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4 **ANALYSIS OF VORTEX CORE GENERATION IN PIPE FLOWS UNDER DIFFERENT REYNOLDS**  
5 **NUMBER CONDITIONS**

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15 **ABSTRACT**

16 Pipe flow is a well-documented case widely studied in both theoretical and practical applications. The present work aims  
17 at studying the influence of the Reynolds number on turbulent vortex distribution using Large Eddy Simulations (LES).  
18 Features such as the mean velocity profiles and root-mean squared velocity are first numerically investigated for different  
19 fluid properties involving Reynolds numbers ranging from 5,925 to 15,190 in order to verify the law-of-the-wall and  
20 turbulence statistics with experimental and DNS data. Once the simulations are validated, the vortex core generation  
21 within the flow is studied through a detection algorithm based on the  $\lambda_2$  criterion with two different approaches, first  
22 using an absolute threshold value and then using a relative threshold value depending on the turbulent intensity. Results  
23 are compared in terms of number of structures and Probability Density Functions for both the size and the radial  
24 distributions. Finally, results are compared for one condition with the Q-criterion to assess the results obtained resulting  
25 in practically identical volume and radial distributions. These results are deemed to shed light on the vortex formation  
26 and location to generate proper inflow boundary conditions to highly resolved simulations in varied engineering  
27 applications.

28

29 **KEYWORDS**

30 Computational, LES, pipe flow, turbulence, vortex detection

31

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42 **Code availability:** The code used is OpenFOAM v3.0.0.

43

## 44 **LIST OF NOTATIONS**

45  $u_\tau$  friction velocity

46  $R$  pipe radius

47  $Re_\tau$  Kármán number

48  $Re_D$  Reynolds number

49  $L$  pipe length

50  $D$  pipe diameter

51  $\bar{u}_i$  resolved velocity vector

52  $\bar{p}$  modified kinetic pressure

53  $C_w$  WALE constant

54  $\bar{S}_{ij}$  resolved strain rate tensor

55  $S$  symmetric strain rate tensor

56  $U_b$  bulk velocity

57  $y^+$  non-dimensional distance to the wall

58  $r_{wall}^+$  cell size in radial direction on the wall in wall units

59	$x^+$	cell size in axial direction in wall units
60	$r$	radial position
61	$u_x$	axial velocity
62	$u_{x,cl}$	axial velocity on the centerline
63	$u_x^+$	axial velocity in wall units
64	$u_{x,rms}^+$	axial root mean squared velocity in wall units
65	$f$	friction factor
66	$N_s$	number of structures
67	<i>GREEK SYMBOLS</i>	
68	$\tau_w$	wall shear stress
69	$\rho$	density
70	$\nu$	kinematic viscosity
71	$\nu_t$	turbulent viscosity
72	$\Delta$	width of the LES filter
73	$\Omega$	antisymmetric rate-of-rotation tensor
74	$\mu$	dynamic viscosity
75	$\omega_{wall}^+$	cell size in azimuthal direction on the wall in wall units
76	$\xi$	non-dimensional radial position
77	$\xi^+$	non-dimensional radial position in wall units
78		
79	<i>ABBREVIATIONS</i>	
80	DNS	Direct Numerical Simulations
81	ECN	Engine Combustion Network
82	SGS	SubGrid Scale
83	WALE	Wall Adapting Local Eddy-viscosity
84	LES	Large Eddy Simulations
85	CFL	Courant-Friedrich-Lewis
86	PDF	Probability Density Function
87		

## 88 1. INTRODUCTION

89 Most turbulent flows in engineering applications are wall-bounded, at least partially. Over the years, the interaction  
90 between viscous turbulent flows and solids has been studied in several problems. Regarding wall-bounded flows, there  
91 are three canonical flows that represent the purest interaction between these two worlds: the spatially evolving boundary  
92 layer, the channel flow, and the pipe flow.

93 The importance of the presence of a solid within the flow is that the behaviour of the mean velocity profile is affected:  
94 near the wall, viscous effects are important and the scaling factor depends on the friction velocity  $u_\tau = \sqrt{\tau_w/\rho}$  and the  
95 wall length scale  $\nu/u_\tau$ , where  $\tau_w$  is the wall shear stress whereas  $\rho$  and  $\nu$  are the fluid density and kinematic viscosity,  
96 respectively. In the outer region, the appropriate length scale is the pipe radius ( $R$ ), whereas the velocity scale remains  
97 being  $u_\tau$  since it is the inner boundary condition for the outer flow [1, 2]. Therefore, a specific Reynolds number based  
98 on those parameters and known as the Kármán number may be defined and particularized for pipe flows as  $Re_\tau = u_\tau R/\nu$ .  
99 Traditionally, the study of the turbulence in these flows has been carried out by means of experimental works, as done by  
100 Eggels [3] and den Toonder [4] back in mid 90s. More recently, there have been experimental works about high Reynolds  
101 pipe flows in the so-called “Superpipe” located at Princeton [5][6] or the CICLoPE project [7] where Örlü et al. [8]  
102 performed How-wire measurements and conclude that their results supported the attached-eddy hypothesis for the scaling  
103 of the Reynolds stress tensor. But the increasing capabilities in terms of computational processing power are also driving  
104 advances in the theoretical approach of the inner flow turbulence study from a numerical standpoint. First Direct  
105 Numerical Simulations were performed by Eggels et al. [3] who compared DNS results against their own experimental  
106 data for a bulk based Reynolds number ( $Re_D$ ) of 5,300. Even though their results matched their experimental data, they  
107 also compared them with the work of Kim et al. [9], who worked with the same conditions on turbulent channel flow.  
108 The following studies of this kind of flows analyzed two different parameters that control the turbulence spectra. On the  
109 one hand, the pipe length ( $L$ ) to pipe radius ( $R$ ) ratio; on the other hand, the Reynolds number. Several studies [10–12]  
110 investigated the influence of the domain on capturing all the turbulent structures. These studies used pipe lengths ranging  
111 from  $5R$  to  $30R$ . Kim et al. [10] found that a length of  $7.5 L/D$  should be enough to capture all the turbulent processes in  
112 pipe flows. Additionally, there are several studies about the influence of the Reynolds number on the turbulent statistics.  
113 For instance, El Khoury et al. [13] performed DNS simulations from low to moderate Reynolds numbers and compared  
114 them with other results of DNS on pipe flows [14, 15] and with the other two canonical wall-bounded flows [16, 17].  
115 Even though there exist many works on this canonical flow, there are still many insights that remain unclarified. For  
116 instance, the characteristic peak of the root-mean-squared of the streamwise velocity fluctuations seems to be nearly

117 constant with the Reynolds number, but there is no conclusive evidence to support it, as reviewed by some authors [2,  
118 18].

119 These uncertainties bring up new different approaches to the study of turbulence in this kind of flows. Hellström et al.  
120 [19] applied Proper Orthogonal Decomposition (POD) to experimental data in order to study the self-similarity behaviour  
121 of the radial POD resulting in a single length scale representing the complete structure. A similar approach was conducted  
122 by Abreu et al. [20], who applied Spectral POD and resolvent analysis to DNS results from El Khoury et al. [13] to study  
123 the characteristic elongated structures corresponding to near-wall streamwise vortices and streaks. Also, some studies  
124 have been performed on structure detection, as the one performed by Hwang et al. [21] where they demonstrate the  
125 logarithmic region by statistically studying the coherent structures attached to the wall.

