,13 | 0,12 | 0,11 |
| 2,72) | 0,13 | 0,13 | 0,12 | 0,12 | 0,11 |
| 2,8(2) | 0,12 | 0,12 | 0,11 | 0,11 | 0,10 |
| 2,921 | 0,11 | 0,11 | 0,11 | 0,10 | 0,09 |
| 3,012) | 0,11 | 0,10 | 0,10 | 0,10 | 0,09 |
| (10) |  |  |  |  |  |

Figure 8. Lateral-torsional buckling coefficients and buckling curves
The coefficient $C_{1}$ depends on the load case and the effective length factor $k$. It takes into account the bending moment distribution along the beam.
General cases are shown in table 3

| Loads and support conditions | Bending moment diagram | k | $\mathrm{C}_{1}$ |
| :---: | :---: | :---: | :---: |
|  |  | 1, 0 0 5 | 1, 13 0, 97 |
|  |  | 1, 0 0 5 | 1, 28 0 71 |

Table 4. Transverse loading cases

Considering the bending moment diagram of the beam and the position of the points where the out of plane movement is restrained is shown in figure 9, it can be assumed that $L_{c}=1000 \mathrm{~mm}, \mathrm{k}=1$, and therefore $\mathrm{C}_{1}=1,13$ and


Figure 9. Bending moments and bracing system

Operating $\quad M_{L T, v}=247.810 \cdot 10^{6} \cdot \frac{1,13}{1.000^{2}}=280.025 .300 \mathrm{~N} \cdot \mathrm{~mm}$

$$
M_{L T, w}=580.441 \cdot 10^{9} \cdot \frac{1,13}{1.000^{2}}=655.898 .330 \mathrm{~N} \cdot \mathrm{~mm}
$$

$$
M_{c r}=\sqrt{280.025 .300^{2}+655.898 .330^{2}}=713.173 .743 \mathrm{~N} \cdot \mathrm{~mm}
$$

And $\quad \bar{\lambda}_{L T}=\sqrt{\frac{100.650 .000}{713.173 .743}}=0,3756<0,40$ (lateral buckling cannot occur)

### 4.3.3 Shear buckling of the web

Considering that the web is unstiffened, shear buckling of the web cannot occur if:

$$
\frac{d}{t_{w}} \leq 70 \cdot \varepsilon
$$

Being in this case $\frac{240-2 \cdot 9,8-2 \cdot 15}{6,2}=34,7 \leq 70 \cdot \sqrt{\frac{235}{275}}=64,7$
therefore, the shear buckling of the web cannot occur.

### 4.3.4 Local effects, concentrated loads

Considering that the reactions are those shown in figure 10, The condition will be that every punctual load:
$R<R_{b, R d}$
equation 15. Punctual loads condition
$q_{\mathrm{d}}=12 \mathrm{kN} / \mathrm{m}$
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Figure 10. Reactions
where $R_{b, R d}$ is the design buckling resistance of the web of the beam considered as a short column:
$R_{b, R d}=N_{b, R d,}$ with $N_{b, R d}=\frac{\chi_{\min } \cdot A \cdot f_{y}}{\gamma_{M 1}}$
Being $N_{b, R d}$ the buckling resistance of the portion of the web bearing the punctual load.

Where $A=\left(20 \cdot t_{w} \cdot \varepsilon\right) \cdot t_{w}=20 \cdot 6,2 \cdot \sqrt{\frac{235}{275}} \cdot 6,2=710 \mathrm{~mm}^{2}$

$$
\begin{aligned}
& I=\frac{\left(20 \cdot t_{w} \cdot \varepsilon\right) \cdot t_{w}^{3}}{12}=\frac{20 \cdot 6,2 \cdot \sqrt{\frac{235}{275}} \cdot 6,2^{3}}{12}=2.265 \mathrm{~mm}^{4} \\
& i_{\min }=\sqrt{\frac{l}{A}}=\sqrt{\frac{2.265}{710}}=1,78 \mathrm{~mm}
\end{aligned}
$$

$$
\lambda=\frac{0,8 \cdot d}{i_{\min }}=\frac{0,8 \cdot(240-2 \cdot 9,8-2 \cdot 15)}{1,78}=85,57
$$

being $\bar{\lambda}=\frac{\lambda}{\lambda_{R}}$ and considering that the steel is $S 275, \lambda_{R}=86,8$

$$
\bar{\lambda}=\frac{85,57}{86,8}=0,98 \simeq 1 \xrightarrow{\text { curve } c} \chi_{\text {min }}=0,54
$$

Then $\quad N_{b, R d}=\frac{0,54 \cdot 710 \cdot 275}{1,05}=100.414 \mathrm{~N}$

As $\quad N_{E d}=R_{c}=75.000 \mathrm{~N}<N_{b, R d}=100.414 \mathrm{~N}$ it is not necessary to stiffen the beam.

### 4.4 Final design of the beam

As the S 275 hot rolled IPE 240 cross section fulfils all the Ultimate Limit States conditions, it can be designed with that cross section.

## 5 Conclusion

This document explains, with a worked example, how to check Ultimate Limit States conditions of a simple beam with a cantilever loaded with a uniform load

With this purpose, the student has to follow the following steps:

1. Obtain the internal forces in the analysed member
2. Check the bending moments resistance condition
3. Check the shear forces resistance condition
4. Check the bending moments-shear forces interaction
5. Check the axial forces-bending moments interaction
6. Check the lateral-torsional condition
7. Check the shear buckling of the web
8. Check the local effects of the concentrated loads.
9. Finally, once al the conditions have been fulfilled, the initial design has been accepted.

Finally, it is important to point out that the initial design of the cross section is not always going to fulfil all the conditions. In those cases, the cross section must be redesigned and checked until all the conditions are fulfilled.
$q_{\mathrm{d}}=12 \mathrm{kN} / \mathrm{m}$

## 6 Proposed exercise

Considering the frame in the figure, where the overhanging has been eliminated from the initial design, it is requested to check if the same cross section, with the same bracing conditions, will fulfil the Ultimate Limit State conditions.

## 



## 7 Bibliography

### 7.1 Books:

Guardiola-Víllora, A; Pérez-García, A: "Steel structures. Worked examples according to the Spanish Code", Ed. Universitat Politècnica de València, 2017, chapter 4

### 7.2 Codes:

Steel structures Spanish code "Documento Básico, Seguridad Estructural, Acero". Ministerio de Fomento 2006. It can be downloaded from:
https://www.codigotecnico.org/

## 8 Solution

Being the maximum bending moment, at the middle of the span equal to:
$M_{E d}=\frac{q \cdot L^{2}}{8}=96 \mathrm{kNm}$, and $M_{p l, R d}$ equal to 95857142 Nmm (table 2) the cross section fails the first condition, so, BC beam cannot be designed with an S 275 IPE 240