126 LES and DNS have proven their key role on the study of turbulent flows for fundamental and engineering applications.  
127 Their versatility allows studying theoretical situations that simplify the problems and help isolating the effects that in  
128 reality use to take place combined with other processes (e.g. using flat velocity profiles to purely study solid or liquid  
129 interactions). However, in most real applications the different processes that take place in one problem feedback with  
130 each other and lead to a very different behaviour that cannot be predicted on the isolated study.

131 Over the last years, different techniques have been used to feed the LES and DNS simulations with coherent velocity  
132 fields capable to correctly trigger the turbulence within the domain as reviewed by Dhamankar et al. [22]. From their  
133 work, two approaches can be highlighted: the use of synthetic boundary conditions through digital filters, as proposed by  
134 Klein et al. [23]; or mapping a turbulent database from a prior computation or experiment. Both approaches have been  
135 applied to wide range of problems. For instance, synthetic boundary conditions were used on [24, 25] to study primary  
136 breakup and particle laden respectively, and mapped boundary conditions were used in [26, 27]. In particular, Payri et  
137 al. [26] studied the atomization process of the ECN Spray A [28] through DNS using both methods. When comparing  
138 them, they noticed that the injected turbulent structures showed a very different shape, importantly influencing droplet  
139 generation. While the synthetic vortex had annular shape, the turbulence mapped from a previous pipe flow LES showed  
140 axially elongated structures.

141 In this framework, considering that using fair synthetic turbulent statistics does not ensure the proper shape of the  
142 generated structures, the main objective of this paper is to investigate the influence of the Reynolds conditions on the  
143 vortex structures formation through widely known tools. Also, since DNS are currently limited to low to moderate  
144 Reynolds numbers, a parametric study has been carried out for the 5,925 to 15,190 range. These results can be used to

145 compare the structures generated by a pipe flow synthetic boundary condition when using them or when generating new  
 146 synthetic turbulence models for these cases.

147 The paper is structured as follows. **Section 2** is a description of the numerical method and sub-grid model used to carry  
 148 out the simulations. In **Section 3**, the computational and physical parameters are presented for all the simulations,  
 149 including the spatial domain and the computational grid parameters. **Section 4** discusses the results through two main  
 150 approaches. First, a validation of the results by comparing them with the law of the wall theory and the friction factor is  
 151 done. Once the data are validated, a deeper study on the turbulence side intends to shed light on the vortex size and radial  
 152 distribution. Finally, **Section 5** wraps up the conclusions and points out the directions for future studies.

153

## 154 2. NUMERICAL METHODS

155 As stated in Section 3, the study focuses on isothermal and incompressible flow conditions. Considering a Newtonian  
 156 fluid, the governing equations of the problem are then the continuity (1) and the momentum (2) equations:

$$\frac{\partial}{\partial x_i}(\bar{u}_i) = 0 \quad (1)$$

$$\frac{\partial \bar{u}_i}{\partial t} + \frac{\partial}{\partial x_j}(\bar{u}_i \bar{u}_j) = -\frac{1}{\rho} \frac{\partial \bar{p}}{\partial x_j} + \frac{\partial}{\partial x_j} \left( (\nu + \nu_t) \left( \frac{\partial \bar{u}_i}{\partial x_j} + \frac{\partial \bar{u}_j}{\partial x_i} \right) \right) \quad (2)$$

157 Where  $\bar{u}_i$  is the resolved velocity field and  $\bar{p}$  is the modified kinetic pressure. The closure of Eq. (2) is given by the  
 158 Subgrid Scale (SGS) viscosity ( $\nu_t$ ), which is a non-linear term that needs to be modelled. There are many subgrid models  
 159 to estimate the energy dissipation in the subgrid range. Given the conditions of the problem and the objective of the study,  
 160 the Wall Adapting Local Eddy-viscosity (WALE) model [29] is chosen. This model is based on the square of the velocity  
 161 gradient tensor to estimate the local eddy viscosity through Eq. (3) and is deemed to properly handle the transition from  
 162 laminar to turbulent flow:

$$\nu_t = (C_w \Delta)^2 \frac{(\bar{S}_{ij}^d \bar{S}_{ij}^d)^{3/2}}{(\bar{S}_{ij} \bar{S}_{ij})^{5/2} + (\bar{S}_{ij}^d \bar{S}_{ij}^d)^{5/4}} \quad (3)$$

163 where  $C_w$  is the WALE model constant (which allows calibrating the dissipation),  $\Delta$  is the width of the LES filter and the  
 164 last term depends on the traceless symmetric part of the squared gradient tensor  $\bar{S}_{ij}^d$  and the resolved strain rate tensor  $\bar{S}_{ij}$ .  
 165 About this last term, it is important to point out that it behaves as a cubic function of the wall-distance ( $y^3$ ) and it is  
 166 function of both the rotation and the strain rates. Hence, this model is capable of reproducing the near-wall scaling without  
 167 any dynamic procedure and has proved its suitability to reproduce the turbulent flow behaviour in pipes [29]. The constant  
 168  $C_w$  is set to 0.5 as proposed by Nicoud and Ducros [29] for this flow topology.

169 All the simulations present in the document are performed using the standard PISO (Pressure Implicit with Splitting  
170 Operator) solver proposed by Issa [30] from the open source C++ library OpenFOAM 3.0.0 [31]. Second-order centred  
171 accurate discretization schemes are employed to compute gradients and Laplacian terms, whereas a second-order implicit  
172 scheme is used for time-stepping. A constant time step is used, set to  $4 \cdot 10^{-9}$  s to ensure a CFL-number lower than 0.4 in  
173 the whole domain during the simulation time.

174 Concerning the study of the vortex structures generated within the fluid submitted to pipe flow, there exist several methods  
175 to detect local vortex, most of them being based on the analysis of the velocity gradient tensor  $\nabla \mathbf{u}$ . Among these methods,  
176 the  $\lambda_2$  criterion proposed by Jeong and Hussain [32] has been chosen to discriminate the coherent structures for all  
177 conditions. Additionally, the Q-criterion has been applied to one of the test conditions in order to compare both criteria.  
178  $\lambda_2$  method starts by decomposing  $\nabla \mathbf{u}$  into a symmetric rate-of-strain tensor ( $S$ ) and an antisymmetric rate-of-rotation  
179 tensor ( $\Omega$ ). Neglecting the viscous effects and the unsteady irrotational straining, the symmetric part of the gradient of the  
180 incompressible Navier-Stokes equation can be expressed as established by Eq. (4).

$$S^2 + \Omega^2 = -\frac{1}{\rho} \nabla^2 p \quad (4)$$

181 Therefore,  $S^2 + \Omega^2$  is a real and symmetric matrix and implies the existence of a local minimum of pressure. Due to its  
182 characteristics, the matrix has 3 eigenvalues ( $\lambda_1 \geq \lambda_2 \geq \lambda_3$ ). It is established that a vortex core is a connected zone with  
183 two negative eigenvalues [32]. Since the eigenvalues are sorted in the decreasing order, this is equivalent to saying that a  
184 connected zone with  $\lambda_2 < 0$  can be regarded to as a vortex structure.

185 On the other hand, the Q-criterion [33] defines an eddy as a region with a positive second invariant ( $Q$ ) of  $\nabla \mathbf{u}$ . This term  
186 can be expressed as given by Eq. (5):

$$Q \equiv \frac{1}{2} (\|\Omega\|^2 - \|S\|^2) \quad (5)$$

187 which means that Q is a balance between shear strain rate and vorticity magnitude, becoming 0 at the wall.

188

### 189 3. SIMULATION PARAMETERS

#### 190 3.1 Domain

191 In order to study a fully developed turbulent pipe flow, calculations are performed over a straight pipe under isothermal  
192 and incompressible flow conditions. The computational domain consists of a pipe of constant circular cross-section with  
193 radius  $R$  and length  $16R$ . Its coordinates are defined as  $x$  in the streamwise direction and  $y$  and  $z$  in the lateral directions,  
194 as shown in Figure 1a).



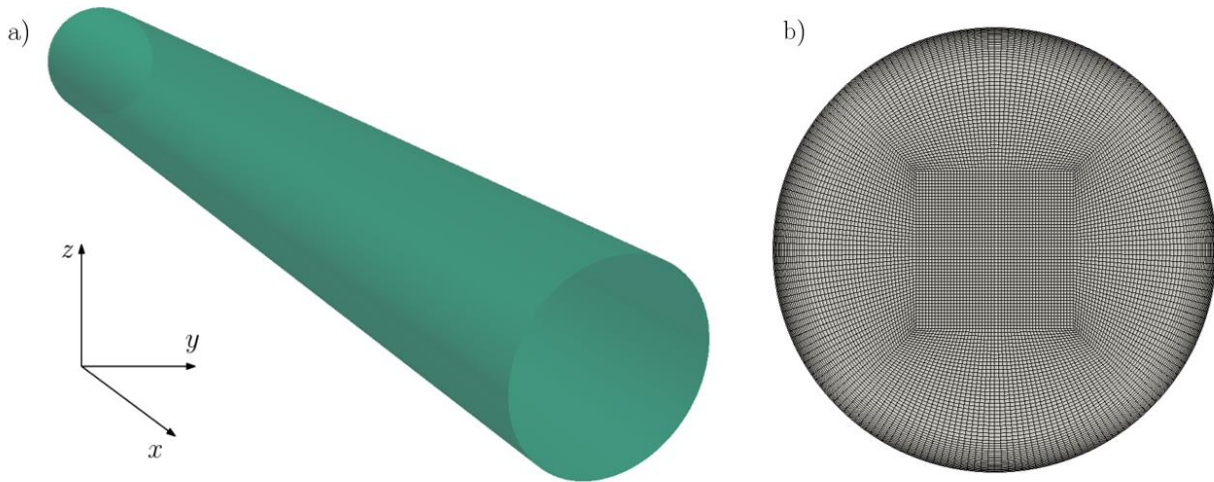


Figure 1: Computational domain and mesh scheme.

195

### 196 3.2. Boundary conditions

197 With the objective of reaching a fully developed turbulent flow within the computational domain, a cyclic boundary  
 198 condition is used both at the inlet and the outlet. This way, the values at the outflow section are used as input conditions  
 199 at the inflow section. Once the simulation completes around 1,000 washouts (a wash-out time being the interval that a  
 200 particle remains in the computational domain before passing through the outflow section), the mean velocity profile is  
 201 checked to match the typical developed turbulent pipe flow profile. Finally, a wall boundary condition is used in the  
 202 cylindrical surface.

### 203 3.3. Initial conditions

204 In order to generate turbulence within the flow, a channel whose length matches the pipe length and whose side and height  
 205 are equal to the pipe diameter is used. Turbulence is triggered in this preliminary domain using the boxTurb tool from  
 206 OpenFOAM [31]. Once the turbulence is achieved in the channel, the result is mapped into the pipe domain. It is worth  
 207 mentioning that, in order to achieve the final coherent turbulent flow in the pipe, it is necessary to simulate for around  
 208 200 washouts until the turbulent flow is fully adapted to the pipe domain. Once the turbulence is adapted to the new  
 209 domain, the initial condition has been reached and the simulation can be started.

Name	$\rho$ [ $kg/m^3$ ]	$\mu$ [ $kg/m \cdot s$ ]	$\nu$ [ $m^2/s$ ]	$Re_D$
Ethanol	790	$1.2 \cdot 10^{-3}$	$1.52 \cdot 10^{-6}$	5,925
Pseudo-fluid	690	$0.69 \cdot 10^{-3}$	$1 \cdot 10^{-6}$	9,000
Isooctane	690	$0.5 \cdot 10^{-3}$	$7.25 \cdot 10^{-7}$	12,420
Heptane	686	$0.41 \cdot 10^{-3}$	$5.92 \cdot 10^{-7}$	15,190

Table 1. Fluid properties for each case of study.

210

211 **3.4. Cases of study**

212 Once the computational domain and boundary conditions are defined, it is important to set the physical conditions that  
213 concern the turbulence behaviour. The most widely non-dimensional number used to define the flow features is the  
214 Reynolds number. In order to vary its values from 5,925 to 15,190, the fluid properties have been modified keeping a  
215 constant bulk velocity  $U_b = 100$  m/s. The fluid properties are listed in Table 1 (please note that a pseudo-fluid with  
216 properties deemed to improve and narrow down the comparison with the literature has been included).

217 **3.5. Computational mesh**

218 Figure 1b) also depicts the o-grid meshing strategy used to set up the numerical grid. When it comes to wall-bounded  
219 flows, a higher resolution is required in the normal direction of the boundary layer than in other parts on the domain. Even  
220 though the same strategy has been used for all the simulations, the cell sizes are different in each case: they have been set  
221 according to the first  $y^+$ , which in turn directly depends on the Reynolds number. Table 2 summarizes the parameters of  
222 the grid, being  $\Delta r_{wall}^+$ ,  $\Delta \omega_{wall}^+$  and  $\Delta x^+$  the sizes (in wall units) of the wall cells in the normal, azimuthal and streamwise  
223 direction respectively. It is important to note that the non-dimensional magnitudes are based on the work by Nicoud and  
224 Ducros [29], which were used as a first mesh set up and were then calibrated with different cell grow factors and  
225 refinement in order to get the final mesh configuration. The mesh study was assessed for  $Re_D = 5,925$  through the law-  
226 of-the-wall, as shown in Figure 2.

227

$Re_D$	5,925	9,000	12,420	15,190
$N_{cells}$	1,650,000	2,250,000	3,729,600	4,669,000
$\Delta r_{wall}^+$	0.86	0.9	0.95	0.91
$\Delta \omega_{wall}^+$	9.5	9	8.1	9.76
$\Delta x^+$	24	24	24	24

228

229

230

231

Table 2. Mesh parameters for each case of study.

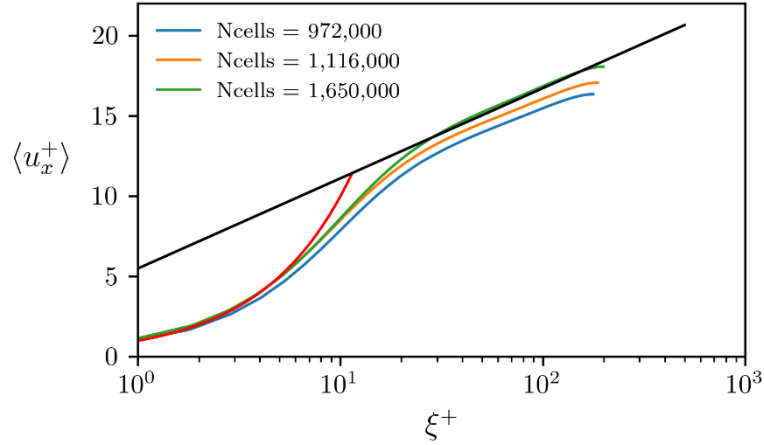


Figure 2: Mesh study for  $Re_D = 5,925$

232

233

### 234 3.6 Mesh index of quality

235 In order to assess the quality of the LES resolution, an index of quality has been computed once each simulation was  
 236 completed. There are three parameters on which the quality index can be based: kinetic energy, length scales and viscosity.

237 In this work, the criterion based on the ratio between the kinetic energy resolved and the total kinetic energy is applied to  
 238 assess the quality of the simulations. A LES is considered to possess good quality when at least the 80% [34] of the kinetic  
 239 energy is resolved by the grid resolution ( $IQ_k > 0.8$ ). The index of quality  $IQ_k$  is expressed at Eq. (6):

$$IQ_k = \frac{k_{res}}{k_{tot}} = \frac{k_{res}}{k_{res} + k_t + k_{num}} \quad (6)$$

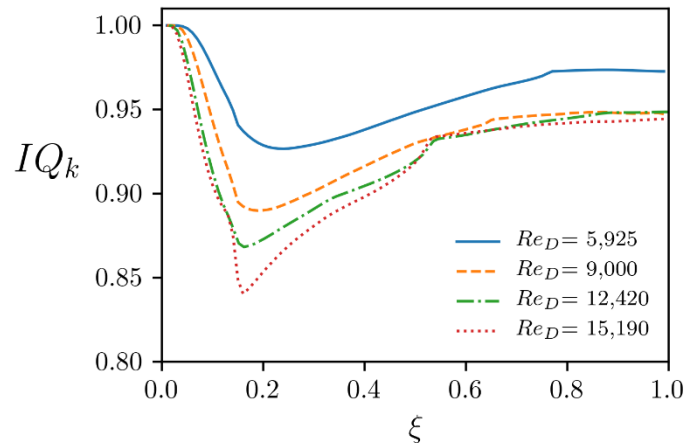
240 where  $k_{res}$  is the kinetic energy resolved,  $k_t$  is the turbulent kinetic energy modelled by the subgrid model and  $k_{num}$  is  
 241 the kinetic energy linked to the numeric error. The kinetic energy resolved can be easily computed with Eq. (7); the  
 242 modelled part is evaluated with Eq. (8) proposed by Sagaut [35]; and the numerical part is estimated as suggested by  
 243 Celik et al. [36] on a single grid according to Eq. (9).

$$k_{res} = \frac{1}{2} (u_{x,rms}^2 + u_{y,rms}^2 + u_{z,rms}^2) \quad (7)$$

$$k_t = \frac{1}{(C_m \Delta)^2} \nu_t \quad (8)$$

$$k_{num} \approx k_t \quad (9)$$

244 where  $C_m$  is a model constant whose value is set to 0.091. As shown in Eq. (7), temporal statistics need to be computed  
 245 in order to obtain  $IQ_k$ . At this point, it is worth mentioning that all the mean parameters included in this document  
 246 correspond to a plane located at half the pipe length, being temporally and azimuthally averaged in order to condense all  
 247 the data in a single curve for each case.



248 Figure 3: Index of Quality based on the turbulence resolution ( $IQ_k$ ) for all tested cases.

249 Figure 3 depicts the evolution of  $IQ_k$  through the radial position defined as  $\xi = (R - r)/R$ . Please note that  $\xi$  is equal to  
 250 0 at the wall, being 1 at the centre of the pipe. It may be seen that all simulations have an  $IQ_k$  higher than 0.8 regardless  
 251 of the radial location, implying a good quality of the calculations in each tested case. All computed cases present the  
 252 minimum values located at the same point, where the grid topology changes. This means that the subgrid model models  
 253 more energy in that region, whereas it has a lower influence on the results in the pipe centre and close to the wall. The  
 254 LES quality has then been assessed based on the resolved kinetic energy in what constitutes the first validation of the  
 255 calculations performed.

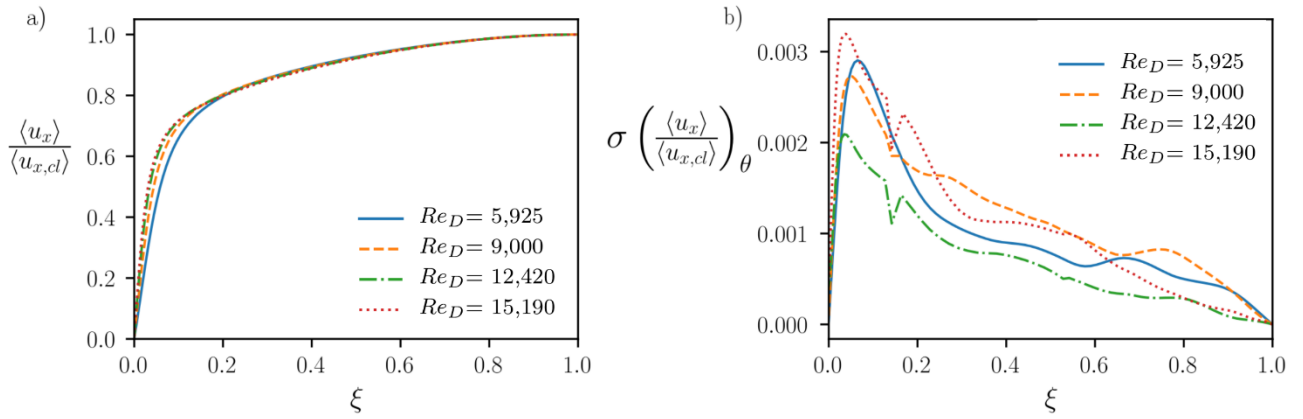
256

## 257 4. RESULTS AND DISCUSSION

### 258 4.1. Validation for fully developed turbulent pipe flow

259 One of the main objectives of this study is the analysis of the turbulent structures within the pipe flow. However, before  
 260 visualizing the vortex distribution, it is important to fully validate the results obtained from the simulations. To this end,  
 261 several parameters based on the mean velocity statistics are assessed in the present Section.

262 First, it is important to check that the mean velocity profile is developed, meaning that the simulation has reached a steady  
 263 state. Figure 4a) shows the mean velocity profile in the streamwise direction normalized by the centreline velocity against  
 264 the radial position  $\xi$ . It can be noticed that the profiles collapse at the centre of the pipe, the discrepancies being mainly  
 265 found near the wall. This behaviour is well-known as addressed in the literature [37]. To ensure that the flow has been  
 266 developed in the azimuthal direction, Figure 4b) displays the radial distribution of the standard deviation of the axial  
 267 component of the velocity along this direction. Major deviations are found close to the wall regardless the case studied.  
 268 Anyway, the values observed are very low for all cases, implying that the mean profiles are practically identical along the  
 269 azimuthal direction and confirming the turbulence has reached a steady state.



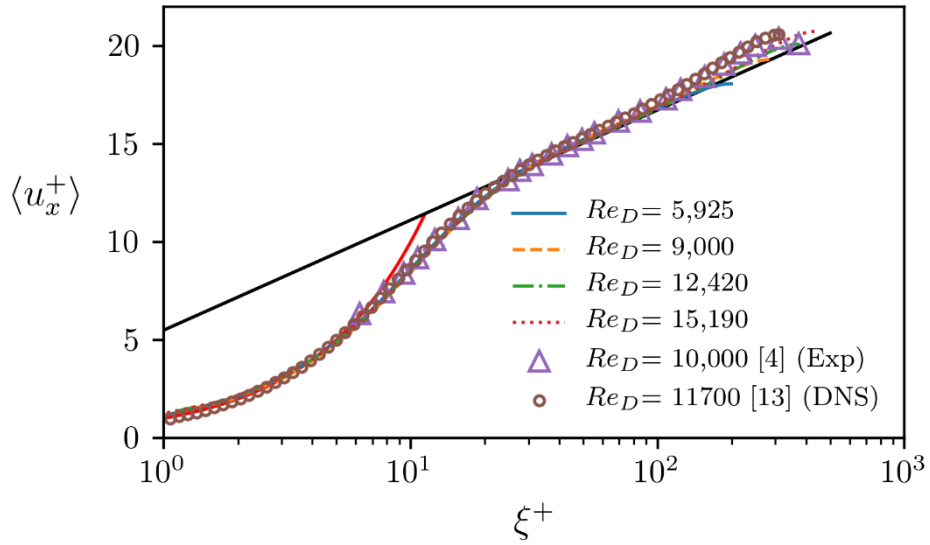
270 Figure 4: a) Mean axial velocity profile over the pipe radius for all tested conditions, b) standard deviation of the mean  
 271 axial velocity profile in the azimuthal direction.

272 Once the steady state is reached, it is important to compute the friction parameters in order to further assess the reliability  
 273 of the obtained results. Table 3 shows the results concerning the friction velocity and the Kármán number ( $Re_\tau$ ). The ratio  
 274 between the friction velocity and the bulk velocity decreases as  $Re$  increases, in agreement with the results presented at  
 275 [13]. This trend will also be seen when computing the friction factor.

$Re_D$	$u_\tau/U_b$	$Re_\tau$
5,925	0.671	199
9,000	0.635	286
12,420	0.607	377
15,190	0.592	445

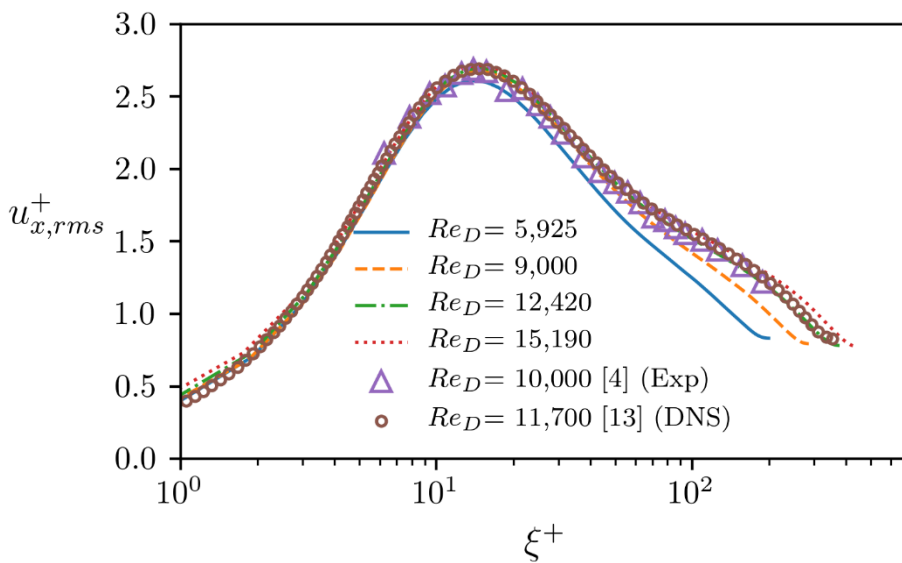
276 Table 3. Friction velocity and Kármán number for each case studied.

277 Figure 5 shows the evolution of the mean axial component of the velocity profile in the radial direction for all the  
 278 simulations carried out. Both  $u_x^+$  and  $\xi^+$  are expressed in wall-units. This result confirms that, for all the studied cases,  
 279 the mean axial velocity profile scales with the friction velocity and collapse according to both the law-of-the-wall and the  
 280 values of DNS data [13] and experimental data [4]. Unlike channel flow, several studies [3, 13] have shown that the pipe  
 281 flow does not follow the logarithmic region close to the pipe centre. This behaviour is more important as the Reynolds  
 282 number increases. Different values of the parameters that define the logarithmic region may be used. In this study,  $B =$   
 283 5.2 and  $\kappa = 0.41$  have been chosen as suggested by Nagib and Chauhan [38].



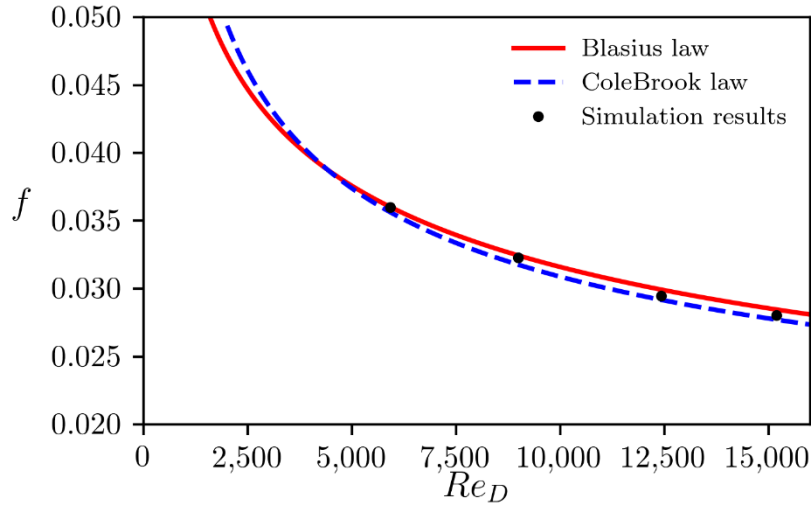
284 Figure 5: Mean axial velocity profile in wall units for each case studied.

285 Together with the velocity profiles, the standard deviation of the velocity field near the wall is important to characterize  
 286 the turbulent intensity. Figure 6 depicts the axial component of the root mean squared velocity ( $u_{x,rms}^+$ ) for all the studied  
 287 cases. Again, results are compared against DNS data from El Khoury et al. [13] and experimental data from den Toonder  
 288 [4]. It is important to note that the maximum value of  $u_{x,rms}^+$  for all the tested conditions is practically obtained at the  
 289 same location (when expressed in wall-units) than the one obtained through DNS and experiments. It may also be noticed  
 290 that the peak value of the  $u_{x,rms}^+$  increases with the Reynolds number. This behaviour has been reported in different studies  
 291 [13–15], but still remains an open question concerning this kind of flows. Nevertheless, the maximum values reached are  
 292 slightly different at low Reynolds number, having a small underestimation of the peak of this parameter.



293 Figure 6: Root mean squared velocity in the streamwise direction for each case studied.

294 Finally, in order to assess the pressure drop inside the pipe, it is relevant to compare the friction factor evaluated from the  
 295 friction velocity ( $f = 8 u_\tau^2 / U_b^2$ ) with the Blasius or the Colebrook laws [39]. Figure 7 plots the friction factor values  
 296 obtained from all numerical simulations against the Reynolds number. Their close resemblance to the values suggested  
 297 by the Blasius and Colebrook (particularized for a roughness factor equal to 0) laws implies that the turbulent behaviour  
 298 near the wall is properly captured. It is interesting to point out that, when the Reynolds number increases, the values  
 299 obtained match the Colebrook law more closely than the Blasius law.



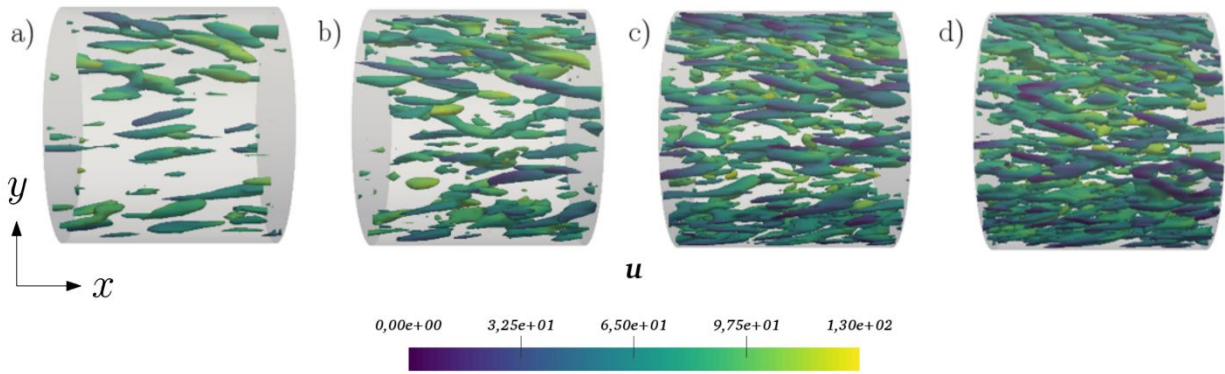
300 Figure 7: Friction Factor obtained from the simulations against the Blasius and Colebrook laws.

#### 301 4.2. Detection and analysis of Vortex

302 By studying the contours of the  $\lambda_2$  criterion, it is possible to study the distribution of vortex cores detected within the  
 303 domain. It must be noted that the values of  $\lambda_2$  usually fall within a wide range of numbers for this kind of studies. Hence,  
 304 in order to compare and distinguish the distribution of the vortex structures for all the conditions tested, two different  
 305 approaches have been performed. On the one hand, the same threshold value has been set for all the operating conditions.  
 306 In this case, if the chosen threshold was too high, no vortex would appear at low Reynolds; if it was too low, the different  
 307 vortex would start merging, obtaining a chaotic pattern not representative of the structure distribution. With this  
 308 considerations, the threshold value for this approach is set to  $\lambda_2 = -1e^{13}$  for all the studied cases. On the other hand,  
 309 being that the values obtained from the  $\lambda_2$  criterion increase with the Reynolds number, a relative threshold has been  
 310 used to have a proportional vortex definition for all cases. An average of the minimum  $\lambda_2$  is first performed for each case.  
 311 Then, the 10% of this value is chosen as a threshold, this value is chosen to coincide for the  $Re_D = 5,925$  condition so  
 312 the chosen values are  $-1e^{13}$ ,  $-1.4e^{13}$ ,  $-2.44e^{13}$  and  $-3.1e^{13}$  when increasing the Reynolds number.

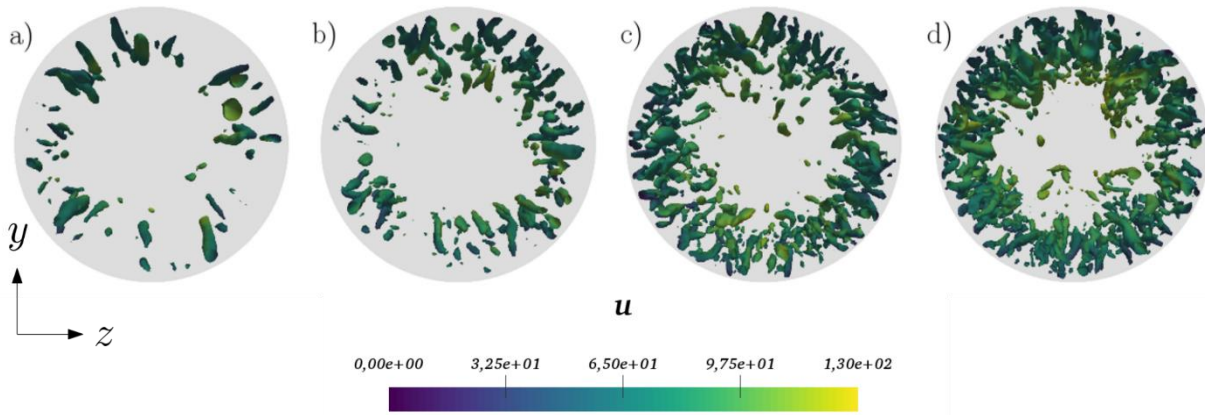
313 First, qualitative results are shown in order to demonstrate the influence of the Reynolds number on the generation of  
 314 turbulent structures. Figure 8 represents the contours of  $\lambda_2$  in a streamwise direction for all the tested conditions. It must

315 be noticed that, as the Reynolds number increases, so does the number of structures. Nevertheless, their individual mean  
 316 volume seems to decrease.



317 Figure 8: Iso-contours of  $\lambda_2 = -1e^{13}$  in the XY plane for a)  $Re_D = 5,925$ , b)  $Re_D = 9,000$ , c)  $Re_D = 12,420$ , d)  $Re_D =$   
 318  $15,190$  coloured by velocity in m/s.

319 In order to visualize the radial distribution, a perpendicular representation is shown in Figure 9. It is interesting to point  
 320 out that the structures are mainly found near the wall and around the half of the pipe radius for all cases, leaving a void  
 321 of turbulent structures at the pipe core. This result confirms the qualitative findings of Figure 8 in terms of number of  
 322 structures and mean volume of the individual structures as a function of  $Re_D$ .



323 Figure 9: Iso-contours of  $\lambda_2 = -1e^{13}$  in the YZ plane for a)  $Re_D = 5,925$ , b)  $Re_D = 9,000$ , c)  $Re_D = 12,420$ , d)  $Re_D =$   
 324  $15,190$  coloured by velocity in m/s.

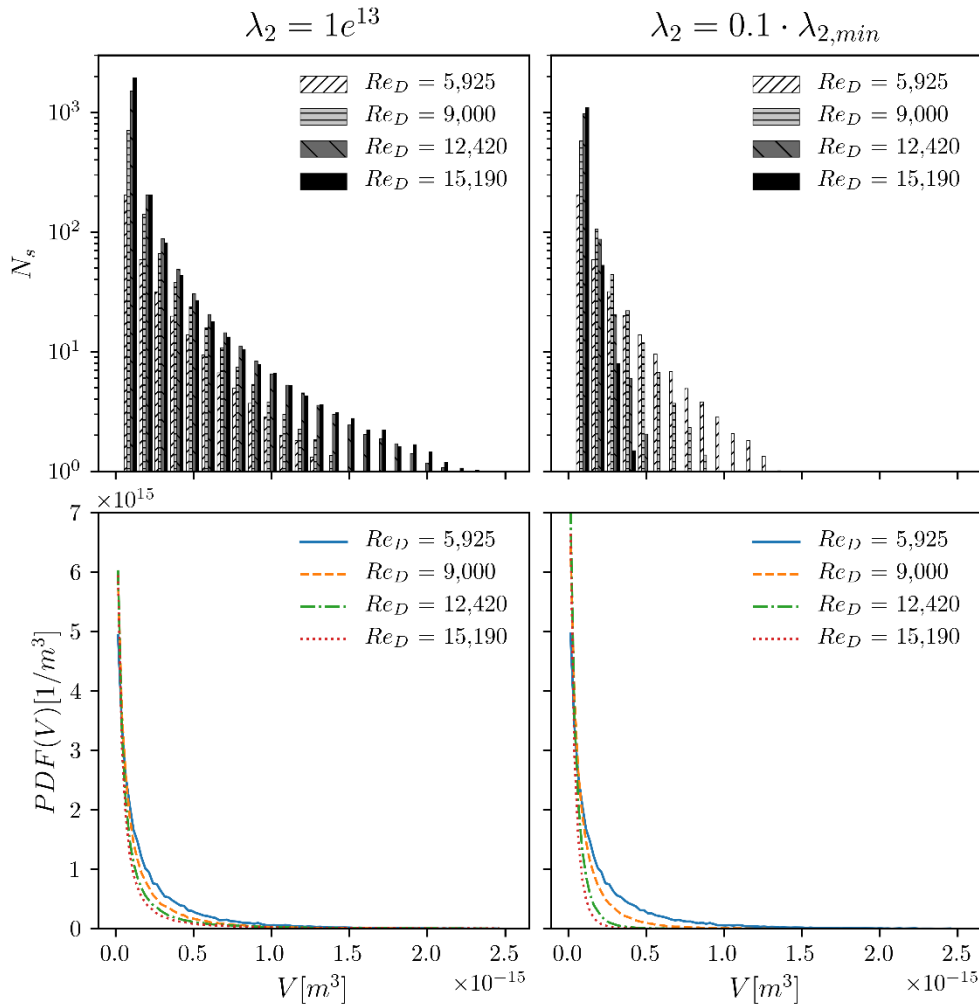
325 Once the qualitative results have been shown, a vortex detection algorithm is used to count these structures and sort them  
 326 by volume and radial position. In order to get a reliable time average distribution, one snapshot per washout has been  
 327 saved along 200 consecutive washouts. When computing the iso-surface extraction, some regions with a very small  
 328 volume appear (as can be seen at Figure 9). Those small contours defined by only a few points have a numerical base. In  
 329 order to clean the results from numerical noise, a minimum volume of two characteristic cells is set. Other than detecting  
 330 the number of structures, Probability Density Functions (PDF) for both the total structures detected by volume and by



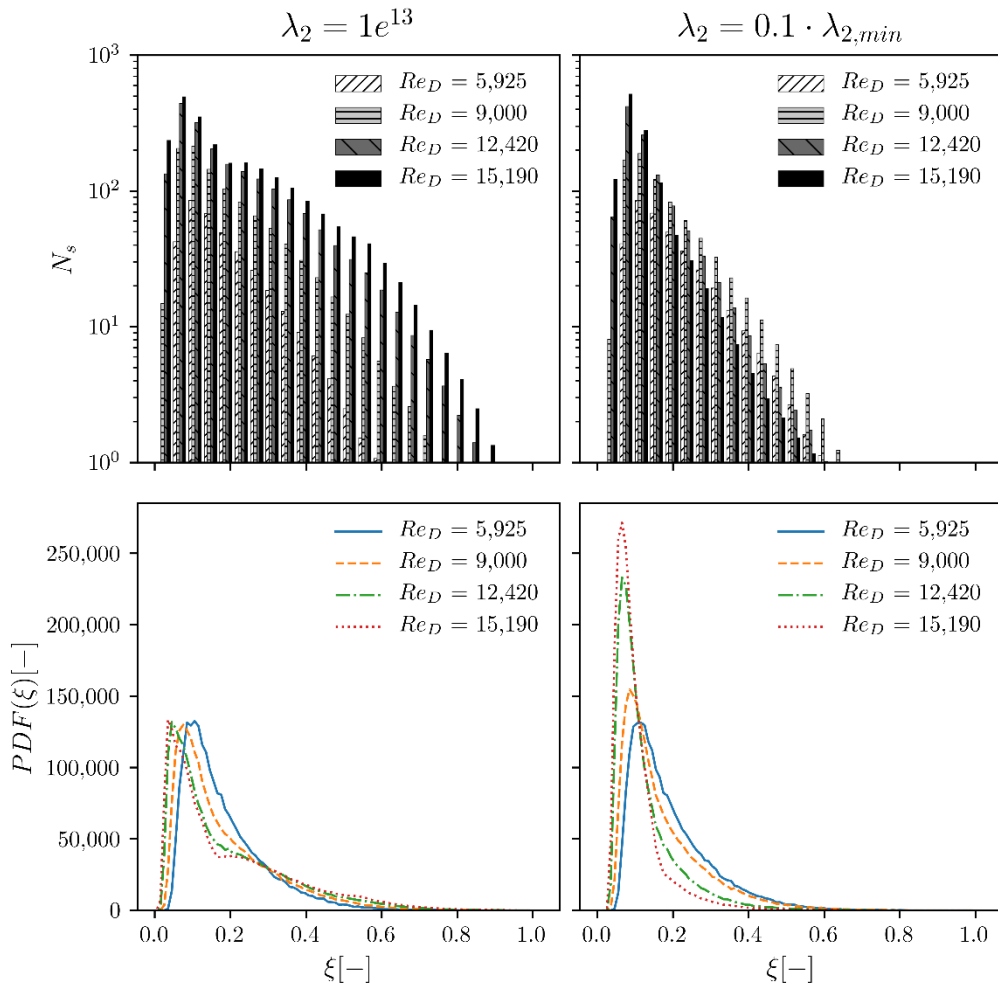
331 radial position have also been obtained. These PDF discard the effect of the total amount of structures and allow directly  
332 comparing the vortex size and radial distribution. Figure 10 shows the results of applying this algorithm in terms of  
333 volume to all the cases studied for both  $\lambda_2$  threshold criteria. Hence, upper figures from Figure 10 show the number of  
334 vortex structures ( $N_s$ ) sorted by volume, whereas bottom figures depict the volume PDFs. Also, left figures correspond  
335 to the absolute threshold approach, whereas the right ones correspond to the relative threshold study. Starting with the  
336 volume distribution, the absolute threshold shows an increase of the total number of structures with the Reynolds number  
337 for the complete range of volumes, as hinted through the iso contours plotted in Figures 8 and 9. On the other hand, the  
338 relative threshold approach dilutes the bigger structures when increasing the Reynolds number. The reason behind these  
339 two different behaviours lies on the fact that the absolute threshold is lower than the relative one used, so that the bigger  
340 structures that appear at higher Reynolds correspond to less turbulent structures that are ‘filtered’ when increasing the  
341 value of this parameter. Nevertheless, bottom figures from Figure 10 show very similar PDF for the structures volume  
342 distribution, where the smallest scales increase its importance as the Reynolds number increases and the bigger ones are  
343 more present at lower Reynolds number. This also agrees with the findings from Figures 8 and 9 that each individual  
344 structure seemed to be qualitatively bigger at low Reynolds. Consequently, as the Reynolds number increases, the number  
345 of structures increases globally (i.e. for any volume size considered), but the probability of a given structure to be big  
346 quantitatively decreases.

347 Analogously, Figure 11 shows the same composition from Figure 10 but in terms of radial distribution (i.e. sorted by the  
348 radial position of the vortex centre). The radial distribution depicted in the upper plots from Figure 11 shows that the  
349 number of structures detected increases with the Reynolds number, regardless of the radial position considered when  
350 applying the absolute threshold. When applying the relative threshold, in turn, the number of structures detected near the  
351 wall is higher at high Reynolds number, but lower when moving towards the pipe centre. Bottom figures from Figure 11  
352 display the probability of finding a structure at each radial position. In contrast with Figure 10, where both approaches  
353 led to similar PDFs, the obtained PDFs show different radial probability patterns depending on the  $\lambda_2$  threshold criterion  
354 used. The absolute threshold approach shows a maximum location shift towards the wall as the Reynold number increases,  
355 and around  $\xi = 0.3$  the higher Reynolds conditions show higher PDF values. This means that, even though the volume  
356 distribution is not highly affected by  $Re_D$ , the radial distribution is more spread the higher the Reynolds number. In  
357 contrast, when applying the relative threshold, the radial distribution tends to gather near the maximum peak and the  
358 dispersion towards the pipe centre disappears when increasing the Reynolds number. Even though this behaviour can  
359 seem contradictory at first sight, it can be understood when comparing both the volume and the radial distributions. The

360 analysis from Figure 10 stated that the bigger structures at higher Reynolds number were less turbulent than the smaller  
 361 ones because they were filtered when the  $\lambda_2$  increased in the relative threshold approach. These filtered structures were  
 362 the ones that spread towards the pipe centre in Figure 11, which is consistent with the fact that the very large structures  
 363 are located near the centre [10] when the turbulence increases. This reasoning is in turn consistent with the vortex  
 364 distributions qualitatively observed in Figure 9.

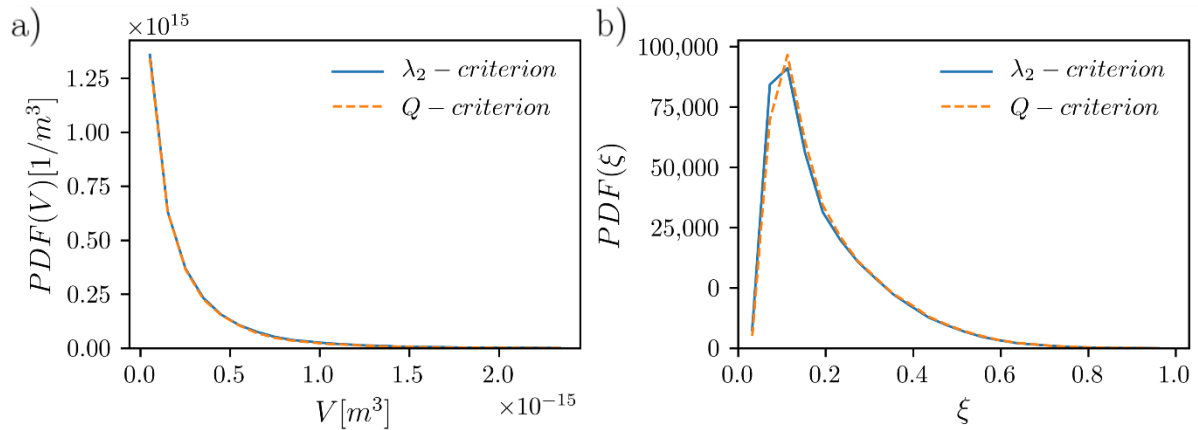


365 Figure 10: Left, absolute threshold approach; right, relative threshold approach. Upper: time averaged number of  
 366 structures sorted by its volume for each case studied; bottom: volume PDF for each case studied.



367 Figure 11: Left, absolute threshold approach; right, relative threshold approach. Upper: time averaged number of  
 368 structures sorted by its radial for each case studied; bottom: radial position PDF for each case studied.

369 Finally, the same approach has been carried out using the Q-criterion to detect and sort the turbulent structures. A  
 370 comparison among the  $\lambda_2$  and Q-criterion has been made for the  $Re_D = 9,000$  condition. Please note that  $Q = 1e^{13}$  has  
 371 been used as the threshold value to detect the contours (same absolute value than the one used for the  $\lambda_2$  criterion). The  
 372 comparison has been done in terms of PDF of volume and radial distribution, as depicted in Figure 12. Figure 12a)  
 373 presents the PDF of the number of structures detected for different volumes, where it can be clearly seen that the trends  
 374 obtained through both criteria collapse perfectly. On the other hand, Figure 12b) shows that the peak shape is slightly  
 375 sharper with the Q-criterion, although its radial position remains at the same distance from the wall and the main shape  
 376 is quite similar.



377 Figure 12: a) Volume PDF for  $Re_D = 9,000$  for both vortex detecting methods, b) Radial PDF for  $Re_D = 9,000$  for both  
 378 vortex detecting methods.

### 379 5. CONCLUSIONS

380 LES of fully developed turbulent pipe flow have been performed at low Reynolds number, from  $Re_D = 5,925$  to  $15,190$   
 381 based on the pipe diameter. The Navier-Stokes equations are solved with the pisoFoam solver (OpenFOAM code) using  
 382 the WALE SGS model. The assessment of the computational LES calculations has been done by evaluating the  $IQ_k$   
 383 quality index, reporting values above 0.8 in all the radial positions for all the simulated cases.

384 Computing the mean statistics for the different cases showed that the steady state was achieved for all the simulations  
 385 carried out. The turbulent processes are found to be well captured, despite small discrepancies in the streamwise inner  
 386 peak of the root mean squared velocity. As the Reynolds number increases, so do the turbulence and the inner  $u_{x,rms}^+$ , as  
 387 stated in the literature [8-10].

388 When it comes to the vortex structures, two different approaches based on the  $\lambda_2$  criterion have been used to get the  
 389 volume and radial distribution of the vortex within the flow. On the one hand, the same threshold has been used to define  
 390 the vortex core at all operating conditions; on the other hand, a relative threshold depending on the minimum value of  $\lambda_2$   
 391 for each tested case has been used. Regarding the number of detected structures, both methods detect an increase in the  
 392 number of structures generated then the turbulence increases. Nevertheless, the absolute threshold procedure exhibits an  
 393 increase of the maximum volume detected with the Reynolds Number, whereas the relative threshold procedure shows  
 394 that the lower Reynolds conditions generate the bigger structures. This means that the lower value of the threshold at high  
 395 Reynolds conditions is linked to bigger structures that are less turbulent. This theory is confirmed when looking at the  
 396 radial distribution of the vortex structures. Using an absolute threshold, the detected structures spread towards the pipe  
 397 centre when increasing the Reynolds number, behaviour that is not showed when using a relative threshold value. This  
 398 means that the bigger structures, which are less turbulent, are located away from the wall. Finally, the peak values at

399 which the structures collapse are maintained using both approaches, their location shifting towards the wall as the  
400 Reynolds number increases. In addition, results obtained using  $\lambda_2$  criterion with the absolute threshold approach and the  
401 ones obtained using the  $Q$  criterion (computed for a single operating condition) are quite similar presenting the same  
402 volume PDF and virtually the same radial distribution.

403 These results can be used to compare the structures generated by a pipe flow synthetic boundary condition when using  
404 them or when generating new synthetic turbulence models for these cases.

405 Lastly, the results presented on this paper expect to be an accessible way of comparing and validating the structures  
406 generated by a pipe flow when using a synthetic boundary condition or when developing new synthetic turbulence models.

407 As reported in the literature [26, 40], these structures play a significant role on spray atomization processes. These works  
408 stated that an increase in the amount of these vortex cores and their proximity to the gas-liquid interface cause an increase  
409 in the spray core instabilities, finally leading to a better atomization. The findings of the present investigation thus open  
410 a new insight on further atomization studies, also being relevant for other applications.

411

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418

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